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**COMPUTER PROGRAMS FOR CALCULATING
TWO-DIMENSIONAL POTENTIAL FLOW
THROUGH DEFLECTED NOZZLES**

**J. Dennis Hawk and Norbert O. Stockman
Lewis Research Center
Cleveland, Ohio**

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16 Abstract <p>Computer programs to calculate the incompressible potential flow, corrected for compressibility, in two-dimensional nozzles at arbitrary operating conditions are presented. Included are a statement of the problem to be solved, a description of each of the computer programs, and sufficient documentation, including a test case, to enable a user to run the program.</p>					
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INTRODUCTION

Many proposed V/STOL aircraft require propulsion systems that deflect the thrust during takeoff, landing, and hovering. This deflection can be accomplished by use of flow turning nozzles downstream of the fans. These nozzles must operate efficiently and effectively over wide ranges of internal flow conditions. An important tool in the design of an efficient nozzle is the capability to theoretically analyze the nozzle flow.

Many of the proposed V/STOL nozzles are three-dimensional (fig. 1). Since the capability for a full three-dimensional nozzle analysis does not exist at this time, simpler but approximate methods for analyzing these nozzles were developed. Even though these nozzles are three-dimensional overall, there are sections of the nozzles for which a two-dimensional analysis is an informative approximation. The analytical procedure described herein is a two-dimensional analysis.

The procedure developed utilizes three computer programs and is similar to the procedure for calculating axisymmetric inlet flows (ref. 1) and two-dimensional inlet flows (ref. 2). The chief program is the higher order Douglas two-dimensional potential flow program (ref. 3) called 24Y at the Lewis Research Center, which calculates the incompressible potential flow about arbitrary two-dimensional bodies. The other programs, original at Lewis, are called SCIRCL and NOZZLEC. Program SCIRCL generates geometric inputs for 24Y from various specified analytical shapes or sets of coordinate points for the nozzle components. Program NOZZLEC takes the basic solutions output by 24Y and combines them into solutions of interest, and, if desired, applies a compressibility correction. Figure 2 is a schematic representation of these solution steps.

This paper consists of a statement of the problem to be solved, description of each program, sample results, and sufficient documentation, including a test case, to enable the user to run the programs.

SYMBOLS

A, B, C, D	combination coefficients
M	Mach number
V	velocity
\dot{W}	mass flow
α	incidence angle

Subscripts:

c	control station
com	compressible
i	incompressible
S1	passage between body 1 and body 2
S2	passage between body 2 and body 3
∞	free stream value

PROBLEM AND SOLUTION DESCRIPTION**Statement of the Problem**

Geometric representation. - A three-dimensional nozzle can be approximated two-dimensionally by using the profiles at the plane of symmetry of the three-dimensional nozzle (fig. 3). The basic assumption for two-dimensionality is that all derivatives in the z-direction vanish, that is, $\partial(\)/\partial z = 0$.

Care should be used when attempting to use a centerbody because in two dimensions the centerbody is actually a flow splitter. In most cases, adequate results can be obtained by omitting the centerbody.

Nozzles. - The basic problem to be solved is to calculate either the compressible or incompressible flow in an arbitrary two-dimensional nozzle at any combination nozzle mass flow rates \dot{W}_c , \dot{W}_{S1} , or \dot{W}_{S2} . Figure 4 shows three possible combinations of nozzle geometries and weight flow specifications to be analyzed. Figure 4(a) shows the single passage nozzle where weight flow rate, \dot{W}_c is specified. Figure 4(b) shows a nozzle with dual passages where the two passage weight flow rates \dot{W}_{S1} and \dot{W}_{S2} are specified. Figure 4(c) shows the same nozzle as figure 4(b). However, in this case the total nozzle and one passage weight flow rate, \dot{W}_c and \dot{W}_{S1} are specified.

Inlets and nozzles. - While most of the nozzles problems are internal flow problems, occasionally external flows about inlets and nozzles are desired (fig. 5). If this is the case the programs can accomplish this by inputting values of free stream velocity V_∞ and nacelle incidence angle, α , along with the required weight flows described previously. A detailed description of inlet calculation methods is given in references 1 and 2.

The two-dimensional flow problem is solved in several steps (programs).

1. Geometric representation (Program SCIRCL)

2. Incompressible potential flow basic solutions (Program 24Y)
3. Combined solutions with compressibility correction (Program NOZZLEC).

Each step and its corresponding program are described in the following section.

Description of Solution Steps and Programs

Geometric representation - Program SCIRCL - The nozzle is assumed to be two dimensional. The profiles are broken into segments at convenient tangential points (fig. 6). These profiles are referred to as bodies in the program nomenclature. Each segment may be defined by an analytic expression or by a set of points. The nozzle walls and outer surfaces (if any) must be extended far downstream (fig. 6) to facilitate obtaining accurate potential flow solutions in the nozzle in the region of interest. The geometry program SCIRCL prepares coordinate-point input for efficient use of the potential flow program and also prints out information such as curvature, wall angles, etc., which is useful in preliminary screening of proposed body shapes.

In addition to the surface points, set of points in the vertical plane, called rakes, are needed at axial locations where velocity profiles or streamlines are desired. At least one set of rake points is required, corresponding to the axial location where the weight flow is specified, as discussed with regard to figure 4. Such rakes are called control stations.

Program SCIRCL generates the coordinates of the rake points for 24Y. Program SCIRCL will also produce a Cartesian plot of the nozzle geometry and rake points.

Incompressible potential flow basic solutions - Program 24Y. - Program 24Y is the Douglas incompressible potential flow computer program for single or multiple two-dimensional bodies. Briefly, the program utilizes a distribution of sources and sinks of initially unknown strength to represent the body profile. The continuous distribution is approximated by representing the body as a finite number of discrete flat elements having constant source strength and characterized by the mid point of the element (called the control point). This approximation results in a set of parabolic algebraic equations for source strength at the control points that are solved by matrix methods. Velocities at the control points and at specified off-body points (rake points) are then calculated from the source/sink distribution.

In the two-passage case, the program is used to obtain five basic solutions which are used in linear combination in order to satisfy the prescribed operating

conditions (fig. 4). The first basic solution is axial flow, the second is a 90° cross flow, the third, fourth, and fifth solutions are vorticity solutions about bodies 1, 2, and 3, respectively. In the single passage case, only four solutions are generated. The first two are the same as the two-passage case while the third and fourth are vorticity solutions about the two bodies.

Combined solution - Program NOZZLEC. - This program combines the basic solutions \bar{V}_j , $j = 1, 2, 3, 4, 5$ from 24Y into any number of solutions of interest. A solution of interest is one having specific values of free stream velocity, V_∞ , angle of attack of the free stream, α , and weight flow, \dot{W} , through the control stations described previously. Optional inputs for \dot{W} are average axial velocity at the control stations (V_c , V_{S1} , or V_{S2}) or average Mach numbers (M_c , M_{S1} , or M_{S2}). If either \dot{W} or M_c are specified, they are converted to velocities for use in the combination solutions. Temperature and pressure must also be specified if other than standard conditions are used.

The method of combination of the basic solutions is shown in figure 7. A compressibility correction, if desired, is then applied to the velocities. The compressibility correction used is

$$V_{\text{com}} = V_1 \left(\frac{\rho_t}{\rho_c} \right)^{V_1 / \bar{V}_c}$$

where the terms on the right-hand side are obtained from the incompressible solution. This correction requires no alteration of the inlet geometry and can handle locally sonic or supersonic flows. Flow properties (Mach number, pressure ratio, etc.) are calculated for either the compressible or incompressible cases depending on the version desired. The compressibility correction is actuated by setting ICOMP1 to 0. The incompressible version is generated by setting ICOMP1 to 1. If two passages are specified, two of the rakes mentioned under SCIRCL are used as control stations by NOZZLEC. A control station is the rake where the average axial velocity of the combined solution is specified. The possible choices of control stations for two passage cases are shown in figures 4(b) and (c). If only one passage is specified (fig. 4(a)), then any rake can be used as a control station. However, it should be noted that the compressible solution is most accurate in the vicinity of the control station since the compressibility correction does not exactly satisfy continuity.

When the velocity on the surface becomes locally supersonic, the agreement between theory and experiment is generally not as good as when the flow

remains subsonic. To improve the agreement, an optional supersonic correction has been incorporated into the program. The supersonic correction is actuated as specifying NX as 1 on input.

Sample Results

To illustrate the results of the present calculation procedure, the graphic output for the Test Case is presented next.

The graphic output of SCIRCL is illustrated in figure 8 and consists of a plot of the nozzle geometry and the computational point spacing, and the location of the rakes and their point spacing. In addition to its reference value the plot is useful for troubleshooting for mistakes in the geometry input and for validating the satisfactory distribution of body points and rake points.

The graphic output of NOZZLEC is illustrated in figures 9 to 11. Figure 9 shows the pressure distribution on the surfaces of body 1. Similar plots can be obtained for bodies 2 and 3.

Figure 10 shows the surface Mach number distribution. The comments made above for the pressure plots apply also to Mach number plots.

Figure 11 shows a plot of the flow field vectors. Whenever rake points are specified (see fig. 8) velocity vectors are calculated and can be plotted.

INPUT AND OUTPUT FORMAT

SCIRCL Input

Card	Format	Column	FORTTRAN name	Description
	9A6	1-54	ARE	= title for area plots. "CROSS-SECTIONAL AREA, SQ. IN."
	9A6	1-54	EX	= title for x-axis. "AXIAL POSITION, IN."
	9A6	1-54	CURVO	= title for curvature plots. "CURVATURE ON SHROUD"

Card	Format	Column	FORTTRAN name	Description
	9A6	1-54	SURFAC	= title for x-axis with surface distance plots. "DISTANCE ALONG SURFACE FROM DOWNSTREAM END OF SHROUD, IN."

These first four cards above will be unchanged for all runs and can be made a part of the execution setup deck, or replaced with data statements.

Card	Format	Column	FORTTRAN name	Description
1	9A6	1-54	TITLE	Description of Case

FOR CALCOMP PLOTTING OF INLET PICTURES

Card	Format	Column	FORTTRAN name	Description
2	8F10.2	1-10	XX	Length, in plot-inches, of X-axis required
		11-20	XMIN	Value, in data-inches, of far left X-point.
		21-30	EXEP	Data-inch per plot-inch along X-axis.
		31-40	YY	Length, in plot-inches, of Y-axis required.
		41-50	YMIN	Value, in data-inches, of bottom Y-point
		51-60	ORD	Data-inch per plot-inch along Y-axis (usually equal to FXFP).

Card	Format	Column	FORTTRAN name	Description
		61-70	ELREF	The X-values in area output data are non-dimensionalized by ELREF. Default value is 1.
		71-80	AREF	The areas in area output data are nondimensionalized by AREF. Default value is 1.
3	4X, F10.2, 1X 3I1	5-14	ALPHER	If only one body is input, ALPHER is the angle of attack (used by 24Y)
		16	IPARA	Element geometry flag used by 24Y = 0 Linear elements = 1 Parabolic elements
		17	IFRST	First-order terms flag = 0 No first-order terms = 1 First derivative terms = 2 Curvature terms = 3 Both first-order terms
		18	ISND	Second-order terms flag = No second-order terms = 1 Second derivative terms = 2 Curvature squared terms = 3 Both second-order terms

Card	Format	Column	FORTTRAN name	Description
		19	IFLLL	= 0 No combination solution will be calculated by 24Y = 1 A combination solution will be calculated by 24Y
4	2A6, I4, 2I1, 2I2, 10X, I1, 9X, 3I1, 2I2	1-6	IDENT	Six-character tag for case I D.
		7-12	PROG	24Y (right justified)
		13-16	NO6	0
	(Flag 'A')	17	LPNCHO	1, Do not save output for 24Y on Unit 17.
	Flag B	18	IPLOTA	Plot area against X-position
	Flag C	19-20	IPLOT C	-1, Plot curvature versus X, +1, Plot curvature versus S
	Flag D	21-22	IREAD	0 (Obsolete)
<u>ALL FLAGS</u> are 'on' when =1, unless otherwise noted (Either E or J or neither can be on but not both).				
	Flag J	33	IAB	Redo geometry from point
	Flag E	43	IREDON(1)	Redo entire geometry via direct interpolation

Card	Format	Column	FORTRAN name	Description
	Flag F	44	IREDON(2)	LPNCHO for any redo
	Flag G	45	IREDON(3)	IPLOTA for any redo
	Flag H	46-47	IREDON(4)	IPLOTC for any redo
	Flag I	48-49	IREDON(5)	IREAD for any redo
Skip card 5 if J=0				
5	4F12.5	1-12	XAA	X position of starting point for partial redo.
		13-24	YAA	Y position of starting point for partial redo.
		25-36	XBB	X position of ending point for partial redo.
		37-48	YBB	Y position of ending point for partial redo.
6	4F10.2	1-10	ANBDYS	Number of bodies
		11-20	DELS	Spacing between points in region of interest.
		21-30	DELSMX	Maximum spacing far from region of interest
		31-40	XRI	Axial distance at which surface distance equals zero.
7	I4	1-4	NRAKE	Number of axial locations at which data across the passage is desired. (Can- not be greater than 25).
8	3F8.5, I3,	1-8	XRAK	Axial location of rake.
	4F8.5	9-16	YLO	Y value of first point (lowest point) on rake at XRAK.

(Note: There is one card for each rake)

Card	Format	Column	FORTTRAN name	Description
		17-24	YHI	Y value of last point (highest point) on rake at XRAK.
		25-27	NY	Number of points in rake at XRAK. Restriction $\Sigma NY \leq 200$. Rake points are equally spaced, ΔY , between YHI and YLO where $\Delta Y = \frac{YHI - YLO}{(NY - 1)}$
		28-35	XTRAN	Value of axial transla- tion of rake
		36-43	YTRAN	Value of vertical trans- lation of rake
		44-51	XSCALE	Value of axial scaling of rake
		52-59	YSCALE	Value of vertical scaling of rake
9	4F10.2, 5F8.2	1-10	TYPBDY	Body number. However, if there is symmetry, then any body can be in- put as a mirror image of any other body. This can be accomplished by setting TYPBDY = -M, where M is the name of the body to be created and N is the number of the body to be copied. A USE is set to the Y value of the are about which body N is to be mirrored. No other

Card	Format	Column	FORTTRAN name	Description
				input is required for this body except for ANLF
		11-20	ANSEG	= Number of segments for the particular body, except as stated in TYPBDY.
		21-30	DELNEW	= -1., Delta S spacing is set to original value of DELS. = 0., Delta S is set to value of DELS from previous body. = + number, Delta S is set to value of input DELNEW.
		31-40	ANLF	= 0 Body is a lifting body, i.e., in 24Y a vorticity solution about this body will be calcu- lated. = 1, Body is a non- lifting body, i.e., no vorticity solution will be calculated
Note: All lifting bodies must be input prior to any non-lifting bodies.				
		41-48	XTRAN	Value of axial transla- tion of this body.
		49-56	YTRAN	Value of vertical trans- lation of this body
		57-64	XSCALE	Axial scaling factor

Card	Format	Column	FORTTRAN name	Description
		65-72	YSCALE	Vertical scaling factor
		73-80	XTMAX	Maximum value of X for which scaling is to be applied.
10	3F10.2	1-10	ENREED	<p>Code indicating type of curve to be fitted through given points.</p> <p>= 0 , for bisuper-ellipses. See Table I and figure 12 for available options. Input 1, 2, 3, 4, 5, or 6 (XIN, YIN) points as directed.</p> <p>= 1000. Same as = 0 but with finer point spacing near one end of segment (two such segments required). Usually used to give finer spacing at the highlight. The superellipse going into the highlight and the one coming out should have this flag</p> <p>For bisuperellipses where the '1000.' option is to be used, the rate at which the point spacing, ds, changes near one end $ds_1 - ds_{1-1} - (Rate)(ds_{1-1})$ can be specified on input.</p>

Card	Format	Column	FORTTRAN name	Description
				<p>The rate (program name = PACE) is entered as the fractional part of ENREED for each segment. For example, if ENREED were input as 1000.06, the spacing for consecutive points would be evaluated as follows:</p> $DS_i = DS_{i-1} - (0.06)DS_{i-1}$ <p>if segment is to go from large-to-small spacing,</p> $\text{or: } DS_i = DS_{i-1} + 1.5(0.06) DS_{i-1}$ <p>if segment is going from small-to-large spacing.</p> <p>If PACE is entered as zero (i.e., ENREED = 1000.), the default value, 0.05, is used.</p> <p>$(PACE \leq 0.133)$</p> <p>*The first '1000' superellipse <u>ON A BODY</u> reduces the point spacing as far as possible, down to a limit of 2 percent of the ds value at the beginning of the segment.</p> <p>*All subsequent '1000' superellipses input will increase ds as far as possible up to the input value of DELS.</p>

Card	Format	Column	FORTTRAN name	Description
				* Any number or types of segments may be input between the first and subsequent '1000' bisuperellipses, with the exception of a normal bisuperellipse (ENREED=0).
				= 1, is a straight line, input 2 coordinates (XIN(1), YIN(1), XIN(2), YIN(2)) (fig. 13(a)).
				The first and last straight lines on bodies 2 and 3 and the last straight line on body 1 will automatically have their spacing increased from approximately DELS near the region of interest to approximately DELSMX away from the region of interest. To get this type of spacing in the first straight line of body 1, ENREED must be specified as 10

Card	Format	Column	FORTTRAN name	Description
				= 10., special straight line used for initial straight line on lower shroud. The straight line starts with large spacing (DEMSMX) and ends with small spacing (DELS), (fig. 13(b)).
				= -1., fits a lemniscate between a straight line and a point. Input is three coordinates (fig. 13(c)).
				= -3., fits a cubic between two straight lines. Input 4 coordinates (fig. 13(d)).
				= -4.0, generates a segment which is a mirrored image of all the points from (XIN(1), YIN(1)) to XIN(2), YIN(2)) about the line $Y = YIN(3)$. See cards 11 and 12 for XIN and YIN formats.
				= 99., for direct interpolation option over one segment (see input instructions for card 12)

Card	Format	Column	FORTTRAN name	Description
		11-20	REEDEN(1)	(See table I). Input exponent of x-term for bisuperellipse equation Blank for all other segment types.
		21-30	REEDEN(2)	(See table I). Input exponent for y-term of bisuperellipse.
11	6F12.5	1-72	XIN(I) I=1, 2, 3, 6, 4, 5	X-coordinate for specified points.
12	6F12.5	1-72	YIN(I) I=1, 2, 3, 6, 4, 5	Y-coordinate for specified points. Note. If ENREFD = 99, instead of cards 11 and 12, input the following cards.
11a	Namelist/ \$BODYIN/		Z(I)	Z is a complex array containing the X value (in the real part) and Y value (imaginary part) of each given point along the segment. The namelist will normally be longer than one card The program will use the input points to generate new points with finer point spacing near regions of high curvature
12a	Namelist/ \$AUXIN/		DONE	A logical variable which should be input as .TRUE

Card	Format	Column	FORTTRAN name	Description
			BYPASS	= .TRUE. if no refinement of input points is required.
				Note: If ANSEG=0 and TYPBDY \neq 0, skip card No. 10, and substitute 11a for 11 and 12a for 12.

Input Deck Structure

Card	
1	
2	
3	
4	
5	(only if flag J > 0)
6	
7	
8	} Number of '8' cards = NRAKES
.	
.	
8	
9	Number of '9' cards = ANBDYS
10	} Number of '10-11-12' groups for each '9' card = ANSEG
11	
12	
.	
.	*If ENREED = 99 on card 10, use 11a and 12a instead of 11 and 12
.	*If ANSEG = 0 and TYPBDY \neq 0 on card 9, skip 10 and substitute 11a and 12a for 11 and 12
10	
11	
12	

Figure 14 shows an SCIRCL input form, reflecting the above instructions.

SCIRCL Output

Printed Output

Input file dump (a list of input cards)

Case number and title

Input card 3

Input card 4 (case I.D. and SCIRCL flags)

Input card 6 (number of bodies, DELS, DELSMX, and XRI)

Total number of points for all bodies should not exceed 500.

Total for any one segment of a body should not exceed 200.

An error message will indicate if these limits have been exceeded.

Total number of off-body points must not exceed 200.

Total number of rakes must not exceed 25.

Body 1 segment data, body 2 segment data, and body 3 segment data. For each segment:

ENREED (as input) and type of segment

Data depending on type of segment.

Straight Line

X X(1) X(2)

Y Y(1) Y(2)

Last point data*

Bisuperellipse

Exponents

P as read in X X(1) X(2) X(3) X(6) X(4) X(5)

Q as read in Y Y(1) Y(2) Y(3) Y(6) Y(4) Y(5)

P as used A XO

P as used B YO OMEGA

where A and B = Semimajor and minor axes of transformed superellipse

XO and YO = Center of the transformed superellipse

OMEGA = The difference (in radians) between the slopes of the end point slope lines minus $\pi/2$ (i.e., OMEGA is a measure of the non-perpendicularity of the slope lines).

Number of iterations**

Iteration data***

Last data point*

'Magic triangle' messages refer to the triangle formed by extending the superellipses slope lines toward each other and drawing a chordline between input points number (X2, Y2) and (X4, Y4). Input points (X3, Y3) and (X6, Y6) must lie within this triangle, or, for certain special cases, may lie in a similar triangle on the opposite side of the chordline.

If input points (X3, Y3) and/or (X6, Y6) fall outside the magic triangle, the following message (or messages) is/are printed:

"This point is below the magic triangle..."

"This point is outside the magic triangle..."

Cubic

X	X(1)	X(2)	X(3)	X(4)
Y	Y(1)	Y(2)	Y(3)	Y(4)
A	B	C	D	

where A, B, C, D are the coefficients of the cubic equation.

Number of iterations**

Iteration data***

Last point data*

Lemniscate

X	X(1)	X(2)	X(3)
Y	Y(1)	Y(2)	Y(3)
THETMX	CALC	ACALC	

where THETMX CALC = angle between line 1 - 3 and line 1 - 2.

ACALC from equation $R^2 = 2(ACALC)^2 \sin 2\theta$ where

$$\theta = \text{THETMX} \text{ and } R^2 = [XIN(3) - XIN(2)]^2 + [YIN(3) - YIN(2)]^2$$

Number of iterations**

Iteration data***

Last point data*

*Last point data. This is the coordinate point data for the last point of the segment. It is presented here because it is overwritten by the first point of the following segment and therefore does not appear in the point-by-point array below.

**Number of iterations is the number of iterations required to achieve a satisfactory point spacing for 24Y and have the calculated last point of the segment coincide with the input endpoint (to within a prescribed tolerance of 0.1 DELS). If this satisfactory spacing and end point matching is not achieved within 150 iterations, the following message is printed out above the number of iterations:

"This set of data exceeded 150 iterations. Calculations stopped XBRK YBRK.

X(1), Y(1), X(2), Y(2), X(3), Y(3), X(4), Y(4), X(5), Y(5)."

***Iteration data

DELS IN	Value of ds at end of previous segment
DELS	Value used to start final iteration for this segment
DELS OUT	Value of ds at end of this segment, to be passed on to the next segment
DSTEST	Distance from last calculated point to input segment end-point
FINAL PACE	The value of PACE at the conclusion (or termination) of the point-spacing iteration for bisuperellipses with ENREED 1000

"INPUT FOR THE NOZZLE COMBINATION PROGRAM"

For three body case

NT(1) = , NT(2) = , NT(3) = , NP =

For two body case

NT(1) = , NT(2) = , NP =

Body coordinates (a separate set for each body)

Point number

X-axial distance

Y-radial distance

KAPPA-curvature

DY/DX-slope

ALPHA-slope angle in degrees

S-surface distance measured from first point of each body

S-S(2) - surface distance measured from XRI

DELTAS-distance between points

Rake Information

XRAK - axial location of rake

YLO - Y value of first location on rake

YHI - Y value of last location on rake

NDY - number of Y points on the rake, as input

File output. - The file output (UNIT 17), written by SCIRCL, is used directly as input to 24Y. This file consists of the on-body point coordinates and rake points.

Graphic output. - Standard Calcomp can be produced for each geometry run as follows:

(1) For all cases: an X-Y 'picture' of the nozzle with SCIRCL-generated on-body points denoted by the '+' symbol and connected with straight lines; segment end points have a large circular symbol; off-body rake points are denoted by a small square symbol (see fig. 8).

(2) For cases with IPLOT $\neq 0$: a graph of upper shroud body curvature against axial position (PILOT $= -1$) or against distance along shroud (IPLOT $= +1$).

24Y INPUT

Card	Format	Column	FORTTRAN name	Description
1	3(I1, 2X), 1X	1	ID	Body number.
	7 A4, 5X,	4	ISV	Flag to control the saving of geometry data for future use again by the Two-Dimensional Program. = 0 Do not save data. = 1 Save the input geometry data for future use.
	9(I1, 2X), 1X, I1	7	ILIFT	Lift control flag. ILIFT = 1-ANLF (from SCIRCL) = 0 This is not a lifting body. = 1 This is a lifting body.
		11-38	TTILE	Body description
		44	IPARA	Element geometry flag = 0 Linear elements = 1 Parabolic elements

Card	Format	Column	FORTTRAN name	Description
		47	IFIRST	First-order terms flag. = 0 No first-order terms = 1 First derivative term = 2 Curvature term = 3 Both first-order terms
		50	ISEC	Second-order terms flag. = 0 No second-order terms = 1 Second derivative term = 2 Curvature squared term = 3 Both second-order terms
		53	ITR	Geometry transformation flag. = 0 Transformation card will not be input. = 1 Geometry transformation card will be input. = 2 Ellipse generation. Ellipse generation card will be input. Transformation card will not be input. = 3 Ellipse generation. Ellipse generation card will be input. Transformation card will be input.

Card	Format	Column	FORTTRAN name	Description
		56	INORM	<p>Geometry normalization flag.</p> <p>= 0 Geometry will not be normalized</p> <p>= 1 All of the geometry data (X and Y) will be divided by the chord length before use by the potential flow program.</p>
		59	IBOD	<p>Body disposition flag. This flag together with the IDOLD parameter controls the sequence of shapes that are to be presented to the potential flow analysis part of the program. With the use of these two flags and the ISV parameter it is possible to perform a variety of multi-element analysis problems with a minimum of input data. For normal useage when all the geometry data are input only the IBOD = 1 and =2 inputs are used.</p> <p>1 New geometry is being input. The storage of geometry data for the potential flow solution will start with this body.</p>

Card	Format	Column	FORTTRAN name	Description
				= 2 New geometry is being input but this is not the first body. This body will be added to the sequence of body data already input.
				3 New geometry is being input but it is to be added to an old sequence of data
				= 4 All previously saved geometry will be used.
				= 5 The geometry for this body will be selected from the previously saved data (body IDOLD will be selected). This selected body will be added to the current string of bodies
				= 6 Previously saved geometry data will be used with the body number indicated by the IDOLD parameter removed from the solution
		62	IDOLD	Old body ID number. This parameter is used in conjunction with the IBOD parameter in selecting which previously saved shape is to be retrieved as the present body

Card	Format	Column	FORTTRAN name	Description
		65	IPVOR	Vorticity distribution flag. = 0 Use constant vorticity between body elements. = 1 Use variable vorticity distribution
		68	LAST	Last body flag. = 0 This is not the last body. After this body is input the program will return to read another Body Title and Control Card for the next body. = 1 This is the last body.
		72	ITYPE	= 1
2	This card is input only when ITR = 1 or 3.			
	7 (F8.0, 1X), F8.0, I1	1 - 8	CHORD	Body chord length. If input as = 0.0 the program will calculate CHORD. This parameter used only when INORM = 1.
		10-17	XMULT	X-coordinate multiplier.
		19-26	YMULT	Y-coordinate multiplier.
		28-35	DX	Delta-X increment applied to the X-coordinates.
		37-44	DY	Delta-Y increment applied to the Y-coordinates.

Card	Format	Column	FORTTRAN name	Description
		46-53	THETA	Angle of rotation (positive for a counterclockwise rotation).
		55-62	XTO	X-coordinate of the center of rotation.
		64-71	YTO	Y-coordinate of the center of rotation.
		72	ITYPE	- 2
Note: These transformations are applied in the following order.				
1. Rotate by angle THETA.				
2. Shift by DX and DY.				
3. Apply multiplier factors, XMULT/CHORD, and YMULT/CHORD.				
3	6F10.0, 4X, I1, 2X, I1, 3X, I1	1-10	X(1)	X-coordinate of the geometry. Up to six points may be input on each card depending upon how the INO flag is set
		11-20	X(2)	
		21-30	X(3)	
		31-40	X(4)	
		41-50	X(5)	
		51-60	X(6)	
		65	INO	Number of data points per card. If there are 6 values per card, INO may be left blank.
		68	ISTAT	Last card flag. 0 This is not the last X-coordinate card. More cards will follow. 1 This is the last X-coordinate card.
		72	ITYPE	3

Card	Format	Column	FORTTRAN name	Description
4	6F10.0, 4X I1, 2X, I1, 3X, I1	1-10	Y(1)	Y-coordinate of the geometry. Up to six points may be input on each card depending upon how the INO flag is set.
		11-20	Y(2)	
		21-30	Y(3)	
		31-40	Y(4)	
		41-50	Y(5)	
		51-60	Y(6)	
		65	INO	Number of data points per card. If there are six values per card, INO may be left blank.
		68	ISTAT	Last card flag. = 0 This is not the last Y-coordinate card. More cards will follow. = 1 This is the last Y-coordinate card.
5	2X, I3, 5X F10.0, 51X, I1	72	ITYPE	= 4
		This card is input only when ITR = 2 or = 3.		
		3-5	LX	Number of points to be generated.
		11-20	ELPSTH	Ellipse thickness ratio.
		72	ITYPE	= 5

Note: At this point in the input the geometry for one body has been read in and stored. If another body is to be loaded (i.e., a multiple element case) and the LAST parameter on the Body Title and Control Card was = 0, then another Body Title and Control Card will be expected next. If LAST = 1 then the geometry load has been completed and the program execution cards will be expected next.

Two-Dimensional Program Execution Cards

All of the following cards are associated with the potential flow solution and are input after the geometry data has been loaded. For most normal problems only the Flow Title Card and the Flow Control Card are required to complete the analysis. All of the other flow execution cards are associated with other options normally not used such as non-uniform flow.

Card	Format	Column	FORTTRAN name	Description
6	15A4, 11X, I1	1-60	FTITLE	Title or description of case
		72	ITYPE	8
7	I1, 4X, F10.5, 2X, I1, 2X, F10.5, 5(4X, I1), 9X, I1, 4X, I1, I2, I2	1	INCLT	C_L - α flag 0 Angle of attack, α , is input. = 1 Total lift coefficient, C_L is input.
		6-15	CLT	Value of angle of attack or lift coefficient depend- ing upon how the INCLT flag was set.
		18	ICHORD	Reference length flag. = 0 The reference length used in calculating the C_L is set = 1 0 1 The reference length used in calculating the C_L will be input as the CCL parameter
		21-30	CCL	The input value for the reference length (chord) used in calculating the C_L

Card	Format	Column	FORTTRAN name	Description
		35	IND	Print control flag. = 0 Minimum print of output solutions = 1 Print the individual solutions (0^0 , 90^0 , Γ , etc.).
		40	ISOL	Matrix solution method control flag. = 0 Use routine SOLVIT for the matrix solution (used when a very large number of geometry points have been input). = 1 Use routine QUASI for the matrix solution. = 2 Use routine MIS1 for the matrix solution. Maximum number of geo- metry points is = 101. If number of points is great- er than 101 the program will automatically shift to use SOLVIT.
		45	IOFF	Off-body calculation flag. = 0 Off-body points will not be calculated. = 1 Off-body points will be calculated.
		50	NONU	Non-uniform flow flag. = 0 Non-uniform flow is not input.

Card	Format	Column	FORTTRAN name	Description
				<p>$\neq 0$ Non-uniform flow will be input. The number of flows input is = NONU (maximum of 6 permitted). When this option is used the program automatically sets the parameter ISOL = 1.</p>
		55	NBNU	<p>The number of bodies for which the non-uniform flows are input.</p>
		65	IPRINT	<p>Print/punch flag.</p> <p>= 0 Normal output.</p> <p>= 2 Print the individual matrices.</p> <p>7 Punch the output on cards.</p>
		70	MORE	<p>Last case flag.</p> <p>= 0 This is the last solution case.</p> <p>- 1 This is not the last solution case. Another set of Flow Title and Flow Control Cards (and any non-uniform or off-body cards) will be expected next after this case is completed</p>
		71	IFLLL	<p>0 No combination solution calculated</p>

Card	Format	Column	FORTTRAN name	Description
				= 1 Combination solution calculated
		72	ITYPE	= 9
8	6F10.0, 10X, I2	1-10	CNU(1)	Combination constant for the first non-uniform flow.
		11-20 etc.	CNU(2)	Combination constant for the second non-uniform flow. etc.
		51-60	CNU(6)	Combination constant for the sixth non-uniform flow.
		71-72	ITYPE	= 10
9	I1, 1X, 3I1 10(2X, I3) F10.5, 5X, I2	This card is input only when NONU \neq 0.		
		1	IBOD	Body ID number for which non-uniform flow is input
		3	IN	Normal velocity input flag. = 0 Normal velocities are not input. = 1 Normal velocities are input.
		4	IT	Tangential velocity input flag = 0 Tangential velocities are not input

Card	Format	Column	FORTTRAN name	Description
		5	NN	= 1 Tangential velocities are input.
		8-10	I1(1)	I1 and IE specify the element range for which the non-uniform flow is input. I1 is the first element of the range and IE is the last ele- ment of the range. Up to five ranges are per- mitted per body.
		13-15	IE(1)	
		18-20	I1(2)	
		23-25	IE(2)	
		28-30	I1(3)	
		33-35	IE(3)	
		38-40	I1(4)	
		43-45	IE(4)	
		48-50	I1(5)	
		53-55	IE(5)	
		56-65	CB	Flow scaling coefficient for this body.
		71-72	ITYPE	- 11
10	This card is input only when NONU \neq 0 and IN = 1. One set of these cards must be present for each body for which non-uniform flow is to be input (when NONU \neq 0 and IN = 1). The number of data points in each set is determined by the number of element ranges input and by the number of elements in each element range.			
	6F10.0, 10X, I2	1-10	VNUF(I)	Input values of non- uniform normal veloc- ity. Input six values per card.
		11-20	VNUF(I+1)	
		21-30	VNUF(I+2)	
		etc.		
		51-60		
		71-72	ITYPE	12
11	This card is input only when NONU \neq 0 and IT = 1. One set of these cards must be present for each body for which non-uniform flow is to be input			

(when $\text{NONU} \neq 0$ and $\text{IT} = 1$). The number of data points in each set is determined by the number of element ranges input and by the number of elements in each element range.

Card	Format	Column	FORTTRAN name	Description
12	6F10.0, 10X, I2	1-10	VTUF(I)	Input values of non-uniform tangential velocity. Input six values per card.
		11-20	VTUF(I+1)	
		21-30	VTUF(I+2)	
		etc.		
		51-60		
		71-72	ITYPE	= 13

Note: The cycle of input for the cards that follow the Flow Control Card is controlled by the NONU and NBNU parameters. The order of cards is as follows:

Flow Control Card (NONU and NBNU are input)

Non-Uniform Flow Coefficients

Non-Uniform Flow Body Card

Non-Uniform Normal Velocities

Non-Uniform Tangential Velocities

these cards are
cycles for
NBNU bodies

these cards
are cycled
for NONU
flows

Card	Format	Column	FORTTRAN name	Description
12	The following cards related to off-body calculations are input only if IOFF = 1.			
	I1, 9X, 7 A4, 12X, 2(2X, I1), 2(5X, I1), 2X, I2	1	ID	Identification number for this group of off-body points. Off-body points are read in groups of up to 100 maximum at a time. There is no limit on the number of groups.
		11-38	TITLE	

Card	Format	Column	FORTTRAN name	Description
		53	ITR	Coordinate transformation flag. See the ITR parameter on card 1.
		56	INORM	Coordinate normalization flag. = 0 Off-body coordinates will not be normalized = 1 Normalize the coordinates by the input chord or by the chord for body with ID = IDOLD.
		62	IDOLD	Body selection flag for normalizing off-body points. = 0 Use input chord (Type 22 Card) to normalize off-body points. ≠ 0 Use chord for body with ID = IDOLD to normalize the off-body points.
		68	LAST	Off-body group termination flag. 0 Additional group(s) of off-body points will be read in after this group is completed. 1 This is the last group of off-body points.
		71-72	ITYPI	21

Card	Format	Column	FORTTRAN name	Description
13	This card is input if ITR = 1 or = 3 on the Off-body Title and Control Card.			
	7 (F8.0, 1X), F8.0, I1			
		1-8	CHORD	Body chord length This parameter used only when INORM = 1 and IDOLD = 0
		10-17	XMULT	X-coordinate multiplier.
		19-26	YMULT	Y-coordinate multiplier.
		28-35	DX	Delta-X increment applied to the X-coordinates.
		37-44	DY	Delta-Y increment applied to the Y-coordinates.
		46-53	THETA	Angle of rotation (positive for a counterclockwise rotation).
		55-62	XTO	X-coordinate of the center of rotation.
		64-71	YTO	Y-coordinate of the center of rotation.
		72	ITYPE	= 2
14	6F10.0, 4X, I1, 2X, I1, 3X, I1	1-10	X(1)	X-coordinate of the off-body point. Up to six points may be input on each card, depending upon how the INO flag is set.
		11-20	X(2)	
		21-30	X(3)	
		31-40	X(4)	
		41-50	X(5)	
		51-60	X(6)	

Card	Format	Column	FORTTRAN name	Description
15	6F10.0, 4X, I1, 2X, I1, 3X, I1	65	INO	Number of data points per card. If there are six values per card, INO may be left blank.
		68	ISTAT	Last card flag = 0 This is not the last X-coordinate card. More cards will follow. = 1 This is the last X-coordinate Card.
		72	ITYPE	= 3
		1-10	Y(1)	Y-coordinate of the off- body point. Up to six points may be input on each card depending upon how the INO flag is set.
		11-20	Y(2)	
		21-30	Y(3)	
		31-40	Y(4)	
		41-50	Y(5)	
		51-60	Y(6)	
		65	INO	Number of data points per card. If there are six values per card, INO may be left blank.
		68	ISTAT	Last card flag. 0 This is not the last Y-coordinate card. More cards will follow. 1 This is the last Y-coordinate Card.
		72	ITYPE	4

Card	Format	Column	FORTTRAN name	Description
16	This card is input only when ITR = 2 or = 3 on the Off-Body Title and Control Card.			
	2X, I3, 5X, F10.0, 51X, I1	3-5	LX	Number of points to be generated.
		11-20	ELPSTH	Ellipse thickness ratio.
		72	ITYPE	5

24Y OUTPUT

Printed Output

For each body the following data are output: UNTRANSFORMED COORDINATE DATA FOR BODY xx and ELEMENT COORDINATE DATA FOR BODY xx. The untransformed data are the geometric points which were input from SCIRCL. For the element data, the midpoint of each segment is calculated. Also printed are the linear distance and surface distance between each segment. For each element, the sine, cosine, and local curvature are also output. Following the element data is the Antal surface distance for that body.

After the input and transformed coordinates are output a BODY GEOMETRY SUMMARY table is output. The following is output in the summary:

BODY DESCRIPTION	Title of test case
BODY ID	Body number
LIFT TYPE	= Yes, vorticity solution about this body is calculated = No, no vorticity solution calculated
N/O	New or old body data
SID	= S, body geometry data saved
T FORM	= Yes, body coordinates have been transformed (see ITR in input) = No, body coordinates have not been transformed

NORM	= Yes, body coordinates have been normalized (see INORM in input) = No, body coordinates have not been normalized
CHORD	If INORM = 1, the CHORD is the body chord length. If INORM = 0, CHORD = 000
TYPE	Type of solution to be generated. = L, only linear terms = P, parabolic terms
SIGMA	If TYPE = P, SIGMA describes which parabolic terms considered
F	= 1, First derivative term
C	= 1, Curvature term
S	= 1, Second derivative term
C	= 1, Curvature squared term
ELEMENT STORAGE	
N/O	See N/O above
FIRST	Number of first element for given body
NO	Number of elements for given body.
Total Number of Bodies	
Total Number of Elements	Self explanatory

Following the GEOMETRY SUMMARY table is a table of the vorticity for each element. If IPVOR does not equal zero then a variable vorticity for each element is calculated. If IPVOR equals zero then the vorticity is a constant for each element.

Following the vorticity functions is a printout of the combination coefficients used internally by 24Y for its version of a combined solution based upon input alpha or total lift if INCLT = 1.

After the combination coefficients of INDIVIDUAL FLOW SOLUTIONS for each body are printed. The first solution is axial flow and the second individual solution is for crossflow. There is also an individual solution based upon each body

which is a lifting body. Each individual solution contains the following:

POINT NUMBER	Element number
VN	Normal velocity at the point
VT	Tangential velocity at the point
SIGMA	Source strength density

The individual solutions are also printed on a disc file to be used as input to NOZZLEC as basic solutions.

The next section of output is the COMBINED SOLUTION if IFILL = 1. This solution is the result of the combination of individual solutions into a solution which satisfies the input conditions of either alpha or total lift. Also included are the integrated values of lift for each body.

If there are off-body points specified in the input, the next section is the individual solutions at each off-body point. These solutions correspond to those for the on-body points. Also included is a combined solution for the off-body points (if IFILL = 1) using the same combination coefficients as those for the on-body points.

NOZZLEC Input

English engineering units are used throughout the program.

Length, in.
 Velocities, ft/sec
 Angles, deg
 Pressure, lb/ft²
 Temperatures, °R
 Densities, slug/ft³
 Force, lb
 Weight flow, lb/sec

Card	Format	Column	FORTTRAN name	Description
1	3A6	1-18	TITLE	Title card
2	9I4	1-4	NT	Total number of on-body points. [NT(3) from

Card	Format	Column	FORTTRAN name	Description
				SCIRCL (3 body case), or NT(2) from SCIRCL (2 body case)] .
		5-8	NS1	Number of on-body points on body 1. (NT(1) from SCIRCL).
		9-12	NH	Number of on-body points on bodies 1 and 2 (NT(2) from SCIRCL) If there are only 2 bod- ies, NH - number of on- body points on body 1 only. (NT(1) from SCIRCL).
		13-16	NP	Total number of off-body points. (NP from SCIRCL).
		17-20	IW	Flag for type of input through the control sta- tion: 0, weight flow 1, Mach number 2, velocity
		21-24	NX	1, apply supersonic velocity correction to data
		25-28	KND	Flag for scaling variables. All input lengths are divi- ded by ELND If KND -1, ELND YCU 0, FLND 1 1, FLND YCU - YCL

Card	Format	Column	FORTTRAN name	Description
				= 2, ELND= the read-in value from card 4.
		29-32	ICOMP1	Type of solution to be computed: = 0, compressible = 1, incompressible
		33-36	IHUB	IHUB = 0, 2 body case ≠ 0, 3 body case
3	10F8.0	1-8	VC	Average axial velocity at the control station be- tween bodies 1 and 3 up- stream of the body 2.
		9-16	VS1	Average axial velocity at the control station be- tween bodies 1 and 2.
		17-24	VS2	Average axial velocity at the control station be- tween bodies 2 and 3. Note if IW=2, then two of the three velocities must be input for the two pas- sage case
		25-32	VINF	Free stream velocity
		33-40	ALFA	Angle between free stream velocity and X-axis of the nozzle
		41-48	MC	Average Mach number at the control station be- tween bodies 1 and 3.
		49-56	MC1	Average Mach number at the control station be- tween bodies 1 and 2.

Card	Format	Column	FORTTRAN name	Description
		57-64	MC2	Average Mach number at the control section between bodies 2 and 3. If IW=1 then two of the three Mach numbers must be input for the two passage case
		65-72	TTOTAL	Total temperature. If TTOTAL = TSTAT=0, then TTOTAL = 518.67 will be used.
		73-80	PT	Total pressure. If PT=0.0 and PSTAT \neq 0.0, the program will calculate PT. If PT = 0.0 and PSTAT=0.0, PT is set to 2116.
4	10F8.0	1-8	ELND	ELND is the arbitrary length used for scaling or normalizing. Refer to KND input.
		9-16	WDOTC	Weight flow at the control station between bodies 1 and 3.
		17-24	WDOTC1	Weight flow at the control station between bodies 1 and 2.
		25-32	WDOTC2	Weight flow at the control station between bodies 2 and 3. If IW=0, then two of the three weight flows must be input for the two passage case.

Card	Format	Column	FORTTRAN name	Description
		33-40	PSTAT	Static pressure
		41-48	TSTAT	Static temperature. If PSTAT and TSTAT are not 0.0, total tem- perature and total pres- sure will be calculated using PSTAT and TSTAT.
		49-56	CUTOF1	If CUTOF1 \neq 0, then the pressure ratio P_S/P_T on body 1 will be plotted against a dimensionless surface distance $S/CUTOF1$ starting at $X = XR1$ for a distance of $S = CUTOF1$.
		57-64	CUTOF3	Same as CUTOF1 except for body 3.
		65-72	CUTOF2	Same as CUTOF1 except for body 2.
		73-80	VPERIN	If VPERIN is greater than zero then a CALCOMP plot of the nozzle showing rake point flow field data will be produced. VPERIN is the value of a unit vector in ft/sec/in. If VPERIN is not equal to zero, card '4a' must be input.
4A	6F10.0	1-10	XX	The length in plot inches of the abscissa of the vel- ocity plot.
		11-20	XMIN	Value, in data inches, of far left X-point.

Card	Format	Column	FORTTRAN name	Description
		21-30	EXEP	Data inch per plot inch along X-axis.
		31-40	YY	Length, in plot inches, of the ordinate.
		41-50	YMIN	Value, in data-inches, of bottom Y point.
		51-60	ORD	Data inch per plot inch along Y-axis.
5	3F10.0	1-10	XTEST	Axial location of the control station up- stream of body 2 be- tween bodies 1 and 3.
		11-20	YCL	Y on body 1 at XTEST
		21-30	YCU	Y on body 3 at XTEST
6	3F10.0	1-10	XTEST1	Axial location of the control station be- tween bodies 1 and 2.
		11-20	YCL1	Y on body 1 at XTEST1.
		21-30	YCU1	Y on body 2 at XTEST1.
7	3F10.0	1-10	XTEST2	Axial location of the control station between bodies 2 and 3.
		11-20	XCL2	Y on body 2 at XTEST2
		21-30	YCU2	Y on body 3 at XTFST2
Note: Cards '6' and '7' are not used for a single-passage case.				
8	3F10.0	1-10	XR1	Axial location on body 1 where $S = 0$.
		11-20	XR3	Axial location on body 3 where $S = 0$.

Card	Format	Column	FORTTRAN name	Description
		21-30	XR2	Axial location on body 2 where $S = 0$. Leave blank for two body case.
9	3F10.0	1-10	YR1	Y on body 1 at XR1.
		11-20	YR3	Y on body 3 at XR2.
		21-30	YR2	Y on body 2 at XRH.

Figure 15 is the input form for NOZZLEC.

NOZZLEC Output

Printed Output

Input file dump

TITLE - followed by 2-D COMBINATION SOLUTION

Version of run (i.e., compressible or incompressible)

A list of the basic flow solutions obtained from 24Y

In the table that follows, several functions of four different velocities are given.

The velocities are:

Control: VC, average axial velocity at upstream control station

Lower passage: VS1, average axial velocity at control station between bodies
1 and 2

Upper passage: VS2, average axial velocity at control station between bodies
2 and 3

Free stream: V_∞ , free stream velocity

The rest of the table is self-explanatory perhaps with the exception of the terms INC and COMP. INC means calculated from the incompressible equations and COMP means calculated from the compressible equations. The rest of the output will be defined by its name.

ALPHA Angle of attack of nozzle

VINF/VC V_∞/V_c

VSONIC Critical velocity uncorrected for compressibility

VSONICC Critical velocity

WDOTCR	Corrected weight flow at upstream control station $\frac{WDOT \times \sqrt{THET}}{DEL}$
WDOTLCR	Corrected weight flow at control station between bodies 1 and 2
WDOTUCR	Corrected weight flow at control station between bodies 2 and 3
TSTAT	Free stream static temperature
PSTAT	Free stream static incompressible pressure
PSTATC	Free stream static compressible pressure
ASTAT	Free stream static speed of sound
RHOSTAT	Free stream static density
WDOTC	Input mass flow at upstream control station
WDOTL	Input mass flow at downstream lower control station
WDOTU	Input mass flow at downstream upper control station
VIC	Incompressible average velocity at upstream control station
VICL	Incompressible average velocity at downstream lower control station
VICU	Incompressible average velocity at downstream upper control station
TTOT	Free stream total temperature
PTOT	Free stream total pressure incompressible
PTOTC	Free stream total pressure compressible
ATOT	Free stream stagnation speed of sound
RHOTOT	Free stream stagnation density
THET	$TTOT/518.67$
DEL	$PTOTC/2116.22$
XRI1	Input
YRI1	Input

XRI2	Input
YRI2	Input
XTEST	Input
YCL	Input
YCU	Input
LND	Length used for scaling
XTEST1	Input
YCL1	Input
YCU1	Input
XTEST2	Input
YCL2	Input
YCU2	Input
P - S CUTOFF1	Input
P - S CUTOFF2	Input
P - S CUTOFF3	Input
NT	Input
NP	Input
NS1	Input
NH	Input
KND	Input
IW	Input
NX	Input
ICOMP1	Input
IHUB	Input
V1	Average axial velocity at specified control station for basic solution 1 from 24Y
V2	Same as V1 except for basic 24Y solution 2
V ₃	Same as V1 except for basic 24Y solution 3

V4 Same as V1 except for basic 24Y solution 4

V5 Same as V1 except for basic 24Y solution 5

24Y basic solutions 1 and 2 are axial and crossflow solutions. For a three-body case, solutions 3, 4, and 5 are vorticity solutions about bodies 1, 2, and 3, respectively. For a two-body case, solutions 3 and 4 are the vorticity solutions about bodies 1 and 3. Note there is no 5 solution for this case.

A, B, C, D Coefficients of combination

VINFP Incompressible free stream velocity "uncorrected" for compressibility if the input value was compressible

OTHER MESSAGES: "VRESN = _____ IS GREATER THAN VMAX.
VCONC = _____. " The velocity at a certain on-body point exceeds the allowable value for the local expansion condition so that the isentropic ratio term:
1. - VCONC is less than zero. Where,

$$VCONC = \frac{\gamma - 1}{2} \left(\frac{VRESN}{a_{tot}} \right)^2.$$

"I EXCEEDS 20 ITERATIONS FOR RHOBAR.

VBAR = _____, VCOMP = _____, RHOBAR = _____. VBAR HAS BEEN REDUCED TO VCOMP * RHOBAR/RHOTOT". Subroutine VBARIT attempts to find the average density at each axial location using the isentropic density ratio, the stagnation density (RHOTOT) and the average incompressible VBAR (based on weight flow and the cross section). It has failed. VCOMP is the 20th attempt at finding the compressible velocity and has been used to compute the RHOBAR that will be returned. The normally unchanged VBAR is adjusted to agree with these abbreviated results.

For NX = +1, supersonic velocity correction is operating and a message to that effect will appear each time a region of local supersonic flow is encountered on the body, and also when it ends. The body point number where these transitions occur will also be printed.

ON-BODY POINTS

Body 1, 2, or 3

For both compressible and incompressible versions:

I	The index number of the point
X	Axial distance
Y	Height
S	Surface distance from XRI

Compressible Version

VCOM	Resultant velocity with compressibility correction applied
VBAR	Average incompressible velocity at a given axial location
MACH	Mach number

CP	Compressible pressure coefficient $\left(\frac{p_i - p_o}{q_o} \right)$
----	--

RB/RT	$\bar{\rho}_c / \rho_t$
-------	-------------------------

PS/PT	Static to total pressure ratio, $\left[1 - 0.2 \left(\frac{V}{a_t} \right)^2 \right]^{3.5}$
-------	---

Incompressible Version

VINC	Resultant incompressible velocity
MACH	Mach number (based on incompressible velocity and compressible flow equations)

CP	Incompressible pressure coefficient $\left[1 - \left(\frac{V_i}{V_o} \right)^2 \right]$
----	---

PS/PT	Static to total pressure ratio, $1.0 - \frac{1/2 \rho V^2}{P_t}$
-------	--

OFF-BODY POINTS

For both incompressible and compressible versions:

Rake number

I	Number of the point (points without numbers are interpolated values at a body)
X	Axial location
V	Vertical location
THETA	Flow angle, $\tan^{-1} \left(\frac{VY}{VX} \right)$
MACH	Mach number
WFRACT	Local cumulative weight flow at a given point on a rake divided by total weight flow at the rake

Compressible Version

VX	Axial velocity corrected for compressibility
VY	Vertical velocity corrected for compressibility
VRE	Resultant velocity = $\sqrt{VX^2 + VY^2}$
VBL	Average incompressible velocity at given axial location
RB/RT	$\bar{\rho}/\rho_t$
PS/PT	Same as ON-BODY POINTS

Incompressible Version

VX	Incompressible axial velocity
VY	Incompressible vertical velocity
VRE	Resultant velocity = $\sqrt{VX^2 + VY^2}$
PS/PT	Same as ON-BODY POINTS

RAKE WEIGHT FLOW DATA

For each rake the following data are given:

I	Number of rake
X	Axial location of rake
IRAK	Parameter to describe location of rake on the inlet = 1; upstream of nozzle = 3; upstream of body 2 between bodies 1 and 3 = 4; downstream of body 2 highlight between bodies 1 and 2 = 5; downstream of body 2 highlight between bodies 2 and 3
WDOT	Integrated weight flow for rake
WDOTCA	Specific corrected weight flow at each rake
MACH	One-dimensional Mach number at each rake based on WDOTCA

Graphic Output

Standard CALCOMP plots of PS/PT (fig. 9) and Mach number (fig. 10) distribution against S/CUTOF1, S/CUTOF2, or S/CUTOF3 are made for any body whose value of cutoff is not zero.

If VPERIN not equal to zero, then a plot will be made of the velocity flow field at the off-body points (see fig. 11). At each off-body point a vector will be drawn showing flow angularity and magnitude. Magnitude will be shown by relative size of the vectors (a unit vector will have a magnitude of VPERIN).

DESCRIPTION OF SUBROUTINES

Figure 16 illustrates the calling relations between the main program and their subroutines.

Program SCIRCL

(A) MAIN SCIRCL	Read all input, call required subroutines for each segment as requested, plot each segment after points are generated by subroutine; list points; test for reworking of geometry if required
-----------------	--

Straight Lines

- | | |
|------------|---|
| (B) STRAIT | Generate points on a general straight line segment |
| (C) FNSTRH | Generate points on final straight segment of a body |
| (D) FRSTSH | Generate points on first straight segment of a body |

Bisuper Ellipses

- | | |
|------------|---|
| (E) TEST | Test superellipse input to see if mirroring about y-axis is required |
| (F) PRELPS | Mirror superellipse input data about y-axis so that slope (1, 2) is greater than the slope (1, 4) (fig. 12 (a)) |
| (G) SUPERC | Generated points on a general bisuperellipse (table I) |
| (H) FONISØ | Iterate on input conditions to find bisuperellipse exponents |

Other Curves

- | | |
|------------|---|
| (I) CUBIC | Fit a cubic polynomial between two nonvertical parallel lines |
| (J) SIMQ | Simultaneous solution of equations to obtain coefficients of the cubic polynomial |
| (K) LEM | Generates points on a general Lemniscate |
| (L) MIRROR | Mirror the hub points to obtain the shroud |

Direct Interpolation

- | | |
|------------|---|
| (M) XYCALC | Executive routine for the following modules purpose is to generate points "correctly" spaced along the curve defined by a list of input points. Inputs are used to develop double 3-point interpolating polynomial in successive regions along the curve. Polynomials are then used to suggest points, derivatives, etc. which can be tested for spacing as defined by standard criterion (see comments in SPGEN listing) |
|------------|---|

(N) SGEN
 (O) DSTRP
 (P) SPGEN
 (Q) DNTRPC
 (R) FNTRP
 (S) FSTRP
 (T) FNTRPC
 (U) FNTRPA
 (V) TLU
 (W) LIMIT

Refer to listing for comment card description

Special Calculations; Output to 24Y

(X) WPUNCH Generate rake points at requested positions; plot rakes
 (Y) WRTXY Write all X, Y coordinates
 (Z) AREAA Compute area

Picture Plotting

(Z1) DRAW Plot X-Y meridional plane picture of each inlet segment
 (Z2) PLOXIS Plot frames for inlet picture and label axis

Utility

(Z3) SINTP Lagrange three-point interpolation
 (Z4) SORTXY Rearrange the values in an array, x, to increase with
 increasing index (ascending order); sort y accordingly

System Library

ERTRAN Routine which gives FORTRAN access to several
 UNIVAC 1110 operating system commands. Can be
 eliminated by defining unit 25 before execution

SYMBOL
SCALE
LINE
NUMBER
PLOT

Standard CALCOMP routines needed for all plotting

Program 24Y

- | | |
|-------------|--|
| A) MAIN 24Y | Executive calls to MAIN1, SOLVE, and MAIN3 also, this program rewinds all disc files used. |
| B) ABFORM | This subroutine calculates induced velocities array elements in arrays A and B. |
| C) ASSEMB | This subroutine reads in normal and tangential onset flows in row order and assembles them in column order. Also it reads in non-uniform onset flows if any. |
| D) COMBO | This subroutine calculates combination coefficients if required. |
| E) ELFORM | This subroutine reads in body coordinates and associate data and calculates element data. |
| F) FILES | This subroutine assigns temporary file numbers. |
| G) FLOWS | This subrouting calculates individual and combined flow for on-body points. It also writes the individual flows on a tape file for input to NOZZLEC. |
| H) GEOMCF | This subroutine calculates required geometry combination coefficients associated with parabolic distribution. |
| I) MAFORM | This subroutine forms and stores matrices A and B and calculates alpha and circulatory onset velocities. |
| J) MAIN1 | This subroutine calls subroutine ELFORM and MAFORM and reads in the flow control parameters. |
| K) MAIN3 | This subroutine calls COMBO (if required), FLOWS, and OFFBOD. |

L) MIS1	This subroutine inverts a matrix and solves simultaneous equations. This is used when ISOL = 2 and number of points less than 102. If more than 101 points are input program shifts to SOLVIT.
M) OFFBOD	This subroutine calls OFFPTS, VXYOFF, and VPROFF.
N) OFFPTS	This subroutine reads in the off-body points and prints them back out.
O) PRINTG	This subroutine writes out the body coordinate data.
P) PRNTEL	This subroutine writes out the body summary data.
Q) QUASI	This subroutine performs direct matrix solution. Program uses this to solve matrix when ISOL = 1.
R) RMAX	This subroutine calculates maximum radius or chord.
S) SOLVE	This subroutine obtains the sigma solutions from QUASI.
T) SOLVIT	This subroutine performs direct matrix solution (used when a very large number of points have been input).
U) TYPE	This subroutine checks input card type to verify proper data card is being input.
V) VXYOFF	This subroutine calculates the individual and combined solutions for each off-body point.
W) VPROFF	This subroutine prints out the individual and combined solutions (in blocks of 100) for each off-body point.
X) WEIGHT	This subroutine calculates the vorticity weighting functions for each element.
Y) XYFORM	This subroutine calculates the indexed velocity array elements A and B.

Program NOZZLEC

(A) MAIN NOZZLEC	Executive calls to INPTR, SEARCH, ANGLEF, SOLVE, OFBDY, and if compressibility correction desired COMCOR. Also, this program calls PLTER if output plots are desired
------------------	--

(B) INPTR	Reads input parameters from unit 5. Also reads 24Y output coordinates and basic flow solution velocities.
(C) CONST	Calculated most constants and intermediate parameters and prints results.
(D) SEARCH	Finds the highlight on each body.
(E) ANGLEF	Finds the body surface angle for each point on the body.
(F) SURF	Calculates surface distance along body as a function of X.
(G) SOLVE	Computes linear combination coefficients A, B, C, and D to satisfy input flow conditions. Also calculates the incompressible velocities and average velocity \bar{V}_1 (used in compressibility correction) for each point.
(H) COMCOR	Applies compressibility correction.
(I) ONBODY	Uses velocities and densities to calculate: pressure ratios, Mach number, flow angles and list all results for points on the body.
(J) OFBDY	Same as ONBODY for rakes. Also calculates local fractional weight flows for each rake point.
(J) VBARIT	Calculates average density ratio for compressibility correction.
(L) INTER	See SCIRCL routine SINTP (duplicate)
(M) SORTXY	See SCIRCL routine SORTXY (duplicate)
(N) INTER2	Calls INTER
(O) INTER3	Calls INTER
(P) INTEG	Performs trapezoidal integration
(Q) CALTIT	Titles plots of pressure and Mach number versus surface distance.
(R) PLTER	Plots pressure and Mach number versus surface distance.

INPUT/OUTPUT UNITS

SCIRCL	5 - Standard card input
	6 - Standard output list
	4 - Temporary storage; if flag J or E is 1, input to direct interpolation routines is written here
	17 - Saved. If flag A is 0, input for 24Y consisting of X, Y points is written here
24Y	5 - Input (=7 from SCIRCL)
	6 - Standard output list
	7 - Saved. Input for NOZZLEC is written here (6E13.8) X, Y, V1, V2, V3, etc.
	<div style="display: inline-block; vertical-align: middle;"> 2-4 8-18 </div> <div style="display: inline-block; vertical-align: middle; font-size: 3em; margin: 0 5px;">}</div> Temporary storage
NOZZLEC	5 - Standard card input, flow conditions, etc.
	6 - Standard output list
	7 - Input from 24Y
	12 - Saved. Input data (for body 1) for boundary layer program
	13 - Saved. Input data (for body 3) for boundary layer program
	14 - Temporary storage

LISTING OF PROGRAMS

Program SCIBCL

HAWK-SCIBCL(1).SCIBCL

1	C	PREPARE INPUT DATA FOR DOUGLAS POTENTIAL FLOW PROGRAMS 220 AND 234A	0000
2	C		A 0010
3	C		A 0030
4		DIMENSION REEDEN(2),ARE(9),EX(9),CURVO(9),SURFAC(9),MUTSY	A 0040
5		DIMENSION CAPPER(20),DIST(200),RAY(9),TYP(9),JREDON(9)	
6		DIMENSION S(500),NY(25)	
7		COMMON /SPREP/ KPREP,NIN	A 0070
8		COMMON /MNSD/ MNSD,MNSBOY(10)	A 0080
9		COMMON /NHIGH/ MSPH,NLAST,XLAST(500),YLAST(500)	A 0090
10		COMMON /MURTE/ IFLAG,NDY4,PROG,TITLE(9),BODIES(4),IDENT,YLO(25),YHA	0100
11		II(25),NDY(25),XRAK(25),MDDPTS(9),NO6,NRAKES	
12		COMMON /FORSS/ IO,DELS,XBK(20),YBK(20),XONI(500),YONI(500),DYDXO(500)	0120
13		II,ALPHA(500),CAPPA(500),SONI(500),PIO180	A 0130
14		COMMON /FOREOD/ IGEOMF,ISIGF,ICURVN,NONEWF,YVORY,ALPHER	A 0140
15		COMMON /SUPP/ IFLD	A 0150
16		COMMON/SEGNO/MSEG,J,ENREED	A 0160
17		COMMON /MAIN/ XIN(10),YIN(10),DELSMX,PIO2,DELS1,IMUB,DELNEW	A 0170
18		COMMON /SS/ NBOY1,MDDV2,TYPBOY,NBDYS,D2TEST	
19		CCCCC THE GENERAL PLOTTING VARIABLES	A 0190
20		COMMON/ LOT/XMIN,YMIN,ORD,EXEP,XX,YY,LPNCNO,IPLOTA,NH	A 0200
21		COMMON/TITL/ TITL(9,6)	A 0210
22		COMMON/YOL/BAGS(15),BAGX(15),ZAP(15),NZAP(15)	A 0220
23		COMMON/SENSE/ X(2),Y(2),A,B	A 0230
24		COMMON/PAC/ PACE,DELSHL	A 0240
25		COMMON/BURLEY/ ELREF,AREF	
26		COMMON/OMVORT/ HLF(10)	
27		LOGICAL TRANSX,TRANSY,SCALEX,SCALEY,CHANGE	
28		COMMON/TRANST/ XTRAN,YTRAN,XSCALE,YSCALE,	
29		1 TRANSX,TRANSY,SCALEX,SCALEY,CHANGE,XMAX	
30		COMMON/V24V/ IPAR,IFST,ISND,IFLL	
31		DATA D2TEST/6H 234/	
32		DATA BLANK/6H /	A 0250
33		DATA EODFF/6HENDOFF/	A 0260
34		DATA REDONE/6HREDONE/	A 0270
35		PI=3.14159265	A 0280
36		PIO180=PI/180.	A 0290
37		PIO2=PI/2.	A 0300
38	C		A 0310
39	C	WHEN NO6 = 1, A FLAG IN CARD COLUMN 6 IS PUNCHED FOR EOD	A 03
40	C	ONLY BASIC DATA WILL BE GIVEN IN EOD PROGRAM	
41	C		A 0340
42		CALL PLOYID	A 0350
43		CCCCC READ AXIS LABELS FOR THE MOST POPULAR PLOTTED VARIABLES	A 0360
44		READ (5,50)TARE	A 0370
45		READ (5,50)DEX	A 0380
46		READ (5,50)CURVO	A 0390
47		READ (5,50)SURFAC	A 0400
48		CALL ECHO	A 0410
49		10 NIN=25	A 0420
50		READ (25,500,END=E20)TITL	A 0430
51		IF (IFLD(10,36),TITLE(11),FO,FLDI(1,36),EODFF) GO TO 630	A 0440
52		REWIND 4	A 0450
53		LOWF=0	A 0460
54		CCCCC READ GENERAL PLOTTING VARIABLES	A 0470
55		READ(25,505) XX,XMIN,FX(P,YY,YMIN,ORD,ELREF,AREF	A 0480
56		T(1)DEF(10,0,0) FIDFF = T.	

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57 IF (AREF.EQ.0.0) AREF = 1.0
58 READ(25,15) ICFOMF, ISIGF, ICURVN, NONWF, ALPHER, IVORT, IPAR, IFST, ISND
59 I, IFLLL
60 15 FORMAT(4I1,F10.2,5I1)
61 20 READ (IN,40) IDENT, PROG, NO6, LPNCHO, IPLOTA, IPLOTG, IPEAD, IAR, (IPEDOA
62 2N(I), I=1,5)
63 PROG1=BLANK
64 IF (IPROG.EQ.' 23Y') PROG1=' 23Y'
65 IF (IPROG.EQ.' 24Y') PROG1=' 24Y'
66 IF (PROG1.NE.' 23Y'.AND.PROG1.NE.' 24Y') PROG1=PROG
67 IF (IPROG.EQ.' 24Y') PROG=' 23Y'
68 LPDUM=LPNCHO
69 IF (LPDUM.EQ.0) LPNCHO=1
70 IF (LPDUM.EQ.1) LPNCHO=0
71 25 JSTART=0
72 JSTOP=0
73 PACE=0.
74 IF (IAB.LE.0) GO TO 30
75 READ (IN,625) XAA, YAA, XRB, YRB
76 IBUMB=0
77 WRITE (4,495) IDENT, PROG, NO6, (IREDON(I), I=2,4), IBUMB, (IREDON(I), I=2,4)
78 2,4)
79 CIIIIII LEENI WILL CONTAIN THE VALUE OF N AT THE HIGHLIGHT. NEEDED TO SPA
80 CIIIIII THE CURVATURE VS. X PLOTS INTO INTERNAL AND EXTERNAL PORTIONS
81 CIIIIII MM COUNTS THE NO. OF SEGMENTS ON SHROUD AS PLOTTING PROCEEDS
82 30 MM=0
83 LEENI=0000
84 CIIIIII LOAD AXIS LABELS INTO COMMON
85 IF (ININ.EQ.4) GO TO 40
86 DO 35 I=1,9
87 TTITL(I,1)=TITL(I)
88 TTITL(I,2)=BLANK
89 TTITL(I,3)=ARE(I)
90 TTITL(I,4)=EX(I)
91 TTITL(I,5)=CURV(I)
92 35 TTITL(I,6)=SURFAC(I)
93 40 WRITE (6,525) IDENT, TITL
94 IF (ININ.EQ.4) WRITE (6,470)
95 IF (ININ.EQ.4) GO TO 55
96 IF (IVORT.EQ.0) GO TO 50
97 IVORT=0
98 GO TO 55
99 50 IVORT=1
100 55 CONTINUE
101 IF (LPNCHO.GY.0) WRITE (6,631)
102 IF (LPNCHO.EQ.0) WRITE (6,632)
103 WRITE (6,633) PROG1
104 IF (IGOM+ISIGF+ICURVN+NONWF.EQ.0) WRITE (6,634)
105 IF (IGOM+ISIGF+ICURVN+NONWF.NE.0) WRITE (6,635)
106 IF (IVORT.EQ.1) WRITE (6,636)
107 IF (IVORT.NE.1) WRITE (6,637)
108 WRITE (6,638)
109 IF (IPLOTA.NE.0) WRITE (6,639)
110 IF (IPLOTA.EQ.0) WRITE (6,640)
111 IF (IPLOTG.EQ.0) WRITE (6,641)
112 IF (IPLOTG.NE.0) WRITE (6,642)
113 IF (IAR+IREDON(I).NE.0) WRITE (6,643)

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114 IF (IA4+IFLON(1),FL) WRITE (6,544)
115 IF (IA5(XX),GT,100.) WRITE (6,60) XX
116 64 FORMAT(17H, YLEN (XX) = ,F10.2,30H,CT. 100.. CHECK FORMATS,INPUT) A 0014
117 IF (IA5(XX),GT,100.) STOP
118 C(1111) PLOT THE AXES NEEDED FOR THE INLET PICTURE, AND LABEL THE CASE A 091R
119 65 CALL PLOXIS(XX,YY,IXEP,ORD,XMIN,YPIN,,25,,25,0,,0,1,2,1,1) A 0920
120 CALL SYMBOL(-3,YY-.5,,25,IDENT,,6) A 0930
121 CALL SYMBOL(XX-1.5,,.5,,25,IDENT,,6) A 0940
122 IF (NIN,FO,4) CALL SYMBOL(XX/2.,YY-.5,,25,REFDONE,,6) A 0950
123 IF (NIN,LO,4) GO TO 75 A 0960
124 C A 0970
125 C READ INPUT CARDS FOR SUPERCIRCLE A 0980
126 C 1 -- CASE HEADER CARD -- NO. OF BODIES,CASE NO.,DELS,DELSMX A 0990
127 C OFF-BODY A 1000
128 C 2 -- NRAKES = MINMER OF RAKES (TOTAL NUMBER CANNOT EXCEED 25) A 1010
129 C 3 -- X,YLO,YHI, NY (DATA FOR EACH RAKE) A 1020
130 C X = X OF THE RAKE, A 1030
131 C YLO = Y OF THE FIRST PT. ON RAKE CLOSEST TO THE HUB - SHOULD BE A 1040
132 C ABOUT DS GREATER THAN Y ON HUB A 1050
133 C YHI = Y OF THE LAST PT ON RAKE CLOSEST TO THE SHROUD - SHOULD BE A 1060
134 C ABOUT DS LESS THAN Y ON SHROUD A 1070
135 C NY = NO. OF PTS TO GENERATE FOR THAT RAKE A 1080
136 C ON-BODY A 1090
137 C FOR EACH SEGMENT A DESCRIPTION CARD IS NEEDED, A 1100
138 C THIS CARD DENOTES THE TYPE OF LINE, AND THE A 1110
139 C COORDINATES OF THE LINE (UP TO 6 SETS) A 1120
140 C A 1130
141 READ (NIN,555,END=630)ANBODYS,DELS,DELSMX,XRI,ANNSD A 1140
142 HOLYDS=DELS
143 READ (NIN,445)NRAKES A 1150
144 C
145 C RAKE POINTS ARE TRANSFORMED INDEPENDENT OF ONE ANOTHER AND
146 C INDEPENDENT OF BODY TRANSFORMATIONS. ZERO VALUES FOR YLO AND
147 C YHI ARE TRANSFORMED AUTOMATICALLY (BY CALCULATION FROM TRANSFORMED
148 C BODY POINTS) IN SUBROUTINE WPURCH.
149 C
150 DO 67 I=1,NRAKES
151 READ(NIN,550) XRAK(I),YLO(I),YHI(I),NY(I),
152 XTRAN,YTRAN,XSCALE,YSCALE
153 TRANSX = XTRAN .NE. 0.0
154 TRANSY = YTRAN .NE. 0.0
155 SCALEX = XSCALE .NE. 0.0
156 SCALEY = YSCALE .NE. 0.0
157 CHANGE = TRANSX .OR. TRANSY .OR. SCALEX .OR. SCALEY
158 IF (.NOT. CHANGE) GO TO 67
159 IF (.NOT. SCALEX) YSCALE = 1.0
160 IF (.NOT. SCALEY) XSCALE = 1.0
161 C
162 C TRANSFORM THE RAKE POINTS
163 XRAK(I) = XRAK(I) * XSCALE + XTRAN
164 YLO(I) = YLO(I) * YSCALE + YTRAN
165 YHI(I) = YHI(I) * YSCALE + YTRAN
166 CONTINUE
167 WRITE (6,480)ANBODYS,DELS,DELSMX,XRI
168 GO TO 75,NDP=65
169 NY(I)=NY(I)

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A 1170
A 1180
A 1190
A 1200

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171 75 NLAST=0 A 1210
172 MSPH=0 A 1220
173 DSAVE=DELS A 1230
174 DELS2=DELS A 1240
175 DELS1=DELS A 1250
176 MBDS=AMBDYS A 1260
177 MNSD=AMNSD A 1270
178 C A 1280
179 K = COUNTER FOR THE NUMBER OF ONBODY POINTS GENERATED
180 C IFLAG = 0, IF THERE IS MORE THAN ONE BODY A 1290
181 C A 1300
182 K=0 A 1310
183 IMUB=0 A 1320
184 IFLAG=1 A 1330
185 C A 1340
186 C A 1350
187 C
188 C MZ LOOP IS FOR THE NUMBER OF BODIES
189 DO 225 MZ=1,MBDS A 1360
190 IFLR=0 A 1370
191 IF INZ.6E.21 IFLAG=0 A 1380
192 IF INZ.6E.21 IMUB=1 A 1390
193 C A 1400
194 C A 1410
195 READ(ININ,55R) TYPRDY,ANSEG,DELNEW,AMLF,XTRAM,YTRAN,XSCALE,YSCALE A 1420
196 MSG=ANSEG A 1430
197 NCFINZ)=ONLY A 1440
198 NUMBOD=IABS(IFIXITYPBOY)) A 1450
199 WRITE(6,626) NUMBOD A 1460
200 C
201 C SET LOGICALS RELATED TO TRANSLATING OR SCALING INPUT BODY POINTS
202 C
203 CHANGE = .FALSE.
204 TRANX = XTRAM * MF * 0.0
205 TRANSY = YTRAN * MF * 0.0
206 SCALEX = XSCALE * ME * 0.0
207 SCALEY = YSCALE * ME * 0.0
208 IF (TRANX .OR. TRANSY .OR. SCALEX .OR. SCALEY) CHANGE = .TRUE.
209 IF (IARGGT.O) WRITE (4,555)TYPRDY,ANSEG,DELNEW,AMLF
210 IF (DELNEW.LY.O.D)FLS1=HOLYDS
211 IF (DELNEW.GY.O)DELS1=DELNEW
212 K=0
213 IF (TYPRDY.LL.O) GO TO 215
214 SOMIK)=0.0
215 C
216 IF INZ.EO.MROINSEG=INSEG-LOWER
217 C SEGMENT LOOP
218 C
219 DO 200 JET=NSFG
220 C
221 C
222 C
223 C
224 C
225 C
226 C
227 C
228 C
229 C
230 C
231 C
232 C
233 C
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228		READ (4IN,625)(VIN(I),I=1,3),VIN(6),(VIN(I),I=4,5)	A	1750
229	C			
230	C	TRANSFORM POINTS IF CALLED FOR (CHANGE = .TRUE.)		
231	C			
232		IF(CHANGE) CALL XYTRAN(J,NSEG,ENREED,NS)		
233		IF (IAB.GT.0) WRITE (4,625)(XIN(I),I=1,3),XIN(6),(XIN(I),I=4,5)	A	1760
234		IF (IAB.GT.0) WRITE (4,625)(VIN(I),I=1,3),VIN(6),(VIN(I),I=4,5)	A	1770
235	90	NSV=N	A	1780
236		CAPPA(K)=0.0	A	1790
237		IF (ENREED.EQ.1.0.AND.ENREED.NE.10.) GO TO 105	A	1800
238		XON(K)=XIN(I)	A	1810
239		YON(K)=VIN(I)	A	1820
240		WRITE (6,570)ENREED,(XIN(I),I=1,2),(VIN(I),I=1,2)	A	1830
241			A	1890
242		IF (ENREED.EQ.10.,OR.(NUMROD.EE.2.AND.J.EQ.1))CALL FRSTSH(K)		
243		IF (ENREED.EQ.1.0.AND.J.EQ.NSEG) CALL FNSTRM (K)	A	1870
244		IF (K.EQ.NK) CALL STRAIT (K,0.0)		
245	95	DYDXO(KSV)=DYDXO(KSV+1)	A	1920
246		ALPHA(KSV)=ALPHA(KSV+1)	A	1930
247		GO TO 160	A	2020
248			A	1900
249	105	IF(ENREED.LT.-3.0)GO TO 216		
250		IF (ENREED.LT.-2.0) GO TO 150		
251		IF (ENREED.LT.-1.0) GO TO 140	A	2080
252			A	2090
253		IF (ENREED.LT.0.0) GO TO 135	A	2100
254	C		A	2110
255			A	2120
256	C	SET-UP SUPER ELLIPSE	A	2130
257	C		A	2140
258		WRITE (6,85)ENREED		
259		KPREP=0	A	2150
260		ENRD=ENREED-1000.	A	2160
261		IF (ENRD.LT.0.0) GO TO 110	A	2170
262		PAGE=ENREED-1000.	A	2180
263		IF(PAGE.LE.0.)PAGE=.05	A	2190
264		ENPEED=0.	A	2200
265		IFLD=IFLD+1	A	2210
266		GO TO 115	A	2220
267	110	IFLD=0	A	2230
268	115	WRITE (6,575)ENREEDPA(I),(XIN(I),I=1,3),XIN(6),(XIN(I),I=4,5),REEDENA	A	2240
269		1(2),(VIN(I),I=1,3),VIN(6),(VIN(I),I=4,5)	A	2250
270		CALL TEST (5)	A	2260
271		ISTART=N	A	2270
272		KI=N	A	2280
273		CALL SUPERC(XIN,VIN,REEDEN,DELS1,ISTART)	A	2300
274		WEIN=I	A	2310
275		IF (N2.EQ.1) GO TO 125	A	2320
276		KKKK=N-I	A	2330
277		DO 120 JE=ISTART,KKKK	A	2340
278		IF (XON(JE).EQ.XON(JE+1)) GO TO 130	A	2350
279		IF ((DYDXO(JE)-DYDXO(JE+1))/(XON(JE)-XON(JE+1))*CAPPA(JE).LT.0.)CAPA	A	2360
280		IPAT(JE)=CAPPA(JE)	A	2370
281	120	CONTINUE	A	2380
282	125	K2=N	A	2390
283		IDUM=0	A	2400
284		IF (KPREP.EQ.N) GO TO 170		

285	CALL PRELPS (IUM,...,K1,K2)	A	2420
286	GO TO 160	A	2480
287	C	A	2490
288	C SET-UP LEMNISCATE	A	2500
289	C	A	2510
290	135 WRITE (6,560)ENREEL,(YIN(I),I=1,5),(YIN(I),I=1,3)	A	2520
291	CALL LEM (K)	A	2540
292	K=K+1	A	2550
293	GO TO 160	A	2610
294	C	A	2620
295	C SET-UP ELLIPSE	A	2630
296	C	A	2640
297	140 WRITE (6,585)ENREEL,(XIN(I),I=1,4),(YIN(I),I=1,4)	A	2650
298	KPREP=0	A	2660
299	CALL TEST (4)	A	2670
300	K1=K	A	2680
301	CALL ELIPSE (K)	A	2700
302	K=K+1	A	2710
303	K2=K	A	2720
304	IDUM=0	A	2730
305	IF (KPREP.EQ.0) GO TO 160		
306	CALL PRELPS (IDUM,1,4,K1,K2)	A	2750
307	GO TO 160	A	2810
308	C	A	2820
309	C SET-UP CURIC	A	2830
310	C	A	2840
311	150 WRITE (6,605)ENREEL,(XIN(I),I=1,4),(YIN(I),I=1,4)	A	2850
312	CALL CURIC (K)	A	2870
313	K=K+1	A	2880
314	GO TO 160	A	2940
315	CCCCC NEW LINEAR INTEGRATION OPTION, BODY OR FULL INLET	A	2950
316	155 4SV=K	A	2970
317	CAPPA(K)=0.0	A	2980
318	IF (INSEG.EQ.0) DELS1=DELSMX	A	2990
319	CALL XYCALCTR(K2,MIN)	A	3000
320	K=K2	A	3010
321	GO TO 160		
322	216 KTOT=0		
323	KBEGIN=0		
324	KSTOP=C		
325	DO 219 JBOP=1,K		
326	IF (XIN(1).EQ.XON(JBOP).AND.YIN(1).EQ.YON(JBOP)) KBEGIN=JBOP		
327	IF (XIN(2).EQ.XON(JBOP).AND.YIN(2).EQ.YON(JBOP)) KSTOP=JBOP		
328	IF (KSTOP*KBEGIN.NE.0) GO TO 221		
329	219 CONTINUE		
330	WRITE (6,222) KBEGIN,KSTOP		
331	221 KTOT=KSTOP-KBEGIN+1		
332	YCL=YIN(3)		
333	WRITE (6,223) YCL,ENREEL,(XIN(L2),L2=1,2),(YIN(L2),L2=1,2)		
334	CALL MIRROR(K,KTOT,KBEGIN,YCL)		
335	CCCCC KR= TOTAL NO. OF POINTS TO BE PLOTTED FOR THIS SEGMENT	A	1940
336	160 KR=K-KK	A	1950
337	CCCCC PLOT CURRENT SEGMENT	A	1960
338	CALL DRAW(KR,KK)	A	1970
339	IF (IAB.LE.0) GO TO 200		
340	IF (ABS(XON(KK)-XAA+YON(KK)-YAA).LE.1.E-7) JSTART=KK	A	2910
341	IF (ABS(XON(KK)-YBB+YON(KK)-YBB).LE.1.E-7) JSTOP=K	A	2920

342		IF (JSTART.EQ.KK) MPL=N7		
343		IF ((JSTART+JSTOP).FC.O.OP.(JSTOP.EC.O.AND.K.GT.JSTOP)) GO TO 200	A	293n
344		DO 165 K9=1,7	A	305n
345	165	BACKSPACE 4	A	306n
346		IF (JSTART.NE.KK) GO TO 180	A	307n
347		WRITE (4,17n)	A	308n
348	170	FORMAT (XX,3H9.,74X)	A	309n
349		WRITE (4,445)XON(KK),YON(KK)	A	310n
350		KK1=KK+1	A	311n
351		DO 175 K9=KK1,K	A	312n
352	175	WRITE (4,450)XON(K9),YON(K9)	A	313n
353		BACKSPACE 4	A	314n
354		IF (JSTOP.NE.K) GO TO 180	A	315n
355		WRITE (4,455)XON(K),YON(K)	A	316n
356		GO TO 200	A	317n
357	180	IF (JSTOP.NE.K) GO TO 190	A	318n
358		LOWER=LOWER+1	A	319n
359		KK1=K-1	A	320n
360		DO 185 K9=KK,KK1	A	321n
361	185	WRITE (4,450)XON(K9),YON(K9)	A	322n
362		WRITE (4,455)XON(K),YON(K)	A	323n
363		GO TO 200	A	324n
364	190	IF (JSTART.EQ.KK.AND.JSTOP.EQ.O) GO TO 200	A	325n
365		LOWER=LOWER+1	A	326n
366		KK1=K-1	A	327n
367		DO 195 K9=KK,KK1	A	328n
368	195	WRITE (4,450)XON(K9),YON(K9)	A	329n
369	200	WRITE (4,205)K,XON(K),YON(K),CAPPA(K),DYDXO(K),ALPHA(K)	A	330n
370	205	FORMAT (IH,IX,	A	331n
371	1	11HLAST POINT ,2MK=,15,4H, X=,E12.5,4H, Y=,E12.5,7H,A		
372		1KAPPA=,F12.4,7H,DY/DX=,E12.5,7H,ALPHA=,E12.5)	A	332
373	C		A	333n
374	C	END OF SEGMENT LOOP	A	334n
375	C		A	335n
376	210	GO TO 220	A	336n
377	215	MIRROD=FIX((ARS(TYPRDY)-FLOAT(NUMROD))*10.1)		
378		YCL=ANSEG		
379		KTOT=NRDPTS(MIRROD)-NRDPTS(MIRROD-1)		
380		KBEGIN=NRDPTS(MIRROD-1)+1		
381		IF (MIRROD.EQ.1) KPF CIN=1		
382		IF (MIRROD.EQ.1) KTOT=NRDPTS(1)		
383		WRITE (6,216)NUMROD,MIRROD,YCL		
384		KK=K		
385		CALL MIRROR(K,KTOT,KBEGIN,YCL)		
386		KR=K-KK		
387		CALL DRAW(KR,KK)		
388	220	NRDPTS(INZ)=K		
389		NBODY1=NRDPTS(1)	A	339n
390		NBODY2=NRDPTS(2)	A	340n
391		TYPINZ)=TYPRDY	A	341n
392		KAYINZ)=K	A	342n
393	225	CONTINUE	A	343n
394	C		A	344n
395	C	END OF BODY LOOP	A	345n
396	C		A	346n
397		ITOP12=K	A	347n
398		IF (YFLAG.EC.1) NRBDY2=ITOP12	A	348n
			A	349n

399		DELSND =DELS		
400		IF (IFLAG.EQ.1) NREV)=0	A	3500
401	C9		A	3510
402		ITOPT4=K+1	A	3520
403		ITOPT5=K+1		
404		ITOPT6=K+2		
405		ITOPT7=K+2		
406	C		A	3530
407	C		A	3540
408	C	CO-ORDINATES OF POINTS ON DOWNSTREAM CLOSURE		
409	C	(GENERATED FOR NON-VORTICITY CASES ONLY)		
410		IF(IIVORT.EQ.1) GOTO 255		
411	C		A	3560
412	C		A	3570
413	C	STRAIGHT SECTION BETWEEN HUB AND SHROUD OR SPLITTER	A	3570
414		IF (NBDY1.EQ.0) GO TO 230	A	3600
415		YNBDY1=YON(NBDY1)	A	3610
416		Y4SAVE=YON(NBDY1)	A	3620
417		Y5SAVE=YON(NBDY1+1)	A	3630
418		Y6SAVE=YON(NBDY2)	A	3640
419		Y7SAVE=YON(NBDY2+1)	A	3650
420		GO TO 235	A	3660
421	230	YNBDY1=0.0	A	3670
422		Y4SAVE=0.0	A	3680
423	235	NDY4=(YON(NBDY1+1)-YNBDY1)*1.5/DELSMX	A	3690
424		ENDY4=NDY4	A	3700
425		NPTS=NDY4+1	A	3710
426		NBDPTS(NBDYS+1)=NPTS+NBDPTS(NBDYS)	A	3720
427		DY4=(YON(NBDY1+1)-YNBDY1)/ENDY4	A	3730
428		DO 240 I=1,NPTS	A	3740
429		AYEM=I-1	A	3750
430		IPN=I+K	A	3760
431		XON(IPN)=XON(NBDY1+1)	A	3770
432		YON(IPN)=YNBDY1+AYEM*DY4	A	3780
433	240	CONTINUE	A	3790
434	245	ITOPT4=K+1	A	3800
435		ITOPT5=ITOPT4+NDY4	A	3810
436		IF (NBDYS.LE.2) GO TO 255	A	3820
437	C		A	3830
438	C	STRAIGHT SECTION BETWEEN FLOW SPLITTER AND SHROUD	A	3850
439	C		A	3870
440		YNBDY2=Y6SAVE	A	3880
441		NDY5=(Y7SAVE-Y6SAVE)*1.5/DELSMX	A	3890
442		ENDY5=NDY5	A	3900
443		NPTS=NDY5+1	A	3910
444		NBDPTS(NBDYS+2)=NPTS+NBDPTS(NBDYS+1)	A	3920
445		DY5=(Y7SAVE-Y6SAVE)/ENDY5	A	3930
446		DO 250 I=1,NPTS	A	3940
447		AYEM=I-1	A	3950
448		IPN=I+ITOPT5	A	3960
449		XON(IPN)=XON(NBDY2+1)	A	3970
450		YON(IPN)=YNBDY2+AYEM*DY5	A	3980
451	250	CONTINUE	A	3990
452		ITOPT6=ITOPT5+1	A	4000
453		ITOPT7=ITOPT6+NDY5	A	4010
454	C		A	4020
455	C		A	4020

456	C		A	3840
457	C	CALL SUBROUTINE TO WRITE AND PUNCH CARDS	A	4030
458	C		A	4040
459		255 CALL WPUNCH	A	4050
460		IF (NBDYS.EQ.2) GO TO 260	A	4060
461		NT1=ITOPTS-3	A	4070
462		NT2=N-4	A	4080
463		GO TO 265	A	4090
464	260	NT1=ITOPTS-NBDYS-(1-IVORT)*(NBDYS-1)-2*IVORT		
465		NT2=ITOPTS-NBDYS-(1-IVORT)*(NBDYS-2)-IVORT		
466	265	NHUBMX=NBDY1-1	A	4140
467		NP=0	A	4150
468		DO 270 I=1,NRAKFS	A	4160
469		NP=NP+NBY(I)+1	A	4170
470	270	CONTINUE	A	4180
471		NHBMXD=NHUBMX	A	4190
472		IF (PROG.NE.02TEST) GO TO 280		
473		DO 281 I=1,NBDYS		
474		NB(I)=NBDPTS(I)-1		
475	281	CONTINUE		
476		IF (NBDYS.EQ.2) WRITE (6,541) NB(1),NP(2),NP		
477		IF (NBDYS.EQ.3) WRITE (6,542) NB(1),NB(2),NP(3),NP		
478		GO TO 285		
479	280	WRITE (6,540) NT1,NT2,NHUBMX,NP	A	4270
480	C		A	4280
481	C		A	4300
482	C	CALCULATION OF SURFACE DISTANCE ALONG EACH BODY FROM X=XN1		
483	C		A	4400
484	285	DO 339 I=1,NBDYS		
485		WRITE(6,545) I,IBOD		
486		NBP1=NBDPTS(1BOD-1)+1		
487		IF (1BOD.EQ.1) NBP1=1		
488		NBP2=NBDPTS(1BOD)		
489		DO 325 I=NBP1,NBP2		
490		JJ=I	A	4770
491		IF (XON(I).LT.XON(I+1)) GO TO 330	A	4780
492	325	CONTINUE	A	4790
493	330	NSTR1=NBP1		
494		XSEQ=XRI		
495		NTOT1=JJ-NBP1+1		
496		IF (1BOD.EQ.1.AND.NBDYS.NE.1) NTOT1=NBP2-JJ+1		
497		IF (1BOD.EQ.1.AND.NBDYS.NE.1) NSTR1=JJ		
498		IF (XRI.LT.XON(JJ)) XSEQ=XON(JJ)		
499		SDEL=0.0	A	4820
500	C			
501		C.... THE FOLLOWING LOOP IS A REDUNDANT CALCULATION OF THE SON ARRAY.		
502		C... PURPOSE IS TO ELIMINATE A 'JUMP' IN THE OUTPUT VALUES OF S, SON, AND SDEL.		
503		C... THE 'JUMP' HAS BEEN NOTICED AT THE FIRST POINT OF A PI-ELLIPSE WHEN		
504		C... PRECEDED BY A STRAIGHT LINE. THE 'JUMP' IN ARC LENGTH QUANTITIES WAS NOT		
505		C... ACCOMPANIED BY A 'JUMP' IN XON,YON VALUES. THE 'JUMP' WAS APPROXIMATELY		
506		C... EQUAL TO THE LENGTH OF THE BI-ELLIPSE. (874778)		
507		DO 334 I=NBP1,NBP2		
508		IF (1.GY.NBPT) GO TO 333		
509		SON(I) = 0.0		
510		GO TO 334		
511	333	ADELX2 = (AP5(XON(I) - XON(I-1)))**2		
512		ADELY2 = (AP5(YON(I) - YON(I-1)))**2		

```

513      SON(I) = SON(I-1) + SORT(ADELX2 + ADELY2)
514      334 CONTINUE
515      CALL SINTP(XON(NSTRT), SON(NSTRT), NTOT, XSEQO, S33)
516      DO 335 I=NBP1, NBP2
517          IF (I.NE.NBP1) SOFL=SON(I)-SON(I-1)
518          S(I)=S33-SON(I)
519      335 WRITE (6,520) I, XON(I), YON(I), CAPPA(I), DYDXO(I), ALPHA(I), SON(I), S(I)
520      1), SOEL
521      C
522      C
523      339 NRDSV=NRDYS
524      C
525      C WRITE OUT CLOSURE COORDINATES
526      C
527      C
528      350 IBD=NBDYS+1
529      IF (IVORT.EQ.1) GO TO 355
530      WRITE (6,515) IBD, (I, XON(I), YON(I), I=ITOPT4, ITOPT5)
531      IF (NRDSV.NE.3) GO TO 355
532      IBD=IBD+1
533      WRITE (6,515) IBD, (I, XON(I), YON(I), I=ITOPT6, ITOPT7)
534      355 WRITE (6,545) (XRAN(I), YLO(I), YHI(I), NY(I), I=1, NRANKS)
535      IF (XON(1).GE.XON(3).OR.NBDYS.NE.1)
536      1CALL AREA
537      C
538      C
539      C
540      C CURVATURE PLOTS
541      C
542      LEL=6
543      C
544      C
545      IF (IPLOT.CE.O) GO TO 405
546      IF (IPLOT.CE.O) LEL=4
547      LEL=0
548      C
549      LEA = LAST PT. ON SHROUD
550      LEA=NBDPTS(NBDYS)
551      LL=1
552      C
553      C
554      C
555      C
556      360 LEE=LL+NBDYS1
557      IF (XON(LEE).GT.XX*EXEP+XMIN) GO TO 375
558      IF (XON(LEE).LT.XON(LEE+1).AND.LEE.LT.LEEH) GO TO 365
559      IF (CAPPA(LEE).EQ.99999.) GO TO 375
560      LE=LEE+1
561      DIST(LE)=SON(LEE)
562      CAPPER(LE)=CAPPA(LEE)
563      C
564      C
565      C
566      C
567      C
568      C
569      C

```

570	CCCCC PLOT THE FIRST FRAME (INTERNAL +POUND PTS.)	A	5820
571	IF (IPLOT.C.LT.O..AND.C.FE.EO.LEEHI) GO TO 380	A	5820
572	375 LE=LL+1	A	5840
573	IF (LEE.LE.LEA		
574	380 CALL PLOT(XX,O.,-?)	A	5860
575	CCCCC IF THE SECOND X-CURVE (EXTERNAL PTS.) IS BEING PLOTTED, DO NOT	A	5870
576	CCCCC GENERATE NEW SCALE FACTORS. USE THOSE OF THE INTERNAL PLOT.	A	5880
577	IF (LFE.GY.LEEHI.AND.IPLOT.C.LT.O) GO TO 385	A	5890
578	CALL CSCALE(CAPPER,YY,LE,1,1,C,EXMIN,DFEX)	A	5500
579	385 CAPPER(LE+1)=EXMIN	A	5510
580	CAPPER(LE+2)=DFEX	A	5520
581	IF (LEE.GY.LEEHI.AND.IPLOT.C.LT.O) GO TO 390	A	5530
582	CALL CSCALE(DIST,YY,LE,1,1,C,EXMIND,DEEXD)	A	5540
583	390 DIST(LE+1)=EXMIND	A	5550
584	DIST(LE+2)=DFEXD	A	5560
585	IF(IPLOT.C.LT.O) DIST(LE+1)=XMIN	A	5570
586	IF(IPLOT.C.LT.O) DIST(LE+2)=EXEP	A	5580
587	CCCCC DRAW AXES FOR CURVATURE PLOT	A	5590
588	CALL PLOT(XX,YY,DIST(LE+2),CAPPER(LE+2),DIST(LE+1),CAPPER(LE+1),	A	5600
589	1.25,.25,0,0,LEE,5,1,2)	A	5610
590	CALL LINE(DIST,CAPPER,LE,1,1,3,DIST(LE+1),DIST(LE+2),CAPPER(LE+1),	A	5620
591	1CAPPER(LE+2))	A	5630
592	CCCCC DRAW SEGMENT MARKERS	A	5640
593	DO 400 MEM=1,MH	A	5650
594	IF(IPLOT.C.LT.O) BAGS(MEM)=BAGX(MEM)	A	5660
595	IF(ZAP(MEM).LT.O..AND.NZAP(MEM).GT.LEEHI) ZAP(MEM)=-ZAP(MEM)	A	5670
596	IF(ZAP(MEM).EQ.99999.) ZAP(MEM)=YY*CAPPER(LE+2) + CAPPER(LE+1)	A	5680
597	IF (LEE.GY.LEEHI.AND.IPLOT.C.LT.O) GO TO 395	A	5690
598	BAGS(MEM)=(BAGS(MEM)-DIST(LE+1))/DIST(LE+2)	A	5700
599	ZAP(MEM)=(ZAP(MEM)-CAPPER(LE+1)) /CAPPER(LE+2)	A	5710
600	IF(IPLOT.C.GY.O.AND.NZAP(MEM).LT.LEEHI) CALL SYMBOL(BAGS(MEM),ZAP(M	A	5720
601	MEM),.2,1,0.,-1)	A	5730
602	395 IF (IPLOT.C.LT.O.AND.BAGS(MEM).GY.(XX*FXEP+XMIN)) GO TO 400	A	5740
603	IF((NZAP(MEM).LE.LEEHI.AND.LEE.FQ.LEEHI).OR.(NZAP(MEM).GE.LEEHI.AND	A	5750
604	10.LEE.NE.LEFHI)) CALL SYMBOL(BAGS(MEM),ZAP(MEM),.2,1,0.,-1)	A	5760
605	400 CONTINUE	A	5770
606	IF (LEE.GY.LEA		
607	1LEA)) GO TO 405	A	5780
608	LE=0	A	5800
609	CALL SYMBOL(XX-.5,YY-.5,.3,52,0.,-1)	A	5810
610	GO TO 375	A	5820
611	C	A	5830
612	CC	A	5840
613	C	A	5850
614	CCCCC ADVANCE THE PLOT ORIGIN FOR THE NEXT CASE	A	5860
615	405 CALL PLOT(XX,O.,-3)	A	5870
616	C	A	5880
617	CCCCC IF THE CASE IS NOT TO BE REWORKED VIA FESSLER, BEGIN NEXT JOB	A	5890
618	IF (IAB.LE.O) GO TO 410	A	5900
619	REWIND 4	A	5910
620	NIN=4	A	5920
621	DELS=DSAVE	A	5930
622	GO TO 20	A	5940
623	410 IF (IREDON1).EQ.1) GO TO 10	A	5950
624	REWIND 8	A	5960
625	LPNCH= IREDON(2)	A	5970
626	IPLOT= IPREDON(3)	A	5980

627	IPLOT= IPLOT(4)		A	598
628	IHEAD = IHEAD(1)		A	599
629	IREDON(1) = IREDON(1)-1		A	600
630	C(1:10) WRITE THE ORIGINAL COORDINATE OUTPUT AS INPUT		A	601
631	C BY DIRECT INTERPOLATION OPTION(YCALC,PSSLER)		A	602
632	WRITE (A,475)IDENT,PROG,NOC,LFNCHN,IPLOTA,IPLUTC,IAR,(IPEDON(1)),IEN		A	603
633	21,4)		A	604
634	DO 425 I=1,MNDYS		A	605
635	IF(MNDYS(I).EQ.0,I.FO.2) TYP(1)=*		A	606
636	IF (I.LO.1) GO TO 415		A	607
637	KAE= NAY(1-1)*2		A	608
638	GO TO 420		A	609
639	415 KA=2		A	610
640	420 WRITE (A,440)TYP(1)		A	611
641	KB= NAY(1)-1		A	612
642	KAB=KA-K1		A	613
643	WRITE (A,445)(YON(1),YON(KAB))		A	614
644	WRITE (A,450)(XON(1),YON(J),J=KA,KB)		A	615
645	IC=NAY(1)		A	616
646	WRITE (A,455)(XON(IC),YON(IC))		A	617
647	425 CONTINUE		A	618
648	435 MIN=4		A	619
649	REWIND 4		A	620
650	GO TO 40		A	621
651	440 FORMAT(F10.2,10H 0,		A	622
652	445 FORMAT(13X,11HBODY IN Z=(F10.6,1M,F10.6,4M),)		A	623
653	450 FORMAT(12Z,11H,F10.6,1M,F10.6,3M),)		A	624
654	455 FORMAT(12X,1M,F10.6,1M,F10.6,3M) \$/1X,20MSAUXIM DONE=.TRUE., \$)		A	625
655	460 FORMAT(BF10.2)		A	626
656	465 FORMAT(3X,7H-1 /1X,11HBODY IN Z=(F10.6,1M,F10.6,4M),)		A	627
657	470 FORMAT(1H,,13X,6HREDOONE)		A	628
658	475 FORMAT(2A6,14,211, 12,12X,11,9X,311, 12)		A	629
659	C FORMATS		A	630
660	C		A	631
661	C		A	632
662	C		A	633
663	480 FORMAT (10H,10X,15NO. OF BODIES = ,F2.0,5X,HODELS = ,F6.3,5X,9HDELA		A	634
664	ILSMX = ,F6.3,5X,6HXOI = ,F10.6)		A	635
665	485 FORMAT (20H)		A	636
666	490 FORMAT(2A6,14,211,212,10X,11,9X,311,212)		A	637
667	495 FORMAT(2A6,14,211, 12,12X,11H,9X,311, 12)		A	638
668	500 FORMAT (9A6)		A	639
669	FORMAT('O',2,X,		A	640
670	1 BODY NO.,14,' WILL BE GENERATED AS A MIRROR IMAGE OF ',		A	641
671	1 BODY NO.,14,' ABOUT AN AXIS AT Y=',E15,4)		A	642
672	FORMAT('SEARCH FOR SEGMENT TO RE PIRROED HAS FAILED',21H)		A	643
673	FORMAT(10H,7X,18H--- MIRROR IMAGE ?		A	644
674	1 25X,EMKED THIS SEGMENT IS A MIRROR IMAGE ABOUT THE'		A	645
675	1, V= ,1P1E15,4,LINE'735X, THE ORIGINAL SEGMENT LIES BETWEEN',		A	646
676	2, THE FOLLOWING POINTS,/26X,F6.3,5X,1MX,1P2F15,4/37X,1MY,1P2E15,4)		A	647
677	505 FORMAT(17X, BODY *12,* CO-ORDINATES- X',12X11YION,5HMAPPAIOIXSHOV/,		A	648
678	10X10X		A	649
679	2 5HALPHSX1158XGHS-S121,4X,6HDELVAS/1X)		A	650
680	FORMAT(8H BODY 12,*,19I(*****))		A	651
681	FORMAT('COORINTRY OUTPUT IS SAVD ON UNIT 17 FOR')		A	652
682	FORMAT('COORDINTRY OUTPUT IS NOT SAVED FOR')		A	653
683	FORMAT('PROGRAM *A6, WHICH WILL GENERATE')		A	654

[illegible]

1	CK	00C0
2	C	K	0010
3	C	K	0020
4	C	K	0030
5	C	K	0040
6	C	K	0050
7	C	K	0060
8	C	K	0070
9	C	K	0080
10	C	K	0090
11	C	K	01C0
12	C	K	0110
13	C	K	0120
14	C	K	0130
15	C	K	0140
16	C	K	0150
17	C	K	0160
18	C	K	0170
19	C	K	0180
20	C	K	0190
21	C	K	02C0
22	C	K	0210
23	C	K	0220
24	C	K	0230
25	C	K	0240
26	C	K	0250
27	C	K	0260
28	C	K	0270
29	C	K	0280
30	C	K	0290
31	C	K	03C0
32	C	K	0310
33	C	K	0320
34	C	K	0330
35	C	K	0340
36	C	K	0350
37	C	K	0360
38	C	K	0370
39	C	K	0380
40	C	K	0390
41	C	K	04C0
42	C	K	0410
43	C	K	0420
44	C	K	0430
45	C	K	0440
46	C	K	0450
47	C	K	0460
48	C	K	0470
49	C	K	0480
50	C	K	0490
51	C	K	05C0
52	C	K	0510
53	C	K	0520
54	C	K	0530
55	C	K	0540
56	C	K	0550

SUBROUTINE SIMQ
 PURPOSE
 OBTAIN SOLUTION OF A SET OF SIMULTANEOUS LINEAR EQUATIONS,
 $AX=B$
 USAGE
 CALL SIMQ(A,B,N,KS)
 DESCRIPTION OF PARAMETERS
 A - MATRIX OF COEFFICIENTS STORED COLUMNWISE. THESE ARE
 DESTROYED IN THE COMPUTATION. THE SIZE OF MATRIX A IS
 N BY N.
 B - VECTOR OF ORIGINAL CONSTANTS (LENGTH N). THESE ARE
 REPLACED BY FINAL SOLUTION VALUES, VECTOR X.
 N - NUMBER OF EQUATIONS AND VARIABLES. N MUST BE .GT. ONE.
 KS - OUTPUT LIGHT
 0 FOR A NORMAL SOLUTION
 1 FOR A SINGULAR SET OF EQUATIONS
 REMARKS
 MATRIX A MUST BE GENERAL.
 IF MATRIX IS SINGULAR, SOLUTION VALUES ARE MEANINGLESS.
 AN ALTERNATIVE SOLUTION MAY BE OBTAINED BY USING MATRIX
 INVERSION (MINV) AND MATRIX PRODUCT (MMPD).
 SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED
 NONE
 METHOD
 METHOD OF SOLUTION IS BY ELIMINATION USING LARGEST PIVOTAL
 DIVISOR. EACH STAGE OF ELIMINATION CONSISTS OF INTERCHANGING
 ROWS WHEN NECESSARY TO AVOID DIVISION BY ZERO OR SMALL
 ELEMENTS.
 THE FORWARD SOLUTION TO OBTAIN VARIABLE N IS DONE IN
 N STAGES. THE BACK SOLUTION FOR THE OTHER VARIABLES IS
 CALCULATED BY SUCCESSIVE SUBSTITUTIONS. FINAL SOLUTION
 VALUES ARE DEVELOPED IN VECTOR B, WITH VARIABLE 1 IN B(1),
 VARIABLE 2 IN B(2),....., VARIABLE N IN B(N).
 IF NO PIVOT CAN BE FOUND EXCEEDING A TOLERANCE OF 0.0,
 THE MATRIX IS CONSIDERED SINGULAR AND KS IS SET TO 1. THIS
 TOLERANCE CAN BE MODIFIED BY REPLACING THE FIRST STATEMENT.

 SUBROUTINE SIMQ (A,B,N,KS)
 DIMENSION A(1), B(1)
 FORWARD SOLUTION
 TOL=0.0
 KS=0
 JJ=-N
 DO 45 J=1,N
 JY=J+1

57	JJ=JJ+N+1	K	0560
58	BIGA=0	K	0570
59	IT=JJ-J	K	0580
60	DO 15 I=J,N	K	0590
61	C	K	0600
62	C SEARCH FOR MAXIMUM COEFFICIENT IN COLUMN	K	0610
63	C	K	0620
64	IJ=IT+1	K	0630
65	IF (ABS(BIGA)-ABS(A(I,J))) 10,15,15	K	0640
66	10 BIGA=A(I,J)	K	0650
67	IMAX=I	K	0660
68	15 CONTINUE	K	0670
69	C	K	0680
70	C TEST FOR PIVOT LESS THAN TOLERANCE (SINGULAR MATRIX)	K	0690
71	C	K	0700
72	IF (ABS(BIGA)-TOL) 20,20,25	K	0710
73	20 K5=I	K	0720
74	RETURN	K	0730
75	C	K	0740
76	C INTERCHANGE ROWS IF NECESSARY	K	0750
77	C	K	0760
78	25 I1=J+N*(J-2)	K	0770
79	IT=IMAX-J	K	0780
80	DO 30 K=J,N	K	0790
81	I1=I1+N	K	0800
82	I2=I1+IT	K	0810
83	SAVE=X(I1)	K	0820
84	A(I1)=A(I2)	K	0830
85	A(I2)=SAVE	K	0840
86	C	K	0850
87	C DIVIDE EQUATION BY LEADING COEFFICIENT	K	0860
88	C	K	0870
89	30 A(I1)=A(I1)/BIGA	K	0880
90	SAVE=B(IMAX)	K	0890
91	B(IMAX)=B(IJ)	K	0900
92	B(IJ)=SAVE/BIGA	K	0910
93	C	K	0920
94	C ELIMINATE NEXT VARIABLE	K	0930
95	C	K	0940
96	IF (J-N) 35,50,35	K	0950
97	35 IQS=N*(J-1)	K	0960
98	DO 45 IX=JY,N	K	0970
99	IXJ=IQS+IX	K	0980
100	IT=J-IX	K	0990
101	DO 40 JX=JY,N	K	1000
102	IXJX=N*(JX-1)+IX	K	1010
103	JJX=IXJX+IT	K	1020
104	40 A(IXJX)=A(IXJX)-A(IXJ)*A(JJX)	K	1030
105	45 B(IXJ)=B(IXJ)-B(IJ)*A(IXJ)	K	1040
106	C	K	1050
107	C BACK SOLUTION	K	1060
108	C	K	1070
109	50 NY=N-1	K	1080
110	IT=N+N	K	1090
111	DO 55 J=1,NY	K	1100
112	IA=IT-J	K	1110
113	IB=N-J	K	1120
114	IC=N	K	1130
115	DO 55 M=1,J	K	1140
116	B(IIB)=B(IIB)-A(IA)*B(IC)	K	1150
117	IA=IA-N	K	1160
118	55 IC=IC-1	K	1170
119	RETURN	K	1180
120	EKD	K	1190

1		SUBROUTINE STRAIT (K,ISHR)	B	0000
2	C		B	0010
3	C	A REGULAR STRAIGHT SEGMENT	B	0020
4	C		B	0030
5		COMMON /MAIN/ XIN(10),VIN(10),DELSMX,PI02,DELS1,INUB	B	0040
6		COMMON /FORSS/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)	B	0050
7		1),ALPHA(500),CAPPA(500),SON(500),PI0180	B	0060
8		COMMON /SS/ KBDY1,KBDY2,TYPBDY,KPDYS	B	0070
9		COMMON /FNST/ NFIRST	B	0080
10		COMMON /NHIGH/ NSPHG,NLAST,XLAST(500)	B	0090
11		NFIRST=K	B	0100
12		XTEST=XIN(2)-XIN(1)	B	0110
13		YTEST=VIN(2)-VIN(1)	B	0120
14		IF (XTEST.EQ.0.0) GO TO 10	B	0130
15		IF (XTEST.EQ.0.0) GO TO 10	B	0130
16		DYDX=XTEST/YTEST	B	0140
17		ALPHAC=ATAN(YTEST/XTEST)	B	0150
18		GO TO 15	B	0160
19		10 DYDX=99999.	B	0170
20		ALPHAC=PI02	B	0180
21	C		B	0190
22	C	CALCULATE DELSNW	B	0200
23	C		B	0210
24		15 STOT=SQRT(XTEST**2+YTEST**2)	B	0220
25		ANOS=STOT/DELS1	B	0230
26		AINDS=AINTTANDS	B	0240
27		TEST=ANOS-AINDS	B	0250
28		IF (TEST.GE..5) AINDS=AINDS+1.0	B	0260
29		DELSNW=STOT/AINDS	B	0270
30		DELSNW=ABS(DELSNW)	B	0280
31		DELS1=DELSNW	B	0290
32		DELS=DELS1	B	0300
33		IF (YTEST) 20,35,20	B	0310
34		20 IF (XTEST.EQ.0.0) GO TO 50	B	0320
35		DYDX0(K+1)=DYDX	B	0330
36		ALPHA(K+1)=ALPHAC	B	0340
37		SIGN=1.0	B	0350
38		IF (XTEST.LT.0.0) SIGN=-1.0	B	0360
39		YON(K+1)=YON(K)+SIGN*DELSNW*SIN(ALPHA(K+1))	B	0370
40		IF (NSPHG.EQ.0) GO TO 25	B	0380
41		XON(K+1)=XON(MLAST-1)	B	0390
42		MLAST=MLAST-1	B	0400
43		GO TO 30	B	0410
44		25 XON(K+1)=XON(K)+SIGN*DELSNW*COS(ALPHA(K+1))	B	0420
45		30 SON(K+1)=SON(K)+SQRT((XON(K+1)-XON(K))**2+(YON(K+1)-YON(K))**2)	B	0430
46		CAPPA(K+1)=0.0	B	0440
47		IF (XTEST.LT.0.0.AND.XON(K+1).LE.XIN(2).OR.XTEST.GT.0.0.AND.XON(K+1).GE.XIN(2)) GO TO 55	B	0450
48		IF (ABS(XON(K+1)-XIN(2)).LE.1.0E-4*DELS.AND.ABS(YON(K+1)-VIN(2)).LE.1.0E-4*DELS) GO TO 60	B	0460
49		IF (ABS(XON(K+1)-XIN(2)).LE.1.0E-4*DELS.AND.ABS(YON(K+1)-VIN(2)).LE.1.0E-4*DELS) GO TO 60	B	0470
50		IF (1.0E-4*DELS) GO TO 60	B	0480
51		K=K+1	B	0490
52		GO TO 20	B	0500
53		35 DYDX0(K+1)=0.0	B	0510
54		ALPHA(K+1)=0.0	B	0520
55		SIGN=1.0	B	0530
56		IF (XTEST.LT.0.0) SIGN=-1.0	B	0540

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57 IF (NSPH6.EQ.0) GO TO 40
58 XOM(K+1)=XOM(MLAST-1)
59 MLAST=MLAST-1
60 GO TO 45
61 40 XOM(K+1)=XOM(K)+SIGN*DELSMU
62 45 YOM(K+1)=YOM(K)
63 SORT(K+1)=SORT(K)+SORT(XOM(K+1)-XOM(K))002+YOM(K+1)-YOM(K))002)
64 CAPPA(K+1)=0.0
65 IF (YTEST.LT.U.U.AND.XOM(K+1).LE.XIM(2).OR.YTEST.GT.U.U.AND.YOM(K+1).LE.YIM(2)) GO TO 55
66 11).GE.XIM(2)) GO TO 55
67 IF (ABS(XOM(K+1)-XIM(2)).LE.1.0E-4*DELS.AND.ABS(YOM(K+1)-YIM(2)).LE.1.0E-4*DELS) GO TO 60
68 1E-1.0E-4*DELS) GO TO 60
69 K=K+1
70 GO TO 35
71 50 OYDXOM(K+1)=99999.
72 ALPHA(K+1)=PI02
73 SIGN=1.0
74 IF (YTEST.LT.0.0) SIGN=-1.0
75 XOM(K+1)=XOM(K)
76 YOM(K+1)=YOM(K)+SIGN*DELSMU
77 SORT(K+1)=SORT(K)+SORT(XOM(K+1)-XOM(K))002+YOM(K+1)-YOM(K))002)
78 CAPPA(K+1)=0.0
79 IF (YTEST.LT.U.U.AND.YOM(K+1).LE.YIM(2).OR.YTEST.GT.U.U.AND.YOM(K+1).LE.YIM(2)) GO TO 55
80 11).GE.YIM(2)) GO TO 55
81 IF (ABS(XOM(K+1)-XIM(2)).LE.1.0E-4*DELS.AND.ABS(YOM(K+1)-YIM(2)).LE.1.0E-4*DELS) GO TO 60
82 1E-1.0E-4*DELS) GO TO 60
83 K=K+1
84 GO TO 50
85 55 IF (ABS(XOM(K+1)-XIM(2)).LE.1.0E-3*DELS.AND.ABS(YOM(K+1)-YIM(2)).LE.1.0E-3*DELS) GO TO 60
86 1E-1.0E-3*DELS) GO TO 60
87 K=K-1
88 60 K=K+1
89 DO 65 K=K+1,N
90 65 ALPHA(K)=ALPHA(K)/PI0100
91 RETURN
92 END

```

1	SUBROUTINE FNSTRH (K)	C	00C0
2		C	0010
3	C FINAL STRAIGHT SEGMENT ON THE HUB AND SHROUD	C	0020
4	C	C	0030
5	COMMON /MAIN/ XIN(10),YIN(10),DELSMX,PIO2,DELS1,INLB	C	0040
6	COMMON /FORSSS/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)	C	0050
7	1),ALPHA(500),CAPPA(500),SON(500),PIO180	C	0060
8	COMMON /SS/ NBDY1,NBDY2,TYPBDY,KPDYS	C	0070
9	COMMON /FNST/ NFIRST	C	0080
10	NFIRST=N	C	0090
11	DS=DELS1	C	01C0
12	DELSTR=DELSMX	C	0110
13	YTEST=YIN(2)-YIN(1)	C	0120
14	XTEST=XIN(2)-XIN(1)	C	0130
15	ASIGN=1.0	C	0140
16	IF (XTEST.EQ.0.0) ASIGN=-1.0	C	0150
17	ISTAR=0	C	0160
18	SSEG=SQRT(XTEST**2+YTEST**2)	C	0170
19	IF (XTEST.EQ.0.0) GO TO 10	C	0180
20	IF (YTEST.EQ.0.0) GO TO 15	C	0190
21	DYDX=YTEST/XTEST	C	0200
22	ALPHAC=ATAN(YTEST/XTEST)	C	0210
23	SINAL=SIN(ALPHAC)	C	0220
24	COSAL=COS(ALPHAC)	C	0230
25	GO TO 20	C	0240
26	10 DYDX=SIGN(99999.,YTEST)	C	0250
27	ALPHAC=SIGN(PIO2,YTEST)	C	0260
28	SINAL=SIN(ALPHAC)	C	0270
29	COSAL=COSAL	C	0280
30	GO TO 20	C	0290
31	15 DYDX=0.0	C	03C0
32	ALPHAC=0.0	C	0310
33	SINAL=0.0	C	0320
34	COSAL=1.0	C	0330
35	20 DYDX0(K+1)=DYDX	C	0340
36	ALPHA(K+1)=ALPHAC	C	0350
37	IF (DS.GT.DELSMX) GO TO 25	C	0360
38	GO TO 45	C	0370
39	25 IF (ISTAR.NE.0) GO TO 45	C	0380
40	DS=AST	C	0390
41	XON(K+1)=XON(K)	C	04C0
42	YON(K+1)=YON(K)	C	0410
43	ICOUNT=0	C	0420
44	30 XSAVE=XON(K+1)-XIN(1)	C	0430
45	YSAVE=YON(K+1)-YIN(1)	C	0440
46	SSTAR=SQRT(XSAVE**2+YSAVE**2)	C	0450
47	ASTAR=(SSEG-SSTAR)/DELSTR	C	0460
48	ATTEST=ASTAR*FLOAT(1)/X(ISTAR)	C	0470
49	IF (ATTEST.GT..5) ASTAR=ASTAR+1.0	C	0480
50	NSTAR=IFIX(ATTEST)	C	0490
51	ISTAR=1	C	05C0
52	IF (NSTAR.EQ.0) GO TO 35	C	0510
53	DS=(SSEG-SSTAR)/FLOAT(NSTAR)	C	0520
54	IF (DS.GT.DELAST) GO TO 35	C	0530
55	IF (ICOUNT.(1.0) K+1)	C	0540
	GO TO 45	C	0550

57	35 K=K-1	C	0560
58	IF (K.GT.NFIRST) GO TO 40	C	0570
59	K=NFIRST	C	0580
60	CALL STRAIT (K,0)	C	0590
61	K=K-1	C	0600
62	GO TO 50	C	0610
63	40 DSLAST=SQRT((XON(K-1)-XON(K))**2+(YON(K-1)-YON(K))**2)*1.2	C	0620
64	DELSL=DSLAST	C	0630
65	ICOUNT=ICOUNT+1	C	0640
66	GO TO 30	C	0650
67	45 XON(K+1)=XON(K)+ASIGN*DS*COSAL	C	0660
68	YON(K+1)=YON(K)+ASIGN*DS*SINAL	C	0670
69	SON(K+1)=SON(K)+SQRT((XON(K+1)-XON(K))**2+(YON(K+1)-YON(K))**2)	C	0680
70	CAPPA(K+1)=0.0	C	0690
71	IF (ABS(XON(K+1)-XIN(2))) LE .001*DS .AND. XTEST.NE.0.0) GO TO 50	C	0700
72	IF (ABS(YON(K+1)-YIN(2))) LE .001*DS .AND. XTEST.EQ.0.0) GO TO 50	C	0710
73	C	C	0720
74	CIIIIII IS THE CURRENT POINT PAST THE NEAREST ENDOPOINT OF SEGMENT? (PRC	C	0730
75	CIIIIII VIUUS TESTS WERE ONLY FOR ABSOLUTE PROXIMITY TO ENDOPOINT)	C	0740
76	IF (ABS(YON(K+1)-YIN(1))) GT .ABS(YTEST)) GO TO 50		
77	IF (ABS(XON(K+1)-XIN(1))) GT .ABS(XTEST)) GO TO 50		
78	K=K+1	C	0770
79	IF (ISTAR.EQ.0) DS=DS*1.2	C	0780
80	GO TO 20	C	0790
81	50 DELSI=DELS	C	0800
82	XON(K+1)=XIN(2)	C	0810
83	YON(K+1)=YIN(2)	C	0820
84	NBDY1=K+1	C	0830
85	K=K+1	C	0840
86	DO 55 KAL=NFIRST,K	C	0850
87	55 ALPHAKAL=ALPHAKAL/P10180	C	0860
88	RETURN	C	0870
89	END	C	0880

1		SUBROUTINE FRSTSH (K)	D	00C0
2	C		D	0010
3	C	FIRST STRAIGHT SEGMENT SHROUD	D	0020
4	C		D	0030
5	C	IF THERE IS NO HUB INTERCHANGE PCINTS (X1,Y1) AND (X2,Y2)	D	0040
6	C	AND TREAT LIKE FINAL STRAIGHT SECTION ON THE HUB,	D	0050
7	C	THEN REVERSE XON AND YON APRAYS	D	0060
8	C		D	0070
9	C		D	0080
10		COMMON /MAIN/ XIN(10),YIN(10),DELSH,PI02,DELSI,INHUB,DELNEW		
11		COMMON /FOR3SS/IO,CELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)	D	0100
12		IF,ALPHA(500),CAPPA(500),SONT(500),PI0180	D	0110
13		COMMON /SS/ NBDY1,NBDY2,TYPBDY,NBDYS	D	0120
14		COMMON /FWS/ NFIRST	D	0130
15		DIMENSION XA(2),YA(2),DSV(500),ASV(500),XSV(500),YSV(500)	D	0140
16		SONIK=D.O	D	01E0
17		IF (INHUB.EQ.1.AND.CELNEW.EQ.0.) GOTO25		
18		DO 10 I=1,2	D	0180
19		XA(I)=XIN(I)	D	0190
20	10	YA(I)=YIN(I)	D	0200
21		XIN(I)=XA(2)	D	0210
22		XIN(2)=XA(1)	D	0220
23		YIN(1)=YA(2)	D	0230
24		YIN(2)=YA(1)	D	0240
25		NFB2=K	D	0250
26		YONIK=YIN(1)	D	0260
27		XONIK=XIN(1)	D	0270
28		CALL FNSTRH (K)	D	0280
29		KSV=K	D	0290
30		KTOT=KSV-NFB2+1		
31		DO 15 II=1,KTOT		
32		I=NFB2+II-1		
33		KSR=KSV+1-II		
34		DSV(KSR)=DYDX0(I)	D	0320
35		ASV(KSR)=ALPHA(I)	D	0330
36		XSV(KSR)=XON(I)	D	0340
37	15	YSV(KSR)=YON(I)	D	0350
38		SONINFB2I = D.O		
39		DO 20 I=NFB2,KSV	D	0370
40		DYDX0(I)=DSV(I)	D	0380
41		ALPHA(I)=ASV(I)	D	0390
42		XON(I)=XSV(I)	D	0400
43		YON(I)=YSV(I)	D	0410
44		CAPPA(I)=D.O	D	0430
45	20	CONTINUE	D	0440
46		III=NFB2+1		
47		DO 21 I=III,KSV		
48		SONII = SONII-1+SONI(XON(I)-XON(I-1))*2+(YON(I)-YON(I-1))*2)		
49	21	CONTINUE		
50		DELSI=ABS(SONI(KSV)-SONI(KSV-1))	D	0450
51		RETURN	D	0470
52	C		D	0480
53	C	IF THERE IS A HUB, USE X VALUES FROM FINAL STRAIGHT	D	0490
54	C	SECTION ON THE HUB FOR FIRST STRAIGHT SECTION ON	D	0500
55	C	SHROUD	D	0510
56	C		D	0520

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57 25 XTEST=XIN(1)-XIN(2)
58 YTEST=YIN(1)-YIN(2)
59 IF YTEST.EQ.0.U) 60 TO 30
60 DYDX=YTEST/XTEST
61 ALPHAC=ATAN2(YTEST,XTEST)
62 60 TO 35
63 30 DYDX=99999.
64 ALPHAC=PI/2
65 K=K-1
66 NBOC=NBDY1
67 IF (TYPE.DY.EQ.3.U.AND.NBDYS.EQ.3) NBOC=NBDY2
68 DO 50 I=FIRST,NBOC
69 KEEP=NBOC-NFIRST-I
70 XON(K+1)=XON(KEEP)
71 DYDXON(K+1)=DYDX
72 ALPHAC(K+1)=ALPHAC
73 IF (I.EQ.NFIRST) 60 TO 40
74 YON(K+1)=YON(K)+XON(K+1)-XON(K)*DYDX
75 IF (COS(ALPHAC).EQ.-C.) STOP
76 44 YON(K+1)=SON(K)+XON(K)-XON(K)*COS(ALPHAC)
77 SON(K+1)=SON(K)+XON(K+1)-XIN(1)*DYDX
78 60 TO 45
79 40 YON(K+1)=YIN(1)+XON(K+1)-XIN(1)*DYDX
80 45 CAPPA(K+1)=0.0
81 ALPHAC(K+1)=ALPHAC(K+1)/PI/180
82 K=1
83 50 CON=4*ME
84 DELST=SON(K)-SON(K-1)
85 RETURN
86 7
87 END

```

1	SUBROUTINE TEST (IA)	E	0000
2	COMMON /MAIN/ XIN(10),VIN(10),DELSM,XPI02,DELSI,INHUB	E	0010
3	COMMON /FOR355/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDXO(500)	E	0020
4	IF,ALPHA(500),CAPPA(500),SON(500),PI0180	E	0030
5	COMMON /SPREP/ KPREP	E	0040
6	M=IA-1	E	0050
7	IF (XIN(2).EQ.XIN(1)) GO TO 10	E	0060
8	SLP1=(VIN(2)-VIN(1))/(XIN(2)-XIN(1))	E	0070
9	GO TO 15	E	0080
10	10 TEST1=(VIN(2)-VIN(1))/(XIN(M)-XIN(1))	E	0090
11	SLP1=SIGN(99999.,TEST1)	E	0100
12	15 IF ((XIN(1).LT.XIN(M).AND.XIN(1).LE.XIN(2)).OR.(XIN(1).GE.XIN(2).AND.XIN(1).GT.XIN(M))) GO TO 20	E	0110
13	TIP=XIN(1)-XIN(2)	E	0120
14	XIN(1)=XIN(1)+SIGN(50.,TIP)	E	0130
15	VIN(1)=(VIN(2)-VIN(1))+	E	0140
16	1(XIN(1)-XIN(2))/(XIN(2)-XIN(1)+SIGN(50.,TIP))+VIN(2)	E	0150
17	20 SLP2=(VIN(M)-VIN(1))/(XIN(M)-XIN(1))	E	0160
18		E	0170
19	C ROTATION ONLY	E	0180
20		E	0190
21		E	0200
22	25 IF (SLP1.GT.SLP2) RETURN	E	0210
23		E	0220
24	C MIRROR INTO XIN(1)	E	0230
25		E	0240
26	30 CALL PRELPS (2,0,IA,1,1)	E	0250
27	RETURN	E	0260
28	END	E	0270

1	SUBROUTINE PRELPS (MODE,KAT,IA,K1,K2)	F	00C0
2	COMMON /HAIN/ XIN(10),YIN(10),DELSMX,PI02,DELS1,THUR	F	00D0
3	COMMON /FOR355/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)	F	00E0
4	IT,ALPHA(500),CAPPA(500),SON(500),PI0180	F	00F0
5	COMMON /SPREP/ KPREP	F	00A0
6	KPREP=1	F	0050
7	IL=IA	F	0060
8	IF (KAT.EQ.1) GO TO 60	F	0070
9	KID=KODE	F	0080
10	XI=XIN(1)	F	0090
11	YI=YIN(1)	F	01C0
12	IF (IA.EQ.5) GO TO 10	F	0110
13	XC=XIN(4)	F	0120
14	YC=YIN(4)	F	0130
15	GO TO 15	F	0140
16	10 XC=XIN(5)	F	0150
17	YC=YIN(5)	F	0160
18	IL=IA+1	F	0170
19	15 DO 45 IB=1,IL	F	0180
20	IF (IB.NE.6) GO TO 20	F	0190
21	IF (ABS(XIN(6))+ABS(YIN(6)).LE.1.E-15.OR.YIN(6).EQ.200.) GO TO 45	F	02C0
22	IF (YIN(6).NE.-100.) GO TO 20	F	0210
23	XIN(6)=-XIN(6)	F	0220
24	GO TO 45	F	0230
25	20 GO TO (25,30,35,40,30,40),KODE	F	0240
26	25 YIN(18)=YIN(14)-(YIN(18)-YIN(14))	F	0250
27	GO TO 45	F	0260
28	30 XIN(18)=XIN(11)-(XIN(18)-XIN(11))	F	0270
29	GO TO 45	F	0280
30	35 YIN(18)=YIN(11)-(YIN(18)-YIN(11))	F	0290
31	GO TO 45	F	03C0
32	40 XIN(18)=XIN(14)-(XIN(18)-XIN(14))	F	0310
33	45 CONTINUE	F	0320
34	IF (KODE.EQ.5) GO TO 50	F	0330
35	IF (KODE.EQ.6) GO TO 55	F	0340
36	RETURN	F	0350
37	50 KODE=1	F	0360
38	GO TO 15	F	0370
39	55 KODE=3	F	0380
40	GO TO 15	F	0390
41	60 DO 90 IB=K1,K2	F	04C0
42	GO TO (65,70,75,80,85,75),KID	F	0410
43	65 YON(18)=YC-(YON(18)-YC)	F	0420
44	GO TO 85	F	0430
45	70 XON(18)=XI-(XON(18)-XI)	F	0440
46	GO TO 85	F	0450
47	75 YON(18)=YI-(YON(18)-YI)	F	0460
48	GO TO 85	F	0470
49	80 XON(18)=XC-(XON(18)-XC)	F	0480
50	85 DYDX0(18)=-DYDX0(18)	F	0490
51	90 CONTINUE	F	05C0
52	IF (KID.EQ.5) GO TO 95	F	0510
53	IF (KID.EQ.6) GO TO 100	F	0520
54	RETURN	F	0530
55	95 KID=2	F	0540
56	GO TO 60	F	0550
57	100 KID=4	F	0560
58	GO TO 60	F	0570
59	END	F	0580

1		SUBROUTINE ELIPSE (K)	6	0000
2	C		6	0010
3	C	THIS SUBROUTINE FITS A SEGMENT OF AN ELLIPSE TO TWO ARBITRARILY	6	0020
4	C	ORIENTED STRAIGHT LINES NOT MORE THAN 90 DEGREES APART	6	0030
5	C		6	0040
6		COMMON /MAIN/ XIN(10),YIN(10),DELSHX,PI02,DELSI,IN08	6	0050
7		COMMON /FOR3SS/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)	6	0060
8		17,ALPHA(500),CAPPA(500),SOM(500),PI0180	6	0070
9		COMMON /SS/ NBDY1,NBDY2,TYPBDY,AR0VS	6	0080
10	C		6	0090
11	C	TRANSLATE INPUT BREAK POINTS SO THAT POINT NO. 2 BECOMES	6	0100
12	C	THE ORIGIN	6	0110
13	C		6	0120
14		DELSIN=DELSI	6	0130
15		NOUNT=0	6	0140
16		DELS=DELSI	6	0150
17		P1=3.141592653	6	0160
18		K=K-1	6	0170
19		KSTART=K	6	0180
20		X2=XIN(2)	6	0190
21		Y2=YIN(2)	6	0200
22		DO 10 I=1,4	6	0210
23		XIN(1)=XIN(1)-X2	6	0220
24		10 YIN(1)=YIN(1)-Y2	6	0230
25	C		6	0240
26	C	ROTATE THE TRANSLATED BREAK POINTS SO THAT THE SLOPE OF THE	6	0250
27	C	FIRST STRAIGHT LINE IS ZERO	6	0260
28	C		6	0270
29		IF (XIN(2).NE.XIN(1)) GO TO 15	6	0280
30		SLOPE=99999.	6	0290
31		PHI=-PI02	6	0300
32		IF (YIN(1).GT.YIN(2)) PHI=PI02	6	0310
33		GO TO 20	6	0320
34		15 SLOPE=(YIN(2)-YIN(1))/(XIN(2)-XIN(1))	6	0330
35		PHI=ATAN(SLOPE)	6	0340
36		IF (XIN(1).LT.XIN(2)) PHI=PI+ATAN(SLOPE)	6	0350
37		20 DO 25 I=1,4	6	0360
38		XA=XIN(1)	6	0370
39		XIN(1)=XA+COS(PHI)*YIN(1)*SIN(PHI)	6	0380
40		25 YIN(1)=-XA*SIN(PHI)+YIN(1)*COS(PHI)	6	0390
41	C		6	0400
42	C	DETERMINE THE ELLIPSE	6	0410
43	C		6	0420
44		IF (XIN(4).NE.XIN(3)) GO TO 30	6	0430
45		B=YIN(3)	6	0440
46		A=ABS(XIN(3))	6	0450
47		PHIAB=PI02	6	0460
48		GO TO 35	6	0470
49		30 SLOP2=(YIN(4)-YIN(3))/(XIN(4)-XIN(3))	6	0480
50		C3=XIN(3)*SLOP2/YIN(3)	6	0490
51		RAD = (C3 - 2.0)/C3	6	0500
52	CW			
53		WRITE(6,1000) C3,RAD,SLOP2,SLOPE,PHI,YIN(1),XIN(1),YIN(2),XIN(2),		
54		1 YIN(3),XIN(3),YIN(4),XIN(4)		
55		1000 FORMAT(//1X,46HC3 RAD SLOP2 SLOPE PHI Y1 X1 Y2 X2 Y3 X3 Y4 X4, //		
56		1 13E10.3)		

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57 C**
58 IF (RAD .LT. 0.0) GO TO 135
59 PHIAB = 2.0 * ATAN(SQRT(RAD))
60 A=-XIN(3)/SIN(PHIAB)
61 B=YIN(3)/YI.0-COS(PHIAB)
62 C**
63 WRITE(6,1100) PHIAB,A,B
64 1100 FORMAT(1X//9HPHIAB A B,3E10.3)
65 C**
66 35 THETMX=PHIAB-PI02
67 THETMXD=THETMX/PI0180
68 WRITE (6,140)A,B,XIN(1),YIN(1),P+I,THETMXD
69 C
70 C INITIALIZE THE FIRST POINT ON THE ELLIPSE
71 C
72 40 N=KSTART
73 XON(K+1)=XIN(2)
74 YON(K+1)=YIN(2)
75 CAPPA(K+1)=-B/(A**2)
76 ALPHA(K+1)=0.0
77 DYDXO(K+1)=0.0
78 KOUNT=KOUNT+1
79 THET=-PI02
80 OSSAVE=DELS
81 DS=DELS/PI.0+.2*YANHTABSTCAPPA(1))
82 DTHET=DS/ABS(A)
83 THET=THET+DTHET
84 C
85 C GENERATE THE POINTS ON THE ELLIPSE
86 C
87 45 K=K+1
88 50 XON(K+1)=-A*COS(THET)
89 YON(K+1)=B*PI.0+SIN(THET)
90 SON(K+1)=SON(K)+SQRT((XON(K+1)-XON(K))**2+(YON(K+1)-YON(K))**2)
91 IF (ABSTSON(K+1)-SCN(K)).GT.1.05*DS) GO TO 55
92 IF (ABS(SON(K+1)-SCN(K)).LT..95*DS) GO TO 60
93 GO TO 65
94 55 THET=THET-.02*DTHET
95 GO TO 50
96 60 THET=THET+.02*DTHET
97 GO TO 50
98 65 IF (THET.EQ.0.0) GO TO 70
99 DYDXO(K+1)=B*COTAN(THET)/A
100 ALPHA(K+1)=ATAN(DYDXO(K+1))
101 GO TO 75
102 70 DYDXO(K+1)=99999.
103 ALPHA(K+1)=PI02
104 CAPPA(K+1)=-A*B/(B*B*COS(THET)**2+A*A*SIN(THET)**2)**.5
105 DS=DELS/PI.0+.2*YANHTABSTCAPPA(1))
106 IF (ABS(DS-DELS).GT..20*DELS) DS=DELS*SIGN(.20*DELS,DS-DELS)
107 OSSAVE=DS
108 DTHET=DS/SQRT(B*B*COS(THET)**2+A*A*SIN(THET)**2)
109 DTS=DTHET
110 THET=THET+DTHET/2.0
111 DTHET=DS/SQRT(B*B*COS(THET)**2+A*A*SIN(THET)**2)
112 IF (ABS(DTHET-DTS).LT..001*DTS) GO TO 85
113 GO TO 80

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6 0520
6 0530

6 0540
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6 0580

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6 0900

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6 0920

6 0930
6 0940

6 0950
6 0960

6 0970
6 0980

6 0990
6 1000

6 1010

114	85	IF (THEY.LE.THETHX-DTHEY/2.0) 60 TO 45	6	1020
115		IF (KOUNT.GT.100) 60 TO 115	6	1030
116		DELS=DELS	6	1040
117		DSTEST=((XONIK+1)-XIN(3))*2+(YONIK+1)-YIN(3))*2)**.5	6	1050
118		IF (ABSTOS-DSTEST).LT..01*DS) 60 TO 90	6	1060
119		IF (DSTEST.GT.DS) 60 TO 110	6	1070
120		IF (DSTEST.LT..01*DS) 60 TO 95	6	1080
121		IF (DSTEST-.5*DS) 105,90,100	6	1090
122	90	K=K+1	6	1100
123	95	XONIK+1=XIN(3)	6	1110
124		YONIK+1=YIN(3)	6	1120
125		60 TO 115	6	1130
126	100	DELS=(FLOAT(K+1-KSTART)*DELS+DSTEST)/FLOAT(K+2-KSTART)	6	1140
127		IF (KOUNT.GE.10) DELS=(DELS+DELS)/2.0	6	1150
128		60 TO 40	6	1160
129	105	DELS=DELS+DSTEST/FLOAT(K+2-KSTART)	6	1170
130		IF (KOUNT.GE.10) DELS=(DELS+DELS)/2.0	6	1180
131		60 TO 40	6	1190
132	110	DELS=.8*DELS	6	1200
133		60 TO 40	6	1210
134	115	DELS=.05*1.2	6	1220
135		IF (DELS1.GT.DELS) DELS1=DELS	6	1230
136		WRITE (6,145)KOUNT	6	1240
137		WRITE (6,155)DELS1,DELS,DELS1,DSTEST	6	1250
138		KEND=K+1	6	1260
139		KSTART=KSTART+1	6	1270
140	C		6	1280
141	C	ROTATE AND TRANSLATE BACK	6	1290
142	C		6	1300
143		DO 130 KROT=KSTART,KEND	6	1310
144		XX=XONIKROT	6	1320
145		XONIKROT)=XX*COS(PHI)-YONIKROT)*SIN(PHI)+X2	6	1330
146		YONIKROT)=XX*SIN(PHI)+YONIKROT)*COS(PHI)+Y2	6	1340
147		ALPHA(KROT)=ALPHA(KROT)+PHI	6	1350
148		IF (ALPHA(KROT).EQ.PI*2) 60 TO 120	6	1360
149		DYDX(KROT)=TAN(ALPHA(KROT))	6	1370
150		60 TO 125	6	1380
151	120	DYDX(KROT)=99999.	6	1390
152	125	ALPHA(KROT)=ALPHA(KROT)/PI*180	6	1400
153	130	CONTINUE	6	1410
154		RETURN	6	1420
155	135	WRITE (6,150)SLOP2,XIN(3),YIN(3)	6	1430
156		STOP	6	1440
157	C		6	1450
158	C		6	1460
159	140	FORMAT (11H0,25X,4HA =,1PE10.3,5X,4HB =,1PE10.3,5X,5HX0 =,1PE10.6	6	1470
160		13,5X,7HY0 =,1PE10.3/24X,	6	1480
161	1	7HPHI =,1PE10.3,5X,9HTHEYX =,1PE10.36	6	1490
162	21		6	1500
163	145	FORMAT (26X,13,2X,13HITERATIONS---)	6	1510
164	150	FORMAT (11H0,25X,42FCOMBINATION OF SLOPE, X, Y NOT COMPATIBLE/30X,	6	1520
165	1		6	1530
166		1HSLOPE2 =,F7.3,3X,9HXIN(3) =,F7.3,3X,9HYIN(3) =,F7.3)	6	1540
167	155	FORMAT (11X,10HDELS IN =,F8.5,3X,7HDELS =,F8.5,3X,11HDELS OUT =	6	1550
168		F,8.5,3X,9HDSTEST =,F8.5)	6	
169		END	6	

1	SUBROUTINE SUPERC (XBRK,YBRK,ENREED,DELS1,ISTART)	M	0000
2	DIMENSION ENREED(2)	M	0010
3	DIMENSION XBRK(6), YBRK(6), XBRK(13), YBRK(13)	M	0020
4	REAL LOGX0A,LOGY0B	M	0030
5	COMMON /SUPP/ IFLO	M	0040
6	COMMON /FOR355/IO,DELS,XBRK(20),YBRK(20),XON(500),YON(500),DVOXOT500H	M	0050
7	1),ALPHA(500),CAPPA(500),SON(500),PIO180	M	0060
8	COMMON/SUPN/ XOT(2),YOT(2),LOGXOT(2),LOGYOT(2)	M	0070
9	COMMON /NHIGH/ NSPIG,MLAST,XLAST(500),YLAST(500)	M	0080
10	COMMON/SENSE/ XOT(2),YOT(2),A,B,INFLEC	M	0090
11	COMMON /SPREP/ KPREP	M	0100
12	COMMON/PAC/ PALE,DELSHL	M	0110
13	COMMON/TRVFIT/KOUNT	M	0120
14	IP1=0	M	0130
15	PI=3.14159265	M	0140
16	IF (IFLO.EQ.1) DELSHL=DELS1	M	0150
17	DELSIN=DELS1	M	0160
18	DELS2=DELS1	M	0170
19	KOUNT=0	M	0180
20	10 II=ISTART	M	0190
21	IF (IFLO.GE.1) DS=DELS2	M	0200
22	KOUNT=KOUNT+1	M	0210
23	I=ISTART	M	0220
24	X41=XBRK(I)	M	0230
25	Y41=YBRK(I)	M	0240
26	***** FOR SUBSEQUENT ITERATIONS, SKIP LL INITIALIZATION.	M	0250
27	IF (ILL.GE.5.OR.LL.LT.-5).AND.KOUNT.NE.1) GO TO 50	M	0260
28	LL=0	M	0270
29	***** FLAG FOR ENDP. CURVATURE MATCH IS YBRK(6)=200.	M	0280
30	IF (YBRK(6)*200.) 40,15,40	M	0290
31	15 CAP=-ABS(XBRK(6))	M	0300
32	LL=-8	M	0310
33	IF (XBRK(6).NE.999.) GO TO 50	M	0320
34	IF (XBRK(3).NE.XBRK(2)) GO TO 20	M	0330
35	***** IF THE PT. WHERE CURVATURES MUST MATCH HAS INFIN. SLOPE,	M	0340
36	***** OBTAIN CURVATURE FROM PREVIOUS SEGMENT'S EXPONENTS.	M	0350
37	IF (XBRK(2).NE.XBRK(1)) GO TO 25	M	0360
38	LL=-8	M	0370
39	CAP=-2.*A/XB(1)/B/B	M	0380
40	GO TO 50	M	0390
41	20 IF (XBRK(5).NE.XBRK(4)) GO TO 25	M	0400
42	LL=-8	M	0410
43	CAP=-2.*A/XB(1)/B/B	M	0420
44	GO TO 50	M	0430
45	25 IF (XBRK(6).EQ.999.)XBRK(6)= CAPPA(ISTART)	M	0440
46	IF (XBRK(6).EQ.0..OR.ABSTXBRK(6)).GE.99999.) GO TO 30	M	0450
47	CAP=-ABS(XBRK(6))	M	0460
48	GO TO 50	M	0470
49	30 WRITE (6,35)	M	0480
50	35 FORMAT(1H0,3X,99HREQUEST FOR SPECIFIC CURVATURE MUST BE MODIFIED OR	M	0490
51	IF WITHDRAWN. DESIRED CURVATURE CAN'T=0. OR INFINTY)	M	0500
52	STOP	M	0510
53	***** POINT-PLUS-SLOPE FLAG IS YBRK(6)= -100.	M	0520
54	40 IF (YBRK(6)*100.) 50,45,50	M	0530
55	45 LL=5	M	0540
56	***** ADDITIONAL FLAG FOR INFLECTION-POINT-PLUS-SLOPE IS YBRK(13)= 100H	M	0550

57	IF(YBRK(3).EQ.100.) LL=6	H	0560
58	C(1111FOR INITIAL GUESS OF UNKNOWN Y AT INFLECTION POINT,USE ENDP. AVG.	H	0570
59	IF(LL.EQ.6) YBRK(3)= YBRK(2)+(XBRK(2)-XBRK(3))/(XBRK(2)-XBRK(4))*H	H	0580
60	1YBRK(4)-YBRK(2))	H	0590
61	C(11111 CREATE A DUMMY POINT TO SIMULATE THE GIVEN SLOPE THROUGH	H	0600
62	C (XBRK(3),YBRK(3))	H	0610
63	YBRK(6)= XBRK(6)*(XBRK(3)+5.-XBRK(3)) +YBRK(3)	H	0620
64	XBRK(6)= XBRK(3)+5.	H	0630
65	50 DO 55 J=1,6	H	0640
66	XBK(1J+7)=XBRK(J)	H	0650
67	YBK(1J+7)=YBRK(J)	H	0660
68	55 CONTINUE	H	0670
69	IF (XBK(19).NE.XBK(18)) 60 TO 65	H	0680
70	IF (YBK(18).LT.YBK(19)) 60 TO 60	H	0690
71	SLOP=99999.	H	0700
72	SINATD=1.0	H	0710
73	SLOP=99999.	H	0720
74	ATDYDD=90.	H	0730
75	COSATD=0.0	H	0740
76	60 TO 80	H	0750
77	60 SLOP=-99999.	H	0760
78	SINATD=-1.0	H	0770
79	SLOP=-99999.	H	0780
80	ATDYDD=-90.	H	0790
81	COSATD=0.0	H	0800
82	60 TO 80	H	0810
83	65 SLOP=(YBK(19)-YBK(18))/(XBK(19)-XBK(18))	H	0820
84	SLOP=ATAN(SLOP)	H	0830
85	ATDYDD=SLOP/PI*180	H	0840
86	IF (XBK(18)-XBK(19)) 70,75,75	H	0850
87	70 SLOP=PI*SLOP	H	0860
88	ATDYDD=-ATDYDD	H	0870
89	PI=1	H	0880
90	75 SINATD=SIN(SLOP)	H	0890
91	COSATD=COS(SLOP)	H	0900
92	80 CONTINUE	H	0910
93	INFLEC = ABS(INT(SINATD))	H	0920
94	XBK(8)=XBK(18)	H	0930
95	YBK(8)=YBK(18)	H	0940
96	DO 85 J=9,13	H	0950
97	XP=XBK(1J)-XBK(18)	H	0960
98	YP=YBK(1J)-YBK(18)	H	0970
99	XBK(1J)=XBK(18)+XP*(COSATD+YP*SINATD	H	0980
100	YBK(1J)=YBK(18)-XP*SINATD+YP*COSATD	H	0990
101	85 CONTINUE	H	1000
102	Q=1.	H	1010
103	P=1.	H	1020
104	XBK(5)=XBK(8)	H	1030
105		H	1040
106	YBK(5)=YBK(8)	H	1050
107	XBK(6)=XBK(9)	H	1060
108	YBK(6)=YBK(9)	H	1070
	XBK(9)=XBK(10)	H	1080
	YBK(9)=YBK(10)	H	1090
	DELS=DELS	H	1100
	OSSAVE=DELS	H	1110
	XTH=XBK(19)	H	1120

114	YTH=YBK(9)	H	1130
115		H	1140
116	B=YBK(11)-YBK(6)	H	1150
117	TOMEGA=(XBK(12)-XPK(11))/(YBK(12)-YBK(11))	H	1160
118	IF (ABS(TOMEGA)-.LE..0001) TOMEGA=0.	H	1170
119	OMEGA=ATAN(TOMEGA)	H	1180
120	XU=XBK(6)+B*TOMEGA	H	1190
121	YU=YBK(11)	H	1200
122	A=XU-XBK(11)	H	1210
123	X19=XU-XBK(9)	H	1220
124	ETA9=YU-YBK(9)	H	1230
125	Y8(1)=ETA9	H	1240
126	X8(1)=X19-ETA9*OMEGA	H	1250
127	X8(2)=XU-XBK(13)-TOMEGA*(YU-YBK(13))	H	1260
128	Y8(2)=YU-YBK(13)	H	1270
129	B0A=B/A	H	1280
130	IF (LL.GE.5) GO TO 90	H	1290
131	IF (LL.LE.-6) GO TO 105	H	1300
132	LL=0	H	1310
133	IF (ENREED(1).GT.0.) P=ENREED(1)	H	1320
134	IF (ENREED(2).GT.0.) Q=ENREED(2)	H	1330
135	IF (P.EQ.1.) LL=1	H	1340
136	IF (Q.EQ.1.) LL=LL+2	H	1350
137	IF (LL.EQ.0) GO TO 115	H	1360
138	YU=XU(1)-X8(1)/A	H	1370
139	YOB(1)=Y8(1)/B	H	1380
140	LOGXU(1)=ALOG(XU(1))	H	1390
141	LOGYOB(1)=ALOG(YOB(1))	H	1400
142	IF (XBRK(6).EQ.0..AND..LL.EQ.3) GO TO 95	H	1410
143	IF (LL.NE.3) GO TO 100	H	1420
144	XU(2)=X8(2)/A	H	1430
145	YOB(2)=Y8(2)/B	H	1440
146	LOGXU(2)=ALOG(XU(2))	H	1450
147	LOGYOB(2)=ALOG(YOB(2))	H	1460
148	GO TO 100	H	1470
149	95 LL=4	H	1480
150	100 CALL FOMISUTP,Q,LL)	H	1490
151	GO TO 115	H	1500
152	C(1111) FOR CURVATURE MATCH, NO ITERATION REQD. BUT ONE EXPONENT MUST=2H	H	1510
153	105 IF (XBRK(3).EQ.XBRK(2)) P=2.	H	1520
154	IF (XBRK(3).EQ.XBRK(1)) Q=2.	H	1530
155	C(1111) OBTAIN OTHER EXPONENT FROM ENDPOINT CURVATURE RELATION	H	1540
156	110 IF (P.EQ.2.) Q=-2.*B/CAP/A/A	H	1550
157	IF (Q.EQ.2.) .AND. P.NE.2.) P=-2.*A/CAP/B/B	H	1560
158	115 IF (KOUNT.NE.1) GO TO 120	H	1570
159	WRITE (6,475) P,A,XC,Q,B,YU,OMEGA	H	1580
160	120 I=1	H	1590
161	ILO=1	H	1600
162	XU(1)=XU(1)	H	1610
163	C10N=1./P	H	1620
164	B1=B*OMEGA	H	1630
165	DX1=DELS*COSATD	H	1640
166	XP=X1W-XBK(18)	H	1650
167	YP=Y1W-YBK(18)	H	1660
168	X11ROT=XBK(18)+XP*COSATD+YP*SINATD	H	1670
169	X1=XU-X11ROT	H	1680
170	Y=YU-YBK(6)	H	1690

171	X=XI-Y*WOMEGA	H	1700
172	IF (X.LY.O.O) X=O.C	H	1710
173	DSM=SON(I)-SON(I-1)	H	1720
174	125 XOATON=(X/A)**P	H	1730
175	VOBYON=(Y/P)**Q	H	1740
176	C(1111) AVOID (L.E. O.)*(L.E. O.)	H	1750
177	IF (P.GE.1.) GO TO 135	H	1760
178	IF (X.NE.O.) GO TO 130	H	1770
179	XNMOAN=99999.	H	1780
180	GO TO 140	H	1790
181	130 XNMOAN=(1./X)**(1.-P)/A**P	H	1800
182	GO TO 140	H	1810
183	135 XNMOAN=X**IP-1./A**P	H	1820
184	140 IF (Q.GE.1.) GO TO 150	H	1830
185	IF (Y.NE.O.) GO TO 145	H	1840
186	YNMOBN=9999.	H	1850
187	GO TO 155	H	1860
188	145 YNMOBN=(1./Y)**(1.-Q)/B**Q	H	1870
189	GO TO 155	H	1880
190	150 YNMOBN=Y**IQ-1./B**Q	H	1890
191	155 FOFY=XOATON*VOBYON-1.	H	1900
192	IF (ABS(FOFY).LE.1.0E-5) GO TO 160	H	1910
193	FPOFY=Q*YNMOBN-WOMEGA*P*XNMOAN	H	1920
194	YNEW=Y-FOFY/FPOFY	H	1930
195	GO TO 165	H	1940
196	160 YNEW=Y	H	1950
197	165 IF (ABS(Y-YNEW)/YNEW-.1E-4) 175,175,170	H	1960
198	170 Y=YNEW	H	1970
199	X=XI-Y*WOMEGA	H	1980
200	GO TO 125	H	1990
201	175 Y=YNEW	H	2000
202	X=XI-Y*WOMEGA	H	2010
203	180 ETA=1	H	2020
204	DELS=DELS2	H	2030
205	IPN=1	H	2040
206	IF (X.LY.O.O) X=O.C	H	2050
207	C(1111) AVOID (L.E. O.)*(L.E. O.)	H	2060
208	IF (P.GE.1.) GO TO 190	H	2070
209	IF (X.NE.O.) GO TO 185	H	2080
210	XOANMI=99999.	H	2090
211	GO TO 195	H	2100
212	185 XOANMI=(A/X)**(1.-P)	H	2110
213	GO TO 195	H	2120
214	190 XOANMI=(X/A)**IP-1.)	H	2130
215	195 IF (Q.GE.1.) GO TO 205	H	2140
216	IF (Y.NE.O.) GO TO 200	H	2150
217	YOBMMI=99999.	H	2160
218	GO TO 210	H	2170
219	200 YOBMMI=(B/Y)**(1.-Q)	H	2180
220	GO TO 210	H	2190
221	205 YOBMMI=(Y/B)**IQ-1.)	H	2200
222	210 F=(XOANMI/A)**P	H	2210
223	FZ=(YOBMMI/B)**Q	H	2220
224	3=1/OMEGA*F1	H	2230
225	IF (X.GE.O.O) GO TO 225	H	2240
226	IF (P.GE.2.) GO TO 215	H	2250
227	F107-P*(1./Y)**(2.-P)/X**P	H	2260

228	60 TO 220	H	2270
229	215 F10X=P*X** (P-2.) / A**P	H	2280
230	220 60 TO 230	H	2290
231	225 IF (P.EQ.2.) F10X=2.0 / (A**A)	H	2300
232	IF (P.GT.2.) F10X=0.0	H	2310
233	230 IF (Y.EQ.0.0) 60 TO 240	H	2320
234	IF (Q.GE.2.) 60 TO 235	H	2330
235	F20Y=Q*(1./Y)**(2.-Q) / B**Q	H	2340
236	60 TO 245	H	2350
237	235 F20Y=Q*Y** (Q-2.) / B**Q	H	2360
238	60 TO 245	H	2370
239	240 IF (Q.EQ.2.) F20Y=2. / (B**B)	H	2380
240	IF (Q.GT.2.) F20Y=0.	H	2390
241	245 DEN=F2-F3	H	2400
242	IF (DEN.NE.0.0) 60 TO 250	H	2410
243	DETOXI=99999.	H	2420
244	60 TO 255	H	2430
245	250 DETOXI=-F1/DEN	H	2440
246	255 OYUXO(IPN)=DETOXI	H	2450
247	CIMEPT=1.-DETOXI*OMEGA	H	2460
248	IF (IPN.EQ.1) STARTI.AND.LL.EE.-61 60 TO 260	H	2470
249	C111111 ELIMINATE CASES OF UNDEFINED CURVATURE	H	2480
250	IF (X.EQ.0..AND.P.LT.2.) 60 TO 265	H	2490
251	IF (Y.EQ.0..AND.Q.LT.2.) 60 TO 265	H	2500
252	G1=IP-1.) F10X=CIMEPT	H	2510
253	SAND1=DEN*G1	H	2520
254	SAND2=F1*ITQ -1.) F20Y*DETOXI=G1*OMEGA	H	2530
255	IF (ABS(DETOXI).GT.1.E 11) DETOXI=1.E 11	H	2540
256	RKT=11.*DETOXI**2) W1=5	H	2550
257	CAPPA(IPN)=(SAND2-SAND1) / (DEN**2*RKT)	H	2560
258	IF (P.EQ.2..AND.X.EQ.0.) CAPPA(IPN)=-2.*B/Q/A/A	H	2570
259	IF (Q.EQ.2..AND.ABS(Y).LT.1.E-4) CAPPA(IPN)=-2.*A/P/B/B	H	2580
260	60 TO 270	H	2590
261	260 IF (P.EQ.2.) CAPPA(IPN)=-2.*B/Q/A/A	H	2600
262	IF (Q.EQ.2.) CAPPA(IPN)=-2.*A/P/B/B	H	2610
263	60 TO 270	H	2620
264	265 CAPPA(IPN)=99999.	H	2630
265	270 ALPHA(IPN)=ATAN(DYEXO(IPN)) / PI0180	H	2640
266	XONTIPN=XO-XI	H	2650
267	YON(IPN)=YO-ET	H	2660
268	OYI=DELS*SINATD	H	2670
269	IF (IFLD.GE.1) 60 TO 275	H	2680
270	OS=DELS/11.0**20*1ANHVARSTCAPPA(I))	H	2690
271	60 TO 285	H	2700
272	275 IF (IFLD.GT.1.AND.IIPN-ISTARTI).GT.3) 60 TO 280	H	2710
273	DS=DS-PAE*DS	H	2720
274	60 TO 290	H	2730
275	280 DS=DS+1.5*PAE*DS	H	2740
276	IF (OS.GT.DELSHL) ES=DELSHL	H	2750
277	60 TO 290	H	2760
278	285 IF (ABS(DS-DELS).GT.20*DELS) DS=DELS*SIGN1.20*DELS,DS-DELS	H	2770
279	290 IF (DS.LT..02*DELS2) DS=.02*DELS2	H	2780
280	IF (IFLD.GT.0.AND.(I-ISTARTI).GT.200) 60 TO 410	H	2790
281	IF (INSPHG.EQ.0) 60 TO 295	H	2800
282	OXI=ABSTXLASTINLASTI-XLASTINLASTI-1)	H	2810
283	OX11=OX1	H	2820
284	OYI=ABSTYLASTINLASTI-YLASTINLASTI-1)	H	2830

285	NLAST=NLAST-1		
286	295 IF (ABS(DELTXI)-1.) 320,320,300	H	2840
287	300 DYI=DS/SQRT(1.+1./DELTXI**2)	H	2850
288	IF (INSPHG.NE.0) DYI=DXI	H	2860
289	305 YTH=YON(1)+DYI	H	2870
290	IF (YTH-YBK(11)) 310,390,390	H	2880
291	310 ETA=Y0-YTH	H	2890
292	Y=ETA	H	2900
293	C	H	2910
294	C STRAIGHT SECTION BETWEEN POINTS 11 AND 12 MUST HAVE SLOPE ABOVE 1	H	2920
295	C	H	2930
296	C	H	2940
297	C X MAY NOT BE TESTED AGAINST XBK(11)	H	2950
298	C	H	2960
299	X=X(11)-(Y/B)**Q)**CION	H	2970
300	XI=X+Y*OMEGA	H	2980
301	XTH=X0-XI	H	2990
302	DXI=XTH-XON(11)	H	3000
303	DELTAS=SQRT(DYI**2+DXI**2)	H	3010
304	IF (DELTAS.GT.1.02*DS.AND.IPN.NE.1.AND.NSPHG.EQ.0) 60 TO 315	H	3020
305	60 TO 390	H	3030
306	315 DYI=DS*DYI/DELTAS	H	3040
307	60 TO 305	H	3050
308	320 DXI=DS/SQRT(1.+DELTXI**2)	H	3060
309	IF (INSPHG.NE.0) DXI=DYI	H	3070
310	IF (INSPHG.NE.0.AND.IPI.EQ.1) DXI=DXI1	H	3080
311	325 XTH=XON(11)+SIGN(DXI,DELTXI)	H	3090
312	IF (DELTXI.EQ.0.) XTH=XON(11)-DXI	H	3100
313	XI=X0-XTH	H	3110
314	Y=Y0-YON(11)	H	3120
315	330 X=XI-Y*OMEGA	H	3130
316	IF (X.LT.0.0) X=0.0	H	3140
317	XOATON=IX/X**P	H	3150
318	YOBTON=(Y/B)**Q	H	3160
319	CITTTT AVOID (L.E. 0.)**1.L.E. 0.)	H	3170
320	IF (P.GE.1.) 60 TO 340	H	3180
321	IF (X.NE.0.) 60 TO 335	H	3190
322	XNMOAN=99999.	H	3200
323	60 TO 345	H	3210
324	335 XNMOAN=(1./X)**(1.-P)/A**P	H	3220
325	60 TO 345	H	3230
326	340 XNMOAN=X**(P-1.)//A**P	H	3240
327	345 IF (Q.GE.1.) 60 TO 355	H	3250
328	IF (Y.NE.0.) 60 TO 350	H	3260
329	YNMOBN=99999.	H	3270
330	60 TO 360	H	3280
331	350 YNMOBN=(1./Y)**(1.-Q)/B**Q	H	3290
332	60 TO 360	H	3300
333	355 YNMOBN=Y**Q-1./B**Q	H	3310
334	360 FOFY=XOATON+YOBTON-1.	H	3320
335	IF (ABS(FOFY).LE.1.E-5) 60 TO 365	H	3330
336	FOFY=Q+YNMOBN-OMEGA*P*XNMOAN	H	3340
337	YNEW=Y+FOFY/FOFY	H	3350
338	60 TO 370	H	3360
339	365 YNEW=Y	H	3370
340	370 IF (ABS(Y-YNEW)/YNEW-.1E-4) 380,380,375	H	3380
341	375 Y=YNEW	H	3390
		H	3400

342	GO TO 330	H	3410
343	380 Y=YNEW	H	3420
344	X=XI-Y*YONEGA	H	3430
345	YTM=Y0-Y	H	3440
346	XI=X+Y*YONEGA	H	3450
347	C DYI=YTM-YONI(1-1)	H	3460
348	DYI=YTM-YONI(1)	H	3470
349	DELTA=SQRT(DYI**2+DXI**2)	H	3480
350	IF (DELTA*.6T+.1.02*DS.AND.IPW.NE.1.AND.WSPWG.EQ.0) GO TO 385	H	3490
351	GO TO 390	H	3500
352	385 DXI=DS*DXI/DELTA	H	3510
353	GO TO 325	H	3520
354	390 SONTI=SONTI-1+DSF	H	3530
355	DSH=DS	H	3540
356	IF (WSPWG.NE.0) DS=DXI	H	3550
357	IF (ABS(YTM-YBK(11)).LT..001*DS) GO TO 400	H	3560
358	I=I+1	H	3570
359	IF (YTM-YBK(11)) 180,395,395	H	3580
360	395 IHI=I-1	H	3590
361	GO TO 405	H	3600
362	400 IHI=I	H	3610
363	I=I+1	H	3620
364	405 XIM=XBK(11)	H	3630
365	IIMI=IHI	H	3640
366	410 IF (IFLO.EQ.0) (I-1)START).LT.2) GO TO 415	H	3650
367	IF (IFLO.EQ.1)	H	3660
368	IF (IFLO.EQ.2)	H	3670
369	I=ILO-1	H	3680
370	GO TO 10	H	3690
371	415 DO 455 J=ILO,IHI	H	3700
372	XP=XONI(J)-XBK(11)	H	3710
373	YP=YONI(J)-YBK(11)	H	3720
374	XONI(J)=XBK(11)+XP*COSATD+YP*SINATD	H	3730
375	YONI(J)=YBK(11)+YP*COSATD+XP*SINATD	H	3740
376	DEL22=DEL2	H	3750
377	IF (J.NE.IHI) GO TO 445	H	3760
378	DTEST=(XONI(IHI)-XONI(IHI-1))*2+(YONI(IHI)-YONI(IHI-1))*2**5	H	3770
379	IF (INONT.6T.150) GO TO 445	H	3780
380	IF (ABS(DS-DTEST)) .LT..1*DS) GO TO 420	H	3790
381	IF (DTEST.LT..01*DS) GO TO 425	H	3800
382	IF (IHI.EQ.ILO) GO TO 440	H	3810
383	IF (ABS(DEL22-DTEST).LT..001*DS) GO TO 435	H	3820
384	IF (DTEST.LT..5*DS) GO TO 435	H	3830
385	IF (DTEST.6T..5*DS) GO TO 430	H	3840
386	***** VIA BOTTOM 17574	H	3850
387	420 IIMI=IIMI+1	H	3860
388	I=I+1	H	3870
389	IONE=IIMI-1	H	3880
390	SONI(IHI)=SONI(IONE)+DTEST	H	3890
391	425 IIMI=IIMI+1	H	3900
392	IONE=IIMI-1	H	3910
393	XONI(IONE)=XNI	H	3920
394	YONI(IONE)=YNI	H	3930
395	GO TO 445	H	3940
396	430 IF (IFLO.6T.1) GO TO 435	H	3950
397	DEL22=(FLOAT(IHI-ILO)*DEL22+DTEST)/FLOAT(IHI+1-ILO)	H	3960
398	IF (INONT.6T.10) DEL22 (DEL22+DEL22)/2.0	H	3970

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399 I=ILO-1 M 3980
400 60 TO 10 M 3990
401 435 DELS2=DELS2+DSTEST*FLOAT(IMI-IL0) M 4000
402 IF (KOUNT-6E-10) DELS2=(DELS2+DEL22)/2.0 M 4010
403 IF(ILO-6T-1-AND-1DS/DEL22).GT.2.1 M 4020
404 IPACE=PACE+1.1+DSTEST/DEL22+(FLOAT(IMI-IL0)*1.1+1.5*PACE)*FLOAT(IMI) M 4030
405 2I-IL0-11-(11.1+1.5*PACE)*FLOAT(IY-IL0)-1.171.57*PACE)) M 4040
406 IF(ILO-6T-1-AND-1DS/DEL22).GT.2.1)DELS2=DEL22 M 4050
407 I=ILO-1 M 4060
408 60 TO 10 M 4070
409 440 DELS2=2*DELS2 M 4080
410 I=ILO-1 M 4090
411 60 TO 10 M 4100
412 445 ALPHA(IJ)=ALPHA(IJ)-ATODDD M 4110
413 IF (ABSTABSTALPHA(IJ)-90.7).LE-1E-4) 60 TO 450 M 4120
414 DYDX(IJ)=TAN(ALPHA(IJ))*PI0100) M 4130
415 60 TO 455 M 4140
416 450 DYDX(IJ)=SIGN(999.9,ALPHA(IJ)) M 4150
417 455 CONTINUE M 4160
418 IMI=10NE M 4170
419 IF (KOUNT-6T-1507) WRITE (6,4807)YBRKT(I08),YBRKT(I08),I08=1,5) M 4180
420 DELS1=DS M 4190
421 IF (ILO-6T-1)DELS1=1.1*DS M 4200
422 IF (DELS1-6T-DELS2-AND-IFLO-LE-1.1)DELS1=DELS2 M 4210
423 DO 465 J=ILO,IMI M 4220
424 IF (J-6E-1) 60 TO 460 M 4230
425 SUM(IJ)=SUM(IJ-1)+SUM(IYBRKT(IJ)-XOM(IJ-1))42+IYOM(IJ)-YOM(IJ-1))42) M 4240
426 60 TO 465 M 4250
427 460 SUM(IJ)=0.0 M 4260
428 465 CONTINUE M 4270
429 WRITE (6,485)KOUNT M 4280
430 WRITE (6,490)DELS1,DELS2,DELS1,DSTEST M 4290
431 IF (ILO-6T-1) WRITE (6,470)IPACE M 4300
432 470 FORMAT(1H,30X,'FINAL PACE= ',F8.5) M 4310
433 10-1 M 4320
434 X8(11)=P M 4330
435 X8(12)=Q M 4340
436 X0IF=XBRK(4)-XBRK(5) M 4350
437 IF(ABSTX0IF).LT-1E-15)X0IF=SIGN(X0IF,-15*X0IF) M 4360
438 DYDX(10-1)=(YBRK(4)-YBRK(5))/X0IF M 4370
439 IF(ABS(DYDX(10-1))-177.6T-99999.70YDX(10-1)-17)ST6M199999.9UYDX(10-1)-17)M 4380
440 ALPHA(10-1)=ATAN(0YDX(10-1))/PI0180 M 4390
441 IF(ICAPPA(10-1)-EQ-1E-6-AND-P-LE-2.2)XRD=2.2ICAPPA(10-1))-2.2*P/P78M 4400
442 1/B M 4410
443 RETURN M 4420
444 C M 4430
445 C M 4440
446 475 FORMAT(1X/19X, M 4450
447 1 MHP = ,E16-8,X4,MH = ,E16-8,7X,5HXU = ,E16-8,7/19X, M 4460
448 2 M 4470
449 12 ,E16-8,X4,MH = ,E16-8,7X,5HYU = ,E16-8,3X,8HME6M = ,E16-8,7/19X M 4480
450 480 FORMAT (1H0,15X, M 4490
451 1 BUNTHIS SET OF DATA EXCEEDED 150ITERATIONS CALCULATION M 4500
452 15 STOPPED/20X,4HXBRK-5X,4HVBK/2CX,1PI0E10.3) M 4510
453 485 FORMAT (21X, 13,2X,13HITERATIONS---) M 4520
454 490 FORMAT(121X,10HDELS IM = ,F8.5,3X,7HDELS = ,F8.5,3X,11HDELS OUT = ,M 4530
455 1F8.5,3X,9HDELS-EST = ,F8.5) M 4540
456 END M 4550

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57	DFDQA2 = Y0BTQ2 * LOGY0B(2)	I	0560
58	F2 = XOATP2 + Y0BTQ2 - 1.	I	0570
59	UFOPA2 = XOATP2*LOGX0A(2)	I	0580
60	40 QN = (F1*DFDPA2/(DFDQA2*DFDPA1)-F2/DFDQA2)/(1. - DFDPA2*DFDQA1/(DFI	I	0590
61	IDPA1*DFDQA2)) + Q	I	0600
62	PN = ((Q-QN)*DFDQA1 - F1)/DFDPA1 + P	I	0610
63	45 TESTP=ABS(PN-P)/P -TOL	I	0620
64	TESTQ=ABS(QN-Q)/Q -TOL	I	0630
65	Q=QN	I	0640
66	P=PN	I	0650
67	IF (TESTP.GT.0..QN.TESTQ.GT.0.) 60 TO 25	I	0660
68	50 RETURN	I	0670
69	55 PN= -F1/DFDPA1 + P	I	0680
70	TESTP=ABS(PN-P)/P -TOL	I	0690
71	P=PN	I	0700
72	IF (TESTP) 50,25,25	I	0710
73	60 QN= -F1/DFDQA1 + Q	I	0720
74	TESTQ=ABS(QN-Q)/Q -TOL	I	0730
75	Q=QN	I	0740
76	IF (TESTQ) 50,25,25	I	0750
77	65 DFDQA1= Y0BTQ1*LOGY0B(1)	I	0760
78	PN=-F1/(DFDPA1+DFDQA1) +P	I	0770
79	TESTP= ABS(PN-P)/P -TOL	I	0780
80	P=PN	I	0790
81	Q=P	I	0800
82	IF (TESTP) 50,25,25	I	0810
83	70 DYDX= (Y(2)-Y1)/X(2)-X1	I	0820
84	F2=DYDX+ P*XOATP1*Y(1)/Q/Y0BTQ1/X(1)	I	0830
85	E0CALD= - P*XOATP1*Y(1)/Q/Y0BTQ1/X(1)	I	0840
86	DFDQA2= - (ALOG(B)-1./Q-ALOG(Y(1))) *E0CALD	I	0850
87	UFOPA2= - (1./P-ALOG(A)+ALOG(X(1))) *E0CALD	I	0860
88	IF (LL.EQ.6) 60 TO 75	I	0870
89	60 TO 40	I	0880
90	75 E0CALD= - P*XOATP1*Y(1)/Q/Y0BTQ1/X(1)	I	0890
91	G=F2	I	0900
92	D6DP= DFDPA2	I	0910
93	UG0Q= DFDQA2	I	0920
94	H = E0CALD*(1P-1./X(1)+(1.-Q)/Y(1))*E0CALD)	I	0930
95	DHDP = E0CALD*(1P-1./X(1)+(1.-Q)/Y(1))*E0CALD)*(1./P-ALOG(A)*AL	I	0940
96	LOG(X(1))*E0CALD*(1./X(1) +(1.-C)/Y(1))*E0CALD*(1./P-ALOG(A)+ALOG	I	0950
97	2(X(1)))	I	0960
98	DH0Q = E0CALD*(1P-1./X(1)+(1.-Q)/Y(1))*E0CALD)*(-1./Q+ALOG(B)-A	I	0970
99	LOG(Y(1))*E0CALD*(1-E0CALD/Y(1)+(1.-Q)/Y(1))*E0CALD*(1./Q+ALOG(B)	I	0980
100	2)-ALOG(Y(1)))	I	0990
101	DFDY= Q/Y(1)*Y0BTQ1	I	1000
102	IF(INFLEC.EQ.1)DFDY=P/X(1)*XOATP1	I	1010
103	D6DY= E0CALD*U=-1./Y(1)	I	1020
104	IF(INFLEC.EQ.1)	I	1030
105	D6DY=E0CALD*(1.-P)/X(1)	I	1040
106	E0CAL2=2.*E0CALD	I	1050
107	DHNY= E0CALD*(1.-1./X(1)+(1.-C)/Y(1))*E0CAL2*(1.-Q)/Y(1)+E0CAL	I	1060
108	10*(Q-1./Y(1)/Y(1))	I	1070
109	IF(INFLEC.EQ.1)	I	1080
110	DHNY=E0CALD/X(1)*(1P-1.)*(1P-2.)/X(1)+2.*E0CALD*(1.-Q)/Y(I	1090
111	Y(1))	I	1100
112	1 QN51 DFDPA1*(LOG(Q/DHNY-D6DY*DHC0)-DFDQA1*(D6DP*D+DY-D6DY*DNDP)+	I	1110
113	10*Q+10*(Q+UG0Q*DNDP)	I	1120

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114 QN=Q*(F1*(D6DP*DNDY-DGDY*DNDP)-G*(DFDPAI*DNDY-DFDY*ENDP)+H*(DFDPAI
115 1*DGDY-DFDY*D6DP))/WRONSK I 1130
116 PN=P*(F1*(D6DQ*DNDY-DGDY*DNDU)+G*(DFDQAI*DNDY-DFDY*ENDU)-H*(DFDQAI
117 11*DGDY-DFDY*D6DQ))/WRONSK I 1140
118 YN=Y*(F1*(DNDP*(P-PN)+DNDG*(Q-QN)-H)/DNDY I 1150
119 IF(INFLEC.EQ.1) I 1160
120 1 YN=X*(F1*(DNDP*(P-PN)+DNDG*(Q-QN)-H)/DNDY I 1170
121 TESTY=ABS(YN-Y(1))/Y(1) - 1.E-05 I 1180
122 IF(INFLEC.EQ.1) I 1190
123 1 TESTY=ABS(YN-X(1))/X(1) - 1.E-05 I 1200
124 IF(INFLEC.NE.1) Y(1)=YN I 1210
125 IF(INFLEC.EQ.1) I 1220
126 IX(1)=YN I 1230
127 IF (TESTY) 45,45,80 I 1240
128 80 Q=QN I 1250
129 P=PN I 1260
130 GO TO 25 I 1270
131 85 FORMAT(1H0,42H THIS POINT IS OUTSIDE THE MAGIC TRIANGLE,,2E15.4/ I 1280
132 2 52H I 1290
133 1 THIS CONDITION IS NOT VALID FOR ANY CASE. REVISE I 1300
134 90 FORMAT(1H0,42H THIS POINT IS BELOW THE MAGIC TRIANGLE,,2E15.4/52H I 1310
135 1H THIS CONDITION IS VALID ONLY FOR THE BISUPERELLIPSE/72H WITH INF I 1320
136 ILECTIONIP OR Q LESS THAN 1.). SUCH A CURVE HAS BEEN GENERATED I 1330
137 CIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIII I 1340
138 END I 1350

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1      SUBROUTINE CUBIC (N)
2      C
3      C FIT A CUBIC BETWEEN 2 STRAIGHT LINES -- RESTRICTION -- THE STRAIGHT
4      C LINES CANNOT BE VERTICAL
5      C
6      DIMENSION A(4,4), B(4)
7      COMMON /MAIN/ XIN(10),YIN(10),DELSX,P102,DELS1,INLB
8      COMMON /FORSS/IO,DELS,XB(120),YB(120),XONT500,YONT500,DYDX(1500)
9      10,ALPHA(1500),CAPPA(1500),SON(500),P10180
10     COMMON /SS/ MBDY1,MBDY2,TPBDY,MBDYS
11     DELSIN=DELS1
12     DELS=DELS1
13     NOUNT=0
14     N-K-1
15     KSTART=K
16     X2=XIN(12)
17     X3=XIN(13)
18     Y2=YIN(12)
19     Y3=YIN(13)
20     SLOP2=(YIN(11)-Y3)/(XIN(11)-X3)
21     C
22     C SETUP A X Y MATRIX OF COEFFICIENTS
23     C
24     A(1,1)=1.0
25     A(1,2)=X2
26     A(1,3)=X2**2
27     A(1,4)=X2**3
28     A(2,1)=0.0
29     A(2,2)=1.0
30     A(2,3)=X3**2
31     A(2,4)=3.0*X2**2
32     A(3,1)=1.0
33     A(3,2)=X3
34     A(3,3)=X3**2
35     A(3,4)=X3**3
36     A(4,1)=0.0
37     A(4,2)=1.0
38     A(4,3)=2.0*X3
39     A(4,4)=3.0*X3**2
40     DO 10 I=1,N
41     10 CONTINUE
42     C
43     C SETUP VECTOR OF ORIGINAL CONSTANTS -- BB
44     C
45     BB(1)=Y2
46     BB(2)=(Y2-YIN(1))/(X2-XIN(1))
47     BB(3)=Y3
48     BB(4)=(YIN(1)-Y3)/(XIN(1)-X3)
49     MSIM=4
50     KSIM=0
51     CALL SIMQ (AA,BB,MSIM,MSIM)
52     D=BB(1)
53     C=BB(2)
54     X=BB(4)
55     C=BB(12)
56     B=BB(13)

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[illegible]

1	SUBROUTINE LEM (K)	L	0000
2	C	L	0010
3	C SUBROUTINE TO CALCULATE POINTS ON A LEMNISCATE	L	0020
4	C	L	0030
5	COMMON /FOR355/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDXO(500)	L	0040
6	1),ALPHA(500),CAPPA(500),SON(500),PI0180	L	0050
7	COMMON /SS/ NBDY1,MBDY2,TYPBDY,NBDYS	L	0060
8	COMMON /HAIN/ XINT(10),YINT(10),DELSMX,PI02,DELS1,THUB	L	0070
9	DELSIN=DELS1	L	0080
10	K=K-1	L	0090
11	KSTART=K	L	0100
12	DELS=DELS1		
13	KOUNT=0	L	0120
14	IF (YINT(1).EQ.YINT(2)) GO TO 30	L	0130
15	IF (XINT(1).EQ.XINT(2)) GO TO 10	L	0140
16	SLOPE=(YINT(2)-YINT(1))/(XINT(2)-XINT(1))	L	0150
17	AROT=-TAN(SLOPE)	L	0160
18	GO TO 15	L	0170
19	10 SLOPE=99999.	L	0180
20	AROT=-PI02	L	0190
21	15 DO 20 IROT=1,3	L	0200
22	XN=XINT(IROT)	L	0210
23	XIN(IROT)=XN+COS(AROT)-YIN(IROT)*SIN(AROT)	L	0220
24	20 YIN(IROT)=XN*SIN(AROT)+YIN(IROT)*COS(AROT)	L	0230
25	25 K=KSTART	L	0240
26	30 XONTK(1)=XINT(2)	L	0250
27	THETMX=ATAN(ABS((YIN(3)-YIN(2))/(XIN(3)-XIN(2))))	L	0260
28	K=SORTE((XINT(3)-XINT(2))*2+(YINT(3)-YINT(2))*2)/((2.*C*SINT2.0*THETMX	L	0270
29	1)))	L	0280
30	YONTK(1)=YIN(2)	L	0290
31	CAPPA(K+1)=0.0	L	0300
32	DYDXOK(K+1)=0.0	L	0310
33	ALPHA(K+1)=0.0	L	0320
34	KOUNT=KOUNT+1	L	0330
35	DSSAVE=DELS	L	0340
36	DS=DELS	L	0350
37	DTMET=DS**2/A**2	L	0360
38	THET=DTMET*.5	L	0370
39	35 R=A*SORT(2.0*SIN(2.0*THET))	L	0380
40	DSCHK=RCOS(THET)	L	0390
41	IF (DSCHK.GT.1.1*DS) GO TO 40	L	0400
42	IF (DSCHK.LT..9*DS) GO TO 45	L	0410
43	DELS=DS	L	0420
44	GO TO 50	L	0430
45	40 THET=THET+.02*DTMET	L	0440
46	GO TO 35	L	0450
47	45 THET=THET+.02*DTMET	L	0460
48	GO TO 35	L	0470
49	50 K=K+1	L	0480
50	55 R=A*SORT(2.0*SINT2.0*THET))	L	0490
51	XON(K+1)=XIN(2)-R*COS(THET)	L	0500
52	YON(K+1)=YIN(2)+R*SINT(THET)	L	0510
53	SON(K+1)=SON(K)+SQRT((XON(K+1)-XON(K))**2+(YON(K+1)-YON(K))**2)	L	0520
54	IF (ABS(SON(K+1)-SON(K)).GT.1.05*DS) GO TO 60	L	0530
55	IF (ABS(SON(K+1)-SON(K)).LT..95*DS) GO TO 65	L	0540
56	GO TO 70	L	0550

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57 60 THEY=THEY-.02*OTHEY L 0560
58 60 TO 55 L 0570
59 THEY=THEY+.02*OTHEY L 0580
60 60 TO 55 L 0590
61 DYDUTKRTT=-TAN(1.0*THEY) L 0600
62 ALPHA(K+1)=-3.0*THEY L 0610
63 CAPTRKRTT=3.0*SQR(TSINT2-OTHEY**2-.077A L 0620
64 DS=DELS/SORT(1.0+ABS(CAPTRK+1)) L 0630
65 IF TBSSTDS=DS*SAVE) .01+.25*DS*SAVE) DS=DS*SAVE,DS=DS* L 0640
66 1AVE) L 0650
67 DS*SAVE=DS L 0660
68 OTHET=DS*SQR(TSINT2-OTHEY**2-.077A L 0670
69 THEY=THEY+OTHEY L 0680
70 IF (THEY-LE.THEYMX) 60 TO 60 L 0690
71 IF TROUNT-67.501 60 TO 95 L 0700
72 DSTEST=((XOM(K+1))-XIM(3))**2+((YOM(K+1))-YIM(3))**2)*.5 L 0710
73 IF (DSTEST-67.05) 60 TO 90 L 0720
74 IF (DSTEST-67.0001005) 60 TO 75 L 0730
75 IF (DSTEST-67.0001005) 60 TO 75 L 0740
76 75 YOM(K+1)=YIM(3) L 0750
77 XOM(K+1)=XIM(3) L 0760
78 60 TO 95 L 0770
79 80 DELS=DELS-DSTEST/FLQAT(K+1)*KSTART) L 0780
80 60 TO 25 L 0790
81 85 DELS=DELS-DSTEST/FLQAT(K+1)*KSTART) L 0800
82 60 TO 25 L 0810
83 90 DELS=.8*DELS L 0820
84 60 TO 25 L 0830
85 95 DELS=.5*DELS L 0840
86 IF (DELS-67.05) DELS=DELS L 0850
87 WRITE (6,120)DELS,DELS,DELS L 0860
88 WRITE (6,120)DELS,DELS,DELS L 0870
89 KEND=X+1 L 0880
90 KSTART=KSTART+1 L 0890
91 IF (YIM(27)-67.05) 60 TO 105 L 0900
92 DO 100 KROT=KSTART,KEND L 0910
93 KX=XOM(KROT) L 0920
94 XOM(KROT)=XOM(COS(AROT)+YOM(KROT)*SIN(AROT)) L 0930
95 YOM(KROT)=YOM(KROT)*COS(AROT)-XOM(KROT)*SIN(AROT) L 0940
96 ALPHA(KROT)=ALPHA(KROT)-AROT L 0950
97 DYDUTKRTT=-TAN(1.0*ALPHA(KROT)) L 0960
98 100 CONTINUE L 0970
99 105 DO 110 KAL=KSTART,KEND L 0980
100 110 ALPHA(KAL)=ALPHA(KAL)/PI*10180 L 0990
101 RETURN L 1000
102 C L 1010
103 C L 1020
104 115 FORMAT(10X,13,2X,13HITERATION#---,3X,13HTHEYXCALC = ,F10.5,3X,8HAL L 1030
105 ICALC = ,F10.5) L 1040
106 120 FORMAT(10X,10HDELS IN = ,F8.5,3X,7HDELS = ,F8.5,3X,11HDELS OUT = ,F L 1050
107 1060 L 1060
108 END L 1070

```

```

1 SUBROUTINE MIRROR (K,KTOT,KSTART,YCL)
2 C
3 C USED FOR 22Y - 2-D INLETS
4 C
5 COMMON /FOR355/IO,DELS,YBK(120),YBK(120),XON(1500),YON(1500),DYDXO(1500),
6 Y1,ALPHAT(500),CAPPAT(500),SON(500),PIOT(60)
7 COMMON /55/ NBDY1,NBDY2,TYPBDY,NBDYS
8 DO 10 J=1,KTOT
9 ISTAR=KSTART+KTOT-J
10 XON(I)=XON(ISTAR)
11 YON(I)=2-D*YCL-YON(ISTAR)
12 CAPPAT(I)=-CAPPAT(ISTAR)
13 DYDXO(I)=-DYDXO(ISTAR)
14 ALPHAT(I)=-ALPHAT(ISTAR)
15 SON(I)=SON(ISTAR)
16 IF(12.NE.KTOT)K=K+1
17 10 CONTINUE
18 RETURN
19 C
20 END

```

```

1      SUBROUTINE XYCALC(KSTART,M2,MIN)
2      C      -- XYCALC --
3      C.....GENERATES DATA FILES FOR ON-BODY POINTS.
4
5      INTEGER SGEN
6      REAL X(1300),Y(1300), C(1300),SI(1300),SP(1400)
7      COMPLEX Z(1300),DZ(1300),DZ2,ZZ,FZTRP,DZTRP
8      LOGICAL THIN,EVEN,SPGEN,DONE,BYPASS
9      COMMON/SEGNO/NSEG,J,ENREFD
10     COMMON /MAIN/ XINT(10),YINT(10),DELSHX,PI02,DELSI,INCB
11     COMMON /SPGEN/ A,DSMAX,RMAX,TMIN,B,TMIN,DSENC
12     COMMON /FORSS/IO,DELS,XBR(20),YBR(20),XON(1500),YON(1500),DVOX(1500)
13     1),ALPHA(1500),CAPPA(1500),SON(1500),PI0180
14     LOGICAL TRANSX,TRANSY,SCALEX,SCALEY,CHANGE
15     COMMON/TRANSF/ XTRAN,YTRAN,XSCALE,YSCALE,
16     1 TRANSX,TRANSY,SCALEX,SCALEY,CHANGE,XTHAX
17     COMMON/TRANXY/ 2
18     DATA NSMAX,NSS,EMPTY,ONE/300,200,1.0E20,1.0001/
19     NAMELIST /BODYIN/ Z,SI
20     NAMELIST /AUXIN/ A,DSMAX,RMAX,TMIN,B,MFIN,SP,NSP,DSEND,DONE,
21     1 EVEN,TMIN,BYPASS
22
23     C.....INITIALIZE PROGRAM.
24     DO 15 I=1,NSMAX
25     15 Z(I)=EMPTY
26     SI=0.0
27     NSMAX=400
28     A=.50
29     DSMAX=DELSHX
30     TMIN=.FALSE.
31     RMAX=1.2
32     DSEND=DSMAX
33     DONE=.FALSE.
34     EVEN=.FALSE.
35     BYPASS=.FALSE.
36     B=0.3
37     TMIN=0.1
38     MFIN=0
39     NSP=0
40
41     C.....INPUT BODY POINTS AND BODY TYPE.
42     DO 25 I=1,NSMAX
43     25 READ (MIN,BODYIN)
44     IF (REAL(Z(I)).EQ.EMPTY) GO TO 30
45     25 NS=I
46     30 SI(I)=SI
47
48     C
49     C TRANSFORM THE COORDINATES,Z,IF REQUIRED
50     IF (CHANGE) CALL XYTRANS(J,NSEG,ENREFD,NS)
51     IBAD=SGEN(1,Z,NS)
52     IF (IBAD.NE.0) WHILE (16,125)IBAD
53
54     C.....SET UP DERIVATIVES & CURVATURES.
55     DO 35 I=1,NS
56     35 DZ(I)=DZTRP(I),Z,SI(I),NS)

```

```

N 0000
N 0010
N 0020
N 0030
N 0040
N 0050
N 0060
N 0070
N 0080
N 0090
N 0100
N 0110
N 0120
N 0130
N 0140
N 0150
N 0170
N 0180
N 0190
N 0200
N 0210
N 0220
N 0230
N 0240
N 0250
N 0260
N 0270
N 0280
N 0290
N 0300
N 0310
N 0320
N 0330
N 0340
N 0350
N 0360
N 0370
N 0380
N 0390
N 0400
N 0410
N 0420
N 0430
N 0440
N 0450
N 0460

```

57	DO 40 I=1,NS	N	0470
58	DZ2=DZTRP(S,DZ,S(I),NS)	N	0480
59	40 C(I)=AIMAG(CONJG(DZ2(I)))DZ2(I)/CABS(DZ2(I))**3	N	0490
60		N	0500
61	C.....INPUT AUXILIARY (CONTROL) DATA.	N	0510
62	C		
63	C IF BYPASS = .TRUE. THE INPUT POINTS ARE THE OUTPUT POINTS		
64	C		
65	45 READ (WIN,AUXIN)	N	0520
66	IF (NFIN.EQ.0) NFIN=NS	N	0530
67	SFIN=SFNFIN	N	0540
68	DSMAX=AMAX1(DSMAX,DSEND)	N	0550
69	IF (.NOT. BYPASS) GO TO 47		
70	NSP=NS		
71	DO 46 I=1,NS		
72	X(I)=REAL(Z(I))		
73	Y(I)=AIMAG(Z(I))		
74	46 CONTINUE		
75	C		
76	C SET DELS1 FOR NEXT SEGMENT. SPGEN DOES THIS WHEN BYPASS = .FALSE.		
77	DXLAST = X(NS) - X(NS-1)		
78	DYLAST = Y(NS) - Y(NS-1)		
79	DSLST = SQRT(DXLST**2. + DYLAST**2.)		
80	DELS1 = DSLST		
81	GO TO 91		
82	47 IF (EVEN) GO TO 50	N	0560
83		N	0570
84	C.....GENERATE BODY POINTS ON A SEGMENT.	N	0580
85	IF (.NOT. SPGENIS,Z,C,NS,SP,NSP,SFIN,NMAX)) GO TO 130	N	0590
86	GO TO 80	N	0600
87		N	0610
88	C.....GENERATE UPPER AND LOWER SURFACES TOGETHER (EVEN).	N	0620
89	50 SHALF=SFIN/2.0	N	0630
90	IF (NSP.LT.1) GO TO 55	N	0640
91	IF (SP(NSP).GE.SHALF/ONE) GO TO 60	N	0650
92	55 IF (.NOT. SPGENIS,Z,C,NS,SP,NSP,SHALF,NMAX)) GO TO 140	N	0660
93	60 SREH=(SFIN-SP(NSP))/ONE	N	0670
94	DO 65 I=1,NSP	N	0680
95	IF (SP(I).GE.SREH) GO TO 70	N	0690
96	65 CONTINUE	N	0700
97	GO TO 140	N	0710
98	70 IF (NSP-I-1.GT.NMAX) GO TO 150	N	0720
99	75 I=I-1	N	0730
100	IF (I.LT.1) GO TO 80	N	0740
101	NSP=NSP+1	N	0750
102	SP(NSP)=SFIN-SP(I)	N	0760
103	GO TO 75	N	0770
104		N	0780
105	C.....TEST FOR FINISH.	N	0790
106	80 IF (DONE) GO TO 85	N	0800
107	GO TO 45	N	0810
		N	0820
	1. OUTPUT RESULTING OR-BODY POINTS.	N	0830
0	85 DO 90 I=1,NSP	N	0840
111	Z2=DZTRP(S,Z,SP(I),NS)	N	0850
112	DZ(I)=DZTRP(S,Z,SP(I),NS)	N	0860
113	DZ2=DZTRP(S,DZ,SP(I),NS)	N	0870

```

114 C(I)= AIMAG(CONJG(C2(I))*D2Z)/CABS(DZ(I))*3
115 X(I)=REAL(ZZ)
116 90 Y(I)=AIMAG(ZZ)
117 91 CONTINUE
118 K1=KSTART
119 K2=NSP*KSTART-1
120 DO 110 I=K1,K2
121 II=I-KSTART+1
122 XONT(I)=X(II)
123 YON(I)=Y(II)
124 CAPPA(I)=C(II)
125 IF (ABS(CAPPA(I)).LT..0001)CAPPA(I)=0.
126 IF (REAL(DZ(I)).EQ.0.) GO TO 100
127 DYDXO(I)=AIMAG(DZ(I))/REAL(DZ(I))
128 IF (ABS(DYDXO(I)).GT.999.) GO TO 100
129 IF (ABS(DYDXO(I)).LT..0001) GO TO 95
130 ALPHA(I)=ATAN(DYDXO(I))*180./3.14157
131 GO TO 105
132 95 DYDXO(I)=0.
133 ALPHA(I)=0.
134 GO TO 105
135 100 DYDXO(I)=999.
136 ALPHA(I)=90.
137 105 SON(I)=0.
138 IF ((NE.1)SON(I)-SON(I)-1)*SQRT((XONT(I)-XONT(I-1))**2+(YON(I)-YON(I-1))**2)
139 11) **2)
140 IF ((NSEG.EQ.0.AND.II.EQ.1).OR.(IJ.EQ.1.AND.II.EQ.1)SON(I)=0.
141 IF (I.EQ.K1) GO TO 110
142 IF (XONT(I).EQ.XONT(I-1))CAPPA(I)=CAPPA(I-1)
143 IF (XON(I).EQ.XON(I-1)) GO TO 110
144 IF ((DYDXO(I)-DYDXO(I-1))/(XONT(I)-XONT(I-1))CAPPA(I).LT.0.)CAPPA(I)=
145 1)-CAPPA(I)
146 IF (I.EQ.(K1+1))CAPPA(I)=1)-SIGN(CAPPA(I-1),CAPPA(I))
147 IF ((DYDXO(I)*DYDXO(I-1)).LT.0..AND.DYDXO(I).GT.0.)CAPPA(I)=-ABS(CAPPA(I-1))
148 1A(I))
149 IF (1.LT.(K1+2)) GO TO 110
150 IF ((DYDXO(I)*DYDXO(I-2)).LT.0.) GO TO 110
151 IF ((DYDXO(I-1).GT.DYDXO(I)).AND.DYDXO(I-1).LT.DYDXO(I-2)).OR.(DYDXN
152 10(I-1).LT.DYDXO(I)).AND.DYDXO(I-1).GT.DYDXO(I-2)) GO TO 110
153 IF ((CAPPA(I-1).LE.CAPPA(I)).AND.CAPPA(I-1).GE.CAPPA(I-2)).OR.(CAPPA(I-1).GT.
154 1A(I-1).GE.CAPPA(I)).AND.CAPPA(I-1).LE.CAPPA(I-2)) GO TO 110
155 CAPPA(I-1)=-CAPPA(I-1)
156 110 CONTINUE
157 WRITE (6,115)NS,NSP,DSMAX,X(1),Y(1),X(NSP),Y(NSP)
158 115 FORMAT(1H0,7X,5H----,
159 1
160 58HDIRECT INTERPOLATION. FULL POINT-SPACING REQUIREMEN
161 ITS MET./24X,19HNO. OF INPUT PTS.= ,I4,2X,20HNO. OF OUTPUT PTS.= ,I4
162 24,2X,16HDSMAX = DSEND = ,F10.2/24X,14HSTART(X,Y) = 1,F10.6,1H,,F10
163 3.6,1H/24X,12HEND(X,Y) = 1,F10.6,1H,,F10.6,1H)
164 IF (DONE) GO TO 120
165 120 RETURN
166 C.....ERROR MESSAGES.
167 125 FORMAT(20HSGEN FAILED. IBAO= ,I3)
168 130 WRITE (6,135)
169 135 FORMAT(33HSPGEN UNABLE TO COMPLETE SEGMENT)
170 STOP
171 140 WRITE (6,145)
172 145 FORMAT(29HSPGEN UNABLE TO DO EVEN BODY)
173 STOP
174 150 WRITE (6,155)NSP
175 155 FORMAT(14H100 MANY POINTS FOR BOTH SURFACES. NLOW= ,I4)
176 STOP
177 END

```

```

N 0880
N 0890
N 0900
N 0910
N 0920
N 0930
N 0940
N 0950
N 0960
N 0970
N 0980
N 0990
N 1000
N 1010
N 1020
N 1030
N 1040
N 1050
N 1060
N 1070
N 1080
N 1090
N 1100
N 1110
N 1120
N 1130
N 1140
N 1150
N 1160
N 1170
N 1180
N 1190
N 1200
N 1210
N 1220
N 1230
N 1240
N 1250
N 1260
N 1270
N 1280
N 1290
N 1300
N 1310
N 1320
N 1330
N 1340
N 1350
N 1360
N 1370
N 1380
N 1390
N 1400
N 1410
N 1420
N 1430
N 1440
N 1450
N 1460
N 1470
N 1480
N 1490

```

```

1      SUBROUTINE XYTRAN(J,NSEG,ENREED,NS)
2      C
3      C CALLED BY SUBROUTINES SCIRCLE AND XYCALC
4      C THIS SUBROUTINE TRANSLATES AND/OR SCALES INPUT BODY POINTS
5      C
6      C
7      C J - INDEX OF SEGMENT LOOP IN MAIN
8      C NSEG - NO. OF BODY SEGMENTS (INPUT)
9      C ENREED - TYPE OF BODY SEGMENT (INPUT)
10     C NS - NO. OF Z VALUES INPUT (CALCULATED IN XYCALC)
11     C
12     COMPLEX Z(300)
13     COMMON/MAIN/ XIN(10),YIN(10),DELSMX,PI02,DELS1,IHUB,DELNEW
14     LOGICAL TRANSX,TRANSY,SCALEX,SCALEY,CHANGE
15     COMMON/TRANSF/ XTRAN,YTRAN,XSCALE,YSCALE,
16     1 TRANSX,TRANSY,SCALEX,SCALEY,CHANGE,XTMAX
17     COMMON/TRANXY/ Z
18     LOGICAL W060
19     C
20     C ERROR FLAG -- TRANSFORMATION OF INPUT BODY POINTS IS NOT PERMITTED
21     C FOR CERTAIN BISUPERELLIPSE OPTIONS. SEE OPTIONS
22     C E, F, G, AND H IN TABLE 1, TM-73728, P. 37A.
23     C
24     W060 = .FALSE.
25     W060 = YIN(6) .EQ. -100. .OR. YIN(6) .EQ. 200.
26     IF(W060) GO TO 900
27     IF(.NOT. TRANSX) GO TO 20
28     IF(ENREED .EQ. 99) .OR. INSEG .EQ. 0) GO TO 16
29     DO 15 ITX=1,6
30     XI = XIN(ITX)
31     IF(XI .LT. XTMAX) XIN(ITX) = XI + XTRAN
32     15 CONTINUE
33     GO TO 19
34     16 CONTINUE
35     DO 17 ITX=1,NS
36     B = REAL(Z(ITX))
37     IF(B .LT. XTMAX) B = B + XTRAN
38     C = AIMAG(Z(ITX))
39     Z(ITX) = CMPLX(B,C)
40     17 CONTINUE
41     19 IF(J .EQ. 1) WRITE(6,1000) XTRAN
42     20 IF(.NOT. TRANSY) GO TO 30
43     IF(ENREED .EQ. 99) .OR. INSEG .EQ. 0) GO TO 26
44     DO 25 ITY=1,6
45     YIN(ITY) = YIN(ITY) + YTRAN
46     25 CONTINUE
47     GO TO 29
48     26 CONTINUE
49     DO 27 ITY=1,NS
50     B = REAL(Z(ITY))
51     C = AIMAG(Z(ITY)) + YTRAN
52     Z(ITY) = CMPLX(B,C)
53     27 CONTINUE
54     29 IF(J .EQ. 1) WRITE(6,1500) YTRAN
55     30 IF(.NOT. SCALEX) GO TO 40
56     IF(ENREED .EQ. 99) .OR. INSEG .EQ. 0) GO TO 36

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57      DO 35 ISX=1,6
58      XINI(ISX) = XINI(ISX)*XSCALE
59      35 CONTINUE
60      GO TO 39
61      36 CONTINUE
62      DO 37 ISX=1,NS
63      B = REALIZ(ISX)*XSCALE
64      C = AIMAG(ZI(ISX))
65      ZI(ISX) = CMPLX(B,C)
66      37 CONTINUE
67      39 IF(IJ.EQ. 1) WRITE(6,2000) XSCALE
68      40 IF(.NOT. SCALEY) GO TO 50
69      IF(ITERREED.EQ. 99) .OR. INSEG.EQ. 0) GO TO 46
70      DO 45 ISY=1,6
71      YINI(ISY) = YINI(ISY)*YSCALE
72      45 CONTINUE
73      GO TO 49
74      46 CONTINUE
75      DO 47 ISY=1,NS
76      B = REALIZ(ISY)
77      C = AIMAG(ZI(ISY))*YSCALE
78      ZI(ISY) = CMPLX(B,C)
79      47 CONTINUE
80      49 IF(IJ.EQ. 1) WRITE(6,2500) YSCALE
81      50 CONTINUE
82      RETURN
83      900 WRITE(6,3000)
84      STOP
85      C
86      C  FORMAT STATEMENTS
87      C
88      1000 FORMAT(1HD,5X,'* * * X - VALUES OF BODY POINTS TRANSLATED*,
89      1 IPE12.5,' UNITS*)
90      1500 FORMAT(1HD,5X,'* * * Y - VALUES OF BODY POINTS TRANSLATED*,
91      1 IPE12.5,' UNITS*)
92      2000 FORMAT(1HD,5X,'* * * X - VALUES OF BODY POINTS SCALED*,
93      1 BY A FACTOR OF ',IPE12.5)
94      2500 FORMAT(1HD,5X,'* * * Y - VALUES OF BODY POINTS SCALED*,
95      1 BY A FACTOR OF ',IPE12.5)
96      3000 FORMAT(3X,'*** ERROR IN XYTRAN *** - USER ATTEMPTED*,IX,
97      1 *TO TRANSFORM INPUT POINTS FOR AN ILLEGAL CASE*)
98      END

```



```

1      LOGICAL FUNCTION SPGEN (S,Z,C,NS,SP,NSP,SFIN,NMAX)      Q 0000
2      C.....GENERATES TABLE SP HAVING VALUES OF PARAMETER S AS WIDELY SPACED Q 0010
3      C      AS POSSIBLE AND YET SATISFYING THE FOLLOWING CONDITIONS ON DS: Q 0020
4      C      1 NSP .LE. NMAX Q 0030
5      C      2 DS .LE. A/(C*S) (C=CURVATURE) Q 0040
6      C      3 DS .LE. DSPAX Q 0050
7      C      4A DS(I) .LE. DS(I-1)*RMAX Q 0060
8      C      4B DS(I) .GE. DS(I-1)/RMAX Q 0070
9      C      FOR THIN SECTIONS, AN ADDITIONAL CONDITION IS: Q 0080
10     C      DS .LE. B*ITLOC (ITLOC=LOCAL THICKNESS) Q 0090
11     C.....SPGEN = .TRUE. IF ALL CONDITIONS HAVE BEEN SATISFIED. Q 0100
12
13     REAL S(NS),C(NS),SP(NMAX) Q 0110
14     COMPLEX Z(NS),FZTRF Q 0120
15     LOGICAL THIN,FIN Q 0130
16     COMMON /SPGENC/ A,DSMAX,RMAX,THIN,B,THIN,USEND Q 0140
17     COMMON /MAIN/ XIN(10),VIN(10),DELSM,PIO2,DELS1,INUB Q 0150
18     DATA ONE,CMIN/1.0001,1.0E-6/ Q 0160
19
20     C.....INITIALIZATION SECTION. Q 0170
21     SPGEN=.FALSE. Q 0180
22     J1=MAX(0,NSP-2)*1 Q 0190
23     IF (NSP.GT.1) GO TO 15 Q 0200
24     IF (NSP.LT.1) SP(1)=ST1 Q 0210
25     DS1= DELS1 Q 0220
26     10 SP(2)=SP(1)*DS1 Q 0230
27 Q 0240
28     C.....BEGIN MAIN LOOP. Q 0250
29     15 DO 45 J=J1,NMAX Q 0260
30     L=J Q 0270
31     20 I=L Q 0280
32     25 DSLAST=SP(I-1)*SP(I-2) Q 0290
33     SBAR=SP(I-1)+DSLAST/2.0 Q 0300
34     CA=AMAX1(CMIN,ABS(FZTRP(S,Z,C,SBAR,NS))) Q 0310
35     DSLIN=AMIN1(DS1,DSLAST/RMAX) Q 0320
36     IF (.NOT. THIN) GO TO 30 Q 0330
37     TLOC=CABS(FZTRP(S,Z,SBAR,NS))-FZTRP(S,Z,S(NS)-SBAR,NS) Q 0340
38     DSLIN=AMIN1(DSLIN,B*AMAX1(TLOC,THIN)) Q 0350
39     30 DSFIN=SFIN-SP(I-1) Q 0360
40     NEVEN=DSFIN/DSLIN/ONE*1.0 Q 0370
41     DSEVEN=DSFIN/FLOAT(NEVEN) Q 0380
42     DS=MIN1(CA/DSEVEN) Q 0390
43     IF (I.NE.J) DS=AMIN1(DS,DSLAST/RMAX) Q 0400
44 Q 0410
45     C.....CALCULATED VALUE OF DS SATISFIES CONDITIONS 2 THRU 4A.TEST FOR 4B.Q 0420
46     IF (DS.GE.DSLAST/RMAX) GO TO 40 Q 0430
47 Q 0440
48     C.....IF CONDITION 4B IS NOT SATISFIED, RE-DO EARLIER INTERVALS Q 0450
49     C.....USING SMALLER VALUES OF DS. IF RE-DOING ALL INTERVALS WON'T Q 0460
50     C.....WORK, START OVER USING SMALLER STARTING VALUE OF DS (DS1). Q 0470
51     35 L=L-1 Q 0480
52     IF (L.GE.J1) GO TO 20 Q 0490
53     IF (NSP.GT.1) RETURN Q 0500
54     DS1=DS1/RMAX Q 0510
55     GO TO 10 Q 0520
56 Q 0530

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```

57 C.....IF CONDITIONS 2 THRU 48 ARE SATISFIED, TEST FOR FINISH.
58 40 SP1:=SP1-1)DS
59 FIN:=FIN/SP1)DS
60 IF (FIN.AND.DS.GT.CEND) 60 TO 35
61 IF (FIN) 60 TO 50
62 IF (I.EE.J) 60 TO 45
63 I:=I+1
64 60 TO 25
65 45 CONTINUE
66 C.....SPEN = FALSE. IF CONDITION 1 CANNOT BE SATISFIED.
67 RETURN
68
69 C.....IF CONDITIONS ARE SATISFIED, UPDATE MSP.
70 50 MSP:=1
71 DELET=DS
72 SPGEN:=TRUE.
73 RETURN
74 END

```

```

1  INTEGER FUNCTION SGM (S,F,MS) 0 00C0
2  C.....GENERATES THE PARAMETER ARRAYS FOR THE SET OF POINT-PAIRS F SUCH 0 0010
3  C.....THAT S(I) GIVES THE LINE INTEGRAL ON THE CURVE OF F2IRP (S,F,X,MS) 0 0020
4  C.....WHEN X=I(1). 0 0030
5  REAL SINS) 0 0040
6  COMPLEX F(1),D2IRP 0 0050
7  DATA MAX,N,FN,TEST/4,10,10,0,0,0,1/ 0 0060
8  DO 10 I=2,NS 0 0070
9  10 S(I)=S(I-1)+CABS(F(I))-F(I-1)) 0 0080
10 DO 30 K=1,MAX 0 0090
11 SGEN=0 0 01C0
12 DO 25 I=2,NS 0 0110
13 DS=S(I)-S(I-1) 0 0120
14 DARG=DS/FN 0 0130
15 ARG=S(I-1)-DARG/2.0 0 0140
16 SUM=0.0 0 0150
17 DO 15 J=1,N 0 0160
18 ARG=ARG+ATAN2(FN,DS) 0 0170
19 15 SUM=SUM+CABS(D2IRP(S,F,ARG,MS))-1.0) 0 0180
20 SUM=SUM/FN 0 0190
21 ERROR=ABS(SUM) 0 0200
22 DS=DS*SUM 0 0210
23 DO 20 J=1,NS 0 0220
24 20 S(I)=S(I)+DS 0 0230
25 IF (ERROR.GT.TEST.AND.SGEN.EQ.0) SGEN=1 0 0240
26 25 CONTINUE 0 0250
27 IF (SGEN.EQ.0) RETURN 0 0260
28 30 CONTINUE 0 0270
29 C.....SGEN=INDEX IF IT DOESN'T CONVERGE. 0 0280
30 RETURN 0 0290
31 END 0 03C0

```

107

```

1  COMPLEX FUNCTION D2IRP (A,F,X,NA) P 00C0
2  C.....COMPLEX DERIVATIVE EVALUATION FOR DOUBLE 3-POINT INTERPOLATION. P 0010
3  COMPLEX F(NA) P 0020
4  COMMON /NTRPC37 IT,I2,C(4) P 0030
5  C.....FIRST EVALUATE FUNCTION COEFFICIENTS. P 0040
6  CALL FTRPA (A,X,NA) P 0050
7  CALL DMRPC P 0060
8  C.....THEN EVALUATE FUNCTION VALUE. P 0070
9  D2IRP=0.0 P 0080
10 J=0 P 0090
11 DO 10 I=1,I2 P 0100
12 J=J+1 P 0110
13 10 D2IRP=D2IRP+C(J)*F(I) P 0120
14 RETURN P 0130
15 END P 0140

```

1	SUBROUTINE DNTRPC	R	0000
2	C.....CALCULATION OF C COEFFICIENTS FOR DERIVATIVES OF DOUBLE	R	0010
3	C.....3-POINT INTERPOLATION.	R	0020
4	COMMON /NTRPC1/ L,I,A11,A12,A13,A14,A22,A23,A24,A33,A34,A44	R	0030
5	COMMON /NTRPC3/ I1,I2,C1,C2,C3,C4	R	0040
6	IF (L-LE.I) 60 TO 25	R	0050
7	IF (I-L-3) 20,15,10	R	0060
8	C.....FOR DOUBLE 3-POINT INTERPOLATION.	R	0070
9	10 C1=+(A22+A33+A22)/A23+A33/A12/A13	R	0080
10	C4=-A33+A22+A33/A23+A22/A34/A24	R	0090
11	P=A23+A23	R	0100
12	C2=-A11+A33+A11/A33/A12+A33/A44+A22+A33/A24/P	R	0110
13	C3=+((A44+A22+A44)+A22/A34+(A22+A11+A33+A11+A33+A22)/A13)/P	R	0120
14	60 TO 30	R	0130
15	C.....FOR SIMPLE 3-POINT INTERPOLATION.	R	0140
16	15 C1=+A33+A22/A12/A13	R	0150
17	C2=-A33+A11/A12/A23	R	0160
18	C3=+A22+A11/A13/A23	R	0170
19	60 TO 30	R	0180
20	C.....FOR 2-POINT INTERPOLATION.	R	0190
21	20 C1=1.0/A12	R	0200
22	C2=-C1	R	0210
23	60 TO 30	R	0220
24	C.....ONLY ONE TABLE VALUE GIVEN.	R	0230
25	25 C1=0.0	R	0240
26	30 I1=I	R	0250
27	I2=I+L-1	R	0260
28	RETURN	R	0270
29	END	R	0280

108

1	FUNCTION FNTRP (A,F,X,NA)	S	0000
2	C.....FUNCTION EVALUATION FOR DOUBLE 3-POINT INTERPOLATION.	S	0010
3	REAL FNA)	S	0020
4	COMMON /NTRPC2/ I1,I2,C1,C4	S	0030
5	C.....FIRST EVALUATE FUNCTION COEFFICIENTS.	S	0040
6	CALL FNTRPA (A,X,NA)	S	0050
7	CALL FNTRPC	S	0060
8	C.....THEN EVALUATE FUNCTION VALUE.	S	0070
9	ENTRY FNTRP1 (F)	S	0080
10	FNTRP=0.0	S	0090
11	J=0	S	0100
12	DO 10 I=I1,I2	S	0110
13	J=J+1	S	0120
14	10 FNTRP=FNTRP+C(I,J)*F(I)	S	0130
15	RETURN	S	0140
16	END	S	0150

1	COMPLEX FUNCTION FZTRP (A,F,X,NA)	T	0000
2	C.....COMPLEX FUNCTION EVALUATION BY DOUBLE 3-POINT INTERPOLATION.	T	0010
3	COMPLEX F(NA)	T	0020
4	COMMON /NTRPC2/ I1,I2,C(1)	T	0030
5	C.....FIRST EVALUATE FUNCTION COEFFICIENTS.	T	0040
6	CALL FNTRPA (A,X,NA)	T	0050
7	CALL FNTRPC	T	0060
8	C.....THEN EVALUATE FUNCTION VALUE.	T	0070
9	FZTRP=0.0	T	0080
10	J=0	T	0090
11	DO 10 I=I1,I2	T	0100
12	J=J+1	T	0110
13	10 FZTRP=FZTRP+C(I)*F(I)	T	0120
14	RETURN	T	0130
15	END	T	0140

1	SUBROUTINE FNTRPC	U	0000
2	C.....CALCULATION OF C COEFFICIENTS FOR FUNCTION VALUES BY DOUBLE	U	0010
3	C.....3-POINT INTERPOLATION.	U	0020
4	COMMON /NTRPC1/ L,I1,A11,A12,A13,A14,A22,A23,A24,A33,A34,A44	U	0030
5	COMMON /NTRPC2/ I1,I2,C1,C2,C3,C4	U	0040
6	IF (L.LE.1) GO TO 25	U	0050
7	IF (L-3) 20,15,10	U	0060
8	C.....FOR DOUBLE 3-POINT INTERPOLATION.	U	0070
9	10 C1=+A33/A23+A22/A12+A33/A13	U	0080
10	C4=-A22/A23+A33/A34+A22/A24	U	0090
11	P2=A33/A23+A11/A23	U	0100
12	P3=A22/A23+A44/A23	U	0110
13	C2=-A33*(P2/A12+P3/A24)	U	0120
14	C3=+A22*(P3/A34+P2/A13)	U	0130
15	GO TO 30	U	0140
16	C.....FOR SIMPLE 3-POINT INTERPOLATION.	U	0150
17	15 C1=+A22/A12+A33/A13	U	0160
18	C2=-A11/A12+A33/A23	U	0170
19	C3=+A11/A13+A22/A23	U	0180
20	GO TO 30	U	0190
21	C.....FOR 2-POINT INTERPOLATION.	U	0200
22	20 C1=+A22/A12	U	0210
23	C2=-A11/A12	U	0220
24	GO TO 30	U	0230
25	C.....ONLY ONE TABLE VALUE GIVEN.	U	0240
26	25 C1=1.0	U	0250
27	30 I1=I	U	0260
28	I2=I*L-1	U	0270
29	RETURN	U	0280
30	END	U	0290

```

1 SUBROUTINE FTRPA (A,X,MA)
2 C.....COMMON SUBROUTINE EVALUATES A COEFFICIENTS IN DOUBLE
3 C.....3-POINT INTERPOLATIONS.
4 C.....L=NO. OF POINTS IN THE FIT
5 C.....I=INDEX TO FIRST POINT
6 REAL X(M),Y(M)
7 COMMON /FTRPA/ L1,A11,A12,A13,A10,A22,A23,A20,A33,A30,A00
8 C.....GET I AND L BY TABLE LOOK-UP.
9 L=LIM1(I,MA,3)
10 M=MAX(1,MA-2)
11 CALL TLU (A,X,MA,J)
12 IF (J.EQ.LIM1(I,MA,3)) L=L1
13 I=LIM1(I,0-1,M)
14 C.....CALCULATE A-ARRAY.
15 A11=A11
16 A22=A11*I
17 A33=A11*I*I
18 IF (L.EQ.4) IF (L-2) 20,15,10
19 A00=A11*I*I*I
20 A10=A11*I*I
21 A20=A22-A00
22 A30=A33-A00
23 A00=X-A00
24 10 A10=A11-A33
25 A20=A22-A33
26 A30=X-A33
27 15 A10=A11-A22
28 A20=X-A22
29 A10=X-A11
30 20 RETURN
31 END

```

```

1 SUBROUTINE TLU (TABLE,ARG,M,I)
2 C.....TABLE LOOK UP: FINDS I SUCH THAT
3 C ARG-GE.TABLE(I),AND,ARG-LT.TABLE(I+1)
4 C IF I=0, ARG-LT.TABLE(I)
5 C IF I=M, ARG-GE.TABLE(M)
6 C NEAR TABLE(N)
7 I=LIMIT(I,I,M)
8 IF (ARG-GE.TABLE(I)) GO TO 15
9 C.....DESCEND IN TABLE.
10 I=I-1
11 IF (I.LE.0) RETURN
12 IF (ARG-GE.TABLE(I)) RETURN
13 GO TO 10
14 C.....ASCEND IN TABLE.
15 IF (I-GE.M) RETURN
16 IF (ARG-LT.TABLE(I)) RETURN
17 I=I+1
18 GO TO 15
19 END

```

W 00C0
W 0010
W 0020
W 0030
W 0040
W 0050
W 0060
W 0070
W 0080
W 0090
W 0100
W 0110
W 0120
W 0130
W 0140
W 0150
W 0160
W 0170
W 0180

```

1 FUNCTION LIMIT (I,J,M)
2 C.....INTEGER FUNCTION LIMITS J BETWEEN I AND M.
3 LIMIT=I
4 IF (J.LT.LIMIT) RETURN
5 LIMIT=M
6 IF (J.GT.LIMIT) RETURN
7 LIMIT=J
8 RETURN
9 END

```

X 00C0
X 0010
X 0020
X 0030
X 0040
X 0050
X 0060
X 0070
X 0080

```

1      SUBROUTINE WPUNCH
2      COMMON /FOREOD/ IGEOMF,YSIGF,ICURVM,NONEWF,IVORY,AL
3      COMMON /SS/ NBDY1,NBDY2,TYPBDY,NBDYS
4      COMMON /FORSSS/IO,DELS,XBR(20),YBR(20),XOM(500),YOM(500),OVDXO(500)
5      1),ALPHA(500),CAPPA(500),SOM(500),PIO(180)
6      COMMON /HWRYE/ YFLAG,NDY4,PRO6,TITLE(9),BODYES(4),IDENY,VLO(25),VHY
7      11(25),NDY(25),XRAK(25),NBDPTS(9),NO6,NRAKES
8      COMMON /LOT/XMIN,VMIN,ORD,EXEP,VX,VY,LPNCHO,IPLOYA,WH
9      COMMON /OMVORT/NLF(10)
10     COMMON /Y24Y/ IPAR,IFSY,ISND,IFLL
11     DIMENSION YOFF(200), XOFF(200), FI(500)
12     DIMENSION XI(25),YI(25)
13     COMMON /MNSD/ MNSD,MNSDDY(10)
14     DATA BODYD/6H-BODY 7,YFLAG/4H 1117,YFLAG/24H 1 17,Y224/6H 23V/,Y
15     11FL62A/1H /,IFL62B/1H1/
16     C
17     C IF YLO AND VHI ARE READ IN AS ZERO,CALCULATE THEM FOR THAT RAKE
18     C FOR RAKES BETWEEN BODYES =1 AND =2 ONLY
19     C
20     C FIND HIGHLIGHT ON THE SHROUD
21     C
22     NINE=9
23     NEIGHT=8
24     I21=21
25     IONE=1
26     ITWO=2
27     IZER=0
28     NB=NBDPTS(1)+1
29     IF(IFLAG.EQ.1)NB=1
30     NE=NBDPTS(2)
31     C I SET JMIN=LAST SHROUD PT., IN CASE X NEVER INCREASES ON SHROUD(VTOL)
32     JMIN=NE
33     DO 10 I=NB,NE
34     IF (XONTI+1).LT.XONTI) GO TO 10
35     JMIN=I
36     GO TO 15
37     10 CONTINUE
38     15 DO 20 I=I,NE
39     20 FI(I)=1
40     NOFF=0
41     IF (NRAKES.EQ.0) GO TO 70
42     DO 65 I=1,NRAKES
43     NLO=NOFF+1
44     NOFF=NLO+NDY(1)
45     ENDY=NDY(1)
46     IF (YHII).EQ.0.0.OR.YE01).EQ.0.0) GO TO 25
47     GO TO 35
48     25 IF (YHII).NE.0.0) GO TO 30
49     CALL SINTP (XON(10),YON(10),JMIN-NB+1,XRAK(I),YH)
50     CALL SINTP (XON(10),YON(10),JMIN-NB+1,XRAK(I),FI)
51     IF=FI
52     DS=SUM((XONTI)-XONTI+1)*W2+(YONTI)-YONTI+1)*W2)
53     YHII)=YH-DS
54     30 IF (VLO(1).NE.0.0.OR.XRAK(1).LT.XONTI) GO TO 35
55     CALL SINTP (XON,YON,NBDPTS(1),XRAK(1),VL)
56     CALL SINTP (XON,FI,NBDPTS(1),XRAK(1),FI)

```


57	IF=FYI	Y	0480
58	DS=SQRT((XON(IF)-XCN(IF+1))**2+(YON(IF)-YCN(IF+1))**2)	Y	0490
59	YLO(I)=YL+DS	Y	0500
60	35 DYI=(YHI(I)-YLO(I))/ENDY	Y	0510
61	K=0	Y	0520
62	DO 55 J=NLO,NOFF	Y	0530
63	DJM=J-NLO	Y	0540
64	XOFF(J)=XRAK(I)	Y	0550
65	YOFF(J)=YLO(I)+DYI*DJM	Y	0560
66	YMAN=YV*ORD+YMIN		
67	IF (LPCNCH0.EQ.2) YOFF(J)=YMAN-(YOFF(J)-YMAN)		
68	IF (XOFF(J)-XMIN) 55,40,40	Y	0570
69	40 IF (YOFF(J)-XX*EXEP-XMIN) 45,45,55	Y	0580
70	45 IF (YOFF(J)-YV*ORD-YMIN) 50,50,55	Y	0590
71	50 K=K+1	Y	0600
72	X(K)=XOFF(J)	Y	0610
73	Y(K)=YOFF(J)	Y	0620
74	55 CONTINUE	Y	0630
75	C	Y	0640
76	C(IIII PLOT OFF-BODY POINTS (IRAKES)	Y	0650
77	C	Y	0660
78	60 CALL LINE(X,Y,N,1,-1,0,XMIN,EXEP,YMIN,ORD)	Y	0670
79	65 CONTINUE	Y	0680
80	C	Y	0690
81	C(IIII PUNCH OPTION (IIII	Y	0700
82	C	Y	0710
83	IF (LPCNCH0.EQ.0) RETURN		
84	70 NTBDY=NBDYS+MNSD+1-IVORT	Y	0720
85	NLOOP=2-IVORT	Y	0730
86	IF (NBDYS.EQ.3.AND.PROG.NE.T22Y.AND.IVORT.EQ.0) GO 1075		
87	60 TO 80	Y	0740
88	75 NTBDY=NTBDY+1	Y	0750
89	NLOOP=3	Y	0760
90	80 K=0	Y	0770
91	IF (PROG.EQ.T22Y) NLOOP=1		
92	DO 110 I=1,NLOOP	Y	0780
93	N=NTBDY-I+1	Y	0790
94	IFLAGG=IFLAG2	Y	0800
95	IF (N.EQ.NTBDY .OR. NBDYS.EQ.3.AND.N.EQ.3) IFLAGG=IFLY	Y	0810
96	1AG1	Y	0820
97	IF (PROG.EQ.T22Y) 60 TO 92		
98	WRITE (17,115) (I,1),L=1,9,N,BODYD,IDENT	Y	0830
99	WRITE (17,120)H,IFLAGG,N06,IVORT,IDENT	Y	0840
100	90 CONTINUE	Y	0850
101	WRITE (17,125)IDENT		
102	92 NA=1		
103	NS=0	Y	0860
104	IF (1.NE.1) K=1	Y	0870
105	DO 105 J=1,M	Y	0880
106	IF (J.6Y.NBDYS.AND.MNSD.NE.0) GO TO 95	Y	0890
107	NB=NBDPTS(J)	Y	0900
108	60 TO 100	Y	0910
109	95 NS=NS+1	Y	0920
110	IF (NS.6Y.MNSD) NSBDY(NS)=NBDPTS(NBDYS+1)-NBDPTS(NBDYS)	Y	0930
111	NB=NSBDY(NS)+NA-1	Y	0940
112	100 NP=NB-NA+1	Y	0950
113	IF (PROG.NE.T22Y) GO 10102		

```

114      LASBOD=0
115      NOIBOD=2
116      IF(J.EQ.1) NOIBOD=1
117      IF(J.EQ.M) LASBOD=1
118      MNN = 1 - MLFIJ
119      WRITE(17,130) J,IGCOMF,MNN,(TITLE(L),L=1,5),IPAR,IFS,ISNB,
120      I,NOIBOD,IVORY,LASBOD,IONE
121      C      T R A N S F
122      102 CALL WRTXY (NIP,IDENT,J,K,XON,YON,NA,NB,PROG)
123      NA=NB+1
124      105 CONTINUE
125      IF(PROG.NE.Y22Y) GO TO 107
126      WRITE(17,140) (TITLE(L),L=1,6),WEIGHT
127      V2=2.
128      WRITE(17,145) AL,ICNE,IONE,V2,IZER,IFILL,WINE
129      107      M=0
130      NA=1
131      J=0
132      NB=NOFF
133      IF(PROG.NE.Y22Y) GO TO 108
134      IF(NB.EQ.100) NB=100
135      IG=0
136      IF(NOFF.LE.100) IG=1
137      WRITE(17,150) IONE,(TITLE(L),L=1,6),IG,I21
138      C      T R A N S F
139      108 CALL WRTXY (NOFF,IDENT,J,K,XOFF,YOFF,NA,NB,PROG)
140      IF (PROG.NE.Y22Y) GO TO 110
141      IF(NOFF.LE.100) GO TO 110
142      WRITE(17,150) I21,NOFF,(TITLE(L),L=1,6),IONE,I21
143      C      T R A N S F
144      NOF=NOFF-100
145      NA=NB+1
146      NB=NOFF
147      CALL WRTXY (NOF,IDENT,J,K,XOFF,YOFF,NA,NB,PROG)
148      110 CONTINUE
149      RETURN
150      C
151      C      FORMATS
152      C
153      C
154      C
155      115 FORMAT ( 9A6,11,A6,2X,A6)
156      120 FORMAT ( 11,A4,11,8X,11,47X,A6,11X)
157      125 FORMAT ( 62X,A6,11X)
158      130 FORMAT(311,2X,1X,5A6,3X,3(11,2X),6X,11,5X,11,2X,11,3X,11)
159      135 FORMAT ( 3H0.0,7X,3H0.0,7X,3H90.)
160      140 FORMAT(10X,6X6,25X,11)
161      145 FORMAT(5X,F10.2,19X,11,9X,11,10X,F9.3,5X,11,11,11)
162      150 FORMAT(11,8X,6A6,22X,11,2X,12)
163      155 FORMAT(8F10.3)
164      END

```

Y 1070
Y 1080

Y 1100
Y 1120
Y 1110

Y 1150

Y 1200
Y 1210
Y 1220

Y 1230
Y 1240
Y 1250

Y 1260
Y 1270
Y 1280
Y 1290

Y 1310

Y 1320

114

```

1      SUBROUTINE WRTXY (NP,IDENT,J,K,X,Y,NA,NB,PROG)
2
3      C      WRITE X AND Y COORDINATES
4      C
5      DIMENSION X(1), Y(1)
6      COMMON /FOREOF/ I6E0FF,ISIGF,ICURVN,NONEWF,IVORY
7      COMMON/OMVORT/ NLF(10)
8      DIMENSION V(8)
9      DATA T2ZY/6H 23V/
10     DATA V/6HT ,6H ,6HF10.5,,6H ,6HX, ,6H4X,11,,
11     16H2X,11,,6H3X,11//
12     IF (PROG.NE.T2ZY) GO TO 10
13     LASY=1
14     NFUL=(NB-NA+1)/6
15     NFULL=NFUL*6
16     NREST=NA+NFULL-1
17     NDIF=NB-NA+1-NFULL
18     NREST=NREST+1
19     IT=3
20     NAA=NA-6
21     DO 2 LC=1,NFUL
22     NAA=NAA+6
23     NSTOP=NAA+5
24     IF(NSTOP.EQ.NB)GOTO5
25     WRITE(17,40)(X(1),L=NAA,NSTOP),IT
26     2      CONTINUE
27     IF(NDIF.EQ.0)GOTO6
28     NDIFH=(6-NDIF)/10
29     ENCODE(6,56,DUMPH)NDIF
30     ENCODE(6,56,DUMPH)NDIFH
31     DECODE(6,59,DUMPH)VEE
32     DECODE(6,59,DUMPH)VE2
33     V(1)=VEE
34     V(4)=VE2
35     WRITE(17,V) (X(1),LL=NREST,NB),NDIF,LASY,IT
36     GOTO6
37     5      WRITE(17,45)(X(1),L=NAA,NSTOP),LASY,IT
38     45     FORMAT(6F10.5,7X,11,3X,11)
39     6      IT=4
40     NAA=NA-6
41     DO 300 LC=1,NFUL
42     NAA=NAA+6
43     NSTOP=NAA+5
44     IF(NSTOP.EQ.NB)GOTO7
45     WRITE(17,40)(Y(1),L=NAA,NSTOP),IT
46     300    CONTINUE
47     IF(NDIF.EQ.0)RETURN
48     WRITE(17,V) (Y(1),LL=NREST,NB),NDIF,LASY,IT
49     RETURN
50     7      WRITE(17,45)(Y(1),L=NAA,NSTOP),LASY,IT
51     RETURN
52     10     WRITE (17,25)I6E0FF,ISIGF,ICURVN,NONEWF,NP,IDENT
53     IF(J.EQ.0) NNN=0
54     IF(J.EQ.0) GO TO 15
55     NNN=NLF(J)
56     15     WRITE (17,30)J,K,NA,N,IDENT

```

Z0 0000

Z0 0010

Z0 0020

Z0 0030

Z0 0040

Z0 0050

Z0 0060

Z0 0070

Z0 0140

```

57      2D IF INEQ.11 RETURN      20 0150
58      WRITE (17,35) (X11,1,4=NA,NB) 20 0160
59      WRITE (17,35) (Y11,1,4=NA,NB) 20 0170
60      RETURN                     20 0180
61      C                          20 0190
62      C                          20 0200
63      C                          20 0210
64      C                          20 0220
65      C                          20 0230
66      25 FORMAT (9I1,3H,13,52X,A6,11X) 20 0240
67      30 FORMAT (1,3(9X,11),32X,A6,11X) 20 0250
68      35 FORMAT (6E13.0)         20 0260
69      40 FORMAT (F10.5,11X,11)
70      55 FORMAT (1M,6H,11,2X,11,3X,11)
71      66 FORMAT (1J6)
72      59 FORMAT (A6)
73      END
      20 0290

```

```

1      SUBROUTINE AREA
2      COMMON /WRITE/ IFLAG,NDY4,PROG,TITLE(9),BODIES(4),IDENY,VELO(25),YMA
3      II(25),NDY(25),XRAK(25),NBDPTS(9),NO6,NRAKES
4      COMMON /SS/ NBDY1,NBDY2,YYPBDY,NBDYS,DZTEST
5      COMMON /FOR3SS/IO,CELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0(500)
6      IF(ALPHA(500),CAPPA(500),SONT(500),PI0(180)
7      COMMON /MNSD/ MNSD,MNSD8DY(10)
8      COMMON/ LOT/XMIN,YMIN,ORD,EXEP,XY,YY,LPNCHO,IPLOTA,MH
9      COMMON/BURLEY/ELREF,AREF
10     DIMENSION JMAX(20),JMIN(20),AREAS(20),VAR(20)
11     I,XXA(200),DISC(200),ANULUS(200)
12     PI=3.14159265
13     JPLA=0
14     NBP1=NBDPTS/NBDYS-1
15     IF(NBDYS.EQ.1)NBP1=1
16     NBP2=NBDPTS/NBDYS
17     DO 10 J=NBP1,NBP2
18     JJSS=J
19     CIIII(1) .LT. CHANGED TO LE. TO AVOID AREA PLOTS FROM GOING UP VSTOL LIP
20     IF (XON(I).LE.XON(I)*I) GO TO 15
21     10 CONTINUE
22     15 WRITE(6,98)
23     IF(ELREF.NE.1.0) WRITE(6,100) ELREF
24     IF(AREF.NE.1.0) WRITE(6,110) AREF
25     WRITE (6,75)
26     IF(NBDYS.LE.2) GO TO 40
27     C
28     C SEARCH FOR MINIMUM AND MAXIMUM X ON EACH BODY
29     C
30     NBDUMI=NBDYS-1
31     NB=1
32     DO 35 I=1,NBDUMI
33     NE=NBDPTS(I)
34     XMIN=XON(NB)
35     JMIN(I)=NB
36     XMAX=XON(NB)
37     JMAX(I)=NB
38     NEMI=NE-1
39     DO 30 J=NB,NEMI
40     IF (XON(I).GT.XMAX) GO TO 20
41     IF (XON(I).LT.XMIN) GO TO 25
42     GO TO 30
43     20 XMAX=XON(I)
44     JMAX(I)=J
45     GO TO 30
46     25 XMIN=XON(I)
47     JMIN(I)=J
48     30 CONTINUE
49     NB=NE+1
50     WRITE (6,80)XMIN,XMAX,JMIN(I),JMAX(I)
51     35 CONTINUE
52     40 DO 65 J=NBP1,JJSS
53     IA=1
54     YAR(IA)=0.
55     IF(XON(1).GT.XON(3))GO TO 42
56     CALL SINIP (XON,YON,NBDY1,XON(I),YAR(IA))

```

Z1 0060

UICD

Z1 0020

Z1 0030

Z1 0040

Z1 0050

Z1 0060

Z1 0070

Z1 0080

Z1 0090

Z1 0130

Z1 0140

Z1 0150

Z1 0160

Z1 0180

Z1 0200

Z1 0240

Z1 0250

Z1 0260

Z1 0270

Z1 0280

Z1 0290

Z1 0300

Z1 0310

Z1 0320

Z1 0330

Z1 0340

Z1 0350

Z1 0360

Z1 0370

Z1 0380

Z1 0390

Z1 0410

Z1 0420

Z1 0440

Z1 0444

Z1 0450

57 42 IFINBOYS,LE,21 60YCSO

58 DO 45 I=2,NBDSM1

59 JEND=NBDSM1

60 JMI=JMIN(I)

61 JMI=JMAX(I)

62 IF (XON(I),GT,XON(JMA),OR,XON(I),LT,XON(JMI)) 60 TO 45

63 IA=IA+1

64 CALL SINTP (XON(JMA),YON(JMA),JMI-JMA+1,XON(J),YAR(IA))

65 IA=IA+1

66 CALL SINTP (XON(JMI),YON(JMI),JEND-JMI+1,XON(J),YAR(IA))

67 45 CONTINUE

68 50 IA=IA+1

69 YAR(IA)=YON(J)

70 IS=0

71 AREA=0.0

72 DO 55 I=1,IA,2

73 IS=IS+1

74 AREAS(I)=I*VAR(I+1)*2-VAR(I)*2*4*PI

75 IF (PROG=0) 02 TESTTHERSTIS=VART(I+1)-VART(I)

76 AREA=AREA+AREAS(I)

77 55 CONTINUE

78 AREAD=AREA+VAR(I)*2*4*PI

79 IF (PROG=0) 02 TESTTHERED=AREA+VART(I)

80 AREA = AREA/AREF

81 AREAD = AREAD/AREF

82 XPLT = XON(I)/ELREF

83 YPLT = YON(I)/ELREF

84 YMPLT = VAR(I)/ELREF

85 IF (VART(I)/YON(I)) 02 58 TO 55

86 EMSUBK=CAPPA(I)+YON(I)-VART(I)/SORT(I)+VAR(I)/YON(I)

87 56 IF (EMPRAT) 02 58 TO 55

88 IF (INBOYS,LE,21) 60 TO 60

89 WRITE (6,88) THERSTIS,I-1,IS

90 60 WRITE (6,90) J,XPLT,YPLT,YMPLT,AREA,AREAD,EMSUBK

91 WRITE (6,95)

92 IF (XON(I),GT,(X)*EXEP+XMIN) 60 TO 65

93 JPLR=JPLR+1

94 IF (IPLOTA,LE,0) 60 TO 65

95 XXATJPLR=XONTJ

96 ANNULUS(JPLR)=AREA

97 DISCJPLR=AREAD

98 65 CONTINUE

99 C11111 IF AREA PLOT IS NOT REQUIRED, 60 TO 70

100 IF (IPLOTA,LE,0) 60 TO 70

101 CALL CSCHETDISC,VY,JPLR,I,TU,EXFIN,DEEX

102 CALL CSCHETANULUS,VY,JPLR,I,TU,EXMIN,DEEX

103 DISCJPLR=EXFIN

104 DISCJPLR=2*DEEX

105 CALL PLOTJPLR,0,0-37

106 CALL PLOTJPLR,XX,VY,EXEP,DISCJPLR+21,XMIN,DISCJPLR+11,25,25,0,0,21 0870

107 70 50 00

108 C11111 PLOT THE DISC AREA 45. X

109 CALL LINEXXX,DISCJPLR,I,1,3,XMIN,EXEP,DISCJPLR+11,DISCJPLR+21 0910

110 C11111 PLOT THE ANNULUS AREA 45. X

111 CALL LINEXXX,ANNULUS,JPLR,I,1,3,XMIN,EXEP,DISCJPLR+11,DISCJPLR+21 0920

112 111

113 70 RETURN

114	C		21	0950
115		75 FORMAT (1MD//9X,1H1,14X,3HXON,18X,3HYON,16X,4HYONH,12X,4HAREA,14X,21		0960
116		19MDISC AREA,10X,6HENSUBK)	21	0970
117		80 FORMAT (1MD,5X,7HXPIN = ,1PE14.5,5X,7HXMAX = ,1PE14.5,5X,7HJMIN = 21		0980
118		1,16,5X,7HJMAX = ,16)	21	0990
119		85 FORMAT (74X,1PE19.4)	21	1000
120		90 FORMAT (8X,I3,1P6E19.4)	21	1010
121		95 FORMAT (1MD)	21	1020
122		98 FORMAT(11H1)		
123		100 FORMAT(1MD,/,6X,'XON,YON,YONH HAVE BEEN SCALED BY L = ',OPF10.4)		
124		110 FORMAT(1MD,5X,'AREA AND DISC AREA HAVE BEEN SCALED BY AREF = ',		
125		1 OPF10.4)		
126		END	21	1030

1		SUBROUTINE DRAW(KR,KK)	22	0000
2	C(11111	SUBROUTINE ADDED TO DRAW PICTURE OF INLET VIA CALCOMP PLOTTER.	22	0010
3	C(11111	CALLED ONCE FOR EACH SEGMENT	22	0020
4	C		22	0030
5		DIMENSION X(200),Y(200)	22	0040
6		COMMON /SS/ NBDY1,NBDY2,YYPBDY,NBDYS	22	0050
7		COMMON /FOR355/IO,DELS,XBK(20),YBK(20),XON(500),YON(500),DYDX0150022		0060
8		17,ALPHAT500),CAPPA(500),SONT500,PI0180	22	0070
9		COMMON/ LOT/XMIN,YMIN,ORD,EXEP,XX,YY,LPNCHO,IPLOTA,MM	22	0080
10		COMMON/YOL/BAGS(15),BAGX(15),ZAP(15),NZAP(15)	22	0090
11		NL=NR+1	22	0100
12		II=0	22	0110
13		DO 20 I=1,NL	22	0120
14		N=KK+I-1	22	0130
15		IF (11.6E.200.0R.N.6T.500) GO TO 30	22	0140
16	C(11111	TEST EACH (X,Y) PT. EXCLUDE THOSE BEYOND (XX+EXEP+XMIN) INCHES22		0150
17		YMAN=YY+ORD+YMIN		
18		IF (LPNCHO.EQ.2) YONIN=YMAN-(YMIN)-YMAN)		
19		IF (XONIN).LT.XMIN) GO TO 20		
20		IF (YONIN).LT.YMIN) GO TO 20		
21		IF (XONIN)-XX+EXEP-XMIN) 10,10,20	22	0160
22	10	IF (YONIN)-YY+ORD-YMIN) 15,15,20	22	0170
23	15	II=II+1	22	0180
24		X(II)=XONIN)	22	0190
25		Y(II)=YONIN)	22	0200
26		IF (11.NE.1.0R.YYPBDY.EQ.1.) GO TO 20	22	0210
27	C(11111	STORE CURVATURE VALUES OF SEGMENT'S FIRST PT. FOR USE WITH	22	0220
28	C	SUBSEQUENT CURVATURE PLOTS.	22	0230
29		MM=MM+1	22	0240
30		BAGX(MM)=XONIN)	22	0250
31		BAGS(MM)=SONIN)	22	0260
32		ZAP(MM)=CAPPAIN)	22	0270
33		NZAP(MM)=N	22	0280
34	20	CONTINUE	22	0290
35	C(11111	DRAW A SEGMENT MARKER AT FIRST PT. OF SEGMENT	22	0300
36		IF (XON(KK).GT.(XX+EXEP+XMIN).OR.YON(KK).GT.(YY+ORD+YMIN)) GO TO 222		0310
37	*5		22	0320
38		XSYM= (X(1)-XMIN)/EXEP	22	0330
39		YSYM=(Y(1)-YMIN)/ORD	22	0340
40		CALL SYMBOL(XSYM,YSYM,.2,1,0.,-1)	22	0350
41	25	CALL LINE(X,Y,II,1,1,3,XMIN,EXEP,YMIN,ORD)	22	0360
42		RETURN	22	0370
43	30	WRITE (6,35)II,N	22	0380
44	35	FORMAT(1MD,' SCIRCLE ERROR EXIT - DATA POINTS EXCEED 200 ON A SEG	22	0390
45		10R EXCEED 500 ON TOTAL INLET - ' /218)	22	0400
46		STOP	22	0410
47		END	22	0420

```

1      SUBROUTINE PLOXIS (XX,YY,EXEP,ORD,OFSETA,OFSET,SLETRS,SNOSZ,M5,M6,MZ3) 0000
2      I,L,NH,NL) 23 0010
3      C 23 0020
4      CXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX 23 0030
5      CXXXXXXXX SUBROUTINE ADDED TO DRAW AND LABEL AXIS FRAMES FOR ALL PLOTS 23 0040
6      COMMON/ITL/ YITL(9,6) 23 0050
7      UP =11.-YY-2.*SNOSZ 23 0060
8      M1=XX 23 0070
9      M2=YY 23 0080
10     CALL PLOT(4.,-11.,-3) 23 0090
11     CALL PLOT(0.,UP,-3) 23 0100
12     DO 25 I=1,M1 23 0110
13     X=I 23 0120
14     P=EXEP*X+OFSETA 23 0130
15     CALL PLOT(X,0.,2) 23 0140
16     CALL PLOT(X,.2,2) 23 0150
17     M=I/2 23 0160
18     B=FLOAT(I)-FLOAT(M)*X/2. 23 0170
19     IF (B) 10,10,25 23 0180
20     IF (K5) 15,15,20 23 0190
21     15 CALL NUMBER(X-SNOSZ,-SNOSZ-.10,SNOSZ,P,0.,NH) 23 0200
22     GO TO 25 23 0210
23     20 SN = 1.333*SNOSZ 23 0220
24     CALL NUMBER(X-SNOSZ-SNOSZ,-SN-SNOSZ-.10,SN,10.,0.,-1) 23 0230
25     CALL NUMBER(999.0,-SNOSZ -.10,SNOSZ,P,0.,NH) 23 0240
26     25 CALL PLOT(X,0.,3) 23 0250
27     B = (XX-54.*SLETRS)/2. 23 0260
28     CALL SYMBOL(B,-SNOSZ-SNOSZ-SLETRS-.15-.6,SLETRS,YITL(1,M),0.,54) 23 0270
29     CALL PLOT(0.,0.,3) 23 0280
30     DO 45 J=1,M2 23 0290
31     Y=J 23 0300
32     O=ORD*Y+OFSET 23 0310
33     CALL PLOT(0.,Y,2) 23 0320
34     CALL PLOT(.2,Y,2) 23 0330
35     M=J/2 23 0340
36     B=FLOAT(J)-FLOAT(M)*Y/2. 23 0350
37     IF (B) 30,30,45 23 0360
38     IF (K6) 35,35,40 23 0370
39     35 CALL NUMBER(-4.*SNOSZ -.15,Y,SNOSZ,0,0.,NL) 23 0380
40     GO TO 45 23 0390
41     40 SN = 1.333*SNOSZ 23 0400
42     CALL NUMBER(-.15 -SN-SN-SN ,Y-SNOSZ,SN,10.,0.,-1) 23 0410
43     CALL NUMBER(999.0,Y-SN-SNOSZ,SNOSZ,0,0.,NL) 23 0420
44     45 CALL PLOT(0.,Y,3) 23 0430
45     C = (YY-54.*SLETRS)/2. 23 0440
46     CALL SYMBOL(-SNOSZ-SNOSZ-SNOSZ-.15-.6,C,SLETRS,YITL(1,M),90.,54) 23 0450
47     CALL PLOT(0.,Y,3) 23 0460
48     CALL PLOT(XX,YY,2) 23 0470
49     CALL PLOT(XX,0.,2) 23 0480
50     DO 50 J=1,M2,2 23 0490
51     Y=J 23 0500
52     IF (Y.EQ.YY) GO TO 55 23 0510
53     CALL PLOT(XX,Y,3) 23 0520
54     CALL PLOT(0.,Y,2) 23 0530
55     IF ((Y+1.).EQ.YY) GO TO 55 23 0540
56     CALL PLOT(0.,Y+1.,3) 23 0550

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57	50 CALL PLOT(XX,Y+1.,2)	23	0560
58	55 CONTINUE	23	0570
59	DO 60 J=1,M1,2	23	0580
60	X=J	23	0590
61	IF (X.EQ.XX) 60 TO 65	23	0600
62	CALL PLOT(XX-X,YY,3)	23	0610
63	CALL PLOT(XX-X,0.,2)	23	0620
64	IF ((XX-X-1.).EQ.0.) 60 TO 65	23	0630
65	CALL PLOT(XX-X-1.,0.,3)	23	0640
66	60 CALL PLOT(XX-X-1.,YY,2)	23	0650
67	65 RETURN	23	0660
68	END	23	0670

1	SUBROUTINE SINTP (Z,W,N,X1,Y1)	24	00C0
2	CITITT ENLARGED FROM THE ORIGINAL (200)	24	0010
3	DIMENSION A(13)	24	0020
4	DIMENSION XT(250), Y(250), Z(250), W(250)	24	0030
5	DATA EODFF/6HENDOFF/	24	0040
6	DO 10 I=1,N	24	0050
7	X(I)=Z(I)	24	0060
8	10 Y(I)=W(I)	24	0070
9	CALL SORTXY (X,Y,N)	24	0080
10	C	24	0090
11	DO 15 I=1,N	24	0100
12	K=I	24	0110
13	IF (X1.GT.X(I)) 60 TO 15	24	0120
14	IF (X1.EQ.X(I)) 60 TO 20	24	0130
15	IF (X1.LT.X(I)) 60 TO 25	24	0140
16	15 CONTINUE	24	0150
17	20 Y1=Y(K)	24	0160
18	60 TO 30	24	0170
19	25 IF (K.EQ.1) 60 TO 35	24	0180
20	IF (K.EQ.N) K=N-1	24	0190
21	W1=(X1-X(K))*(X1-X(K+1))/(X(K-1)-X(K))/(X(K-1)-X(K+1))	24	0200
22	W2=(X1-X(K-1))*(X1-X(K+1))/(X(K)-X(K-1))/(X(K)-X(K+1))	24	0210
23	W3=(X1-X(K-1))*(X1-X(K))/(X(K+1)-X(K-1))/(X(K+1)-X(K))	24	0220
24	Y1=Y(K-1)+W1*Y(K)+W2*Y(K+1)+W3	24	0230
25	30 RETURN	24	0240
26	35 Y1=0.0	24	0250
27	RETURN	24	0260
28	ENTRY ECHO	24	0270
29	CALL ERTN(16,'@A56,T 25. . .')	24	0280
30	WRITE (6,40)	24	0290
31	40 FORMAT(1H1,23X,' INPUT FILE DUMP//')	24	0300
32	45 READ (5,50,END=60)A	24	0310
33	50 FORMAT(13A6)	24	0320
34	WRITE (6,55)A	24	0330
35	WRITE (25,50)A	24	0340
36	55 FORMAT(1H ,13A6)	24	0350
37	60 TO 45	24	0360
38	60 WRITE (25,50)EODFF	24	0370
39	REWIND 25	24	0380
40	RETURN	24	0390
41	END	24	04C0

```

1  SUBROUTINE SORTXY(X,Y,MPTS)
2  DIMENSION X(100),Y(100)
3  10 M=MPTS
4  15 N=N-1
5  20 DO 55 K=1,M
6  YMIN=X(K)
7  JAD=K
8  JLC=K
9  25 DO 45 JK=JL,M
10 30 IF YMIN=X(JK) 45,45,35
11 35 KMIN=X(JK)
12 40 JAD=JK
13 45 CONTINUE
14 50 YMIN=Y(JAD)
15 X(JAD)=X(K)
16 Y(JAD)=Y(K)
17 X(K)=XMIN
18 Y(K)=YMIN
19 55 CONTINUE
20 RETURN
21 END

```

Program 24Y

1Y

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1 C NEWMAN PROGRAM - TWO DIMENSIONAL MULTIELEMENT AIRFOILS EXEC001
2 C EXEC002
3 C THIS IS THE EXECUTIVE ROUTINE FOR THE NEW NUEMANN EXEC003
4 C EXEC004
5 C EXEC005
6 COMMON /BFLA6/ IDP(10), INL(10), IFL(10), NL(10), LIFT(10), EXEC006
7 1 IDMF(10), ISAV1(10), ISAV2(10), ISAV3(10), EXEC007
8 2 BTITLE(10, 7), IPT, IBST, IBTOT, MELTOT, EXEC008
9 3 ITRB(10), INMB(10), CHORQB(10), IDQ(10), LIFTOT EXEC009
10 4 ,IPRF(10), IFST(10), ISEC(10), FTITLE(15), IPVR(10) EXEC010
11 C EXEC011
12 C EXEC012
13 C EXEC013
14 COMMON /COMBOD/CCL, INCLT, CLT, ALPHA, SUMDS(10), TLU(10,12),IND EXEC014
15 1 ,ALPHA0, CMU(10), SMDSWF(10), MIQ(10) EXEC015
16 C EXEC016
17 COMMON /FILEID/ IFILE1, IFILE2, IFILE3, IFILE4, IFILE5, EXEC017
18 1 IFILE6, IFILE7, IFILE8, IFILE9, IFIL10, EXEC018
19 2 IFIL11, IFIL12, IFIL13, IFIL14, IFIL15 EXEC019
20 3 ,IFIL16, IFIL17, IFIL18, IFIL19, IFIL20 EXEC020
21 COMMON /MDATA/ ISOL,IOFF,NOMU,NBNU,IPRINT,MORE,M EXEC021
22 COMMON/ROTAT/PROT, ROTRAD(10) EXEC022
23 C EXEC023
24 C SET UP THE VARIOUS STORAGE UNITS REQUIRED BY THE PROGRAM EXEC024
25 CALL FILES EXEC025
26 10 CONTINUE EXEC026
27 REWIND IFILE1 EXEC027
28 REWIND IFILE2 EXEC028
29 REWIND IFILE3 EXEC029
30 REWIND IFILE4 EXEC030
31 REWIND IFILE8 EXEC031
32 REWIND IFILE9 EXEC032
33 REWIND IFIL10 EXEC033
34 REWIND IFIL11 EXEC034
35 REWIND IFIL12 EXEC035
36 REWIND IFIL13 EXEC036
37 REWIND IFIL14 EXEC037
38 REWIND IFIL15 EXEC038
39 REWIND IFIL16 EXEC039
40 REWIND IFIL17 EXEC040
41 CALL TSETV EXEC041
42 C EXEC042
43 C EXEC043
44 CALL MAIN1 EXEC044
45 C EXEC045
46 C EXEC046
47 C OBTAIN SIGMA SOLUTIONS EXEC047
48 ISIZE = 11413 EXEC048
49 CALL SOLVE (MELTOT, M, ISIZE, ISOL) EXEC049
50 C EXEC050
51 C EXEC051
52 CALL TIMEVIT) EXEC052
53 WRITE(6,70) T EXEC053
54 70 FORMAT (1MD, 'SOLVE COMPLETE. READ FLOW TITLE & CONTROL CARD, ', EXEC054
55 1 'CALL COMBO, T = ', F9.3, 'SECONDS. ') EXEC055
56 C EXEC056
57 CALL MAIN3 EXEC057
58 C EXEC058
59 C EXEC059
60 IF (MORE .EQ. 1) GO TO 10 EXEC060
61 STOP EXEC061
62 END EXEC062

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1      FUNCTION ABFORM (XI, YI, OSI, SINI, COSI, IEST,
2      1      XJ, YJ, OSJ, SINJ, COSJ, ISEC,
3      2      AI, AC, AP, BQ, BI, BC, BP)
4      C
5      C
6      C THIS ROUTINE ACTUALLY CALCULATES THE INDUCED VELOCITY
7      C ARRAY ELEMENTS A AND B.
8      C
9      C
10     DATA E1,E2/169.0,11.111/.EQ,EY/O.OQB1,Q.O14
11     C
12     C
13     DX = XI - XJ
14     DY = YI - YJ
15     ROSQ = DX**2 + DY**2
16     DSJSQ = OSJ**2
17     C
18     IF (ROSQ .LT. DSJSQ*E1) GO TO 18
19     C
20     C USE FAR FIELD FORMULAS
21     VX = 2.*OSJ/ROSQ
22     VY = VX*DY
23     VZ = VX*DX
24     AO = -VX*SINI + VY*COSI
25     ABFORM = AO
26     BO = VX*COSI + VY*SINI
27     IF (IEST .EQ. 0) .AND. (ISEC .EQ. 0) RETURN
28     SC = SINI*COSJ
29     SS = SINI*SINJ
30     CS = COSI*SINJ
31     CC = COSI*COSJ
32     SSCC = ISS - CC*VX*VY/12.0
33     CSSC = ICS + SC*VX*VY/12.0
34     DXR = 1.0 - 2.0*DX**2/ROSQ
35     DYR = 1.0 - 2.0*DY**2/ROSQ
36     DDR = DSJSQ/(ROSQ*6.0)
37     C
38     AI = DDR*(SC*DXR - CS*DYR) - SSCC
39     BI = -DDR*(CC*DXR + SS*DYR) + CSSC
40     AC = -DDR*(ISS*DXR + CC*DYR) - CSSC
41     BC = DDR*(ICS*DXR - SC*DYR) - SSCC
42     AP = AO/12.0
43     BP = BO/12.0
44     RETURN
45     C
46     C USE NEAR FIELD FORMULAS
47     ID X = DX*COSJ + DY*SINJ
48     Y = DY*COSJ - DX*SINJ
49     S = SINI*COSJ - COSI*SINJ
50     C = COSI*COSJ + SINI*SINJ
51     C
52     IF (ROSQ .GT. DSJSQ*E2) GO TO 2C
53     C
54     C USE EXACT FORMULAS
55     XB = X/OSJ
56     YB = Y/OSJ

```

```

ABF0001
ABF0002
ABF0003
ABF0004
ABF0005
ABF0006
ABF0007
ABF0008
ABF0009
ABF0010
ABF0011
ABF0012
ABF0013
ABF0014
ABF0015
ABF0016
ABF0017
ABF0018
ABF0019
ABF0020
ABF0021
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ABF0023
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ABF0035
ABF0036
ABF0037
ABF0038
ABF0039
ABF0040
ABF0041
ABF0042
ABF0043
ABF0044
ABF0045
ABF0046
ABF0047
ABF0048
ABF0049
ABF0050
ABF0051
ABF0052
ABF0053
ABF0054
ABF0055
ABF0056

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57      R0 = XB*2 + YB*2      ABF0057
58      RD = XB*2 - YB*2      ABF0058
59      R1 = P0 + XR + 0.25    ABF0059
60      R2 = R0 - XR + 0.25    ABF0060
61      C                        ABF0061
62      VX = +ALOG(LIROSQ+CS*XR+0.25*DSJSLJ/IR0SQ-DSJ*XR+0.25*DSJSDJ) ABF0062
63      Y = YDSJ              ABF0063
64      X = ROSQ - 0.25*DSJSDJ ABF0064
65      VY = 2.0*ATAN2(V,X)    ABF0065
66      IF (IIFST.EQ.0) .AND. (ISEC.EQ.0) 60 TO 30 ABF0066
67      C                        ABF0067
68      VXQ = VY              ABF0068
69      VY0 = VY              ABF0069
70      VX1 = YB*VY0 + XB*VY0 - 2.0 ABF0070
71      VY1 = YB*VY0 - YB*VY0 ABF0071
72      VXC = -2.0*VY1 + XB*YB/IR1*21 ABF0072
73      VYC = 2.0*VX1 + 1.01-2.0*IR0*2 - 0.25*RD1/IR1*21 ABF0073
74      VXP = RD*VX0 + 2.0*XB*YB*VY0 - 1.0 ABF0074
75      VYP = RD*VY0 - 2.0*YB*XB*VY0 - 1.0 ABF0075
76      C                        ABF0076
77      60 TO 30              ABF0077
78      C                        ABF0078
79      C                        ABF0079
80      C  HSF MULTIPOLY FORMULAS ABF0080
81      20 AE = XDSJ/ROSQ      ABF0081
82      BE = YDSJ/ROSQ      ABF0082
83      ASQ = X*2/ROSQ        ABF0083
84      ESQ = DSJSDJ/ROSQ     ABF0084
85      VX = 2.0*AE*(1.0 + (ASQ - 0.75)*ESQ/3.0) ABF0085
86      VY = 2.0*BE*(1.0 + (ASQ - 0.25)*ESQ/3.0) ABF0086
87      IF (IIFST.EQ.0) .AND. (ISEC.EQ.0) 60 TO 30 ABF0087
88      A4 = 0.15*16.0*ASQ*ASQ - 1.01 + 1.01*ESQ ABF0088
89      VX1 = ESQ*(2.0*ASQ - 1.0 + A4)/6.0 ABF0089
90      VY1 = AE*BE*(1.0 + 0.3*12.0*ASQ - 1.01*ESQ)/3.0 ABF0090
91      VXC = AE*BE*(1.0 + 0.3*16.0*ASQ - 1.01*ESQ)/3.0 ABF0091
92      BSQ = Y*2/ROSQ        ABF0092
93      VYC = ESQ*(2.0*BSQ - 1.0 - 0.5*A4)/6.0 ABF0093
94      VYP = 11.0 + (ASQ - 0.25)*ESQ*0.61/6.0 ABF0094
95      CVP = AE*VYP          ABF0095
96      VYP = BE*VYP          ABF0096
97      C                        ABF0097
98      C                        ABF0098
99      30 CONTINUE          ABF0099
100     ABFORM = -S*VX + C*VY ABF0100
101     B0 = C*VX + S*VY      ABF0101
102     IF (IIFST.EQ.0) .AND. (ISEC.EQ.0) RETURN ABF0102
103     A1 = -S*VX1 + C*VY1    ABF0103
104     AC = -S*VXC + C*VYC     ABF0104
105     AP = -S*VXP + C*VYP     ABF0105
106     C                        ABF0106
107     B1 = C*VX1 + S*VY1     ABF0107
108     BC = C*VXC + S*VYC     ABF0108
109     BP = C*VXP + S*VYP     ABF0109
110     C                        ABF0110
111     C                        ABF0111
112     RETURN                  ABF0112
113     END                      ABF0113

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1  FUNCTION ARSIN(X)
2  ARSIN=ASIN(X)
3  RETURN
4  END

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57	30 READ(IF12) (VN(1,K), K = 1,M)	ASEM057
58	REWINO IF12	ASEM058
59	DO 40 K = 1,M	ASEM059
60	40 CALL SAVE(IF12, 1, 1, 1, MELTOT, VN(1,K), 1, VNA)	ASEM060
61	C	ASEM061
62	C	ASEM062
63	C CHECK IF NON-UNIFORM ONSET FLOWS INPUT	ASEM063
64	340 IF (INMU .LE. 0) RETURN	ASEM064
65	C	ASEM065
66	DO 560 I = 1,MONU	ASEM066
67	M = M + 1	ASEM067
68	C	ASEM068
69	C PRESET ALL VELOCITIES TO ZERO	ASEM069
70	DO 350 I = 1, MELTOT	ASEM070
71	VNUF(I) = 0.0	ASEM071
72	350 VNUF(I) = 0.0	ASEM072
73	CALL SAVE(IF11, 1, 1, MELTOT, VNUF, 1, VN)	ASEM073
74	C	ASEM074
75	C	ASEM075
76	C PRESET ILU ARRAYS TO ZERO	ASEM076
77	DO 355 I = 1, IRTOT	ASEM077
78	355 ILU(I,M) = 0.0	ASEM078
79	C	ASEM079
80	C	ASEM080
81	C READ IN COMBINATION CONSTANTS FOR NON-UNIFORM FLOWS	ASEM081
82	ITVP = 10	ASEM082
83	360 READ(5,440) (CMU(I), I = 1, 6), ITYPE	ASEM083
84	IF (ITYPE .NE. ITVP) CALL TYPE(ITVP, ITYPE)	ASEM084
85	C	ASEM085
86	DO 460 MB = 1, MBNU	ASEM086
87	LB = 0	ASEM087
88	C READ BODY CONTROL CARD FOR NON-UNIFORM FLOW	ASEM088
89	ITVP = 11	ASEM089
90	READ(5,370) I800, IM, ITMM, I(1:1), IE(1), I = 1, 5), CB, ITYPE	ASEM090
91	370 FORMAT (11,1311, 102H13), F10.5, 5X12)	ASEM091
92	IF (ITYPE .NE. ITVP) CALL TYPE(ITVP, ITYPE)	ASEM092
93	C	ASEM093
94	C SEARCH FOR BODY ID AND SET LIFTING BODY COUNTER	ASEM094
95	DO 380 IB = 1, IRTOT	ASEM095
96	IF (LIFT(I8) .NE. 0) LB = LB + 1	ASEM096
97	IIB = IB	ASEM097
98	IF (I800 .EQ. IOR(IIB)) GO TO 400	ASEM098
99	380 CONTINUE	ASEM099
100	WRITE(16,390) I800	ASEM100
101	390 FORMAT(1H0, 'NON-UNIFORM FLOW INPUT, BODY WITH ID = ', I1,	ASEM101
102	1	ASEM102
103	STOP	ASEM103
104	C	ASEM104
105	400 MO = MIN(IIB) - 1	ASEM105
106	IF (IMRT .GT. 0) GO TO 455	ASEM106
107	C	ASEM107
108	DO 450 IL = 1, MN	ASEM108
109	IO = MO + I(ILL)	ASEM109
110	IF = MO + IE(ILL)	ASEM110
111	C	ASEM111
112	IF (IM .EQ. 0) GO TO 420	ASEM112
113	C READ IN NORMAL VELOCITIES	ASEM113

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114      ITYP = 12                                ASEM114
115      M2 = IO - 1                                ASEM115
116      410 M1 = M2 + 1                            ASEM116
117      M2 = M1 + 5                                ASEM117
118      READ(5,940) (VMUF(I), I = M1,M2), ITYPE    ASEM118
119      IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE) ASEM119
120      IF (M2 .LT. IF) GO TO 413                    ASEM120
121      420 IF (IT .EQ. 0) GO TO 450                ASEM121
122      C                                           ASEM122
123      C READ IN TANGENTIAL VELOCITIES             ASEM123
124      ITYP = 13                                    ASEM124
125      M2 = IO - 1                                ASEM125
126      430 M1 = M2 + 1                            ASEM126
127      M2 = M1 + 5                                ASEM127
128      READ(5,940) (VTUF(I), I = M1,M2), ITYPE    ASEM128
129      940 FORMAT(6E10.0, 10X12)                 ASEM129
130      IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE) ASEM130
131      IF (M2 .LT. IF) GO TO 451                  ASEM131
132      450 CONTINUE                                ASEM132
133      GO TO 457                                    ASEM133
134      C                                           ASEM134
135      C SPECIAL ROTATING FLOW - INPUT GENERATED ASEM135
136      455 CONTINUE                                ASEM136
137      C COUNTERS FOR MATRIX STORAGE              ASEM137
138      IO = MO + 1                                ASEM138
139      IF = MO + NL(IIB)                          ASEM139
140      C                                           ASEM140
141      C COUNTER FOR BODY GEOMETRY                 ASEM141
142      JI = INL(IIB) - 1                          ASEM142
143      C ROTATION RADIUS                          ASEM143
144      ROTRAD(IIB) = CB                           ASEM144
145      CB = 1.0/CB                                ASEM145
146      DO 456 I = IO, IF                          ASEM146
147      JI = JI + 1                                ASEM147
148      VMUF(I) = -Y0(JI)*SA(JI) - X0(JI)*CA(JI)    ASEM148
149      VTUF(I) = Y0(JI)*CA(JI) - X0(JI)*SA(JI)    ASEM149
150      456 CONTINUE                                ASEM150
151      WRITE(6,900) NOMU, MROT, MBMU, L, NB, IO, IF ASEM151
152      900 FORMAT(1ND, 7MN0NU = ,13,5X7HNR0T = ,13,5X7HMBMU = ,13, ASEM152
153      1 4ML = ,13,5X5HNB = ,13, 5X5HIO = ,13,5X5HIO = ,13) ASEM153
154      457 CONTINUE                                ASEM154
155      C                                           ASEM155
156      C ALL NON-U VELOCITIES FOR BODY IO=IB00 READ IN. ASEM156
157      C                                           ASEM157
158      C SCALE VELOCITIES                         ASEM158
159      IF (ABS(CB) .LT. 1.E-6) GO TO 480          ASEM159
160      DO 470 I = IO, IF                          ASEM160
161      VMUF(I) = VMUF(I)*CB                       ASEM161
162      470 VTUF(I) = VTUF(I)*CB                   ASEM162
163      480 CONTINUE                                ASEM163
164      C                                           ASEM164
165      C RESET TLU IF LIFTING BODY                ASEM165
166      IF (LIFT(IIB) .EQ. 0) GO TO 460            ASEM166
167      IO = MO + 1                                ASEM167
168      IF = MO + NL(IIB)                          ASEM168
169      TLU(IB, M) = VTUF(IO) + VTUF(IF)           ASEM169
170      460 CONTINUE                                ASEM170

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171	C		ASEM171
172	C	ALL VELOCITIES FOR A GIVEN NON-U FLOW READ IN.	ASEM172
173		CALL SAVE(IFO4, 1, 1, NELTOT, VNLF, 1, VN)	ASEM173
174		CALL SAVE(IFO2, 1, 1, NELTOT, VTUF, 1, VN)	ASEM174
175	C		ASEM175
176	C		ASEM176
177		490 J2 = 0	ASEM177
178		495 WRITE(6,500)	ASEM178
179		500 FORMAT(1H1)	ASEM179
180		WRITE(6,510) 1, M	ASEM180
181		510 FORMAT(1H0, 15X, 'NON-UNIFORM FLOW NUMBER ', I2, ', P = ', I3//	ASEM181
182		1 T15, 'I', I27, ' VN', T88, ' VT'	ASEM182
183		2 T75, 'I', T87, ' VN', T104, ' VT'	ASEM183
184		J1 = J2 + 1	ASEM184
185		JMX = J1 + 49	ASEM185
186		N2 = (NELTOT + 1 + J2)/2	ASEM186
187		IF (JMX .GT. N2) JMX = N2	ASEM187
188		J2 = JMX	ASEM188
189		DO 530 J = J1, JMX	ASEM189
190		J2 = J2 + 1	ASEM190
191		IF (J2 .GT. NELTOT) GO TO 540	ASEM191
192		WRITE(6,520) J, VNLF(J), VTUF(J), J2, VNLF(J2), VTUF(J2)	ASEM192
193		520 FORMAT(1H, 11X I3, 2(5XF12.6), 23X I3, 2(5XF12.6))	ASEM193
194		530 CONTINUE	ASEM194
195		IF (J2 .LT. NELTOT) GO TO 495	ASEM195
196		GO TO 550	ASEM196
197		540 WRITE(6,520) J, VNLF(J), VTUF(J)	ASEM197
198		550 CONTINUE	ASEM198
199	C		ASEM199
200	C		ASEM200
201		560 CONTINUE	ASEM201
202	C		ASEM202
203	C		ASEM203
204		RETURN	ASEM204
205		END	ASEM205

```

1      SUBROUTINE COMBOIN, LT, MT, NONL)
2      C
3      C
4      C CALCULATE COMBINATION CONSTANTS CCK, SYSTEM ANGLE OF ATTACK
5      C AND CL, SYSTEM LIFT CURVE CONSTANTS IN1, K2, & K11, AND CK CONSTANTS
6      C
7      C M = TOTAL NUMBER OF ELEMENTS
8      C LT = NUMBER OF LIFTING BODIES
9      C MT = NUMBER OF ONSET FLOWS
10     C MT1 = MT + 1
11     C CCL = CHORD FOR CLT CALCULATION
12     C INCLT FLAG = 0, ALPHA INPUT (COMES IN AS CLT)
13     C NOT = 0, CLT INPUT
14     C
15     C
16     DIMENSION BLU(500), DV(10,12), A(10,10), CCK(10,12), SIG(100)
17     C
18     C
19     COMMON /FILEID/ IF01, IF02, IF03, IF04, IF05, IF06, IF07, IF08,
20     1 IF09, IF10, IF11, IF12, IF13, IF14, IF15
21     2 IF16, IF17, IF18, IF19, IF20
22     C
23     C
24     COMMON /COMBOD/CCL, INCLT, CLT, ALPHA, SUMDS(10), TLU(10,12), IND
25     1 , ALPHAO, CMU(10), SMDSMF(10), MIO(10)
26     COMMON /SIGMAS/CSTG(500), CK(12)
27     EQUIVALENCE (BLU(1), CSTG(1))
28     C
29     DATA PI, RC/3.1415927, 1.7953293E-2/
30     NRMS = 2
31     IF (NONL .GT. 0) NRMS = 3
32     C
33     PIC = 0.0
34     RK1 = 0.0
35     RK2 = 0.0
36     RK3 = 0.0
37     ALPHAO = 0.0
38     IF (LT .EQ. 0) GO TO 140
39     C
40     C
41     C CALCULATE TRAILING EDGE VELOCITY DIFFERENCE ARRAY
42     C
43     C INITIALIZE DV ARRAY TO TLU VALUES
44     DO 5 L = 1,LT
45     DO 5 K = 1,MT
46     5 DV(L,K) = TLU(L,K)
47     C
48     REMIND IF13
49     DO 30 L = 1,LT
50     CALL GETT (IF13, 1, N, BLU, 1, D)
51     REMIND IF14
52     DO 20 K = 1,MT
53     CALL GETT (IF14, 1, N, SIG, 1, D)
54     DO 10 J = 1,N
55     10 DV(L,K) = DV(L,K) + BLU(J)*SIG(J)
56     20 CONTINUE

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57      30 CONTINUE
58      C
59      C
60      C SET ARRAYS TO OBTAIN COMBINATION CONSTANTS CCK. USE MIS1
61      C AILT,LT) IS COEF. ARRAY
62      C CCK(1,3) IS INPUT AS RHS. RETURNED AS CCK.
63      C
64      DO 70 L = 1,LT
65      CCK(1,1) = -DV(1,1)
66      CCK(1,2) = -DV(1,2)
67      CCK(1,3) = 0.0
68      IF (NONU .EQ. 0) GO TO 50
69      M = LT + 2
70      DO 90 K = 1,NONU
71      M = M + 1
72      90 CCK(1,3) = CCK(1,3) - DV(1,M)*CNU(K)
73      50 CONTINUE
74      C
75      DO 60 N = 1,LT
76      K2 = N + 2
77      60 A(1,N) = DV(1,K2)
78      C
79      70 CONTINUE
80      C
81      IF (LT .GT. 1) GO TO 90
82      C
83      C ONLY ONE LIFTING BODY. CALCULATE COMBINATION
84      C CONSTANTS STRAIGHT AWAY.
85      DO 80 K = 1,NRHS
86      80 CCK(1,K) = CCK(1,K)/A(1,1)
87      GO TO 110
88      C
89      C CALL MIS1 FOR SOLUTION
90      90 CONTINUE
91      D = 1.0
92      L10 = 10
93      CALL MIS1 (A, LT,L10, CCK, NRHS, NERR, 0)
94      C
95      C CHECK FOR SINGULAR CASE
96      WRITE (6,100) NERR
97      100 FORMAT(1HD, 'ON RETURN FROM MIS1, NERR = ', I2)
98      C
99      C CALCULATE SYSTEM ANGLE OF ATTACK (ALPHA) AND TOTAL LIFT (CLT).
100      C
101      110 CONTINUE
102      DO 130 L = 1,LT
103      RK1 = RK1 + SMDSWF(L) * CCK(1,1)
104      RK2 = RK2 + SMDSWF(L) * CCK(1,2)
105      120 RK3 = RK3 + SMDSWF(L) * CCK(1,3)
106      C
107      130 CONTINUE
108      C
109      PIC = 8.*PI/CCL
110      ALPHA0 = ATAN2(RK1, RK2)
111      C
112      C CHECK IF ALPHA OR CLT INPUT
113      IF ((INCLT .EQ. 0) GO TO 140

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114 C
115 C CLT INPUT, DETERMINE ALPHA COMB114
116 RK4 = (CLT/PIC - RK3)/SQRT(RK1**2 + RK2**2) COMB115
117 ALPHA = -ALPHA0 + ARSIN(RK4) COMB116
118 GO TO 150 COMB117
119 C COMB118
120 C ALPHA INPUT, CALCULATE CLT (REPEAT IF INPUT) COMB119
121 140 ALPHA = ALPHA*RC COMB120
122 150 COSA = COS(ALPHA) COMB121
123 SINA = SIN(ALPHA) COMB122
124 IF (INCLT .EQ. 0) COMB123
125 ICLT = PIC*(RK1*COSA + RK2*SINA + RK3) COMB124
126 ALPHA = ALPHA/RC COMB125
127 ALPHA0 = ALPHA0/RC COMB126
128 C COMB127
129 C CALCULATE COEFFICIENTS CN(I) FOR LIFTING BODIES COMB128
130 CN(1) = COSA COMB129
131 CN(2) = SINA COMB130
132 C COMB131
133 IF (LT .LE. 0) GO TO 165 COMB132
134 DO 160 L = 1,LT COMB133
135 N = L + 2 COMB134
136 CN(I) = CCK(L,1)*COSA + CCK(L,2)*SINA + CCK(L,3) COMB135
137 160 CONTINUE COMB136
138 C COMB137
139 C SET NON-U FLOW CN'S TO ASSUMED VALUE OF 1.0 COMB138
140 165 CONTINUE COMB139
141 IF (NONU .LE. 0) GO TO 180 COMB140
142 N = LT + 2 COMB141
143 DO 170 J = 1,NONU COMB142
144 N = N + 1 COMB143
145 170 CN(I) = CN(I,J) COMB144
146 180 CONTINUE COMB145
147 C COMB146
148 C CALCULATE COMBINED SIGMAS COMB147
149 DO 190 J = 1,M COMB148
150 190 CSIG(J) = 0.0 COMB149
151 REMIND IF14 COMB150
152 DO 210 K = 1,MT COMB151
153 CALL GETT (IF14, 1, N, SIG, 1, 0) COMB152
154 DO 200 J = 1,M COMB153
155 200 CSIG(J) = CSIG(J) + SIG(J)*CN(K) COMB154
156 210 CONTINUE COMB155
157 C COMB156
158 C COMB157
159 C PRINT OUT SOME STUFF FOR CHECKOUT PURPOSES COMB158
160 WRITE (6,220) ALPHA COMB159
161 220 FORMAT (1H1, 'COMBINATION CONSTANTS'//T10, 'ALPHA = ', COMB160
162 1 T30, 'D', T50, '90', T64, F12.6) COMB161
163 C COMB162
164 IF (LT .LE. 0) GO TO 245 COMB163
165 DO 230 L = 1, LT COMB164
166 N = L + 2 COMB165
167 230 WRITE (6,240) L, CCK(L,1), CCK(L,2), CN(I) COMB166
168 240 FORMAT (1H0, T14, I2, T24, F12.6, T44, F12.6, T64, F12.6) COMB167
169 C COMB168
170 245 CONTINUE COMB169
171 WRITE (6,250) RK1, RK2, RK3, ALPHA0, ALPHA, CLT COMB170
172 250 FORMAT (1H0, 'LIFT CURVE CONSTANTS'//T10, 'RK1 = ', F12.6, COMB171
173 1 T30, 'RK2 = ', F12.6, T50, 'RK3 = ', F12.6//T10, COMB172
174 2 'ALPHA0 = ', F12.6//T10, 'ALPHA = ', F12.6//T10, COMB173
175 3 'CLT = ', F12.6) COMB174
176 C COMB175
177 C COMB176
178 C COMB177
179 RETURN COMB178
180 END COMB179
COMB180

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1      SUBROUTINE      ELFORM (SUMDS)
2      C
3      DIMENSION X(500),Y(500),XO(500),YO(500),DS(500),SA(500),CA(500),
4      1      ICR(10), INL(10), IFL(10), NL(10), LIFT(10), IRMF(10),
5      2      ISAV(10), ISAV2(10), ISAV3(10), TTITLE( 7),
6      3      PTITLE(10, 7), SUMDS(10)
7      5      XP(500), YP(500)
8      C
9      C
10     COMMON /ELDATA/ XO, YO, DS, SA, CA, CURV(500), DL(500)
11     COMMON/GCOEFS/ CO(500), CF(500), CG(500), CI(500), VE(500)
12     C
13     COMMON /FILEID/ IF1, IF2, IF3, IF4, IF5, IF6, IF7, IF8, IF9,
14     1 IF10, IF11, IF12, IF13, IF14, IF15, IF16, IF17, IF18, IF19, IF20
15     C
16     COMMON /BFLAG/ IDB, INL, IFL, NL, LIFT, IDBF, ISAV1, ISAV2,
17     1      ISAV3, RTITLE, IRT, IRST
18     2      IBOTOT, MELTOT, ITRB(10), INMB(10), CHORDB(10),
19     3      IRD(10), LIFTOT
20     4      IPRB(10), IFST(10), ISEC(10), FTITLE(15), IPVR(10)
21     C
22     COMMON /GEOMD/ X, Y
23     C
24     EQUIVALENCE (XP(1), CO(1)), (YP(1), CF(1))
25     C
26     DATA      IBMAX,MAXEL/      10, 500/, DR/1.74532925E-2/
27     1      , EPS/1.0E-7/
28     C
29     C
30     C
31     C
32     C
33     ISAVU = IF8
34     C
35     C
36     C READ IN BODY TITLE AND CONTROL CARD
37     10 IITY = 1
38     READ (5,20) ID, ISV, ILIFT, TTITLE
39     1      ,IPARA, IFIRST, ISECND
40     2      ,ITR, INORM, IBOD, IDOLD, IPVOR, LAST
41     3      ,IITYP
42     C
43     20 FORMAT (3(11,2X), 1X7A4, 5X9(11,2X), 1X11)
44     IF (IITYP.NE. IITY) CALL TYPE(IITYP, ITYPE)
45     C
46     C READ IN COORDINATE TRANSFORMATION CARD IF REQUIRED
47     IITY = 2
48     CHORD = 0.0
49     IF (ITR.EQ. 0 .OR. ITR.EQ. 2) GO TO 40
50     READ (5,30) CHORD, XMULT, YMULT, CX, CY, THETA, XTO, YTO, ITYPE
51     30 FORMAT (7(F8.0,1X), F8.0, 11)
52     IF (IITYP.NE. IITY) CALL TYPE(IITYP, ITYPE)
53     C
54     C DETERMINE STORAGE SEQUENCE
55     IF (IBOD.LT.1.OR. IBOD.GT.6) GO TO 40
56     40 GO TO (60,70,100,140,140,210), IBOD

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ELF0001
ELF0002
ELF0003
ELF0004
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ELF0053
ELF0054
ELF0055

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57	C		ELF0056
58		99 WRITE (6,50) IROD	ELF0057
59		50 FORMAT (1H1, 'THE OPTION IROD = '.I3, ' IS NOT LEGITIMATE.')	ELF0058
60		60 TO 640	ELF0059
61	C		ELF0060
62	C	NEW GEOMETRY. START A NEW SEQUENCE	ELF0061
63		60 IRT = 0	ELF0062
64		IRST = 0	ELF0063
65		IRS = 0	ELF0064
66		REWIND ISAVU	ELF0065
67	C		ELF0066
68	C	NEW GEOMETRY. CONTINUE SEQUENCE	ELF0067
69		70 IRT = IRT + 1	ELF0068
70		IF (IRT .GT. 10) GO TO 80	ELF0069
71		IR = IRT	ELF0070
72		IRMF(IR) = 1	ELF0071
73		60 TO 250	ELF0072
74		99 WRITE (6,90) IRT, IRMAX	ELF0073
75		90 FORMAT (1H1, 'ATTEMPTED TO LOAD THE '.I2, 'TH BODY. MAXIMUM ',	ELF0074
76		1 'ALLOWABLE NUMBER OF BODIES IS '.I3)	ELF0075
77		60 TO 640	ELF0076
78	C		ELF0077
79	C	NEW GEOMETRY. OLD SEQUENCE	ELF0078
80		100 IF (IRT .LE. 0) GO TO 120	ELF0079
81		DO 110 IR = 1,IRT	ELF0080
82		IF (IDB(IR) .EQ. ICOLD) GO TO 240	ELF0081
83		110 CONTINUE	ELF0082
84		120 WRITE (6,130) IROD, ICOLD	ELF0083
85		130 FORMAT (1H1, 'OPTION IROD = '.I3, ', GEOMETRY WITH IC = ', I3,	ELF0084
86		1 'NOT PREVIOUSLY LOADED.')	ELF0085
87		60 TO 640	ELF0086
88	C		ELF0087
89	C	OLD GEOMETRY. OLD (OR CONTINUED) SEQUENCE	ELF0088
90		100 IF (IRT .LE. 0) GO TO 120	ELF0089
91		DO 150 IR = 1,IRT	ELF0090
92		IF (IDB(IR) .EQ. ICOLD) GO TO 160	ELF0091
93		150 CONTINUE	ELF0092
94		60 TO 120	ELF0093
95	C		ELF0094
96	C	DESIRE OLD GEOMETRY. IF SAVED, RETRIEVE	ELF0095
97		160 IF (ISAV3(IR) .LT. 0) GO TO 190	ELF0096
98		IRS = ISAV3(IR)	ELF0097
99		REWIND ISAVU	ELF0098
100		DO 170 I = 1,IRS	ELF0099
101		LX = ISAV2(IRS)	ELF0100
102		CALL GETT(ISAVU, J, LX, X, LX, Y)	ELF0101
103		170 CONTINUE	ELF0102
104		IF (IROD .EQ. 5) GO TO 180	ELF0103
105		IRMF(IR) = 2	ELF0104
106		60 TO 360	ELF0105
107		100 IRT = IRT + 1	ELF0106
108		IF (IRT .GT. 10) GO TO 80	ELF0107
109		IR = IRT	ELF0108
110		IRMF(IR) = 1	ELF0109
111		ISAV1(IR) = ICOLD	ELF0110
112		ISAV3(IR) = -1	ELF0111
113		GO TO 360	ELF0112

114	C		ELF0113
115		190 WRITE (6,200) IBOD, IDOLD	ELF0114
116		200 FORMAT (1H1,'OPTION IROD = ',I3,'. GEOMETRY WITH IC = ',I3,	ELF0115
117		1 'NOT PREVIOUSLY SAVED.')	ELF0116
118		GO TO 640	ELF0117
119	C		ELF0118
120		DELETE AN EXISTING BODY	ELF0119
121		210 IF (IBJ .LE. 0) GO TO 120	ELF0120
122		DO 220 IB = 1,IBJ	ELF0121
123		IF (IDB(IB) .EQ. ICOLD) GO TO 230	ELF0122
124		220 CONTINUE	ELF0123
125		230 IBMF(IB) = -1	ELF0124
126		GO TO 600	ELF0125
127	C		ELF0126
128		NEW GEOMETRY TO BE READ IN.	ELF0127
129		240 IBMF(IB) = 2	ELF0128
130	C	POSITION SAVE UNIT IF NEW GEOMETRY TO BE SAVED.	ELF0129
131		250 IF (ISV .EQ. 0) GO TO 270	ELF0130
132		IF (IBS .EQ. IBST) GO TO 270	ELF0131
133		DO 260 IT = IBS,IBST	ELF0132
134		LX = ISAV2(IT)	ELF0133
135		260 CALL GETI(ISAVU, 3, LX, X, LX, Y)	ELF0134
136		270 CONTINUE	ELF0135
137	C		ELF0136
138		CHECK IF ELLIPSE TO BE GENERATED	ELF0137
139		IF (ITR .GT. 1) GO TO 320	ELF0138
140	C		ELF0139
141		DATA ON UNIT 5, X-COORDS FIRST	ELF0140
142		L = 0	ELF0141
143		ITYP = 3	ELF0142
144		280 READ (5,290) (X(I+1), I=1,6), INO, ISTAT, ITYPE	ELF0143
145		290 FORMAT (6F10.0, 4X11, 2X11, 3X11)	ELF0144
146		IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE)	ELF0145
147		IF (INO .LE. 0 .OR. INO .GT. 6) INO = 6	ELF0146
148		L = L + INO	ELF0147
149		IF (ISTAT .EQ. 0) GO TO 280	ELF0148
150		LX = L	ELF0149
151	C		ELF0150
152		NOW READ IN Y-COORDS	ELF0151
153		L = 0	ELF0152
154		ITYP = 4	ELF0153
155		300 READ (5,290) (Y(I+1), I=1,6), INO, ISTAT, ITYPE	ELF0154
156		IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE)	ELF0155
157		IF (INO .LE. 0 .OR. INO .GT. 6) INO = 6	ELF0156
158		L = L + INO	ELF0157
159		IF (ISTAT .EQ. 0) GO TO 300	ELF0158
160		LY = L	ELF0159
161	C		ELF0160
162		CHECK FOR INPUT CONSISTENCY	ELF0161
163		IF (LY .EQ. LX) GO TO 350	ELF0162
164		WRITE (6,310) LY, LX	ELF0163
165		310 FORMAT (1H1,'THE NUMBER OF Y-COORDINATES (',I3,') READ DOES ',	ELF0164
166		1 'NOT EQUAL THE NUMBER OF X-COORDINATES READ (',I3,')')	ELF0165
167		GO TO 640	ELF0166
168	C		ELF0167
169		ELLIPSE TO BE GENERATED. READ IN DEFINITION CARD.	ELF0168
170		320 ITYP = 5	ELF0169

171	READ (5,330) LX, ELPSTM, ITYPE	ELF0170
172	330 FORMAT (2X13, 5X16, 5, 5X11)	ELF0171
173	IF (ITYPE .NE. ITP) CALL TYPE(ITYP, ITYPE)	ELF0172
174	ITR = ITR - 2	ELF0173
175	IF (ITR .NE. 1) ITR = 0	ELF0174
176	C	ELF0175
177	DANGLE = 6.2831853072/(LX - 1)	ELF0176
178	ANGLE = DANGLE	ELF0177
179	DO 340 I = 1, LX	ELF0178
180	ANGLE = ANGLE - DANGLE	ELF0179
181	X(I) = COS(ANGLE)	ELF0180
182	340 Y(I) = SIN(ANGLE) + ELPSTM	ELF0181
183	C	ELF0182
184	C	ELF0183
185	C SAVE THE BASIC GEOMETRY IF REQUESTED	ELF0184
186	350 ISAV(I) = -1	ELF0185
187	IF (ISV .EQ. 0) GO TO 360	ELF0186
188	C	ELF0187
189	C	ELF0188
190	IRST = IRST + 1	ELF0189
191	IRS = IRST	ELF0190
192	ISAV(IRS) = ID	ELF0191
193	ISAV2(IRS) = LX	ELF0192
194	ISAV3(IRS) = IRS	ELF0193
195	CALL SAVE(ISAVU, 3, 1, LX, X, LX, Y)	ELF0194
196	C	ELF0195
197	C	ELF0196
198	360 CONTINUE	ELF0197
199	C	ELF0198
200	C WRITE OUT BASIC GEOMETRY DATA	ELF0199
201	IP = 1	ELF0200
202	CALL PRINT6 (IP, LX, ID, ITITLE)	ELF0201
203	C	ELF0202
204	C	ELF0203
205	C TRANSFORM COORDINATES IF REQUESTED	ELF0204
206	IF (ITR .EQ. 1) GO TO 370	ELF0205
207	IF (INORM .EQ. 0) GO TO 410	ELF0206
208	XMULT = 0.0	ELF0207
209	YMULT = 0.0	ELF0208
210	XTO = 0.0	ELF0209
211	YTO = 0.0	ELF0210
212	THETA = 0.0	ELF0211
213	DX = 0.0	ELF0212
214	DY = 0.0	ELF0213
215	370 CONTINUE	ELF0214
216	C	ELF0215
217	IF (ABS(XMULT) .LT. EPS) XMULT = 1.0	ELF0216
218	IF (ABS(YMULT) .LT. EPS) YMULT = 1.0	ELF0217
219	XSF = XMULT	ELF0218
220	YSF = YMULT	ELF0219
221	IF (INORM .EQ. 0) GO TO 390	ELF0220
222	C	ELF0221
223	IF (ABS(CHORD) .GT. EPS) GO TO 380	ELF0222
224	CHORD = RMAX (LX, X, Y, TRAX)	ELF0223
225	380 XSF = XSF/CHORD	ELF0224
226	YSF = YSF/CHORD	ELF0225
227	C	ELF0226


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228      390 COST = COS(THETA*PI)
229      SINT = SIN(THETA*PI)
230      DO 400 I = 1,LX
231      XTOD = X(I) - XTO
232      YTOD = Y(I) - YTO
233      X(I) = XTOD*COST - YTOD*SINT + DXI*XSF
234      Y(I) = YTOD*COST + XTOD*SINT + DYI*YSF
235      400 CONTINUE
236      C
237      C
238      C FORM ELEMENT DATA FOR THIS BODY.
239      C DEFINE STORAGE LOCATIONS AND CROSS CHECK
240      410 IST = 0
241      IF (IB .GT. 1) IST = IFL(1B-1)
242      LXI = LX-1
243      C FIRST CHECK AGAINST EXCEEDING MAXIMUM STORAGE
244      IF ((IST + LXI) .LE. MAXEL) GO TO 430
245      WRITE (6,420) LXI, ID, MAXEL
246      420 FORMAT (1H1,'THE NUMBER OF ELEMENTS (',I4,') FOR BODY ID = ',I2,
247      1      ' WILL EXCEED ALLOWABLE STORAGE (',I4,') WHEN ADDED ',
248      2      ' TO THE DATA SET.')
249      GO TO 640
250      C
251      C NOW, IF USING OLD STORAGE SEQUENCE, CHECK THAT NEW GEOMETRY
252      C DOES NOT RUN INTO THE NEXT BODY.
253      430 IF (1B .EQ. 1BT) GO TO 450
254      IF ((IST + LXI) .LT. INL(1B+1)) GO TO 460
255      WRITE (6,440) LXI, ID, NL(1B), ICOLD
256      440 FORMAT (1H1,'THE NUMBER OF ELEMENTS (',I4,') FOR THE NEW BODY ',
257      1      'ID = ',I2,' EXCEEDS THE NUMBER (',I4,') FOR THE BODY ',
258      2      'IT IS REPLACING, ICOLD = ',I2)
259      GO TO 640
260      C
261      C
262      450 INL(1B) = IST + 1
263      IFL(1B) = IST + LXI
264      460 NL(1B) = LXI
265      LIFT(1B) = IIFT
266      IDB(1B) = ID
267      ITRB(1B) = ITR
268      INMB(1B) = INORM
269      IRD(1B) = IRD
270      CHORDB(1B) = CHORD
271      IPRB(1B) = IPARA
272      IFST(1B) = IFIRST
273      ISEC(1B) = ISECND
274      IPVRI(1B) = IPVOR
275      C
276      DO 470 I = 1,7
277      470 BTITLE(1B,I) = TTITLE(I)
278      C
279      C CALCULATE ELEMENT DATA AND PRINT RESULTS.
280      C
281      DO 480 I = 2,LX
282      IST = IST + 1
283      X(IST) = 0.5*(X(I) + X(I-1))
284      Y(IST) = 0.5*(Y(I) + Y(I-1))

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ELF0227
ELF0228
ELF0229
ELF0230
ELF0231
ELF0232
ELF0233
ELF0234
ELF0235
ELF0236
ELF0237
ELF0238
ELF0239
ELF0240
ELF0241
ELF0242
ELF0243
ELF0244
ELF0245
ELF0246
ELF0247
ELF0248
ELF0249
ELF0250
ELF0251
ELF0252
ELF0253
ELF0254
ELF0255
ELF0256
ELF0257
ELF0258
ELF0259
ELF0260
ELF0261
ELF0262
ELF0263
ELF0264
ELF0265
ELF0266
ELF0267
ELF0268
ELF0269
ELF0270
ELF0271
ELF0272
ELF0273
ELF0274
ELF0275
ELF0276
ELF0277
ELF0278
ELF0279
ELF0280
ELF0281
ELF0282
ELF0283

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285      XD = X(I) - X(I-1)
286      YD = Y(I) - Y(I-1)
287      DL(IIST) = SQRT(XD**2 + YD**2)
288      DS(IIST) = DL(IIST)
289      SA(IIST) = YD/DL(IIST)
290      CA(IIST) = XD/DL(IIST)
291      CURV(IIST) = 0.0
292      *NO CONTINUE
293      C
294      C
295      C PARABOLIC ELEMENT
296      IST = INCLIR
297      LEND = IJ - 2
298      I = 1
299      *NO I = I + 1
300      IST = IST + 1
301      I1 = I - 1
302      I2 = I
303      I3 = I + 1
304      I4 = I + 2
305      L1 = I1
306      L2 = I2
307      L3 = I3
308      C
309      500 DLY = (X(L3)-X(L1))*Y(L2)-Y(L1) - (X(L2)-X(L1))*Y(L3)-Y(L1)
310      DLX = (X(L2)-X(L1))*X(L3)-X(L1) + (Y(L2)-Y(L1))*Y(L3)-Y(L1)
311      DRY = (X(I2)-X(I4))*Y(I3)-Y(I4) - (X(I3)-X(I4))*Y(I2)-Y(I4)
312      DRX = (X(I2)-X(I4))*X(I3)-X(I4) + (Y(I3)-Y(I4))*Y(I2)-Y(I4)
313      DYO = 0.0
314      DLL = SQRT(((X(L2)-X(L1))**2 + (Y(L2)-Y(L1))**2)
315      1 + ((X(L3)-X(L1))**2 + (Y(L3)-Y(L1))**2))
316      DRL = SQRT(((X(I2)-X(I4))**2 + (Y(I2)-Y(I4))**2)
317      1 + ((X(I3)-X(I4))**2 + (Y(I3)-Y(I4))**2))
318      ETAL = 0.5*DL(IIST)*DLY/DLL + DLX
319      ETAR = -0.5*DL(IIST)*DRY/DRL + DRX
320      C
321      C DEFINE AVERAGE OFFSET DYO USING RMS.
322      ETA = ETAL*ETAR
323      IF (ETA .LT. 0.0) 60 TO 510
324      ETA = SQRT(ETA)
325      DYO = SIGN(ETA, ETAL)
326      CURV(IIST) = -4.0*DYO/DL(IIST)**2
327      C
328      C
329      510 CONTINUE
330      XP(IIST) = X(IIST) - DYO*SA(IIST)
331      YP(IIST) = Y(IIST) + DYO*CA(IIST)
332      DS(IIST) = DL(IIST)*(1.0 + (CURV(IIST)*DL(IIST))**2/6.0)
333      C
334      IF (I - LEND)*90,520,530
335      C
336      C SPECIAL HANDLING FOR FIRST AND LAST ELEMENTS
337      C
338      C FIRST ELEMENT
339      520 IST = INCLIR
340      I2 = 1
341      I3 = 2

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ELF0284
ELF0285
ELF0286
ELF0287
ELF0288
ELF0289
ELF0290
ELF0291
ELF0292
ELF0293
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ELF0295
ELF0296
ELF0297
ELF0298
ELF0299
ELF0300
ELF0301
ELF0302
ELF0303
ELF0304
ELF0305
ELF0306
ELF0307
ELF0308
ELF0309
ELF0310
ELF0311
ELF0312
ELF0313
ELF0314
ELF0315
ELF0316
ELF0317
ELF0318
ELF0319
ELF0320
ELF0321
ELF0322
ELF0323
ELF0324
ELF0325
ELF0326
ELF0327
ELF0328
ELF0329
ELF0330
ELF0331
ELF0332
ELF0333
ELF0334
ELF0335
ELF0336
ELF0337
ELF0338
ELF0339
ELF0340

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342	I4 = 3	ELF0341
343	L1 = I2	ELF0342
344	L2 = I3	ELF0343
345	L3 = I4	ELF0344
346	I = LFND + 1	ELF0345
347	GO TO 500	ELF0346
348	C	ELF0347
349	C LAST ELEMENT	ELF0348
350	530 IF (I .GT. LEND+1) GO TO 540	ELF0349
351	IST = IFL(1B)	ELF0350
352	L1 = LX - 2	ELF0351
353	L2 = LX - 1	ELF0352
354	L3 = LX	ELF0353
355	I2 = L1	ELF0354
356	I3 = L2	ELF0355
357	I4 = L3	ELF0356
358	I = I + 1	ELF0357
359	GO TO 500	ELF0358
360	C	ELF0359
361	C SAVE PARABOLIC SURFACE DATA ON UNIT IF17	ELF0360
362	540 CONTINUE	ELF0361
363	C	ELF0362
364	I1 = INL(1B)	ELF0363
365	I2 = NL(1B)	ELF0364
366	C XP AND YP COORDS	ELF0365
367	CALL SAVE(1F17, 3, I2, I2, XP(I1), I2, YP(I1))	ELF0366
368	I2 = IFL(1B)	ELF0374
369	C CHECK IF PARABOLIC ELEMENT DESIRED	ELF0375
370	IF (1PARA .NE. 1) GO TO 560	ELF0376
371	DO 550 I = I1, I2	ELF0377
372	XO(I) = XP(I)	ELF0378
373	550 YO(I) = YP(I)	ELF0379
374	C	ELF0380
375	C PRINT ELEMENT DATA	ELF0381
376	560 I = 1	ELF0382
377	WRITE (6,610) IDB(1B), YTITLE	ELF0383
378	WRITE (6,620) I, X(I), Y(I)	ELF0384
379	LCNT = 1	ELF0385
380	SUMDS(1B) = 0.0	ELF039C
381	ALPH1 = 0.0	ELF0391
382	DO 580 IST = I1, I2	ELF0392
383	I = I + 1	ELF0393
384	IF (LCNT .LT. 49) GO TO 570	ELF0394
385	LCNT = 0	ELF0395
386	WRITE (6,610) IDB(1B), YTITLE	ELF0396
387	570 CONTINUE	ELF0397
388	SUMDS(1B) = SUMDS(1B) + DS(IST)	ELF0398
389	ALPHA = ATAN2(SA(IST), CA(IST))/CR	ELF0399
390	DADS = (ALPHA - ALPH1)/DS(IST)*CR	ELF040C
391	ALPH1 = ALPHA	ELF0401
392	CURV2 = 2.0*CURV(IST)	ELF0402
393	WRITE (6,630) XO(IST), YO(IST), CL(IST), DS(IST),	ELF0403
394	I, SA(IST), CA(IST), CURV2	ELF0404
395	WRITE (6,620) I, X(I), Y(I)	
396	LCNT = LCNT + 2	
397	580 CONTINUE	
398	C	

399	C		ELF0405
400		WRITE (6,500) SUMDS(IH)	ELF0406
401		590 FORMAT(IH0, T13, 'SUMDS = ', F12.6)	ELF0407
402	C		ELF0408
403	C	DETERMINE GEOMETRY COMBINATION COEFFICIENTS	ELF0409
404		CALL GEOMCE (IHL(IH), IFL(IH), OL, TFS(IH), TSEC(IH), DL)	ELF0410
405	C		ELF0411
406	C		ELF0412
407	C	CHECK IF MORE BODIES TO BE INPUT	ELF0413
408		600 IF (LAST.NE., 1) GO TO 10	ELF0414
409	C		ELF0415
410	C		ELF0416
411	C	WRITE OUT BODY SUMMARY	ELF0417
412		CALL PRNTEL	ELF0418
413	C		ELF0419
414		GO TO 660	ELF0420
415	C		ELF0421
416	C		ELF0422
417	C		ELF0423
418		610 FORMAT (IH1, 2DX, 'ELEMENT COORDINATE DATA FOR BODY TO : ', I2,	ELF0424
419	1	' , ' , 744/T10, 'I', T22, 'X(I)', T39, 'Y(I)', T56, 'DL',	ELF0425
420	2	T73, 'DS', T96, 'SIN(1/2) ', T105, 'COS(1/2) ',	ELF0426
421	3	T121, 'CURVATURE'//)	ELF0427
422	C		ELF0428
423		620 FORMAT (IH , 6X13, 2(5XF12.6))	ELF0429
424	C		ELF0430
425		630 FORMAT (IH , 9X, 7(5XF12.6))	ELF0431
426	C		ELF0432
427		640 WRITE (6,650)	ELF0433
428		650 FORMAT (IH0, 'BECAUSE OF THE ABOVE ERROR, THIS RUN IS TERMINATED')	ELF0434
429		STOP	ELF0435
430	C		ELF0436
431	C		ELF0437
432		660 CONTINUE	ELF0438
433	C		ELF0439
434		RETURN	ELF0440
435		END	ELF0441

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1      SUBROUTINE FILES                                FILE001
2      COMMON /FILEID/ IFILE1, IFILE2, IFILE3, IFILE4, IFILE5, FILE002
3      1      IFILE6, IFILE7, IFILE8, IFILE9, IFILE10, FILE003
4      2      IFILE11, IFILE12, IFILE13, IFILE14, IFILE15, FILE004
5      3      IFILE16, IFILE17, IFILE18, IFILE19, IFILE20, FILE005
6      C      FILE006
7      C IFILE1, IFILE2, AND IFILE3 ARE TEMPORARY SCRATCH UNITS USED IN QUASI FILE007
8      IFILE1 = 18 FILE008
9      IFILE2 = 2 FILE009
10     IFILE3 = 3 FILE010
11     C IFILE4 IS INPUT FILE FOR RIGHT SIDE MATRIX IN QUASI FILE011
12     IFILE4 = 4 FILE012
13     C FILE013
14     C IFILE5, IFILE6, AND IFILE7 ARE STANDARD SYSTEM I/O FILE014
15     IFILE5 = 5 FILE015
16     IFILE6 = 6 FILE016
17     IFILE7 = 7 FILE017
18     C FILE018
19     C IFILE8 IS INPUT GEOMETRY SAVE UNIT FILE019
20     IFILE8 = 8 FILE020
21     C FILE021
22     C IFILE9 IS MATRIX A(I,J) FILE022
23     C IFILE10 IS MATRIX B(I,J) FILE023
24     IFILE9 = 9 FILE024
25     IFILE10 = 10 FILE025
26     C IFILE11 IS INDUCED NORMAL VELOCITY N(I). FILE026
27     C IFILE12 IS INDUCED TANGENTIAL VELOCITY T(I) FILE027
28     IFILE11 = 11 FILE028
29     IFILE12 = 12 FILE029
30     C FILE030
31     C IFILE13 IS SPECIAL B ROWS FOR LIFTING BODIES, BLU(I) FILE031
32     IFILE13 = 13 FILE032
33     C FILE033
34     C IFILE14 CONTAINS SIGMA SOLUTIONS ON OUTPUT FROM QUASI FILE034
35     IFILE14 = 14 FILE035
36     C FILE036
37     C IFILE15 CONTAINS BOTH UPPER AND LOWER TRIANGULAR MATRICES FILE037
38     C ON OUTPUT FROM QUASI FILE038
39     IFILE15 = 15 FILE039
40     C FILE040
41     C IFILE16 IS USED FOR OFFBODY CALCULATIONS, /ELDATA/ AND /GCOEFS/ FILE041
42     C ARE STORED (SFE SUBROUTINE ELFORM). FILE042
43     IFILE16 = 16 FILE043
44     C FILE044
45     C IFILE17 IS USED TO SAVE SURFACE COORDS (X, Y, AND DS) AT WHICH FILE045
46     C THE FLOW VELOCITY IS ASSUMED TO ACT. FILE046
47     IFILE17 = 17 FILE047
48     C FILE048
49     C IFILES 18, 19, AND 20 HAVE NOT BEEN ASSIGNED. FILE049
50     C FILE050
51     RETURN FILE051
52     END FILE052

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1      SUBROUTINE FLOWS (N, M, IPUN)
2      C
3      C INDIVIDUAL FLOWS (IND = 1) AND COMBINED FLOWS CALCULATED.
4      C
5      C
6      DIMENSION A(500), B(500), VN(500), VT(500), VNC(500), VTC(500),
7      SIG(500)
8      2
9      DIMENSION EL6C(1350), EL6D(2500), CPJ(500), XP(500), YP(500),
10     V1(500), V2(500), V3(500), V4(500), V5(500)
11     C
12     COMMON /BFLAG/ IDB(10), IML(10), IFL(10), WL(10), LIFT(10),
13     IMHF(10), ISAV(10), ISAV2(10), ISAV3(10),
14     RTITLE(10), 7), IB7, IB37, IB707, MELT07,
15     ITRR(10), INMB(10), CHORD(10), IDI(10), LIFT07,
16     IPRB(10), IFST(10), ISEC(10), FTITLE(10), IPVR(10)
17     C
18     COMMON /SIGMAS/ CSIG(500), CH(12)
19     COMMON /FILEID/ IFK1, IFD2, IFQ3, IFQ4, IFQ5, IFQ6, IFD7, IFQ8,
20     IFQ9, IF10, IF11, IF12, IF13, IF14, IF15
21     2
22     COMMON /COMMOD/ CCL, INCLT, CLT, ALPHA, SUMDS(10), FLU(10, 12), IND
23     1
24     COMMON /ROTAT/ WROT, ROTRAD(10)
25     COMMON /GCF/ CO(500), CF(500), CG(500), CI(500), WF(500)
26     COMMON /ELD/ X, Y, DS, SA, CA, CURV(500), DL(500)
27     COMMON /COM/IFLL
28     EQUIVALENCE (EL6C(1), X(1)), (EL6D(1), CO(1)), (CPJ(1), SIG(1))
29     EQUIVALENCE (A(1), XP(1)), (B(1), YP(1))
30     C
31     C REMIND UNITS FOR NORMAL AND TANGENTIAL ONSET VELOCITIES
32     REMIND IF11
33     REMIND IF12
34     C
35     C REMIND SIGMA UNIT
36     REMIND IF14
37     C
38     C ZERO OUT VNC AND VTC ARRAYS
39     DO 10 I = 1, N
40     VNC(I) = 0.0
41     VTC(I) = 0.0
42     C
43     C
44     C THE PROCEDURE IS TO FIRST CALCULATE THE INDIVIDUAL FLOWS
45     C AND THEN THE COMBINED FLOW. THE ONSET VELOCITIES ARE
46     C COMBINED DURING THE INDIVIDUAL FLOWS CYCLE.
47     C
48     C READ IN SURFACE COORDS AND GEOMETRY COEFFICIENTS FROM CART 16
49     REMIND IF16
50     READ(16) EL6C, EL6D
51     C
52     C
53     C
54     C BEGIN INDIVIDUAL FLOWS CYCLE
55     VNA = 0.0
56     DO 90 N = 1, M

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FLOW001
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FLOW054

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57      C
58      C READ IN NORMAL AND TANGENTIAL ONSET VELOCITIES.
59      CALL GETT(1F11, 1, N, VN, 1, VNA)
60      CALL GETT(1F12, 1, N, VT, 1, VNA)
61      C
62      C CALCULATE COMBINED NORMAL & TANGENTIAL VELOCITIES
63      DO 30 I = 1, N
64          VNC(I) = VNC(I) - VN(I)*CK(I)
65          VTC(I) = VTC(I) + VT(I)*CK(I)
66      C
67      C CHECK IF INDIVIDUAL FLOW DESIRED
68      IF (IND.NE.1) GO TO 90
69      C
70      C REMIND A,B ARRAY UNITS
71      REMIND IF10
72      C
73      C READ IN ROW OF SIGMAS
74      CALL GETT(1F14, 1, N, SIG, 1, VNA)
75      C
76      DO 50 J = 1, N
77      C
78      C READ IN ROW OF A,B ARRAYS
79      CALL GETT(1F10, 1, N, A, 1, VNA)
80      CALL GETT(1F10, 1, N, B, 1, VNA)
81      C
82      VN(I) = -VN(I)
83      VNA = 0.0
84      VTB = 0.0
85      DO 40 J = 1, N
86          VNA = VNA + A(J)*SIG(J)
87          VTB = VTB + B(J)*SIG(J)
88      C
89      VN(I) = VN(I) + VNA
90      VT(I) = VT(I) + VTB
91      C
92      C
93      C PRINT OUT INDIVIDUAL FLOW
94      WRITE (6,60) K
95      60 FORMAT(1H1, 'INDIVIDUAL FLOW NUMBER', I3, '// PT.NO.',
96      1      I19, 'VN', I39, 'VT', I5R, 'SIGMA')
97      DO 70 I = 1, N
98          IF(K.EQ.1) V1(I)=VT(I)
99          IF(K.EQ.2) V2(I)=VT(I)
100         IF(K.EQ.3) V3(I)=VT(I)
101         IF(K.EQ.4) V4(I)=VT(I)
102         IF(K.EQ.5) V5(I)=VT(I)
103         70 WRITE(6,80) I, VN(I), VT(I), SIG(I)
104         80 FORMAT(1H, I4, 3(1X F12.6))
105      C
106      C
107      90 CONTINUE
108      C
109      C INDIVIDUAL FLOWS COMPLETE. NOW DO COMBINED FLOW
110      C
111      IF(IFLLL.EQ.01) GO TO 161
112      REMIND IF10
113      DO 110 I = 1, N

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FLOW105

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114 CALL GETT(1F10, 1, N, A, 1, VNA)
115 CALL GETT(1F10, 1, N, E, 1, VNA)
116 C
117 VNA = 0.0
118 VTP = 0.0
119 DO 100 J = 1, N
120 VNA = VNA + A(J)*CSIG(J)
121 100 VTB = VTB + E(J)*CSIG(J)
122 C
123 VNC(1) = VNC(1) + VNA
124 VTC(1) = VTC(1) + VTB
125 110 CONTINUE
126 C
127 C
128 C PRINT THE OUTPUT DATA (PER BODY).
129 ALPH = ALPHA*0.017453293
130 COSA = COS(ALPH)
131 SINX = SIN(ALPH)
132 CMT = 0.0
133 CHTP = 0.0
134 XM = 0.0
135 YM = 0.0
136 NI = 0
137 NF = 0
138 REMIND IF17
139 DO 160 IB = 1, 1801
140 IF (IB*NF(1B) .LT. 1) GO TO 160
141 NR = NI(1B)
142 NI = NF
143 NF = NF + NR
144 J2 = NI
145 S = 0.0
146 S1 = 0.0
147 I = 0
148 CX = 0.0
149 CN = 0.0
150 CPL = 0.0
151 NI = NI + 1
152 CALL GETT(1F17, 1, NB, XPI(1), AP, YPI(1))
153 IF (INROT .NE. 0) ROT2 = ROTRAD(1)*2
154 120 J1 = J2 + 1
155 J2 = J1 + 40
156 IF (J2 .GT. NF) J2 = NF
157 WRITE(1F06,130) FTITLE, ALPHA, ALPHA0, INTOT, CLT, CCL,
158 1 NELTOT, IOR(1B), (RTITLE(1B,11),11=1,7), NB
159 C
160 130 FORMAT(1H1, ' DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL ',
161 1 'POTENTIAL FLOW PROGRAM/NAME/COMBINED FLOW ', 15A4,
162 2 1H0, 'ALPHA =', F11.6, 130, 'ALPHA C =', F11.6,
163 3 1H0, 'NO. OF BODIES ', 12/,
164 4 1H0, 'CL =', F11.6, 130, 'CHORD =', F11.6,
165 5 1H0, 'TOTAL ELEMENTS ', 13/,
166 6 1H0, 'BODY ID = ', 12, 120, 144, 160, 'NO. OF ELEMENTS ', 13/,
167 7 1H0, 14, '1', 114, 'X', 120, 'Y', 120, 'S', 156, 'VI',
168 8 170, 'CP', 193, '1', 110, 'SIGMA', 1117, 'VN')
169 C
170 DO 140 J = J1, J2

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171      I = I + 1                                FLOW163
172      SD = DS(J)/(2.0*SUMDS(1B))              FLOW164
173      S = S1 + SD                             FLOW165
174      S1 = S + SD                             FLOW166
175      CP = 1.0 - VT(J)**2                     FLOW167
176      CPJ(J) = CP                             FLOW168
177      DCP = 0.0                               FLOW169
178      IF (NROT.EQ.0) GO TO 145                FLOW170
179      DCP = (X(J)**2 + Y(J)**2)/ROT2            FLOW171
180      DCP = DCP - 2.0*(X(J)*SINA - Y(J)*COSA)/ROTRAD(1B) FLOW172
181      145 CP = CP + DCP                       FLOW173
182      XX = XP(J)                              FLOW174
183      YY = YP(J)                              FLOW175
184      T = CP*DS(J)                            FLOW176
185      CN = CN - T*CA(J)                       FLOW177
186      CX = CX + T*SA(J)                       FLOW178
187      CML = CML + T*(CA(J)*(X(J)-XM) + SA(J)*(Y(J)-YM)) FLOW179
188      IF (IPUN.EQ.7) WRITE(7,155) XX, YY, CP, T FLOW180
189      140 WRITE(1F06,150) I,XX,YY, S, VT(J), CP, J, CSIG(J), VNI(J) FLOW181
190      150 FORMAT(1H,13,5(3XF11.6),16X11,2(3XF11.6)) FLOW182
191      C                                         FLOW183
192      155 FORMAT(3F10.5,46X14)                FLOW184
193      C                                         FLOW185
194      IF (J2.NE.NF) GO TO 120                FLOW186
195      CLI = CN*COSA - CX*SINA                  FLOW187
196      CDI = CN*SINA + CX*COSA                  FLOW188
197      WRITE(6,200) CN, CX, CLI, CDI, CML      FLOW189
198      200 FORMAT(1H0,17HINTEGRATED VALUES/   FLOW190
199      1      1H0,5HCY = , F10.5, 5X5HCX = , F10.5/ FLOW191
200      2      1H0,5HCL = , F10.5, 5X5HCD = , F10.5, 5X5HCP = , F10.5) FLOW192
201      CMT = CMT + CML                         FLOW193
202      C                                         FLOW194
203      C CALCULATION OF FORCES & MOMENTS USING TRAPEZOIDAL INTEGRATION FLOW195
204      C COMPLETE.                             FLOW196
205      C NOW CALCULATE FORCES & MOMENTS USING PARABOLIC INTEGRATION. FLOW197
206      C SPECIAL HANDLING FOR FIRST AND LAST ELEMENTS FLOW198
207      C                                         FLOW199
208      C NOTE. DEFINITIONS FOR CD & CF INCLUDE MULTIPLICATION BY DL. FLOW200
209      C FOR CG & CI INCLUDE MULTIPLICATION BY DL**2. FLOW201
210      C                                         FLOW202
211      CX = 0.0                                FLOW203
212      CY = 0.0                                FLOW204
213      CML = 0.0                                FLOW205
214      NF1 = NF - 1                            FLOW206
215      C FIRST ELEMENT                        FLOW207
216      J2 = N1 + 1                            FLOW208
217      J1 = J2 + 1                            FLOW209
218      J3 = J2 + 2                            FLOW210
219      C                                         FLOW211
220      151 CONTINUE                          FLOW212
221      CP1 = CD(J2)*CPJ(J1) - (CG(J2)+CG(J2))*CPJ(J2) + CF(J2)*CPJ(J3) FLOW213
222      CP2 = CG(J2)*CPJ(J1) - (CG(J2)+CI(J2))*CPJ(J2) + CI(J2)*CPJ(J3) FLOW214
223      DLSQ = DL(J2)**2                       FLOW215
224      CXX = CURV(J2)*CP1*DLSQ/6.0            FLOW216
225      CYY = -DL(J2)*(CPJ(J2) + (CP2 /12.0)) FLOW217
226      CMO = -CP1*DLSQ*(1.0 + 3.0*(CURV(J2)*DL(J2)**2)/120.0) FLOW218

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228		DCX = CXX*CA(J2) - CVY*SA(J2)	FLOW22E
229		CX = CX + DCX	FLOW22I
230		DCY = CXX*SA(J2) + CVY*CA(J2)	FLOW22J
231		CY = CY + DCY	FLOW22K
232		CML = CML - DCX*(YM-Y(J2)) - DCY*(X(J2)-XM) - CMO	FLOW22L
233	C		FLOW22M
234	C		FLOW22N
235		IF (J2 - NF) 152,153,154	FLOW22O
236	C		FLOW22P
237		152 J1 = J2	FLOW22Q
238		J2 = J2 + 1	FLOW22R
239		J3 = J2 + 1	FLOW22S
240		GO TO 151	FLOW22T
241	C		FLOW22U
242	C	LAST ELEMENT	FLOW22V
243		153 J2 = NF	FLOW22W
244		J1 = NF - 1	FLOW22X
245		J3 = NF - 2	FLOW22Y
246		GO TO 151	FLOW22Z
247	C		FLOW23A
248	C		FLOW23B
249		154 CL1 = CY*CSA - CX*SINA	FLOW23C
250		CD1 = CY*SINA + CX*CSA	FLOW23D
251		CH1P = CH1P + CL1	FLOW23E
252	C		FLOW23F
253		WRITE(6,205)	FLOW23G
254		205 FORMAT(10D,25N) PARABOLIC INTEGRATION	FLOW23H
255	C		FLOW23I
256		WRITE(6,200) CV, CX, CLY, CD1, CML	FLOW23J
257	C		FLOW23K
258		160 CONTINUE	FLOW23L
259		161 REWIND 7	FLOW23M
260		WRITE(7,1500) (X(J),J=1,N)	
261		WRITE(7,1500) (Y(J),J=1,N)	
262		WRITE(7,1500) (V1(J),J=1,N)	
263		WRITE(7,1500) (V2(J),J=1,N)	
264		WRITE(7,1500) (V3(J),J=1,N)	
265		WRITE(7,1500) (V4(J),J=1,N)	
266		WRITE(7,1500) (V5(J),J=1,N)	
267		IF (IFLLL.NE.0) WRITE(6,210) CMT	
268		210 FORMAT(10D,11M) TOTAL CM = , F10.5)	FLOW251
269		1500 FORMAT(10P6F13.8)	FLOW252
270		IF (IFLLL.NE.0) WRITE(6,215) CMT	
271		215 FORMAT(10D,11M) TOTAL CM = , F10.5, SX11H4PARABOLIC)	FLOW253
272	C		FLOW254
273		RETURN	FLOW255
274		END	FLOW256
			FLOW257

```

1      SUBROUTINE GEOMCF (INL, IFL, DS, IFST, ISEC, DL)
2      C
3      C CALCULATES REQUIRED GEOMETRY COMBINATION COEFFICIENTS
4      C ASSOCIATED WITH PARABOLIC DISTRIBUTIONS.
5      C
6      C IFST = 1 OR 3, NEED CD, CE, AND CF
7      C
8      C ISEC = 1 OR 3, NEED CG, CH, AND CI
9      C
10     C (HOWEVER, ALL ARE CALCULATED FOR NOW)
11     C
12     C
13     C SPECIAL HANDLING FOR FIRST AND LAST ELEMENTS.
14     C (FOR NOW BODY SLOPE ASSUMED CONTINUOUS)
15     C
16     C
17     C DIMENSION DS(1), DL(1)
18     C
19     C COMMON/GEOMCF/ CD(500), CF(500), CG(500), CI(500)
20     C
21     C
22     I1 = IFL
23     I2 = INL
24     C
25     10 I3 = I2 + 1
26     20 C1 = DS(I1) + DS(I2)
27     C2 = (DS(I2) + DS(I2))*DL(I2)/DS(I2)
28     C3 = DS(I2) + DS(I3)
29     C4 = C2/(C1 + C3)
30     C5 = C4/C1
31     C6 = C4/C3
32     C
33     CD(I2) = -C5*C3
34     CF(I2) = C6*C1
35     CG(I2) = C5*C2
36     CI(I2) = C6*C2
37     C
38     I1 = I2
39     I2 = I1 + 1
40     IF (I2 - IFL)10,30,40
41     C
42     30 I3 = INL
43     GO TO 20
44     C
45     40 CONTINUE
46     C
47     C REDO FIRST AND LAST POINTS FOR ONE-SIDED FITS
48     I1 = INL
49     I2 = I1 + 1
50     I3 = I1 + 2
51     C1 = DS(I1) + DS(I2)
52     C3 = DS(I3) + DS(I2)
53     C2 = 2*DS(I1)*DL(I1)/DS(I1)
54     CD(I1) = C2*(1./C1 + 1./C3)
55     CF(I1) = C2*(1./C1 + 1./C3)
56     C2 = (C2/(2.*DS(I2)*DL(I2)/DS(I2)))*2

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57		$CG(I1) = -(C6(I2) + C1(I2)) * C2$	GEOM057
58		$CI(I1) = C1(I2) * C2$	GEOM058
59	C		GEOM059
60		$I3 = IFL$	GEOM060
61		$I2 = I3 - 1$	GEOM061
62		$I1 = I3 - 2$	GEOM062
63		$C1 = DS(I3) + DS(I2)$	GEOM063
64		$C3 = DS(I2) + DS(I1)$	GEOM064
65		$C2 = 2. * DS(I3) * DL(I3) / DS(I3)$	GEOM065
66		$CD(I3) = -(C2 * I1 / C1 + 1. / C3)$	GEOM066
67		$CF(I3) = C2 * I1 / C3 - 1. / (C1 + C3)$	GEOM067
68		$C2 = (C2 / (2. * DS(I2) * DL(I2) / DS(I2))) * C2$	GEOM068
69		$CG(I3) = -(C6(I2) + C1(I2)) * C2$	GEOM069
70		$CI(I3) = C6(I2) * C2$	GEOM070
71	C		GEOM071
72	C		GEOM072
73		RETURN	GEOM073
74		END	GEOM074

1		SUBROUTINE GETT(IU, IY, N1, A1, N2, A2)	GETT001
2	C		GETT002
3		DIMENSION A1(N1), A2(N2)	GETT003
4	C		GETT004
5		GO TO (10, 20, 30, 40), IY	GETT005
6	C		GETT006
7	C	READ A1	GETT007
8		10 READ(IU) A1	GETT008
9		RETURN	GETT009
10	C		GETT010
11	C	READ N1 AND A1	GETT011
12		20 READ(IU) N1, A1	GETT012
13		RETURN	GETT013
14	C		GETT014
15	C	READ A1 AND A2	GETT015
16		30 READ(IU) A1, A2	GETT016
17		RETURN	GETT017
18	C		GETT018
19	C	READ IOUN AND A1	GETT019
20		40 READ(IU) IOUN, A1	GETT020
21		RETURN	GETT021
22		END	GETT022

```

1      SUBROUTINE  MAFORM  ( M, NONU, NBU, ISOL, IPRINT)  MAF0001
2      C                                                    MAF0002
3      C                                                    MAF0003
4      C THIS ROUTINE FORMS AND STORES MATRICES A AND B  MAF0004
5      C ALSO CALCULATES AND STORES ALPHA AND CIRCULATORY ONSET VELOCITIES MAF0005
6      C                                                    MAF0006
7      C THIS IS FIRST ATTEMPT AND IS SIMPLE ONE-TIME PASS. MAF0007
8      C CAPABILITY TO ONLY CHANGE SELECTED ARRAY ITEMS  MAF0008
9      C WILL BE ADDED LATER. MAF0009
10     C                                                    MAF0010
11     C                                                    MAF0011
12     REAL*4  N  MAF0012
13     C                                                    MAF0013
14     C                                                    MAF0014
15     DIMENSION  A(500), B(500), X0(500), Y0(500), MAF0015
16     1          DS(500), SA(500), CA(500), BLU(500), MAF0016
17     2          M(12), T(12) MAF0017
18     C                                                    MAF0018
19     COMMON /COMBOD/CCL, INCLT, CLT, ALPHA, SUMDS(10), TLU(10,12),IND MAF0019
20     1          , ALPHAO, CNU(10), SMOSEF(10), MIO(10) MAF0020
21     C                                                    MAF0021
22     COMMON /BFLAG/ IOB(10), INL(10), IFL(10), ML(10), LIFT(10), MAF0022
23     1          IRMF(10), ISAV1(10), ISAV2(10), ISAV3(10), MAF0023
24     2          RTITLE(10, 7), IBY, IBST, IBTOT, NELTOT, MAF0024
25     3          ITRB(10), IMMB(10), CMORDB(10), IRD(10), LIFTOT MAF0025
26     4          ,IPRB(10), IFST(10), ISEC(10), FTITLE(15), IPVR(10) MAF0026
27     COMMON /FILEID/ IFD1, IFD2, IFD3, IFD4, IFD5, MAF0027
28     1          IFD6, IFD7, IFD8, IFD9, IF10, MAF0028
29     2          IF11, IF12, IF13, IF14, IF15 MAF0029
30     3          ,IF16, IF17, IF18, IF19, IF20 MAF0030
31     COMMON/ROTAT/NROT, ROTRAD(10) MAF0031
32     C                                                    MAF0032
33     COMMON/ELDATA/ X0, Y0, DS, SA, CA, CURV(500), DL(500) MAF0033
34     COMMON/BCOEF5/ CD(500), CF(500), CG(500), CI(500), WF(500) MAF0034
35     C                                                    MAF0035
36     C                                                    MAF0036
37     C BEGIN NEW CASE. II IS ROW COUNT, JJ IS COLUMN COUNT. MAF0037
38     REWIND IF09 MAF0038
39     REWIND IF10 MAF0039
40     REWIND IF11 MAF0040
41     REWIND IF12 MAF0041
42     REWIND IF13 MAF0042
43     M = LIFTOT + 2 MAF0043
44     I2 = 0 MAF0044
45     I1 = 0 MAF0045
46     LB = 0 MAF0046
47     INT = 0 MAF0047
48     NROT = 0 MAF0048
49     C INPUT VALUE OF NONU ,GT. 6 IS USED TO FLAG MAF0049
50     C A ROTATING NON-UNIFORM FLOW MAF0050
51     IF (NONU .LE. 6) GO TO 5 MAF0051
52     NROT = 1 MAF0052
53     NONU = 1 MAF0053
54     5 CONTINUE MAF0054
55     DO 210 IIR = 1,IBTOT MAF0055
56     C CHECK IF BODY IIR IS STILL IN DATA SET. MAF0056

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57      IF (IRMF(IIR) .LT. 0) GO TO 210
58      INT = INT + 1
59      C
60      C
61      IO = INL(IIR)
62      IF = IO + NL(IIR) - 1
63      NIO(IIR) = II + 1
64      RO IO I = IO, IF
65      JJ = 0
66      N = 2
67      II = II + 1
68      IF (IIPRINT .EQ. 2)
69      IWRITE (6,10) II
70      10 FORMAT(1H), T0, 'I = ', I3, '1H0, T4, 'J', T15, 'A0', T11, 'A1',
71      1 T47, 'AC', T63, 'AP', T79, 'B0', T95, 'B1', T111, 'BC', T127, 'BP')
72      C
73      DO 150 IB = 1, IRTOT
74      IF (IRMF(IB) .LT. 0) GO TO 150
75      C
76      C COUNTER FOR ELEMENT GEOMETRY
77      J = INL(IB) - 1
78      C COUNTERS FOR A,B ARRAYS
79      JJ = JJ + 1
80      JF = JJ + NL(IB)
81      C
82      C JJ IS COUNTER FOR THE CURRENT ELEMENT
83      C
84      C ZERO OUT A,B ARRAYS
85      DO 20 JJI = JT, JF
86      A(JJI) = 0.0
87      20 B(JJI) = 0.0
88      C
89      JJ = JJ
90      JJI = JJ + 1
91      JJ3 = JJI + 1
92      GO TO 40
93      30 JJ3 = JJ + 1
94      40 J = J + 1
95      A0 =
96      1 ARFORM(XO(I),YO(I),DL(I),SA(I),CA(I),IFST(I),
97      2 XO(J),YO(J),DL(J),SA(J),CA(J),ISEC(I),
98      A1, AC, AP, B0, B1, BC, BP)
99      B(JJ) = B(JJ) + A0
100     A(JJ) = A(JJ) + A0
101     C
102     IF (IIPRINT .EQ. 2)
103     IWRITE (6,50) JJ, A0, A1, AC, AP, B0, B1, BC, BP
104     50 FORMAT(1H, I3, 81X F12.6))
105     C
106     IF (IFST(IB) .EQ. 0) GO TO 80
107     IF (IFST(IB) - 2165.7R, 60)
108     C
109     C FIRST TERMS OF FIRST ORDER
110     60 A(JJ) = A(JJ) - (CF(J) + CD(J)) * A1
111     B(JJ) = B(JJ) - (CF(J) + CD(J)) * B1
112     A(JJ) = A(JJ) + CD(J) * A1
113     B(JJ) = B(JJ) + CD(J) * B1
114     A(JJ3) = A(JJ3) + CF(J) * A1

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MAF0057
MAF0058
MAF0059
MAF0060
MAF0061
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MAF0070
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MAF0106
MAF0107
MAF0108
MAF0109
MAF0110
MAF0111
MAF0112
MAF0113

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114      B(IJJ3) = B(IJJ2) +      CF(IJ)*B1      MAF0114
115      C      MAF0115
116      70 IF (IFST(IJ) .LT. 2) 60 TO 80      MAF0116
117      C      MAF0117
118      C CURVATURE TERM OF FIRST ORDER      MAF0118
119      A(IJJ1) = A(IJJ1) + DL(IJ)*CURV(IJ)*AC      MAF0119
120      B(IJJ1) = B(IJJ1) + DL(IJ)*CURV(IJ)*BC      MAF0120
121      C      MAF0121
122      80 IF (ISEC(IB) .EQ. C) 60 TO 110      MAF0122
123      IF (ISEC(IB) - 2)9C,100,90      MAF0123
124      C      MAF0124
125      C FIRST TERMS OF SECOND ORDER      MAF0125
126      90 A(IJJ1) = A(IJJ1) - (CG(IJ) + CI(IJ))*AP      MAF0126
127      B(IJJ1) = B(IJJ1) - (CG(IJ) + CI(IJ))*BP      MAF0127
128      A(IJJ1) = A(IJJ1) + CG(IJ)*AP      MAF0128
129      B(IJJ1) = B(IJJ1) + CG(IJ)*BP      MAF0129
130      A(IJJ3) = A(IJJ3) + CI(IJ)*AP      MAF0130
131      B(IJJ3) = B(IJJ3) + CI(IJ)*BP      MAF0131
132      C      MAF0132
133      100 IF (ISEC(IB) .LT. 2) 60 TO 110      MAF0133
134      C      MAF0134
135      C CURVATURE TERM OF SECOND ORDER      MAF0135
136      A(IJJ1) = A(IJJ1) + 2.0*AP*DL(IJ)*CURV(IJ)**2      MAF0136
137      B(IJJ1) = B(IJJ1) + 2.0*BP*DL(IJ)*CURV(IJ)**2      MAF0137
138      C      MAF0138
139      110 JJ1 = JJ      MAF0139
140      JJ = JJ + 1      MAF0140
141      IF (JJ - JF)30,120,130      MAF0141
142      C      MAF0142
143      120 JJ3 = JJ1 - 1      MAF0143
144      60 TO 90      MAF0144
145      130 JJ = JJ - 1      MAF0145
146      C      MAF0146
147      C      MAF0147
148      C IF LIFTING BODY. SAVE FIRST AND LAST B. ALSO CALCULATE N AND Y.      MAF0148
149      IF (LIFT(IB) .EQ. 0) 60 TO 150      MAF0149
150      VN = 0.0      MAF0150
151      VT = 0.0      MAF0151
152      SMDSWF(IB) = 0.0      MAF0152
153      IF (IINT .EQ. 1) .AND. (I .EQ. IC))      MAF0153
154      ICALL WEIGHT(SUMDS(IR), DS, JI, JF, WF, IPVR(IB))      MAF0154
155      DO 140 J = JI,JF      MAF0155
156      SMDSWF(IB) = SMDSWF(IB) + DS(J) * WF(J)      MAF0156
157      VN = VN + B(JI)*WF(J)      MAF0157
158      140 VT = VT + A(JI)*WF(J)      MAF0158
159      C      MAF0159
160      N = N + 1      MAF0160
161      NIKI = VN      MAF0161
162      T(K) = VT      MAF0162
163      C      MAF0163
164      C      MAF0164
165      150 CONTINUE      MAF0165
166      C      MAF0166
167      C A COMPLETE ROW OF BOTH A AND B HAS BEEN GENERATED.      MAF0167
168      C SAVE A ON UNIT 9, A AND B ON UNIT 10.      MAF0168
169      CALL SAVE(10, 1, 1, JJ, A, 1, VN)      MAF0169
170      CALL SAVE(10, 1, 1, JJ, B, 1, VN)      MAF0170

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171	C	MAF0171
172	C	MAF0172
173	C	SET ALPHA ONSET FLOWS
174	C	ALPHA = 0
175		MI11 = SAU11
176		TI11 = CAU11
177	C	MAF0177
178	C	ALPHA = 90
179		MI21 = -CAU11
180		TI21 = -SAU11
181	C	MAF0181
182	C	MAF0182
183		DO 160 K = 1,M
184		JJ = JJ + 1
185		160 A(JJ) = MI(J)
186	C	MAF0186
187	C	THE A-ARRAY IS SAVED ON IFOR IN SUPPLEMENT FORM
188		CALL SAVEIFOR, I, 1, 1, 1, A, 1, 1, 1, 1
189	C	MAF0189
190	C	MAF0190
191	C	ALSO SET UP TLU ARRAY
192		IF ILIE(TLU,1) .EQ. 0, GO TO 203
193		IF (1 - ME, TO) GO TO 201
194		LR = LR + 1
195		DO 200 K = 1,M
196		200 TLU(LR,K) = T(M)
197	C	MAF0197
198		201 IF (1 - ME, IF) GO TO 203
199		DO 202 K = 1,M
200		202 TLU(LR,K) = TLU(LR,K) + I(K)
201		203 CONTINUE
202	C	MAF0202
203	C	MAF0203
204	C	SAVE NORMAL AND TANGENTIAL ONSET VELOCITIES.
205	C	NORMALS ON UNIT 11, TANGENTIALS ON UNIT 12
206	C	WRITE(IF11) M
207		WRITE(IF12) I
208	C	MAF0208
209	C	IF I=10 AND LIFTING BODY, SET BLU
210		IF (1 - ME, TO) GO TO 180
211		IF ILIE(TLU,1) .EQ. 0, GO TO 180
212		DO 170 J = 1, NEL(T)
213		170 BLU(J) = 0(J)
214		215
215	C	MAF0215
216	C	MAF0216
217		180 CONTINUE
218	C	IF LIFTING BODY, SUM BLU
219		IF (LIFTING) .EQ. 0, GO TO 210
220		DO 190 J = 1, NEL(TOT)
221		190 BLU(J) = BLU(J) + 0(J)
222	C	MAF0222
223	C	MAF0223
224	C	STORE BLU ON UNIT 13
225		CALL SAVEIF13, 1, 1, 1, NEL(TOT), BLU, 1, 1, 1
226	C	MAF0226
227		210 CONTINUE
228	C	MAF0228
229	C	MAF0229
230	C	MAF0230
231	C	MAF0231
232	C	MAF0232
233	C	MAF0233
234	C	MAF0234
235	C	MAF0235
236	C	MAF0236
237	C	MAF0237
238	C	MAF0238
239	C	MAF0239
240	C	MAF0240
241	C	MAF0241
242	C	MAF0242
243	C	MAF0243
244	C	MAF0244
245	C	MAF0245
246	C	MAF0246
247	C	MAF0247
248	C	MAF0248
249	C	MAF0249
250	C	MAF0250
251	C	MAF0251
252	C	MAF0252
253	C	MAF0253
254	C	MAF0254
255	C	MAF0255
256	C	MAF0256
257	C	MAF0257
258	C	MAF0258
259	C	MAF0259
260	C	MAF0260
261	C	MAF0261
262	C	MAF0262
263	C	MAF0263
264	C	MAF0264
265	C	MAF0265
266	C	MAF0266
267	C	MAF0267
268	C	MAF0268
269	C	MAF0269
270	C	MAF0270
271	C	MAF0271
272	C	MAF0272
273	C	MAF0273
274	C	MAF0274
275	C	MAF0275
276	C	MAF0276
277	C	MAF0277
278	C	MAF0278
279	C	MAF0279
280	C	MAF0280
281	C	MAF0281
282	C	MAF0282
283	C	MAF0283
284	C	MAF0284
285	C	MAF0285
286	C	MAF0286
287	C	MAF0287
288	C	MAF0288
289	C	MAF0289
290	C	MAF0290
291	C	MAF0291
292	C	MAF0292
293	C	MAF0293
294	C	MAF0294
295	C	MAF0295
296	C	MAF0296
297	C	MAF0297
298	C	MAF0298
299	C	MAF0299
300	C	MAF0300
301	C	MAF0301
302	C	MAF0302
303	C	MAF0

228	C	MAF0228
229	C	MAF0229
230	C	MAF0230
231	C	MAF0231
232	240 CONTINUE	MAF0232
233	IF (IPRINT.NE.21) GO TO 340	MAF0233
234	WRITE (6,250)	MAF0234
235	250 FORMAT(1H1)	MAF0235
236	DO 290 IB = 1,IBTOT	MAF0236
237	IF (IBMF(1B).LT.0) GO TO 290	MAF0237
238	J1 = INL(1B)	MAF0238
239	J2 = IFL(1B)	MAF0239
240	WRITE (6,260) IDB(1B)	MAF0240
241	260 FORMAT(1MD,'GEOMETRY COMBINATION COEFFICIENTS FOR BODY ID = ',I2/MAF0241	
242	1 1MD, T4,'J', T15,'CD',T31,'CF',T47,'CG',T63,'CI')	MAF0242
243	C	MAF0243
244	DO 270 J = J1,J2	MAF0244
245	270 WRITE (6,280) J, CD(J), CF(J), CG(J), CI(J)	MAF0245
246	280 FORMAT(1M, I3, 4(4XF12.6))	MAF0246
247	290 CONTINUE	MAF0247
248	C	MAF0248
249	WRITE (6,250)	MAF0249
250	REWIND IF10	MAF0250
251	JF = MELTOT	MAF0251
252	DO 330 J = 1,JF	MAF0252
253	CALL GETT(1F10, 1, JF, A, 1, VN)	MAF0253
254	CALL GETT(1F10, 1, JF, B, 1, VN)	MAF0254
255	WRITE (6,300) J	MAF0255
256	300 FORMAT(1MD, T4,'A(', I3,' ', J1)'/)	MAF0256
257	WRITE (6,310) (A(I), I=1,JF)	MAF0257
258	310 FORMAT(8(4XF12.6))	MAF0258
259	C	MAF0259
260	WRITE (6,320) J	MAF0260
261	320 FORMAT(1MD, T4,'B(', I3,' ', J1)'/)	MAF0261
262	WRITE (6,310) (B(I), I=1,JF)	MAF0262
263	C	MAF0263
264	330 CONTINUE	MAF0264
265	C	MAF0265
266	C	MAF0266
267	340 CONTINUE	MAF0267
268	C	MAF0268
269	C	MAF0269
270	C	MAF0270
271	RETURN	MAF0271
272	END	MAF0272

```

1      SUBROUTINE MAIN1
2      C
3      DIMENSION X(1500), Y(1500), Z(1500), SA(1500), CA(1500),
4      1      ELGC(1500), ELGD(2500)
5      C
6      COMMON /BFLAG/ IDRI(10), INL(10), IFL(10), NL(10), LIFT(10),
7      1      IAME(10), ISAV(10), ISAV2(10), ISAV3(10),
8      2      BTITLE(10), 71, IRT, IBST, IBTOT, NELTOT,
9      2      ITRD(10), INNH(10), CHORD(10), IRD(10), LIFTOT
10     4      ,IPRB(10), IFST(10), ISEC(10), FTITLE(15), IPVR(10)
11     C
12     C
13     C
14     COMMON /COMBOD/CCL, INCLT, CLT, ALPH4, SUMDS(10), YLV(10,12), IMD
15     1      , ALPHAD, CHVA(10), RDBSNF(10), NIG(10)
16     C
17     COMMON /FILED/ IFIL1, IFIL2, IFIL3, IFIL4, IFIL5,
18     1      IFIL6, IFIL7, IFIL8, IFIL9, IFIL10,
19     2      IFIL11, IFIL12, IFIL13, IFIL14, IFIL15,
20     3      IFIL16, IFIL17, IFIL18, IFIL19, IFIL20
21     COMMON /MDATA/ ISOL, IOFF, NONU, NPMU, IPRINT, MORE, N
22     COMMON /ELDATA/ X0, Y0, Z0, SA, CA, CURV(500), DL(500)
23     COMMON /GCOFFS/ CO(500), CF(500), CG(500), CI(500), WF(500)
24     COMMON /COM/ IFLL
25     EQUIVALENCE (ELGC(1), Z(1)), (ELGD(1), CO(1))
26     C
27     C FORM ELEMENTS
28     CALL TIMEV(1)
29     WRITE(6,20) T
30     20 FORMAT (1MD, 'CALL ELFORM, T = ', F9.3, 'SECONDS.')
31     CALL ELFORM (SUMDS)
32     C
33     C READ IN FLOW TITLE CARD
34     ITYP = 8
35     READ (5,30) FTITLE, ITYPE
36     30 FORMAT (15A4, 11X11)
37     IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE)
38     C
39     C READ IN FLOW CONTROL CARD
40     ITYP = 9
41     READ (5,40) INCLT, CLT, ICHORD, CCL, IMD, ISOL, IOFF, NONU,
42     1      NPMU, IPRINT, MORE, IFLL, ITYPE
43     40 FORMAT (11, 4XF10.5, 2X11, 2XF10.5, 5(9X11), 9X11, 9X11, 11, 11)
44     IF (ITYPE .NE. ITYP) CALL TYPE(ITYP, ITYPE)
45     IF (NONU .NE. 0) ISOL = 1
46     IF (IMCLT .EQ. 0) ALPH4 = CLT
47     IF (ICHORD .EQ. 0) CCL = 1.0
48     C
49     C FORM MATRICES
50     CALL TIMEV(1)
51     WRITE(6,50) T
52     50 FORMAT (1MD, 'ELFORM COMPLETE, CALL MAFORM, T = ', F9.3, 'SECONDS.')
53     CALL MAFORM (N, NONU, NPMU, ISOL, IPRINT)
54     C
55     CALL TIMEV(1)
56     WRITE (6,60) T

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MAN1001
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57      60 FORMAT (1H0, 'MAFORM COMPLETE. CALL SOLVE. T = ',F9.3,'SECONDS.') MAN1056
58      C MAN1057
59      CALL ASSEMR MAN1058
60      C MAN1059
61      C SAVE ELDATA AND GCOEFS DATA MAN1060
62      REWIND IFIL16 MAN1061
63      WRITE(IFIL16) ELGC, ELGD MAN1062
64      C MAN1063
65      RETURN MAN1064
66      END MAN1065

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1      SUBROUTINE MAIN3 MAN3001
2      C MAN3002
3      COMMON /BELAG/ IDRI(10), INLI(10), IFI(10), MI(10), LIFT(10), MAN3003
4      1      IBMF(10), ISAV1(10), ISAV2(10), ISAV3(10), MAN3004
5      2      RYIF(10), T1, TRT, TRST, TRTOT, NFITOT, MAN3005
6      3      ITRB(10), INMB(10), CHORDB(10), IDI(10), LIFTOT MAN3006
7      4      ,IPRB(10), IFST(10), ISEC(10), FTITLE(15), IPWR(10) MAN3007
8      C MAN3008
9      C MAN3009
10     C MAN3010
11     COMMON /COMMOD/CCI, INCLT, CLT, ALPHA, SUMDS(10), TLU(10,12),TND MAN3011
12     1      , ALPHAO, CNU(10), SMO SWF(10), MIO(10) MAN3012
13     C MAN3013
14     COMMON /FILEID/ IFILE1, IFILE2, IFILE3, IFILE4, IFILE5, MAN3014
15     1      IFILE6, IFILE7, IFILE8, IFILE9, IFILE10, MAN3015
16     2      IFIL11, IFIL12, IFIL13, IFIL14, IFIL15 MAN3016
17     3      ,IFIL16, IFIL17, IFIL18, IFIL19, IFIL20 MAN3017
18     COMMON /MDATA/ ISOL,IOFF,NONU,NBNU,IPRINT,MORE,M MAN3018
19     COMMON /COM/ IFILL
20     C MAN3019
21     C CALCULATE COMBINATION CONSTANTS IF IFILL= 1 MAN3020
22     IF (IFILL.EQ.0) GO TO 81
23     CALL COMBO (NELTOT, LIFTOT, M, NONU) MAN3021
24     C MAN3022
25     CALL TIMEVIT) MAN3023
26     WRITE(6,80) T MAN3024
27     80 FORMAT (1H0, 'COMBO COMPLETE. CALL FLOWS. T = ',F9.3,'SECONDS.') MAN3025
28     C MAN3026
29     81 CALL FLOWSINELTOT, M, IPRINT) MAN3027
30     CALL TIMEVIT) MAN3028
31     WRITE(6,90) T MAN3029
32     90 FORMAT(1H0, 'FLOWS COMPLETE. T = ', F9.3, 'SECONDS.') MAN3030
33     C MAN3031
34     C CHECK FOR OFFBODY POINTS MAN3032
35     IF (IOFF.NE.1) GO TO 110 MAN3033
36     C MAN3034
37     CALL OFFBOD(INELTOT, M, CHORDB, IDB, IBTOT) MAN3035
38     CALL TIMEVIT) MAN3036
39     WRITE(6,100) T MAN3037
40     100 FORMAT(1H0, 'OFFBODY POINTS COMPLETE. T = ', F9.3, 'SECONDS.') MAN3038
41     C MAN3039
42     C MAN3040
43     110 CONTINUE MAN3041
44     C MAN3042
45     C MAN3043
46     RETURN MAN3044
47     END MAN3045

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1		SUBROUTINE MIS1 (A, N, MD, B, MD, NERR, D)	MIS1001
2	C		MIS1002
3	C	A REAL*8 SUBROUTINE TO	MIS1003
4	C	INVERT A MATRIX AND/OR SOLVE SIMULTANEOUS EQUATIONS	MIS1004
5	C		MIS1005
6	C	INPUT,	MIS1006
7	C	A = NAME OF INPUT MATRIX, DIMENSIONED A(MD,MD)	MIS1007
8	C	M = IS THE ORDER OF THE MATRIX A	MIS1008
9	C	MD = IS THE DIMENSION OF THE SQUARE ARRAY A	MIS1009
10	C	B = NAME OF INPUT MATRIX, DIMENSIONED B(MD,MD)	MIS1010
11	C	(B(MD) IS ALIGNED IF MD=1)	MIS1011
12	C	MD = IS THE NUMBER OF COLUMNS IN THE RECTANGULAR ARRAY B	MIS1012
13	C	D = SCALE FACTOR FOR VALUE OF DETERMINANT (= 1., FOR MD SQUARE)	MIS1013
14	C		MIS1014
15	C	OUTPUT,	MIS1015
16	C	A(I,J) = GARBAGE	MIS1016
17	C	B(I,J) = A-INVERSE * B	MIS1017
18	C	NERR = 0--OK, 1--A IS SINGULAR	MIS1018
19	C	D = SCALED VALUE OF DETERMINANT	MIS1019
20	C		MIS1020
21	C	REAL*8 A,B,D,AIJMAX,ARRAY	MIS1021
22	C	DIMENSION A(MD,MD), B(MD,MD)	MIS1022
23	C	EQUIVALENCE (I,FI), (M,PM)	MIS1023
24	C	DATA EPS/1.E-25/	MIS1024
25	C	START REDUCTION OF MATRIX A	MIS1025
26	C		MIS1026
27	C	DO 80 I=1,M	MIS1027
28	C		MIS1028
29	C	SEARCH FOR MAXIMUM ELEMENT IN ITH ROW OF A-MATRIX	MIS1029
30	C		MIS1030
31	C	AIJMAX = A(I,1)	MIS1031
32	C	JMAX = 1	MIS1032
33	C	DO 10 J=2,M	MIS1033
34	C	IF (ABS(A(I,J)) .LE. ABS(A(I,JMAX))) GO TO 10	MIS1034
35	C	AIJMAX = A(I,J)	MIS1035
36	C	JMAX = J	MIS1036
37	C	10 CONTINUE	MIS1037
38	C		MIS1038
39	C	IF AIJMAX IS ZERO, THE MATRIX IS SINGULAR	MIS1039
40	C		MIS1040
41	C	IF (ABS(A(I,JMAX)) .GT. EPS) GO TO 20	MIS1041
42	C	D = 0.0E0	MIS1042
43	C	NERR = 1	MIS1043
44	C	RETURN	MIS1044
45	C		MIS1045
46	C	NORMALIZE ITH ROW BY AIJMAX (JMAX ELEMENT OF ITH ROW)	MIS1046
47	C		MIS1047
48	C	20 DO 30 J=1,M	MIS1048
49	C	30 A(I,J) = A(I,J) / AIJMAX	MIS1049
50	C	D = D*AIJMAX	MIS1050
51	C		MIS1051
52	C	NORMALIZE ITH ROW OF B	MIS1052
53	C		MIS1053
54	C	DO 40 J=1,MD	MIS1054
55	C	40 B(I,J) = B(I,J) / AIJMAX	MIS1055
56	C		MIS1056

57	C	USE ROW TRANSFORMATIONS TO GET ZEROS ABOVE AND BELOW THE JMAX	MIS1057
58	C	ELEMENT OF THE ITH ROW OF A. APPLY SAME ROW TRANSFORMATIONS	MIS1058
59	C	TO THE N MATRIX.	MIS1059
60	C		MIS1060
61		DO 70 K=1,M	MIS1061
62		IF (M - EQ. 7) GO TO 70	MIS1062
63		ARAT = -A(K,JMAX)	MIS1063
64		DO 80 J=1,M	MIS1064
65		IF (ABS(A(I,J)) .LT. EPS) GO TO 50	MIS1065
66		A(K,J) = ARAT * A(I,J) + A(K,J)	MIS1066
67		50 CONTINUE	MIS1067
68		A(K,JMAX) = 0.D0	MIS1068
69		DO 60 J=1,M	MIS1069
70		IF (ABS(B(I,J)) .LT. EPS) GO TO 60	MIS1070
71		B(K,J) = ARAT * B(I,J) + B(K,J)	MIS1071
72		60 CONTINUE	MIS1072
73		70 CONTINUE	MIS1073
74	C		MIS1074
75	C	STORE ROW COUNTER (I) IN TOP ELEMENT OF JMAX COLUMN. THUS,	MIS1075
76	C	THE TOP ROW OF A WILL CONTAIN THE LOC OF THE PIVOT (UNITY)	MIS1076
77	C	ELEMENT OF EACH COLUMN (AFTER REDUCTION).	MIS1077
78	C		MIS1078
79		L = I	MIS1079
80		80 A(I,JMAX) = FL	MIS1080
81	C	THIS STORES INTEGER I IN TOP ROW OF A	MIS1081
82	C		MIS1082
83	C	THE REDUCTION OF A IS NOW COMPLETE. PERFORM ROW INTERCHANGES	MIS1083
84	C	AS INDICATED IN THE FIRST ROW OF A.	MIS1084
85	C		MIS1085
86		DO 120 I=1,M	MIS1086
87		K = I	MIS1087
88		90 FM = A(I,M)	MIS1088
89	C	THIS PUTS THE INTEGER VALUE IN A INTO K	MIS1089
90		IF (K-I)90,120,100	MIS1090
91	C		MIS1091
92	C	IF K(I,I) IS LESS THAN I, THEN THAT ROW HAS ALREADY BEEN	MIS1092
93	C	INVOLVED IN AN INTERCHANGE, AND WE USE K(I,K) UNTIL WE GET	MIS1093
94	C	A VALUE OF K GREATER THAN I (CORRESPONDING TO A ROW STORED	MIS1094
95	C	BELOW THE ITH ROW. (CLEAR AS MUCH)	MIS1095
96	C		MIS1096
97		100 DO 110 J=1,M	MIS1097
98		ARAT = B(I,J)	MIS1098
99		B(I,J) = B(K,J)	MIS1099
100		110 B(K,J) = ARAT	MIS1100
101		D = -D	MIS1101
102		120 CONTINUE	MIS1102
103		WERR = 0	MIS1103
104		RETURN	MIS1104
105		END	MIS1105

```

1 SUBROUTINE MIS2 (A, N, MD, B, MD, WL, MERR, DJ) MIS2001
2 C A REAL9A SUBROUTINE TO MIS2002
3 C INVERT A MATRIX AND/OR SOLVE SIMULTANEOUS EQUATIONS MIS2003
4 C MIS2004
5 C INPUT, MIS2005
6 C A = NAME OF INPUT MATRIX, DIMENSIONED A(MD,MD) MIS2006
7 C N = IS THE ORDER OF THE MATRIX A MIS2007
8 C MD = IS THE DIMENSION OF THE SQUARE ARRAY A MIS2008
9 C B = NAME OF INPUT MATRIX, DIMENSIONED B(MD,MD) MIS2009
10 C (BLIND) IS ALLOCATED IF MD=1 MIS2010
11 C MD = IS THE NUMBER OF COLUMNS IN THE RECTANGULAR ARRAY B MIS2011
12 C D = SCALE FACTOR FOR VALUE OF DETERMINANT (21.. FOR NO SCALING) MIS2012
13 C OUTPUT, MIS2013
14 C A(I,J) = GARBAGE MIS2014
15 C B(I,J) = A-INVERSE * B MIS2015
16 C MERR = 0--OK, 1--A IS SINGULAR MIS2016
17 C D = SCALED VALUE OF DETERMINANT MIS2017
18 C REAL9A A,B,D,A1MAX,ARAT MIS2018
19 C DIMENSION A(MD,MD), B(MD,M) MIS2019
20 C EQUIVALENCE (A,EL), (B,EL) MIS2020
21 C DATA EPS/1.E-25/ MIS2021
22 C START REDUCTION OF MATRIX A MIS2022
23 C DO 80 J=1,M MIS2023
24 C DO 80 I=1,M MIS2024
25 C SEARCH FOR MAXIMUM ELEMENT IN ITH ROW OF A-MATRIX MIS2025
26 C A1JMAX = A(I,J) MIS2026
27 C JMAX = J MIS2027
28 C IF A1JMAX IS ZERO, THE MATRIX IS SINGULAR MIS2028
29 C IF (ABS(A1JMAX) > EPS) GO TO 20 MIS2029
30 C D = 0.0E0 MIS2030
31 C MERR = 1 MIS2031
32 C RETURN MIS2032
33 C IF (ABS(A1JMAX) > EPS) GO TO 20 MIS2033
34 C D = 0.0E0 MIS2034
35 C MERR = 1 MIS2035
36 C RETURN MIS2036
37 C IF A1JMAX IS ZERO, THE MATRIX IS SINGULAR MIS2037
38 C IF (ABS(A1JMAX) > EPS) GO TO 20 MIS2038
39 C D = 0.0E0 MIS2039
40 C MERR = 1 MIS2040
41 C RETURN MIS2041
42 C NORMALIZE ITH ROW BY A1JMAX (JMAX ELEMENT OF ITH ROW) MIS2042
43 C DO 40 J=1,M MIS2043
44 C A(I,J) = A(I,J) / A1JMAX MIS2044
45 C NORMALIZE ITH ROW OF B MIS2045
46 C DO 40 J=1,MD MIS2046
47 C B(I,J) = B(I,J) / A1JMAX MIS2047
48 C MIS2048
49 C MIS2049
50 C MIS2050
51 C MIS2051
52 C MIS2052
53 C MIS2053
54 C MIS2054
55 C MIS2055
56 C MIS2056

```

```

57 C USE ROW TRANSFORMATIONS TO GET ZEROS ABOVE AND BELOW THE JMAX MIS2057
58 C ELEMENT OF THE ITH ROW OF A. APPLY SAME ROW TRANSFORMATIONS MIS2058
59 C TO THE B MATRIX. MIS2059
60 C MIS2060
61 DO 70 N=1,M MIS2061
62 IF (A(EB,I)) GO TO 70 MIS2062
63 ARAT = -A(M,JMAX) MIS2063
64 DO 50 J=1,M MIS2064
65 IF (ABS(A(I,J)) < .1) EPS) GO TO 50 MIS2065
66 A(I,J) = ARAT * A(I,J) + A(M,J) MIS2066
67 50 CONTINUE MIS2067
68 A(K,JMAX) = 0.0EN MIS2068
69 DO 60 J=1,M MIS2069
70 IF (ABS(B(I,J)) < .1) EPS) GO TO 60 MIS2070
71 B(K,J) = ARAT * B(I,J) + B(M,J) MIS2071
72 60 CONTINUE MIS2072
73 70 CONTINUE MIS2073
74 C MIS2074
75 C STORE ROW COUNTER (I) IN TOP ELEMENT OF JMAX COLUMN. THUS, MIS2075
76 C THE TOP ROW OF A WILL CONTAIN THE LOC OF THE PIVOT (UNITY) MIS2076
77 C ELEMENT OF EACH COLUMN (AFTER REDUCTION). MIS2077
78 C MIS2078
79 I = I MIS2079
80 A(I,JMAX) = EL MIS2080
81 C THIS STORES INTEGER I IN TOP ROW OF A MIS2081
82 C MIS2082
83 C THE REDUCTION OF A IS NOW COMPLETE. PERFORM ROW INTERCHANGES MIS2083
84 C AS INDICATED IN THE FIRST ROW OF A. MIS2084
85 C MIS2085
86 DO 120 J=1,M MIS2086
87 K = I MIS2087
88 90 FM = A(I,K) MIS2088
89 C THIS PUTS THE INTEGER VALUE IN A INTO M MIS2089
90 IF (K-I) 90,120,100 MIS2090
91 C MIS2091
92 C IF K(I,I) IS LESS THAN I, THEN THAT ROW HAS ALREADY BEEN MIS2092
93 C INVOLVED IN AN INTERCHANGE, AND WE USE K(I,K) UNTIL WE GET MIS2093
94 C A VALUE OF K GREATER THAN I (CORRESPONDING TO A ROW STORED MIS2094
95 C BELOW THE ITH ROW. (CLEAR AS NUC) MIS2095
96 C MIS2096
97 100 DO 110 J=1,M MIS2097
98 ARAT = B(I,J) MIS2098
99 B(I,J) = A(K,J) MIS2099
100 110 B(K,J) = ARAT MIS2100
101 U = -0 MIS2101
102 120 CONTINUE MIS2102
103 NERR = 0 MIS2103
104 RETURN MIS2104
105 FMD MIS2105

```

1			
2	C	SUBROUTINE OFFBOD (N, M, CHORDB, IDB, IBTGT)	OFFB001
3			OFFB002
4		DIMENSION X(100), Y(100), TITLE(7)	OFFB003
5	1	CHORDB(10), IDB(10)	OFFB004
6		COMMON/COMMON/CCI, INCLT, CIT, ALPHA, SUMDS(10), TIU(10, 12), TND	OFFB005
7	1	ALPHA0, CNU(10), SMDSWF(10), MID(10)	OFFB006
8	C		OFFB007
9	C	READ IN OFFBODY POINTS (IN BLOCKS OF UP TO 100)	OFFB008
10		10 CALL OFFPTS(INO, X, Y, TITLE, LAST, CHORDB, IDB, IBTGT)	OFFB009
11	C		OFFB010
12	C	NOW CALCULATE VX AND VY	OFFB011
13		CALL VXYOFF (N, M, NO, X, Y)	OFFB012
14	C		OFFB013
15	C	PRINT OFFBODY POINTS AND VELOCITIES	OFFB014
16		CALL VPROFF (N, M, NO, X, Y, TITLE, IND)	OFFB015
17	C		OFFB016
18	C	CHECK IF MORE POINTS AND CYCLE IF SO	OFFB017
19		IF (LAST .NE. 1) GO TO 10	OFFB018
20	C		OFFB019
21	C		OFFB020
22		RETURN	OFFB021
23		END	OFFB022
			OFFB023


```

1      SUBROUTINE OFFPTSINO, X, Y, TITLE, LAST, CHORDB, ICB, IBTOT)  OFFP001
2      C                                                                OFFP002
3      DIMENSION X(1), Y(1), TITLE( 7)  OFFP003
4      1      ,CHORDB(10), IDB(10)  OFFP004
5      C                                                                OFFP005
6      C                                                                OFFP006
7      DATA EPS/1.0E-7/, CR/1.79532925E-2/  OFFP007
8      C                                                                OFFP008
9      C READ IN BODY TITLE AND CONTROL CARD  OFFP009
10     10 IITYP = 21  OFFP010
11     READ(5,20) ID, TITLE, ITR, IMORN, INOLD, LAST, IITYP  OFFP011
12     20 FORMAT( I1, 9X7A4, 12X, 2(2X11), 2(5X11), 2X12)  OFFP012
13     IF (IITYP .NE. IITYP) CALL TYPE(IITYP, IITYP)  OFFP013
14     C                                                                OFFP014
15     C READ IN COORDINATE TRANSFORMATION CARD IF REQUIRED  OFFP015
16     IITYP = 22  OFFP016
17     CHORD = 0.0  OFFP017
18     IF (ITR .EQ. 0 .OR. ITR .EQ. 2) GO TO 40  OFFP018
19     READ(5,30) CHORD, XMULT, YMULT, DX, DY, THETA, XTQ, YTO, IITYP  OFFP019
20     30 FORMAT (7F8.0,1X), F8.0, 11)  OFFP020
21     IF (IITYP .NE. (IITYP-20)) CALL TYPE(IITYP, IITYP)  OFFP021
22     C                                                                OFFP022
23     40 CONTINUE  OFFP023
24     C                                                                OFFP024
25     C CHECK IF ELLIPSE TO BE GENERATED  OFFP025
26     IF (ITR .GT. 1) GO TO 90  OFFP026
27     C                                                                OFFP027
28     C DATA ON UNIT 5. X-COORDS FIRST  OFFP028
29     L = 0  OFFP029
30     IITYP = 23  OFFP030
31     50 READ(5,60) (X(L+1), I=1,6), INO, ISTAT, IITYP  OFFP031
32     60 FORMAT (6F10.0, 4X11, 2X11, 3X11)  OFFP032
33     IF (IITYP .NE. (IITYP-20)) CALL TYPE(IITYP, IITYP)  OFFP033
34     IF (INO .LE. 0 .OR. INO .GT. 6) INO = 6  OFFP034
35     L = L + INO  OFFP035
36     IF (ISTAT .EQ. 0) GO TO 50  OFFP036
37     LX = L  OFFP037
38     C                                                                OFFP038
39     C NOW READ IN Y-COORDS  OFFP039
40     L = 0  OFFP040
41     IITYP = 24  OFFP041
42     70 READ(5,60) (Y(L+1), I=1,6), INC, ISTAT, IITYP  OFFP042
43     IF (IITYP .NE. (IITYP-20)) CALL TYPE(IITYP, IITYP)  OFFP043
44     IF (INO .LE. 0 .OR. INO .GT. 6) INO = 6  OFFP044
45     L = L + INO  OFFP045
46     IF (ISTAT .EQ. 0) GO TO 70  OFFP046
47     LY = L  OFFP047
48     C                                                                OFFP048
49     C CHECK FOR INPUT CONSISTENCY  OFFP049
50     IF (LY .EQ. LX) GO TO 120  OFFP050
51     WRITE(6,80) LY, LX  OFFP051
52     80 FORMAT (1H1, 'THE NUMBER OF Y-COORDINATES ('.13.') READ DOES '.  OFFP052
53     1      'NOT EQUAL THE NUMBER OF X-COORDINATES READ ('.13.')'  OFFP053
54     GO TO 200  OFFP054
55     C                                                                OFFP055
56     C ELLIPSE TO BE GENERATED. READ IN DEFINITION CARD.  OFFP056

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```

57      90 IITP = 25
58      READ 15,1001 LX, ELPSTH, IITPE
59      100 FORMAT (2X13, 5X1C.5, 5X11)
60      IF (IITPE .NE. (IITP-20)) CALL TYPE(IITP, IITPE)
61      ITR = ITR - 2
62      IF (ITR .NE. 1) ITR = 0
63      C
64      DANGLE = 6.2831853072/(LX - 1)
65      ANGLE = DANGLE
66      DO 110 I = 1,LX
67      ANGLE = ANGLE - DANGLE
68      X(I) = COS(ANGLE)
69      110 Y(I) = SIN(ANGLE)*ELPSTH
70      C
71      120 CONTINUE
72      C
73      C WRITE OUT BASIC GEOMETRY DATA
74      IP = 1
75      C
76      C
77      C TRANSFORM COORDINATES IF REQUESTED
78      IF (ITR .EQ. 1) GO TO 130
79      IF (INORM .EQ. 0) GO TO 220
80      XMULT = 0.0
81      YMULT = 0.0
82      XTO = 0.0
83      YTO = 0.0
84      YMETA = 0.0
85      DX = 0.0
86      DY = 0.0
87      130 CONTINUE
88      C
89      IF (ABS(XMULT) .LT. EPS) XMULT = 1.0
90      IF (ABS(YMULT) .LT. EPS) YMULT = 1.0
91      XSF = XMULT
92      YSF = YMULT
93      IF (INORM .EQ. 0) GO TO 180
94      C
95      IF (ICOLD .LE. 0) GO TO 160
96      DO 140 IB = 1,IBTOT
97      IF (IDB(IB) .EQ. ICOLD) GO TO 150
98      140 CONTINUE
99      GO TO 160
100      150 CHORD = CHORDB(IB)
101      160 IF (ABS(CHORD) .LE. EPS) GO TO 180
102      170 XSF = XSF/CHORD
103      YSF = YSF/CHORD
104      C
105      180 COST = COS(THETA*DR)
106      SINT = SIN(THETA*DR)
107      DO 190 I = 1,LX
108      XTOD = X(I) - XTO
109      YTOD = Y(I) - YTO
110      X(I) = (XTO + XTOD*COST - YTOD*SINT + DX)*XSF
111      Y(I) = (YTO + YTOD*SINT + XTOD*COST + DY)*YSF
112      190 CONTINUE
113      GO TO 220

```

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OFFP057
OFFP058
OFFP059
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OFFP106
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OFFP109
OFFP110
OFFP111
OFFP112
OFFP113

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114	200 WRITE (6,210)	OFFP114
115	210 FORMAT (1H0, 'BECAUSE OF THE ABOVE ERROR, THIS RUN IS TERMINATED')	OFFP115
116	STOP	OFFP116
117	C	OFFP117
118	C	OFFP118
119	220 CONTINUE	OFFP119
120	NO = LX	OFFP120
121	RETURN	OFFP121
122	END	OFFP122

1	SUBROUTINE PRINTG (IP, N, ID, BTITLE)	PRNG001
2	C	PRNG002
3	C THIS SUBROUTINE WRITES OUT THE BODY COORDINATE DATA	PRNG003
4	C	PRNG004
5	C	PRNG005
6	DIMENSION BTITLE (7), A(2)	PRNG006
7	C	PRNG007
8	COMMON /GEOMD/ X(500), Y(500)	PRNG008
9	C	PRNG009
10	DATA A(1),A(2)/'UNTR',' TR'/	PRNG010
11	C	PRNG011
12	C	PRNG012
13	J2 = 0	PRNG013
14	10 WRITE (6,30)	PRNG014
15	WRITE (6,20) A(1P), ID, BTITLE	PRNG015
16	20 FORMAT(1H0,15X4,'TRANSFORMED COORDINATE DATA FOR BODY ID = ',I2,	PRNG016
17	1 ' ' 7A9Z//115,'I', I21,'X(1)', I44,'Y(1)',	PRNG017
18	2 ' ' 775,'I', 787,'X(1)', 7104,'Y(1)')	PRNG018
19	30 FORMAT (1H1)	PRNG019
20	C	PRNG020
21	J1 = J2 + 1	PRNG021
22	JMX = J1 + 49	PRNG022
23	N2 = (N + 1 + J2)/2	PRNG023
24	IF (JMX .GT. N2) JMX = N2	PRNG024
25	J2 = JMX	PRNG025
26	DO 50 J = J1,JMX	PRNG026
27	J2 = J2 + 1	PRNG027
28	IF (J2 .GT. N) 60 TO 60	PRNG028
29	WRITE (6,40) J, X(J), Y(J), J2, X(J2), Y(J2)	PRNG029
30	40 FORMAT (1H ,11X13, 2(5XF12.6), 2X13, 2(5XF12.6))	PRNG030
31	50 CONTINUE	PRNG031
32	C	PRNG032
33	C	PRNG033
34	IF (J2 .LT. N) 60 TO 10	PRNG034
35	RETURN	PRNG035
36	C	PRNG036
37	C	PRNG037
38	60 WRITE (6,40) J, X(J), Y(J)	PRNG038
39	C	PRNG039
40	RETURN	PRNG040
41	END	PRNG041

1		SUBROUTINE PRNTEL	PRNL001
2	C		PRNL002
3	C		PRNL003
4		DIMENSION AN(10)	PRNL004
5		REAL NLV,NO,CHGE	PRNL005
6	C		PRNL006
7		COMMON /BFLAG/ IOR(10), INL(10), IFL(10), NL(10), LIFT(10),	PRNL007
8	1	IRMF(10), ISAV1(10), ISAV2(10), ISAV(10),	PRNL008
9	2	BTITLE(10, 7), IBT, IBST, IBTOT, MELTOT,	PRNL009
10	3	ITRR(10), INMB(10), CHOROB(10), IBD(10), LIFTOT	PRNL010
11	4	IPRR(10), IFST(10), ISEC(10), FTITLE(15), IPVR(10)	PRNL011
12	C		PRNL012
13	C		PRNL013
14		DATA NEW,OLD,YES,NO,SAVED,BLANK/' NEW',' OLD',' YES',' NO ',	PRNL014
15	1	' S ',' ','CHGE','CHGE','CHGE',' NC ','AN',' 0 ',' 1 ',	PRNL015
16	2	' 2 ',' 3 ',' 4 ',' 5 ',' 6 ',' 7 ',' 8 ',' 9 ',	PRNL016
17	3	' AP, AL/' P ',' L ',	PRNL017
18	C		PRNL018
19	C	WRITE HEADING FOR BODY GEOMETRY SUMMARY	PRNL019
20		WRITE (6,10)	PRNL020
21		10 FORMAT (1H1, T51,'BODY GEOMETRY SUMMARY'//	PRNL021
22	1	1H0,T33,'BODY LIFT',T99,'SIGMA',	PRNL022
23	2	T114,'ELEMENT STORAGE'//	PRNL023
24	3	T6,'BODY DESCRIPTION',T34,'ID TYPE',	PRNL024
25	4	T50,'N/O SID TFORM NORM CHORD TYPE',	PRNL025
26	5	T97,'F C S C N/O FIRST NO.')	PRNL026
27	C		PRNL027
28	C		PRNL028
29		IBDTOT = 0	PRNL029
30		MELTOT = 0	PRNL030
31		LIFTOT = 0	PRNL031
32		DO 150 IB = 1,IBT	PRNL032
33		IF (IBD(1B) .GE. 6) GO TO 150	PRNL033
34	C		PRNL034
35		IF (IBD(1B) - 312C,20,30	PRNL035
36	20	A2 = NEW	PRNL036
37		A3 = BLANK	PRNL037
38		IF (ISAV3(1B) .GE. 0) A3 = SAVED	PRNL038
39		GO TO 40	PRNL039
40	C		PRNL040
41	30	A2 = OLD	PRNL041
42		IN = ISAV1(1B) + 1	PRNL042
43		A1 = AN(1B)	PRNL043
44	C		PRNL044
45	40	A4 = NO	PRNL045
46		A5 = NO	PRNL046
47		IF (ITRR(1B) .EQ. 1) A4 = YES	PRNL047
48		IF (INMB(1B) .EQ. 1) A5 = YES	PRNL048
49	C		PRNL049
50		A1 = NO	PRNL050
51		IF (LIFT(1B) .EQ. 1) A1 = YES	PRNL051
52	C		PRNL052
53		A6 = OLD	PRNL053
54		IF (IRMF(1B) - 117C,50,60	PRNL054
55	50	A6 = NEW	PRNL055
56		GO TO 70	PRNL056

57	60 A6 = CMGE	PRNL057
58	C	PRNL058
59	C	PRNL059
60	70 A7 = AP	PRNL060
61	18 = 1	PRNL061
62	19 = 1	PRNL062
63	110 = 1	PRNL063
64	111 = 1	PRNL064
65	C	PRNL065
66	IF (IPRB(IB) .NE. 1) A7 = AL	PRNL066
67	IF (IFST(IB) - 2) RC, 90, 100	PRNL067
68	80 19 = 0	PRNL068
69	IF (IFST(IB) .EQ. 1) 60 TO 100	PRNL069
70	90 18 = 0	PRNL070
71	100 IF (ISEC(IB) - 2) 110, 120, 130	PRNL071
72	110 111 = 0	PRNL072
73	IF (ISEC(IB) .EQ. 1) 60 TO 130	PRNL073
74	120 110 = 0	PRNL074
75	130 CONTINUE	PRNL075
76	C	PRNL076
77	IBDTOT = IBDTOT + 1	PRNL077
78	NELTOT = NELTOT + M(IB)	PRNL078
79	IF (LIFT(IB) .EQ. 1) LIFTOT = LIFTOT + 1	PRNL079
80	WRITE (6,140) (BTITLE(IB, I, I=1,7), IDB(IB), A1, A2, A3,	PRNL080
81	1 A4, A5, CHORDB(IB), A7, 18, 19, 110, 111, A6,	PRNL081
82	2 INL(IB), NL1(IB)	PRNL082
83	140 FORMAT (1H0, 7A4, 4X12, 5X44, 4X44, 2X44, 3X44, 3X44, 2X12.6,	PRNL083
84	1 3X44, 3X11, 2X11, 3X11, 2X11, 3X44, 5X14, 4X14)	PRNL084
85	C	PRNL085
86	150 CONTINUE	PRNL086
87	C	PRNL087
88	C	PRNL088
89	WRITE (6,160) IBDTOT, NELTOT	PRNL089
90	160 FORMAT (1H0, ///T53, 'TOTAL NUMBER OF BODIES = ',I1,	PRNL090
91	1 1H0, ///T51, 'TOTAL NUMBER OF ELEMENTS = ',I1)	PRNL091
92	C	PRNL092
93	C	PRNL093
94	RETURN	PRNL094
95	END	PRNL095

```

1      SUBROUTINE QUASI (A,ND,MD,KD,NI,MM,NO,NAT,NW,LTAPE,PHSTAP,*)
2      C
3      C
4      C
5      C
6      C
7      C
8      C
9      C
10     C
11     C
12     C
13     C
14     C
15     C
16     C
17     C
18     C
19     C
20     C
21     C
22     C
23     C
24     C
25     C
26     C
27     C
28     C
29     C
30     C
31     C
32     C
33     C
34     C
35     C
36     C
37     C
38     C
39     C
40     C
41     C
42     C
43     C
44     C
45     C
46     C
47     C
48     C
49     C
50     C
51     C
52     C
53     C
54     C
55     C
56     C

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DIRECT MATRIX SOLUTION

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C*** **LTAPE IS THE TAPE THE L(I,J) MATRIX WILL BE PUT ON
C*** **RHSTAP IS THE TAPE THAT THE RIGHT HAND SIDES ARE INPUT ON
C*** **NATAPE IS A SCRATCH TAPE
C*** **THE TRIANGULAR MATRIX EXCEPT FOR THE LAST K ROWS WILL BE KEPT ON
C*** **TAPE MM
C*** **THE LAST K ROWS OF THE TRIANGULAR MATRIX WILL BE PUT ON THE
C*** **LTAPE BEHIND THE RHS MATRIX
COMPLEX A,SUM
DIMENSION A ( KD )
LOGICAL JPASS1
LOGICAL LASTRS
LOGICAL LAST
CALL TIMEVIAA1)
10 CONTINUE
REWIND LTAPE
NATAPE = NAT
REWIND NATAPE
N = ND
KORE = KD
C*** * RHSTAP = 0 IF THERE ARE NO RHS TO BE PROCESSED THIS RUN
IF(RHSTAP.NE.0)60 TO 20
MRHS = 0
GO TO 30
20 REWIND RHSTAP
READ(RHSTAP)MRHS
30 M = KORE / N - 1
MMAX = MIN(MRHS,M)
NPM = M + MMAX
IF( (3*NPM).GT. KORE )RETURN 1
M = 0
NPM = N
MT = MM
REWIND MT
MIN = NI
REWIND MIN
NOUT = NO
REWIND NOUT
MP1 = M + 1
NN = N
NEL = NPM
NLCNT = 0

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QUAS001
 QUAS002
 QUAS003
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 QUAS054
 QUAS055
 QUAS056

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57      C                                     QUAS057
58      C - - CALCULATE THE MAXIMUM NO. OF ROWS. *K*                                     QUAS058
59      C                                     QUAS059
60      DO K = (MORE - NEL) / NEL                                     QUAS060
61      C                                     QUAS061
62      C - - TEST TO SEE IF THE REST OF THE MATRIX WILL FIT IN CORE QUAS062
63      C                                     QUAS063
64      LAST = K .GE. MN                                     QUAS064
65      IF (.NOT. LAST) GO TO 50                                     QUAS065
66      K = MN                                     QUAS066
67      B = 3 + MMAX*2                                     QUAS067
68      C = 2 * (1 + MMAX - MORE)                                     QUAS068
69      KTEMP = (1 - B + SQRT(B**2 - 4 * C)) / 2                                     QUAS069
70      IF (KTEMP .GE. N160) GO TO 50                                     QUAS070
71      C                                     QUAS071
72      C*** * WE MUST REDUCE THE FINAL K                                     QUAS072
73      C                                     QUAS073
74      K = KTEMP                                     QUAS074
75      LAST = .FALSE.                                     QUAS075
76      C                                     QUAS076
77      C - - READ *K* ROWS OF THE AUGMENTED *A* MATRIX                                     QUAS077
78      C                                     QUAS078
79      DO NT = 0                                     QUAS079
80      DO 60 IB = 1, K                                     QUAS080
81      NS = NT + 1                                     QUAS081
82      NT = NT + NEL                                     QUAS082
83      60 CALL GETTININ, 1, NEL, A(NS), 1, A(2)                                     QUAS083
84      C                                     QUAS084
85      C - - CHECK TO SEE IF WE WERE UNLUCKY ENOUGH TO END UP WITH ONLY ONE ROW QUAS085
86      C                                     QUAS086
87      IF (K .EQ. 1) GO TO 130                                     QUAS087
88      C                                     QUAS088
89      C - - *K* IS GREATER THAN *J* SO WE CAN START THE TRIANGULARIZATION QUAS089
90      C                                     QUAS090
91      N161 = NEL + 1                                     QUAS091
92      NS = - NEL                                     QUAS092
93      N162 = N161 + 1                                     QUAS093
94      C                                     QUAS094
95      C - - FORM THE *TRAPEZOIDAL* ARRAY (8)                                     QUAS095
96      C                                     QUAS096
97      DO 70 IB = 2, K                                     QUAS097
98      NP = N162 - IB                                     QUAS098
99      NS = NS + N161                                     QUAS099
100     NT = NS                                     QUAS100
101     DO 70 IO = 1B, K                                     QUAS101
102     NT = NT + N161                                     QUAS102
103     MN = NT                                     QUAS103
104     NB = NS                                     QUAS104
105     A(NT) = A(NT) / A(NS)                                     QUAS105
106     DO 70 NF = 2, NP                                     QUAS106
107     MN = MN + 1                                     QUAS107
108     NB = NB + 1                                     QUAS108
109     70 A(MN) = A(MN) - A(NT) * A(NB)                                     QUAS109
110     C*** **WRITE PART OF THE MATRIX ON LTape (TRIANGULAR PART) QUAS110
111     WRITE (LTape)M                                     QUAS111
112     NLCNT = NLCNT + 1                                     QUAS112
113     LBEG = N161                                     QUAS113

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114      NM1 = M - 1
115      DO 80 IB = 1, NM1
116      LEND = LBEG + IB - 1
117      CALL SAVE(LTAPE, 1, IB, IB, A(LBEG), 1, AA2)
118      80 LBEG = LBEG + NM
119      C
120      C - - WRITE THE 'TRAPEZOIDAL' MATRIX ON TAPE
121      C
122      NT = 0
123      NP = NEL
124      NS = - NEL
125      DO 90 IO = 1, M
126      NS = NS + NELP1
127      NT = NT + NEL
128      CALL SAVE(NT, 2, NP, NP, A(NS), 1, AA2)
129      90 NP = NP - 1
130      IF (LAST) GO TO 13C
131      NP = NP - M
132      NS = MORE - NEL + 1
133      C
134      C - - READ ANOTHER ROW
135      C
136      DO 120 IO = 1, NP
137      CALL GET(MIN, 1, NEL, A(NS), 1, AA2)
138      C
139      C - - MODIFY THIS ROW BY THE 'TRAPEZOIDAL' ARRAY
140      C
141      NT = 1
142      MN = NS
143      DO 110 IB = 1, M
144      NB = NT
145      NF = MN + 1
146      A(MN) = A(MN) / A(NT)
147      DO 100 NN = NF, MORE
148      NB = NB + 1
149      100 A(NN) = A(NN) - A(PN) * A(NB)
150      MN = NF
151      110 NT = NT + NELP1
152      C
153      C - - WRITE THE MODIFIED ROW ON TAPE
154      C
155      C*** **WRITE REST OF L MATRIX ON LTAPE
156      MNM1 = MN - 1
157      NN1 = MNM1 - NS + 1
158      CALL SAVE(LTAPE, 1, NN1, NN1, A(NS), 1, AA2)
159      NN1 = MORE - MN + 1
160      120 CALL SAVE(NOUT, 1, NN1, NN1, A(PN), 1, AA2)
161      REWIND NOUT
162      REWIND NIN
163      C
164      C - - SWITCH THE TAPES
165      C
166      NT = NIN
167      NIN = NOUT
168      NOUT = NT
169      C
170      C - - RE-CALCULATE ROW LENGTH AND LOOP PACK

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QUAS114
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QUAS167
QUAS168
QUAS169
QUAS170

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171      C                                     QUAS171
172      NEL = NEL - K                                     QUAS172
173      NN = NEL - M                                     QUAS173
174      GO TO 40                                     QUAS174
175      C                                     QUAS175
176      C -- REWIND ALL TAPES                                     QUAS176
177      C                                     QUAS177
178      130 REWIND NIN                                     QUAS178
179      REWIND NOUT                                     QUAS179
180      C                                     QUAS180
181      140 N1 = MORE - K * M + 1                                     QUAS181
182      REWIND LTAPE                                     QUAS182
183      REWIND MT                                     QUAS183
184      *** **CALCULATE THE NUMBER OF COLUMNS TO BRING OFF OF THE RHS TAPE QUAS184
185      MTOTAL = 0                                     QUAS185
186      M = MMAX                                     QUAS186
187      IFIM .EQ. 0160 TO 520                                     QUAS187
188      *** **TOTAL IS THE TOTAL NUMBER OF RHS COLUMNS ALREADY BROUGHT IN QUAS188
189      150 MTOTAL = MTOTAL + M                                     QUAS189
190      LASTRS = MTOTAL - GE, MRHS                                     QUAS190
191      MTOTAL = MTOTAL - M                                     QUAS191
192      IF (LASTRS)M = MRHS - MTOTAL                                     QUAS192
193      MTOTAL = MTOTAL + M                                     QUAS193
194      *** **BRING IN M COLUMNS OF RHS                                     QUAS194
195      KINIT = MORE - (M*N)                                     QUAS195
196      KINIT = KINIT                                     QUAS196
197      NBEG = KINIT + 1                                     QUAS197
198      NEND = KINIT + M                                     QUAS198
199      DO 160 J = 1, M                                     QUAS199
200      CALL GETI(RHSTAP, 1, M, AINBEG), 1, AA2)                                     QUAS200
201      NBEG = NEND + 1                                     QUAS201
202      160 NEND = NEND + M                                     QUAS202
203      *** **BRING IN L(I,J) MATRIX AND APPLY IT TO RHS                                     QUAS203
204      NBEG = 1 + KINIT                                     QUAS204
205      NEND = 1 + (M-1) * N + KINIT                                     QUAS205
206      KSUM = 0                                     QUAS206
207      *** **DO TRIANGULAR SECTION OF L MATRIX                                     QUAS207
208      170 READ (LTAPE)M                                     QUAS208
209      *** **KSUM IS THE TOTAL NUMBER OF L ROWS THAT WILL                                     QUAS209
210      *** **BE READ AFTER THIS TRIANGULAR SECTION IS FINISHED                                     QUAS210
211      KSUM = KSUM + K                                     QUAS211
212      KM1 = K - 1                                     QUAS212
213      *** **NOTE THAT KM1 CAN'T BE 0 SINCE K CAN'T BE 1 AND STILL HAVE SOME QUAS213
214      *** **ON THE LTAPE                                     QUAS214
215      DO 200 I = 1, KM1                                     QUAS215
216      NBEG = NBEG + 1                                     QUAS216
217      NEND = NEND + 1                                     QUAS217
218      *** **READ 1 ROW OF L(I,J) FROM LTAPE---K-1 TIMES---EACH TIME QUAS218
219      *** **STARTING WITH L(1)                                     QUAS219
220      CALL GETI(LTAPE, 1, I, A, 1, AA2)                                     QUAS220
221      JCNT = -1                                     QUAS221
222      *** **REDUCE THE RHS BY GOING ACROSS A SOLUTION ROW WHICH                                     QUAS222
223      *** **ARE NOT IN CONSECUTIVE ORDER, BUT A(1), AIN+1), A(2N+1) ETC.) QUAS223
224      DO 190 NPP = NBEG, NEND, N                                     QUAS224
225      JCNT = JCNT + 1                                     QUAS225
226      SUM = 0.0                                     QUAS226
227      NROW = KINIT + ( JCNT * N )                                     QUAS227

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278 DO 180 MN=1,1 QUAS228
279 NROW = NROW + 1 QUAS229
280 180 SUM = SUM + (A(MN,MN)) QUAS230
281 190 A(MPP) = A(MPP) - SUM QUAS231
282 200 CONTINUE QUAS232
283 IF (KSUM - EQ - N) GO TO 240 QUAS233
284 C*** KSUM = N IF YOU HAVE READ ENTIRE MATRIX AND QUAS234
285 C*** THERE IS NO CONSTANT SECTION LEFT QUAS235
286 NREG = NREG QUAS236
287 NEND = NEND QUAS237
288 KSUMP1 = KSUM + 1 QUAS238
289 DO 230 I=KSUMP1,M QUAS239
290 NREG = NREG + 1 QUAS240
291 NEND = NEND + 1 QUAS241
292 CALL GETUJAPE, 1, K, A, 1, A2 QUAS242
293 JCNT = -1 QUAS243
294 C*** PARTIALLY REDUCE A RHS ACROSS A RHS ROW BY APPLYING N NUMBER QUAS244
295 C*** OF L(I,J) S QUAS245
296 DO 220 MPP = NREG, NEND, M QUAS246
297 JCNT = JCNT + 1 QUAS247
298 SUM = 0.0 QUAS248
299 NROW = NROW + (JCNT * M) QUAS249
300 DO 210 MN = 1, M QUAS250
301 NROW = NROW + 1 QUAS251
302 210 SUM = SUM + (A(MN) * A(MN)) QUAS252
303 220 A(MPP) = A(MPP) - SUM QUAS253
304 230 CONTINUE QUAS254
305 NREG = NREG + 1 QUAS255
306 NEND = NEND + 1 QUAS256
307 C*** KINIT IS HOW FAR DOWN A COLUMN OF RHS TO START MULTIPLYING BY QUAS257
308 C*** L(I,J) AT EACH PASS THROUGH QUAS258
309 KINIT = KINIT + M QUAS259
310 IF (KSUMP1 - L1 - N) GO TO 170 QUAS260
311 C*** IF KSUMP1 = N THERE ARE NO MORE L(I,J)'S LEFT QUAS261
312 C*** REWRITE OUT ALL BUT LAST N ROWS OF RHS IN ROW ORDER ON TAPE QUAS262
313 240 B = 4*M + 3 QUAS263
314 C = -2 * MORE QUAS264
315 K = 1 - B + SORT(B**2 - 4*C) / 2 QUAS265
316 IF (M - GT - ND) K = ND QUAS266
317 MF = K QUAS267
318 K1 = K - 1 QUAS268
319 KLEFT = M - MF + KINIT QUAS269
320 INITP1 = KINIT + 1 QUAS270
321 NEND = (M - 1) * M + KINIT QUAS271
322 DO 250 MPP = INITP1, KLEFT QUAS272
323 NEND = NEND + 1 QUAS273
324 250 WRITE(MTAP) (A(I,J), J=KINIT, NEND, M) QUAS274
325 REWIND MTAP QUAS275
326 C*** JPASS1 IS TRUE ON 1ST PASS THRU BACK SOLUTION QUAS276
327 JPASS1 = .TRUE. QUAS277
328 C*** REMAINING RHS IN CONTIGUOUS LOCATIONS BY COLUMNS QUAS278
329 FROM MORE - (M * MF) + 1 TO MORE QUAS279
330 NNEW = MORE - MF + 1 QUAS280
331 MM1 = M - 1 QUAS281
332 QUAS282
333 QUAS283
334 QUAS284

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285      C                                     QUAS285
286      C***   ***IF M = 1, THE ELTS OF THE 1 RHS COLUMN ARE ALREADY IN CONTIGUOUS QUAS286
287      C***   ***LOCATIONS QUAS287
288      C                                     QUAS288
289      IF (M.EQ.1) GO TO 265 QUAS289
290      DO 260 I = 1,MH1 QUAS290
291      NOLD = KORE - (I*M) + 1 QUAS291
292      DO 260 J = 1,KF QUAS292
293      NNEW = NNEW + 1 QUAS293
294      NOLD = NOLD - 1 QUAS294
295      A(NNEW) = A(NOLD) QUAS295
296      260 CONTINUE QUAS296
297      265 CONTINUE QUAS297
298      C                                     QUAS297
299      C***   ***NOW NNEW = KORE - (M*KF) + 1 QUAS298
300      C***   ***NOW NOLD = KORE - (M - 1) * M + 1 - KF QUAS299
301      C ***   SKIP 1ST PART OF TRAPEZOIDAL MATRIX + READ LAST K ROWS QUAS300
302      C***   ***ATTACH RHS TO IT SO THAT EVERYTHING IS IN CONSECUTIVE ORDER QUAS301
303      NREMAN = ND - K QUAS302
304      IF(NREMAN.EQ.0) GO TO 280 QUAS303
305      DO 270 I = 1,NREMAN QUAS304
306      270 READ(MT,1) IDUMMY QUAS305
307      280 NEND = 0 QUAS306
308      KCNT = K QUAS307
309      NNEW = NNEW - 1 QUAS308
310      C***   ***NOTE THAT K = KF WHICH IS ALREADY KNOWN IN CORE QUAS309
311      DO 290 JCNT = 1,K QUAS310
312      NBE6 = NEND + 1 QUAS311
313      CALL GETT(MT,4,KCNT,A(NBE6),1,AA2) QUAS312
314      KCNT = KCNT - 1 QUAS313
315      NEND = NBE6 + KCNT QUAS314
316      NNEW = NNEW + 1 QUAS315
317      KEND = (MH1 * KF) + NNEW QUAS316
318      DO 290 NPP=NNEW,KEND,KF QUAS317
319      NEND = NEND + 1 QUAS318
320      290 A(NEND) = A(NPP) QUAS319
321      REWIND LTape QUAS320
322      REWIND MT QUAS321
323      C                                     QUAS322
324      C - - THERE, NOW WE CAN START THE BACK-SOLUTION QUAS323
325      C * * NOTE..THE FIRST AVAILABLE LOCATION FOR THE SOLUTIONS IS A(1) QUAS324
326      C                                     QUAS325
327      C                                     QUAS326
328      C***   ***NL IS THE LAST SUBSCRIPT + 1 OF THE TRAPEZOIDAL A MATRIX THAT QUAS327
329      C***   ***CORE QUAS328
330      C                                     QUAS329
331      NL = NEND + 1 QUAS330
332      NREM = N QUAS331
333      NPM = N + M QUAS332
334      NEL = NPM QUAS333
335      MP1 = M + 1 QUAS334
336      LAST = K.EQ.M QUAS335
337      NPASS = 0 QUAS336
338      C                                     QUAS337
339      C - - SOLVE FOR THE ANSWERS CORRESPONDING TO 'K' ROWS QUAS338
340      C                                     QUAS339
341      300 MH1 = K - 1 QUAS340

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342	MP1 = M + 1	QUAS341
343	NS = NL - MP1	QUAS342
344	NPASS = NPASS + 1	QUAS343
345	DO 330 MN = 1, M	QUAS344
346	NF = NS + MN	QUAS345
347	A(NF) = A(NF) / A(NS)	QUAS346
348	NT = NS	QUAS347
349	IF (KMI .EQ. D) GO TO 330	QUAS348
350	DO 320 IB = 1, KMI	QUAS349
351	NF = NF - IB - M	QUAS350
352	NT = NT - MP1 - IB	QUAS351
353	SUM = 0.0	QUAS352
354	NP = NF	QUAS353
355	N2 = MP1 + IB	QUAS354
356	DO 310 IO = 1, IB	QUAS355
357	MN = NT + IO	QUAS356
358	NP = NP + N2 - IO	QUAS357
359	310 SUM = SUM + A(MN) * A(NP)	QUAS358
360	320 A(NF) = (A(NF) - SUM) / A(NT)	QUAS359
361	330 CONTINUE	QUAS360
362	C	QUAS361
363	C - - MOVE THE SOLUTIONS TO CONTIGUOUS LOCATIONS STARTING AT A(N1)	QUAS362
364	C	QUAS363
365	N1 = MORE + 1	QUAS364
366	DO 350 MN = 1, M	QUAS365
367	DO 340 MN = 1, M	QUAS366
368	NL = NL - 1	QUAS367
369	N1 = N1 - 1	QUAS368
370	340 A(N1) = A(NL)	QUAS369
371	350 NL = NL - MN	QUAS370
372	C	QUAS371
373	C - - WRITE THE SOLUTIONS ON TAPE	QUAS372
374	C	QUAS373
375	WRITE (NIN) K	QUAS374
376	NS = N1 - 1	QUAS375
377	DO 360 MN = 1, M	QUAS376
378	NT = NS + MN	QUAS377
379	360 WRITE (NIN) (A(IO), IO = NT, MORE, M)	QUAS378
380	C	QUAS379
381	C - - TEST IF THIS IS THE LAST PASS	QUAS380
382	C	QUAS381
383	IF (LAST) GO TO 470	QUAS382
384	C	QUAS383
385	C - - WE MUST NOW MODIFY THE TRIANGULAR MATRIX TO REFLECT THE EFFECT OF	QUAS384
386	THE SOLUTIONS OBTAINED SO FAR (EQ 21)	QUAS385
387	C * * NOTE..LOCATIONS A(1) TO A(N1-1) ARE NOW FREE TO USE	QUAS386
388	C	QUAS387
389	C - - CALCULATE THE NEXT VALUES OF 'NEL' AND 'NREM'	QUAS388
390	C	QUAS389
391	NELOLD = NEL	QUAS390
392	NOLD = K	QUAS391
393	NEL = NEL - K	QUAS392
394	NREM = NREM - K	QUAS393
395	C	QUAS394
396	NFOM = NREM - K + 1	QUAS395
397	IF (K .LT. NREM) GO TO 370	QUAS396
398	LAST = .TRUE.	QUAS397

399	NROW = 1	QUAS398
400	M = NREM	QUAS399
401	370 NS = 1	QUAS400
402	MT = MELOLD + 1	QUAS401
403	C	QUAS402
404	C -- READ IN THE ROWS TO BE MODIFIED	QUAS403
405	C	QUAS404
406	DO 450 IB = 1, NREM	QUAS405
407	NT = MT - 1	QUAS406
408	IF (IB .LE. NROW) GO TO 380	QUAS407
409	NS = NS + NM	QUAS408
410	MT = MT + NM	QUAS409
411	380 IF (.NOT. JPASS1) GO TO 390	QUAS410
412	NBEG = MT - P + 1	QUAS411
413	C*** **READ RHS FROM NATAPE	QUAS412
414	CALL GETTINATAPE, 1, M, AINBEG, 1, AA2)	QUAS413
415	NT = MT - M	QUAS414
416	390 CALL GETTINI, 2, NM, AINS), 1, AA2)	QUAS415
417	IF (.NOT. JPASS1) GO TO 400	QUAS416
418	NT = NT + M	QUAS417
419	NM = NM + M	QUAS418
420	400 NP = N1 - 1	QUAS419
421	NF = NT - M - NM1	QUAS420
422	NM = NM - KOLD	QUAS421
423	DO 420 MN = 1, M	QUAS422
424	N2 = NF	QUAS423
425	NA = NP + NM	QUAS424
426	NB = NA	QUAS425
427	SUM = 0.0	QUAS426
428	DO 410 IO = 1, KOLD	QUAS427
429	SUM = SUM + A(IN2) * A(INA)	QUAS428
430	N2 = N2 + 1	QUAS429
431	410 NA = NA + M	QUAS430
432	N2 = N2 + NM - 1	QUAS431
433	420 A(IN2) = A(IN2) - SUM	QUAS432
434	C	QUAS433
435	C -- WRITE THE MODIFIED ROW ON TAPE OR CONDENSE THE ROW	QUAS434
436	C	QUAS435
437	NL = NT - M + 1	QUAS436
438	IF (IB .GE. NROW) GO TO 430	QUAS437
439	NF = NL - NP1	QUAS438
440	NN1 = NF - NS + 1	QUAS439
441	NN2 = NT - NL + 1	QUAS440
442	CALL SAVE(ROUT, 4, NM, NN1, AINS), NN2, A(INL))	QUAS441
443	GO TO 450	QUAS442
444	430 NF = NL - KOLD	QUAS443
445	DO 440 MN = NL, NT	QUAS444
446	A(INF) = A(MN)	QUAS445
447	440 NF = NF + 1	QUAS446
448	450 CONTINUE	QUAS447
449	C*** **IF 1ST TIME THRU BACK SOLN. SWITCH TAPES SO THAT PT WHICH HAS THE	QUAS448
450	C*** **ORIGINAL TRAPEZOIDAL MATRIX ON IT BECOME NATAPE AND IS NOT TO	QUAS449
451	C*** **TAPE PART IN ALTERNATING SHRINKING MATRICES. NATAPE BECOMES MT	QUAS450
452	C*** **AND THIS NOW DOES THE ALTERNATING WITH NOUT.	QUAS451
453	IF (.NOT. JPASS1) GO TO 460	QUAS452
454	NTEMP = MT	QUAS453
455	MT = NATAPE	QUAS454

456	NATAPE = NTEMP	QUAS455
457	JPASS1 = .FALSE.	QUAS456
458	REWIND NATAPE	QUAS457
459	460 REWIND MT	QUAS458
460	REWIND NOUT	QUAS459
461	C	QUAS460
462	C - - SWITCH THE TAPES	QUAS461
463	C	QUAS462
464	MT = MT	QUAS463
465	MT = NOUT	QUAS464
466	NOUT = MT	QUAS465
467	C	QUAS466
468	C - - LOOP BACK THRU THE SOLUTION	QUAS467
469	C	QUAS468
470	NL = NF	QUAS469
471	GO TO 300	QUAS470
472	C	QUAS471
473	C - - START TO WRAP IT UP	QUAS472
474	C	QUAS473
475	470 REWIND MIN	QUAS474
476	N2 = N	QUAS475
477	C	QUAS476
478	C * * NOTE.. AT THIS POINT ALL LOCATIONS A(1) THRU A(NORE) ARE FREE	QUAS477
479	C	QUAS478
480	DO 490 IB = 1, NPASS	QUAS479
481	READ (MIN) N	QUAS480
482	N1 = N2 - K + 1	QUAS481
483	NS = N1	QUAS482
484	NT = N2	QUAS483
485	C	QUAS484
486	C - - READ IN THE SOLUTIONS	QUAS485
487	C	QUAS486
488	DO 480 TO = 1, N	QUAS487
489	CALL GET(MIN, 1, N, A(NS), 1, A(2))	QUAS488
490	NT = NT + N	QUAS489
491	480 NS = NS + N	QUAS490
492	490 N2 = N1 - 1	QUAS491
493	C	QUAS492
494	C --- REWIND ALL INPUT TAPES	QUAS493
495	REWIND MIN	QUAS494
496	REWIND MT	QUAS495
497	REWIND NOUT	QUAS496
498	C - - WRITE THE SOLUTIONS ON TAPE	QUAS497
499	C	QUAS498
500	NT = 0	QUAS499
501	DO 500 TO = 1, N	QUAS500
502	NS = NT + 1	QUAS501
503	MT = NT + N	QUAS502
504	500 CALL SAVE(MN, 1, N, N, A(NS), 1, A(2))	QUAS503
505	C *** IF TAPE WAS NEVER SWITCHED IT WOULD BE FOOLISH TO SWITCH BACK	QUAS504
506	IF(JPASS1)GO TO 51C	QUAS505
507	C	QUAS506
508	C*** ***SWITCH TAPES	QUAS507
509	C*** ***BACK SO THAT MT WILL CONTAIN THE TRAPEZOIDAL MATRIX	QUAS508
510	C*** ***NATAPE WILL HAVE NOTHING USEFUL ON IT.	QUAS509
511	NTEMP = NATAPE	QUAS510
512	NATAPE = MT	QUAS511

513	MT = NTEMP	QUASS12
514	REWIND MATAPE	QUASS13
515	510 IF (.NOT. LASTRS) GO TO 150	QUASS14
516	520 REWIND LTAPE	QUASS15
517	REWIND MT	QUASS16
518	KRED = 0	QUASS17
519	DO 540 I=1,MLCNT	QUASS18
520	READ(ILTAPE)KREAD	QUASS19
521	530 CONTINUE	QUASS20
522	KRED = KRED + KREAD	QUASS21
523	KREAD = KREAD + (N - KRED - 1)	QUASS22
524	DO 540 LREAD=1,KREAD	QUASS23
525	540 READ(ILTAPE)	QUASS24
526	DO 550 MROW = 1,ND	QUASS25
527	CALL GETTIMT, 2, ICNT, A11, 1, AA2)	QUASS26
528	550 CALL SAVEILTAPE, , ICNT, ICNT, A, 1, AA2)	QUASS27
529	C	QUASS28
530	C *** REWIND ALL FILES EXCEPT THE OUTPUT FILE NM	QUASS29
531	REWIND LTAPE	QUASS30
532	REWIND MI	QUASS31
533	REWIND MM	QUASS32
534	REWIND MO	QUASS33
535	REWIND MAT	QUASS34
536	IF (RHSTAP .NE. 0) REWIND RHSTAP	QUASS35
537	CALL TIMEV(AA2)	QUASS36
538	MD = MTOTAL	QUASS37
539	BB = (AA2 - AA1) / 60.	QUASS38
540	WRITE(6,560)N,N,MTOTAL,BB	QUASS39
541	560 FORMAT (4HDTHE IS, 2H X IS, 12H MATRIX WITH IN, 35+ RIGHT SIDES WA	QUASS40
542	IS SOLVED DIRECTLY IN F8.3, 9H MINUTES.)	QUASS41
543	570 CONTINUE	QUASS42
544	RETURN	QUASS43
545	END	QUASS44

1	FUNCTION	RMAX (NL, X, Y, TMAX)	
2	C		RMAX001
3	DIMENSION	X(NL), Y(NL)	RMAX002
4	C		RMAX003
5	ID = 1		RMAX004
6	I1 = NL/2		RMAX005
7	I3 = I1 + ID		RMAX006
8	R1 = SQRT((X(I1)-X(I3))**2 + (Y(I1)-Y(I3))**2)		RMAX007
9	R2 = SQRT((X(I3)-X(I1))**2 + (Y(I3)-Y(I1))**2)		RMAX008
10	IF (R2 .GT. R1) GO TO 20		RMAX009
11	C		RMAX010
12	R3 = R1		RMAX011
13	I3 = I1		RMAX012
14	ID = -ID		RMAX013
15	C		RMAX014
16	10 R1 = R2		RMAX015
17	R2 = R3		RMAX016
18	20 I3 = I3 + ID		RMAX017
19	R3 = SQRT((X(I3)-X(I1))**2 + (Y(I3)-Y(I1))**2)		RMAX018
20	IF (R3 .GT. R2) GO TO 10		RMAX019
21	C		RMAX020
22	C CALCULATE ANGLES		RMAX021
23	I2 = I3 - ID		RMAX022
24	I1 = I2 - ID		RMAX023
25	T1 = ARSIN((Y(I1)-Y(I3))/R1)		RMAX024
26	T2 = ARSIN((Y(I2)-Y(I3))/R2)		RMAX025
27	T3 = ARSIN((Y(I3)-Y(I1))/R3)		RMAX026
28	C		RMAX027
29	C CALCULATE MAXIMUM RADIUS (CHORD)		RMAX028
30	T2 = T2 - T1		RMAX029
31	T3 = T3 - T1		RMAX030
32	A3 = T2*T3*(T3 - T2)		RMAX031
33	R2 = R2 - R1		RMAX032
34	R3 = R3 - R1		RMAX033
35	A1 = (R3*T2 - R2*T3)/A3		RMAX034
36	A2 = (R2*T3**2 - R3*T2**2)/A3		RMAX035
37	C		RMAX036
38	C		RMAX037
39	RMAX = -0.25*A2**2/A1 + R1		RMAX038
40	TMAX = -0.5*A2/A1 + T1		RMAX039
41	C		RMAX040
42	RETURN		RMAX041
43	END		RMAX042
			RMAX043

1	SUBROUTINE SAVE(ILL, II, N, M1, A1, M2, A2)	SAVE001
2	C	SAVE002
3	DIMENSION A1(11), A2(12)	SAVE003
4	C	SAVE004
5	GO TO (10, 20, 30, 40), II	SAVE005
6	C	SAVE006
7	C WRITE(10) A1	SAVE007
8	10 RETURN	SAVE008
9	C	SAVE009
10	C	SAVE010
11	C WRITE(M, A1)	SAVE011
12	20 WRITE(10) M, A1	SAVE012
13	RETURN	SAVE013
14	C	SAVE014
15	C WRITE(A1, A2)	SAVE015
16	30 WRITE(10) A1, A2	SAVE016
17	RETURN	SAVE017
18	C	SAVE018
19	C WRITE(M, A1, AND A2)	SAVE019
20	40 WRITE(10) M, A1, A2	SAVE020
21	RETURN	SAVE021
22	END	SAVE022

```

1      SUBROUTINE SOLVE(N, M, ISIZE, ISOL)
2      C
3      C
4      C THIS ROUTINE OBTAINS THE SIGMA SOLUTIONS
5      C FROM EITHER QUASI OR FULL SOL
6      C
7      C WKAREA MUST BE DIMENSIONED EQUAL TO IMAX.
8      C TO CHANGE THIS, BOTH MUST BE COMPILED.
9      C THE CURRENT ARRAY SIZE IS GIVEN BY
10     C ISIZE, AN INPUT VARIABLE.
11     C
12     C
13     COMMON /SPACER/ WKAREA(11913)
14     C
15     DIMENSION A(101,101), SIG(101,12)
16     C
17     EQUIVALENCE (A(1,1),WKAREA(1)), (SIG(1,1),WKAREA(10202))
18     C
19     COMMON /FILEID/ IFILE1, IFILE2, IFILE3, IFILE4, IFILE5,
20     1 IFILE6, IFILE7, IFILE8, IFILE9, IFIL10,
21     2 IFIL11, IFIL12, IFIL13, IFIL14, IFIL15,
22     3 IFIL16, IFIL17, IFIL18, IFIL19, IFIL20
23     C
24     C
25     IF (ISOL.EQ. 2) GO TO 70
26     MM = M
27     CALL TIMEV(T)
28     10 IF (ISOL.NE. 0) GO TO 30
29     C
30     15 CONTINUE
31     C
32     WRITE(6,20) T
33     20 FORMAT(1HD, 'SOLVIT TIME = ', F9.3, ' SECONDS.')
34     C
35     CALL SOLVIT IMWAREA, N, MM, ISIZE, IFILE9, IFILE1,
36     1 IFILE2, IFIL14, (50)
37     CALL TIMEV(T)
38     WRITE(6,20) T
39     RETURN
40     C
41     C
42     30 WRITE(6,40) T
43     CALL QUASIMWAREA, N, MM, ISIZE, IFILE9, IFILE1, IFILE2,
44     1 IFILE3, IFIL14, IFIL15, IFILE4, (50)
45     CALL TIMEV(T)
46     WRITE(6,40) T
47     40 FORMAT(1HD, 'QUASI TIME = ', F9.3, 'SECONDS.')
48     RETURN
49     C
50     50 WRITE (6,60) ISIZE, IMAX
51     60 FORMAT (1H), 'ISIZE = ', I5, ' TOO SMALL. SET EQUAL TO IMAX (',
52     1 I5, ')')
53     C
54     C
55     IF (ISIZE.GE. IMAX) GO TO 140
56     ISIZE = IMAX

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57      GO TO 10                                SOLV057
58      C                                        SOLV058
59      C                                        SOLV059
60      C                                        SOLV060
61      C THIS IS A DUMMY ROUTINE TO USE MIS1 (MIS2) SOLV061
62      C FOR SOLVING THE VORTEX DISTRIBUTIONS SOLV062
63      C                                        SOLV063
64      C A-ARRAY STORED ON IFIL10 (THE FIRST OF TWO RECORDS) SOLV064
65      C                                        SOLV065
66      C RMS STORED ON IFILE4 SOLV066
67      C                                        SOLV067
68      C SIGMAS SAVED ON IF14 SOLV068
69      C                                        SOLV069
70      C                                        SOLV070
71      70 NMAX = 101 SOLV071
72      MD = 12 SOLV071
73      IF14= IFIL14 SOLV072
74      REWIND IFIL10 SOLV073
75      REWIND IFILE4 SOLV074
76      REWIND IF14 SOLV075
77      IF (N .LE. NMAX) GO TO 90 SOLV076
78      WRITE(6,80) N, NMAX SOLV077
79      80 FORMAT(1HD,'THE SIZE OF ARRAY (',15,') EXCEEDS LIMIT OF ',I4) SOLV078
80      GO TO 15 SOLV079
81      C SOLV080
82      90 CONTINUE SOLV081
83      C SOLV082
84      D = 1.0 SOLV083
85      C SOLV084
86      C READ IN A-ARRAY SOLV085
87      DO 100 I = 1,N SOLV086
88      READ(IFIL10) (A(I,J),J=1,N) SOLV087
89      100 READ (IFIL10) SOLV088
90      C SOLV089
91      C READ IN RMS IN SIG ARRAY SOLV090
92      READ (IFILE4) MRMS SOLV090
93      DO 110 K = 1,M SOLV091
94      110 READ(IFILE4) (SIG(I,K),I=1,N) SOLV092
95      C SOLV093
96      C SOLV094
97      CALL MIS2(A, N, NMAX, SIG, M, MD, NERR, D) SOLV095
98      C SOLV096
99      WRITE (6,120) NERR SOLV097
100     120 FORMAT(1HD,'ON RETURN FROM MIS2, NERR = ',I2) SOLV098
101     C SOLV099
102     C SAVE SIGMAS SOLV100
103     DO 130 J = 1,M SOLV101
104     130 WRITE(IF14) (SIG(I,J), I=1,N) SOLV102
105     RETURN SOLV103
106     C SOLV104
107     C SOLV105
108     140 STOP SOLV106
109     END SOLV107

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1      SUBROUTINE SOLVIT (A, ND, MD, KC, NJ, MM, NO, NW, *)
2      C
3      C
4      C      **** */ ***** */
5      C      * * */ * * * */
6      C      **** */ * * * */
7      C      * * */ * * * */
8      C      * /*** * * **** /*** *
9      C
10     C      DIRECT MATRIX SOLUTION
11     C
12     C      WRITTEN BY J. L. HESS * PROGRAMMED BY T. M. RIDDELL
13     C
14     C      DIMENSION A (ND )
15     C
16     C      LOGICAL LAST
17     C
18     C      CALL TIMEV(AA1)
19     C      IF (AA1 .EQ. D.) CALL TSETV
20     C      N = ND
21     C      M = MD
22     C      MORE = ND
23     C      NPM = N * M
24     C      IF (MAX(3 * NPM, M * N) .GT. NORF) RETURN 1
25     C      NT = MM
26     C      REWIND NT
27     C      MN = NJ
28     C      REWIND MN
29     C      NOUT = NO
30     C      REWIND NOUT
31     C      MPI = M + 1
32     C      NN = N
33     C      NEL = NPM
34     C
35     C      - - CALCULATE THE MAXIMUM NO. OF ROWS, *K*
36     C
37     C      10 K = (MORE - NEL) / NEL
38     C
39     C      - - TEST TO SEE IF THE REST OF THE MATRIX WILL FIT IN CORE
40     C
41     C      LAST = K .GE. MM
42     C      IF (LAST) K = MM
43     C
44     C      - - READ *K* ROWS OF THE AUGMENTED *A* MATRIX
45     C
46     C      20 NT = 0
47     C      DO 30 ID = 1, K
48     C      NS = NT + 1
49     C      NT = NT + NEL
50     C      30 CALL GETTIN(1, NEL, A(NS), 1, AA2)
51     C
52     C      - - CHECK TO SEE IF WE WERE UNLUCKY ENOUGH TO END UP WITH ONLY ONE ROW
53     C
54     C      IF (K .EQ. 1) GO TO 90
55     C
56     C      - *K* IS GREATER THAN 1 SO WE CAN START THE TRIANGULARIZATION

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57      C
58      NHELP1 = NEL + 1
59      NS = - NEL
60      NHELP2 = NHELP1 + 1
61      C
62      C -- FORM THE 'TRAPEZOIDAL' ARRAY (B)
63      C
64      DO 40 IB = 2, K
65      NP = NHELP2 - IB
66      NS = NS + NHELP1
67      NT = NS
68      DO 40 IO = IB, K
69      NT = NT + NEL
70      MN = NT
71      NB = NS
72      A(NT) = (-A(NT)) / A(NS)
73      DO 40 MF = 2, NP
74      MN = MN + 1
75      NB = NB + 1
76      40 A(MN) = A(MN) + A(NT) + A(NB)
77      IF (LAST) GO TO 90
78      C
79      C -- WRITE THE 'TRAPEZOIDAL' MATRIX ON TAPE
80      C
81      NT = 0
82      NP = NEL
83      NS = - NEL
84      DO 50 IO = 1, K
85      NS = NS + NHELP1
86      NT = NT + NEL
87      CALL SAVE(NT, 2, NP, NP, A(NS), 1, AA2)
88      50 NP = NP - 1
89      NP = NP - K
90      NS = KORE - NEL + 1
91      C
92      C -- READ ANOTHER ROW
93      C
94      DO 60 IO = 1, NP
95      CALL GETT(IN, 1, NEL, A(NS), 1, AA2)
96      C
97      C -- MODIFY THIS ROW BY THE 'TRAPEZOIDAL' ARRAY
98      C
99      NT = 1
100     MN = NS
101     DO 70 IB = 1, K
102     NB = NT
103     NF = MN + 1
104     A(MN) = (-A(MN)) / A(NT)
105     DO 60 NM = NF, KORE
106     NB = NB + 1
107     60 A(MN) = A(MN) + A(MN) + A(NB)
108     MN = NF
109     70 NT = NT + NHELP1
110     C
111     C -- WRITE THE MODIFIED ROW ON TAPE
112     C
113     NN1 = KORE - MN + 1

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114	80 CALL SAVEINOUT, 1, NM1, NM1, A(PN), 1, A(2)	SLVT114
115	REWIND NOUT	SLVT115
116	REWIND NIN	SLVT116
117	C	SLVT117
118	C - - SWITCH THE TAPES	SLVT118
119	C	SLVT119
120	NT = NIN	SLVT120
121	NIN = NOUT	SLVT121
122	NOUT = NT	SLVT122
123	C	SLVT123
124	C - - RE-CALCULATE ROW LENGTH AND LOOP RACK	SLVT124
125	C	SLVT125
126	NEL = NEL - N	SLVT126
127	NM = NEL - M	SLVT127
128	GO TO 10	SLVT128
129	C	SLVT129
130	C - - REWIND ALL TAPES	SLVT130
131	C	SLVT131
132	90 REWIND MT	SLVT132
133	REWIND NIN	SLVT133
134	REWIND NOUT	SLVT134
135	C	SLVT135
136	C - - CONDENSE THE MATRIX	SLVT136
137	C	SLVT137
138	NM = NEL	SLVT138
139	NL = NEL + 1	SLVT139
140	IF (K .EQ. 1) GO TO 110	SLVT140
141	NS = 1	SLVT141
142	NT = NEL	SLVT142
143	DO 100 IB = 2, K	SLVT143
144	NS = NS + N(EL)	SLVT144
145	NT = NT + NEL	SLVT145
146	DO 100 IO = NS, NT	SLVT146
147	A(NL) = A(IO)	SLVT147
148	100 NL = NL + 1	SLVT148
149	110 M1 = KORE - K + M + 1	SLVT149
150	C	SLVT150
151	C - - THERE, NOW WE CAN START THE BACK-SOLUTION	SLVT151
152	C * * NOTE..THE FIRST AVAILABLE LOCATION FOR THE SOLUTIONS IS A(N1)	SLVT152
153	C	SLVT153
154	NREM = N	SLVT154
155	NEL = NPM	SLVT155
156	LAST = K .EQ. N	SLVT156
157	NPASS = 0	SLVT157
158	C	SLVT158
159	C - - SOLVE FOR THE ANSWERS CORRESPONDING TO *K* ROWS	SLVT159
160	C	SLVT160
161	120 KM1 = K - 1	SLVT161
162	KP1 = K + 1	SLVT162
163	NS = NL - MP1	SLVT163
164	NPASS = NPASS + 1	SLVT164
165	DO 150 MM = 1, M	SLVT165
166	NF = NS + MM	SLVT166
167	A(NF) = A(NF) / A(NS)	SLVT167
168	NT = NS	SLVT168
169	IF (KM1 .EQ. 0) GO TO 150	SLVT169
170	DO 140 TB = 1, KM1	SLVT170

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171      NF = NF - IR - M
172      NT = NT - MP1 - IP
173      SUM = 0.0
174      NP = NF
175      N2 = MP1 + IP
176      DO 130 IO = 1, IR
177      NN = NT + IO
178      NP = NP + N2 - IO
179      130 SUM = SUM + A(NN) + A(NP)
180      140 A(NF) = (A(NF) - SUM) / A(NT)
181      150 CONTINUE
182      C
183      C - - MOVE THE SOLUTIONS TO CONTIGUOUS LOCATIONS STARTING AT A(N1)
184      C
185      N1 = KORE + 1
186      DO 170 NN = 1, M
187      DO 160 MM = 1, M
188      NL = NL - 1
189      N1 = N1 - 1
190      160 A(N1) = A(NL)
191      170 NL = NL - NN
192      C
193      C - - WRITE THE SOLUTIONS ON TAPE
194      C
195      WRITE (NIN) K
196      NS = N1 - 1
197      DO 180 MM = 1, M
198      NT = NS + MM
199      180 WRITE (NIN) (A(NT), IO = NT, KORE, M)
200      C
201      C - - TEST IF THIS IS THE LAST PASS
202      C
203      IF (LAST) GO TO 26C
204      C
205      C - - WE MUST NOW MODIFY THE TRIANGULAR MATRIX TO REFLECT THE EFFECT OF
206      C THE SOLUTIONS OBTAINED SO FAR (EO 21)
207      C * * NOTE..LOCATIONS A(1) TO A(N1-1) ARE NOW FREE TO USE
208      C
209      C - - CALCULATE THE NEXT VALUES OF *NEL* AND *NREM*
210      C
211      NELOLD = NEL
212      KOLD = K
213      NEL = NEL - K
214      NREM = NREM - K
215      C
216      C**** CALCULATE NEW K. B AND C (REAL) WILL ALWAYS BE INTEGERS.
217      C K WILL BE CALCULATED REAL AND TRUNCATED - - GOOD.
218      C
219      B = 1 + 2*M
220      C = 2*(KOLD*(M+1) - KORE)
221      K = (1-B + SQRT(B**2 - 4*C))/2.0
222      NROW = NREM - K + 1
223      IF (K .LT. NREM) GO TO 190
224      LAST = .TRUE.
225      NROW = 1
226      K = NREM
227      190 NS = 1

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228	MT = NELOLD + 1	SLVT228
229	C	SLVT229
230	C - - READ IN THE ROWS TO BE MODIFIED	SLVT230
231	C	SLVT231
232	DO 250 IB = 1, NREP	SLVT232
233	NT = NT - 1	SLVT233
234	IF (IB .LE. NROW) GO TO 200	SLVT234
235	NS = NS + NN	SLVT235
236	NT = NT + NN	SLVT236
237	200 CALL GETT(MT, 2, NN, AINS), 1, AA2)	SLVT237
238	NP = N1 - 1	SLVT238
239	NK = NT - M - NN1	SLVT239
240	NN = NN - KOLD	SLVT240
241	DO 220 MM = 1, M	SLVT241
242	N2 = NF	SLVT242
243	NA = NP + NN	SLVT243
244	NB = NA	SLVT244
245	SUM = 0.0	SLVT245
246	DO 210 IO = 1, KOLD	SLVT246
247	SUM = SUM + A(IN2) * A(NA)	SLVT247
248	N2 = N2 + 1	SLVT248
249	210 NA = NA + M	SLVT249
250	N2 = N2 + NN - 1	SLVT250
251	220 A(IN2) = A(IN2) - SUM	SLVT251
252	C	SLVT252
253	C - - WRITE THE MODIFIED ROW ON TAPE OR CONDENSE THE ROW	SLVT253
254	C	SLVT254
255	NL = NT - M + 1	SLVT255
256	IF (IB .GE. NROW) GO TO 230	SLVT256
257	NF = NL - NP1	SLVT257
258	NN1 = NF - NS + 1	SLVT258
259	NN2 = NT - NL + 1	SLVT259
260	CALL SAVE(MOUT, 4, NN, NN1, AINS), NN2, A(INL))	SLVT260
261	GO TO 250	SLVT261
262	230 NF = NL - KOLD	SLVT262
263	DO 240 MM = NL, NT	SLVT263
264	A(INF) = A(MM)	SLVT264
265	240 NF = NF + 1	SLVT265
266	250 CONTINUE	SLVT266
267	REWIND MT	SLVT267
268	REWIND MOUT	SLVT268
269	C	SLVT269
270	C - - SWITCH THE TAPES	SLVT270
271	C	SLVT271
272	NT = MT	SLVT272
273	MT = MOUT	SLVT273
274	MOUT = NT	SLVT274
275	C	SLVT275
276	C - - LOOP BACK THRU THE SOLUTION	SLVT276
277	C	SLVT277
278	NL = NF	SLVT278
279	GO TO 120	SLVT279
280	C	SLVT280
281	C - - START TO WRAP IT UP	SLVT281
282	C	SLVT282
283	260 REWIND NIN	SLVT283
284	N2 = N	SLVT284
285	C	SLVT285
286	C * * NOTE * * AT THIS POINT ALL LOCATIONS A(1) THRU A(KORE) ARE FREE	SLVT286
287	C	SLVT287

290		$N1 = M2 - N + 1$	SLVT290
291		$NS = N1$	SLVT291
292		$NT = M2$	SLVT292
293	C		SLVT293
294	C	-- READ IN THE SOLUTIONS	SLVT294
295	C		SLVT295
296		DO 270 IO = 1, M	SLVT296
297		$NM = NT - NS + 1$	SLVT297
298		CALL GETTINR, 1, NM, AINS, 1, AA2)	SLVT298
299		$NT = NT + N$	SLVT299
300		270 $NS = NS + N$	SLVT300
301		280 $N2 = N1 - 1$	SLVT301
302	C		SLVT302
303	C	-- WRITE THE SOLUTIONS ON TAPE	SLVT303
304	C		SLVT304
305		$NT = 0$	SLVT305
306		DO 290 IO = 1, M	SLVT306
307		$NS = NT + 1$	SLVT307
308		$NT = NT + N$	SLVT308
309		290 CALL SAVE(NM, 1, N, N, AINS), 1, AA2)	SLVT309
310	C		SLVT310
311		CALL TIMEV(AA2)	SLVT311
312		$BB = (AA2 - AA1) / 60.$	SLVT312
313		WRITE (6,300) N, N, N, BB	SLVT313
314		300 FORMAT (4H0THE 15, 2H X 15, 12H MATRIX WITH 14, 35+ RIGHT SIDES WAS	SLVT314
315		15 SOLVED DIRECTLY IN F8.3, 9H MINUTES.)	SLVT315
316		RETURN	SLVT316
317		END	SLVT317

1	SUBROUTINE TIMEVAT)	185
2	T=0.0	
3	RETURN	
4	END	

1	SUBROUTINE YSETV
2	RETURN
3	END

1	SUBROUTINE TYPE(IIGOOD, IBAO)	TYPE001
2	C	TYPE002
3	C	TYPE003
4	WRITE(6,10) IIGOOD, IBAO	TYPE004
5	10 FORMAT (1H1, 'AN ATTEMPT HAS BEEN MADE TO READ A TYPE ',I2,	TYPE005
6	1 ' CARD, HOWEVER A TYPE ',I2, ' CARD WAS FOUND.*/1MO,	TYPE006
7	2 'CHECK OVER THE INPUT DATA CARD SEQUENCE FOR ',	TYPE007
8	3 'COMPATIBILITY WITH DESIRED OPTIONS.')	TYPE008
9	C	TYPE009
10	C	TYPE010
11	20 WRITE (6,30)	TYPE011
12	30 FORMAT (1MO, 'BECAUSE OF THE ABOVE ERROR, THIS RUN IS TERMINATED*)	TYPE012
13	C	TYPE013
14	STOP	TYPE014
15	END	TYPE015

```

1      SUBROUTINE VXYOFF(M, N, NO, X, Y)
2      C
3      DIMENSION SIG(500), X(1), Y(1), A(500), B(500)
4      1      , VX(12), VY(12), VXT(12), VYM(12)
5      C
6      COMMON/ELD/   XO(500), YO(500), DO(500), SA(500), CA(500),
7      1      CURV(500), DL(500)
8      C
9      COMMON/COMMON/CCL, INCL1, C11, ALPHA, SUMDS(10), TLN(10, 12), IND
10     1      , ALPHAO, CNV(10), SMDSVF(10), MIO(10)
11     COMMON /SIGMA/ CSIG(500), CN(12)
12     COMMON /GCF/   CD(500), CF(500), CG(500), CI(500), VF(500)
13     C
14     COMMON/BFLAG/  IDB(10), INL(10), IFL(10), NL(10), LIFT(10),
15     1      IBMF(10), ISAV1(10), ISAV2(10), ISAV3(10),
16     2      BTITLE(10, 7), IBT, IBST, IBTOT, NELTOT,
17     3      IIRB(10), IMMR(10), CHORD(10), IND(10), LIFTOT
18     4      , IPRB(10), IFST(10), ISEC(10), FTITLE(15), IPVRI(10)
19     COMMON/COM/IFLL
20     C
21     C
22     C
23     COMMON /FILEID/ IF01, IF02, IF03, IF04, IF05,
24     1      IF06, IF07, IF08, IF09, IF10,
25     2      IF01, IF02, IF13, IF14, IF15
26     3      , IF16, IF17, IF18, IF19, IF20
27     REMIND IF01
28     REMIND IF02
29     C
30     C
31     C SET SOME QUANTITIES
32     20 SAI = 0.0
33     CAI = 1.0
34     OSI = 0.0
35     C
36     DO 230 I = 1, NO
37     JJ = 0
38     K = 2
39     C
40     DO 160 IB = 1, IBTOT
41     IF (IBMF(IB) .LT. C) GO TO 150
42     C
43     C COUNTER FOR ELEMENT GEOMETRY
44     J = INL(IB) - 1
45     C COUNTERS FOR A,B ARRAYS
46     JI = JJ + 1
47     JF = JJ + NL(IB)
48     C
49     C JJ IS COUNTER FOR THE CURRENT ELEMENT
50     C
51     C ZERO OUT A&B ARRAYS
52     DO 30 JJI = JI, JF
53     A(JJI) = 0.0
54     B(JJI) = 0.0
55     C
56     JJ = JI

```

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VXY0001
VXY0002
VXY0003
VXY0004
VXY0005
VXY0006
VXY0007
VXY0008
VXY0009
VXY0010
VXY0011
VXY0012
VXY0013
VXY0014
VXY0015
VXY0016
VXY0017
VXY0018
VXY0019
VXY0020
VXY0021
VXY0022
VXY0023
VXY0024
VXY0025
VXY0026
VXY0027
VXY0028
VXY0029
VXY0030
VXY0031
VXY0032
VXY0033
VXY0034
VXY0035
VXY0036
VXY0037
VXY0038
VXY0039
VXY0040
VXY0041
VXY0042
VXY0043
VXY0044
VXY0045
VXY0046
VXY0047
VXY0048
VXY0049
VXY0050
VXY0051
VXY0052
VXY0053
VXY0054
VXY0055

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57      JJ1 = JJ + 1                                VXY0056
58      JJ3 = JJ1 + 1                                VXY0057
59      GO TO 50                                      VXY0058
60      90 JJ3 = JJ + 1                              VXY0059
61      50 J = J + 1                                  VXY0060
62      AO = XYFORMX(I), Y(I), DST, SAT, CAT, TEST(I), X(I), Y(I),    VXY0362
63      1 DL(I), SA(I), CA(I), ISEC(I), A1, AC, AP, B0, B1, BC, BP) VXY0061
64      B1(J) = B1(J) + B0                            VXY0062
65      A1(J) = A1(J) + A0                            VXY0063
66      C                                             VXY0064
67      C                                             VXY0067
68      IF (IFST(I)) .EQ. C) GO TO A0                 VXY0068
69      IF (IFST(I)) - 2)6C,70,60                     VXY0069
70      C                                             VXY0070
71      C FIRST TERMS OF FIRST ORDER                 VXY0071
72      60 A1(J) = A1(J) - (CF(I) + CD(I))*A1         VXY0072
73      B1(J) = B1(J) - (CF(I) + CD(I))*B1           VXY0073
74      A1(J1) = A1(J1) + CD(I)*A1                  VXY0074
75      B1(J1) = B1(J1) + CD(I)*B1                   VXY0075
76      A1(J3) = A1(J3) + CF(I)*A1                  VXY0076
77      B1(J3) = B1(J3) + CF(I)*B1                   VXY0077
78      C                                             VXY0078
79      70 IF (IFST(I)) .LT. 2) GO TO 80              VXY0079
80      C                                             VXY0080
81      C CURVATURE TERM OF FIRST ORDER              VXY0081
82      A1(J) = A1(J) + DL(I)*CURV(I)*AC             VXY0082
83      B1(J) = B1(J) + DL(I)*CURV(I)*BC             VXY0083
84      C                                             VXY0084
85      80 IF (ISEC(I)) .EQ. C) GO TO 110             VXY0085
86      IF (ISEC(I)) - 2)9C,100,90                   VXY0086
87      C                                             VXY0087
88      C FIRST TERMS OF SECOND ORDER                VXY0088
89      90 A1(J) = A1(J) - (CG(I) + CI(I))*AP         VXY0089
90      B1(J) = B1(J) - (CG(I) + CI(I))*BP           VXY0090
91      A1(J1) = A1(J1) + CG(I)*AP                   VXY0091
92      B1(J1) = B1(J1) + CG(I)*BP                   VXY0092
93      A1(J3) = A1(J3) + CI(I)*AP                   VXY0093
94      B1(J3) = B1(J3) + CI(I)*BP                   VXY0094
95      C                                             VXY0095
96      100 IF (ISEC(I)) .LT. 2) GO TO 110            VXY0096
97      C                                             VXY0097
98      C CURVATURE TERM OF SECOND ORDER             VXY0098
99      A1(J) = A1(J) + 2.0*AP*(DL(I)*CURV(I))*2     VXY0099
100     B1(J) = B1(J) + 2.0*BP*(DL(I)*CURV(I))*2     VXY0100
101     C                                             VXY0101
102     110 JJ1 = JJ                                   VXY0102
103     JJ = JJ + 1                                     VXY0103
104     IF (JJ - JF190,120,130)                       VXY0104
105     C                                             VXY0105
106     120 JJ3 = JJ1 - 1                              VXY0106
107     GO TO 50                                       VXY0107
108     130 JJ = JJ - 1                              VXY0108
109     C                                             VXY0109
110     C                                             VXY0110
111     IF (LIFT(I)) .EQ. C) GO TO 150                VXY0111
112     VN = 0.0                                       VXY0112
113     VT = 0.0                                       VXY0113

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114	DO 140 J = JT,JF	VXY0114
115	VN = VN - B(J)*WF(J)	VXY0115
116	140 VT = VT + A(J)*WF(J)	VXY0116
117	C	VXY0117
118	K = K + 1	VXY0118
119	VXT(K) = VT	VXY0119
120	VYN(K) = VN	VXY0120
121	150 CONTINUE	VXY0121
122	C	VXY0122
123	C SET UNIFORM ONSET FLOWS	VXY0123
124	C ALPHA = 0	VXY0124
125	VXT(1) = CA1	VXY0125
126	VYN(1) = -SA1	VXY0126
127	C	VXY0127
128	C ALPHA = 90	VXY0128
129	VXT(2) = SA1	VXY0129
130	VYN(2) = CA1	VXY0130
131	C	VXY0131
132	C SET INPUT NON-UNIFORM ONSET FLOWS	VXY0132
133	M1 = LIFTOT + 2	VXY0133
134	IF (M.EQ. M1) GO TO 170	VXY0134
135	M2 = M1 + 1	VXY0135
136	DO 160 K = M2, M	VXY0136
137	VXT(K) = 0.0	VXY0137
138	160 VYN(K) = 0.0	VXY0138
139	170 CONTINUE	VXY0139
140	C	VXY0140
141	C CHECK IF INDIVIDUAL FLOWS DESIRED	VXY0141
142	IF (IND.NE. 1) GO TO 200	VXY0142
143	REIND IF14	VXY0143
144	C	VXY0144
145	C CALCULATE INDIVIDUAL FLOWS	VXY0145
146	DO 190 K = 1,M	VXY0146
147	CALL GETT(1F14, 1, M, SIG(1), 1, VN)	VXY0147
148	VX(K) = VXT(K)	VXY0148
149	VY(K) = VYN(K)	VXY0149
150	C	VXY0150
151	DO 180 J = 1,N	VXY0151
152	VX(K) = VX(K) + B(J)*SIG(J)	VXY0152
153	180 VY(K) = VY(K) + A(J)*SIG(J)	VXY0153
154	190 CONTINUE	VXY0154
155	C	VXY0155
156	C SAVE VELOCITIES	VXY0156
157	WRITE(1F02) (VX(K), VY(K), K = 1,M)	VXY0157
158	C	VXY0158
159	C CALCULATE COMBINED FLOW	VXY0159
160	IF(1FLLL.EQ.0) GO TO 230	VXY0160
161	200 VXC = 0.0	VXY0161
162	VYC = 0.0	VXY0162
163	DO 210 K = 1,M	VXY0163
164	VXC = VXC + VXT(K)*CK(K)	VXY0164
165	210 VYC = VYC + VYN(K)*CK(K)	VXY0165
166	C	VXY0166
167	DO 220 J = 1,N	VXY0167
168	VXC = VXC + B(J)*CSIG(J)	VXY0168
169	220 VYC = VYC + A(J)*CSIG(J)	VXY0169
170	C	VXY0170
171	C SAVE VELOCITIES	VXY0171
172	WRITE(1F01) VXC, VYC	VXY0172
173	C	VXY0173
174	C	VXY0174
175	230 CONTINUE	VXY0175
176	C	VXY0176
177	C	VXY0177
178	RETURN	VXY0178
179	END	

1	COMPILER(YM=1)	
2	SUBROUTINE VPROFF(N, M, NO, X, Y, TITLE, IND)	
3	C	VPR0001
4	COMMON /FILE ID/ IFA1, IFA2, IF03, IF04, IF05,	VPR0002
5	IF06, IF07, IF08, IF09, IF10,	VPR0003
6	IF01, IF02, IF13, IF14, IF15,	VPR0004
7	IF16, IF17, IF18, IF19, IF20	VPR0005
8	DIMENSION X(1), Y(1), TITLE(7), V(1200), V1X(200), V2X(200),	VPR0006
9	V3X(200), V4X(200), V5X(200)	VPR0007
10	COMMON/OVER/ V6X(200), V1Y(200), V2Y(200), V3Y(200), V4Y(200),	
11	V5Y(200), V6Y(200)	
12	COMMON/COM/IFLLL	
13	C	VPR0008
14	DATA RD/57.2957797/	VPR0009
15	C	VPR0010
16	C VX AND VY HAVE BEEN SAVED ON UNIT IF02	VPR0011
17	REWIND IF01	VPR0012
18	REWIND IF02	VPR0013
19	C	VPR0014
20	IF (IND.NE. 1) GO TO 40	VPR0015
21	C	VPR0016
22	C INDIVIDUAL FLOWS	VPR0017
23	M2 = M + M	VPR0018
24	I1 = 1 - M2	VPR0019
25	I2 = 0	VPR0020
26	DO 10 J = 1, NO	VPR0021
27	I1 = I1 + M2	VPR0022
28	I2 = I2 + M2	VPR0023
29	C	VPR0024
30	10 READ(IF02) (V(K), K = I1, I2)	VPR0025
31	C	VPR0026
32	DO 30 K = 1, M	VPR0027
33	WRITE(6,60) K, TITLE	VPR0028
34	C	VPR0029
35	DO 20 I = 1, NO	VPR0030
36	IX = (I-1)*M2 + 2*K - 1	VPR0031
37	IY = IX + 1	VPR0032
38	VT = SQRT(V(IX)**2 + V(IY)**2)	VPR0033
39	TH = ATAN2(V(IY), V(IX))*PD	VPR0034
40	IF(K.EQ.1) V1X(I)=V(IX)	
41	IF(K.EQ.1) V1Y(I)=V(IY)	
42	IF(K.EQ.2) V2X(I)=V(IX)	
43	IF(K.EQ.2) V2Y(I)=V(IY)	
44	IF(K.EQ.3) V3X(I)=V(IX)	
45	IF(K.EQ.3) V3Y(I)=V(IY)	
46	IF(K.EQ.4) V4X(I)=V(IX)	
47	IF(K.EQ.4) V4Y(I)=V(IY)	
48	IF(K.EQ.5) V5X(I)=V(IX)	
49	IF(K.EQ.5) V5Y(I)=V(IY)	
50	IF(K.EQ.6) V6X(I)=V(IX)	
51	IF(K.EQ.6) V6Y(I)=V(IY)	
52	C	VPR0035
53	20 WRITE(6,70) I, X(I), Y(I), V(IY), V(IY), VT, TH	VPR0036
54	C	VPR0037
55	30 CONTINUE	VPR0038
56	WRITE(7,1500) (X(J), J=1, NO)	

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57      WRITE(7,1500) (V(I),J=1,N0)
58      WRITE(7,1500) (V1X(J),J=1,N0)
59      WRITE(7,1500) (V2X(J),J=1,N0)
60      WRITE(7,1500) (V3X(J),J=1,N0)
61      WRITE(7,1500) (V4X(J),J=1,N0)
62      WRITE(7,1500) (V5X(J),J=1,N0)
63      IF(K.GE.6) WRITE(7,1500) (V6X(J),J=1,N0)
64      WRITE(7,1500) (V1Y(J),J=1,N0)
65      WRITE(7,1500) (V2Y(J),J=1,N0)
66      WRITE(7,1500) (V3Y(J),J=1,N0)
67      WRITE(7,1500) (V4Y(J),J=1,N0)
68      WRITE(7,1500) (V5Y(J),J=1,N0)
69      IF(K.GE.6) WRITE(7,1500) (V6Y(J),J=1,N0)
70      1500 FORMAT(0P6E13.8)
71      C
72      C
73      C COMBINED FLOW
74      IF(I.F.L1.EQ.0) GO TO 91
75      40 WRITE(6,80) TITLE
76      C
77      DO 50 I = 1,N0
78      READ(17,01) V(1), V(2)
79      VT = SQRT(V(1)**2 + V(2)**2)
80      TH = ATAN2(V(2), V(1))*90
81      50 WRITE(6,70) I, X(1), Y(1), V(1), V(2), VT, TH
82      C
83      60 FORMAT(1H1, 20X 'INDIVIDUAL FLOW NO. ',I2, 5X 'OFFBODY POINTS',
84      1 5X,7A4//T15,'I', T27,'X(I)', T44,'Y(I)', T61,'VX',
85      2 778,'VY', T95,'VT', T109,'THETA(DEG)')//)
86      70 FORMAT(1H , 11X I3, 615XF12.6))
87      80 FORMAT(1H1, 20X 'COMBINED FLOW FOR OFFBODY POINTS', 5X,7A4//
88      1 T15,'I', T27,'X(I)', T44,'Y(I)', T61,'VX',
89      2 778,'VY', T95,'VT', T109,'THETA(DEG)')//)
90      C
91      91 RETURN
92      END

```

```

VPR0039
VPR0040
VPR0041
VPR0042
VPR0043
VPR0044
VPR0045
VPR0046
VPR0047
VPR0048
VPR0049
VPR0050
VPR0051
VPR0052
VPR0053
VPR0054
VPR0055
VPR0056
VPR0057
VPR0058
VPR0059

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```

1      SUBROUTINE WEIGHT ( SUMDS, DS, JI, JF, WF, IPVOR )
2      C
3      DIMENSION DS (JI), WF (JI)
4      IF ( IPVOR .EQ. 0 ) GO TO 20
5      C
6      WRITE ( 6, 1000 )
7      1000 FORMAT ( 'IM1, *VORTICITY WEIGHTING FUNCTION = S/L*411 - S/L1*777 )
8      C
9      S1 = 0.0
10     SDI = 0.0
11     SDN = 1.0
12     C
13     DO 10 J = JI, JF
14         SD = DS (J) / (2.0 * SUMDS)
15         S = S1 + SD
16         S1 = S * SD
17         WF (J) = 15 - SD11 + (SDN - S)
18     10 CONTINUE
19     C
20     WRITE ( 6, 1010 ) ( WF (J), J = JI, JF )
21     1010 FORMAT ( 'IM1, 6F18.6 )
22     C
23     RETURN
24     C
25     20 CONTINUE
26         DO 30 J = JI, JF
27             30 WF (J) = 1.0
28     30 CONTINUE
29     WRITE ( 6, 1020 )
30     1020 FORMAT ( 'IM1, *VORTICITY WEIGHTING FUNCTION CONSTANT = 1.0* /77 )
31     C
32     RETURN
33     END

```

```

1      FUNCTION XYFORM (X1, Y1, DS1, SIN1, COS1, IFST,
2      1      XJ, YJ, DSJ, SINJ, COSJ, ISEC,
3      2      A1, AC, AP, B1, BC, BP)
4      C
5      C
6      C THIS ROUTINE ACTUALLY CALCULATES THE INDUCED VELOCITY
7      C ARRAY ELEMENTS A AND B.
8      C
9      C
10     DATA E1,E2/169.0,11.111/,E0,EY/0.0001,0.01/
11     C
12     C
13     DX = X1 - XJ
14     DY = Y1 - YJ
15     ROSQ = DX**2 + DY**2
16     DSJSQ = DSJ**2
17     C
18     IF (ROSQ .LT. DSJSQ*E1) GO TO 1C
19     C
20     C USE FAR FIELD FORMULAS
21     VX = 2.*DSJ/ROSQ
22     VY = VX*DY
23     VZ = VX*DX
24     AO = -VX*SIN1 + VY*COS1
25     XYFORM = AO
26     BO = VX*COS1 + VY*SIN1
27     IF (IFST .EQ. 0) .AND. (ISEC .EQ. 0) RETURN
28     SC = SIN1*COSJ
29     SS = SIN1*SINJ
30     CS = COS1*SINJ
31     CC = COS1*COSJ
32     SSCC = (SS - CC)*VX*VY/12.0
33     CSSC = (CS + SC)*VX*VY/12.0
34     DXR = 1.0 - 2.0*DX**2/ROSQ
35     DYR = 1.0 - 2.0*DY**2/ROSQ
36     DDR = DSJSQ/ROSQ*E.01
37     C
38     A1 = DDR*(SC*DXR - CS*DYR) - SSCC
39     B1 = -DDR*(CC*DXR + SS*DYR) + CSSC
40     AC = -DDR*(SS*DXR + CC*DYR) - CSSC
41     BC = DDR*(CS*DXR - SC*DYR) - SSCC
42     AP = AO/12.0
43     BP = BO/12.0
44     RETURN
45     C
46     C USE NEAR FIELD FORMULAS
47     ADX = DX*COSJ + DY*SINJ
48     Y = DY*COSJ - DX*SINJ
49     S = SIN1*COSJ - COS1*SINJ
50     C = COS1*COSJ + SIN1*SINJ
51     C
52     IF (ROSQ .GT. DSJSQ*E2) GO TO 2C
53     C
54     C USE EXACT FORMULAS
55     XB = X/DSJ
56     YB = Y/DSJ

```

```

XYF0001
XYF0002
XYF0003
XYF0004
XYF0005
XYF0006
XYF0007
XYF0008
XYF0009
XYF0010
XYF0011
XYF0012
XYF0013
XYF0014
XYF0015
XYF0016
XYF0017
XYF0018
XYF0019
XYF0020
XYF0021
XYF0022
XYF0023
XYF0024
XYF0025
XYF0026
XYF0027
XYF0028
XYF0029
XYF0030
XYF0031
XYF0032
XYF0033
XYF0034
XYF0035
XYF0036
XYF0037
XYF0038
XYF0039
XYF0040
XYF0041
XYF0042
XYF0043
XYF0044
XYF0045
XYF0046
XYF0047
XYF0048
XYF0049
XYF0050
XYF0051
XYF0052
XYF0053
XYF0054
XYF0055
XYF0056

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57	R0 = X1**2 + YB**2	XYF0057
58	RD = XB**2 - YB**2	XYF0058
59	R1 = R0 + XB + 0.25	XYF0059
60	R2 = R0 - XB + 0.25	XYF0060
61	C	XYF0061
62	VX = +ALOG((R0SQ+DSJ*(X+0.25*DSJSC)/(R0SQ-DSJ*(X+0.25*DSJSC)))	XYF0062
63	Y = Y*DSJ	XYF0063
64	X = R0SQ - 0.25*DSJSC	XYF0064
65	VY = 2.0*ATAN2(Y,X)	XYF0065
66	IF (IIFST.EQ. 0) .AND. (ISEC.EQ. 0) GO TO 30	XYF0066
67	C	XYF0067
68	VX0 = VX	XYF0068
69	VY0 = VY	XYF0069
70	VX1 = YB*VY0 + XB*VX0 - 2.0	XYF0070
71	VY1 = XB*VY0 - YB*VX0	XYF0071
72	VXC = -2.0*VY1 + XB*YB/(R1+R2)	XYF0072
73	VYC = 2.0*(VX1 + 1.0)-2.0*(R0**2 - 0.25*RD)/(R1+R2)	XYF0073
74	VXP = RD*VX0 + 2.0*XB*(YB*VY0 - 1.0)	XYF0074
75	VYP = RD*VY0 - 2.0*YB*(XB*VX0 - 1.0)	XYF0075
76	C	XYF0076
77	GO TO 30	XYF0077
78	C	XYF0078
79	C	XYF0079
80	C USE MULTIPOLE FORMULAS	XYF0080
81	20 AE = X*DSJ/R0SQ	XYF0081
82	BE = Y*DSJ/R0SQ	XYF0082
83	ASQ = X**2/R0SQ	XYF0083
84	ESQ = DSJSC/R0SQ	XYF0084
85	VX = 2.0*AE*(1.0 + (ASQ - 0.75)*ESQ/3.0)	XYF0085
86	VY = 2.0*BE*(1.0 + (ASQ - 0.25)*ESQ/3.0)	XYF0086
87	IF (IIFST.EQ. 0) .AND. (ISEC.EQ. 0) GO TO 30	XYF0087
88	A4 = 0.15*(R0*ASQ*(ASQ - 1.0) + 1.0)*ESQ	XYF0088
89	VX1 = ESQ*(2.0*ASQ - 1.0 + A4)/6.0	XYF0089
90	VY1 = AE*BE*(1.0 + 0.3*(2.0*ASQ - 1.0)*ESQ)/3.0	XYF0090
91	VXC = AE*BE*(1.0 + 0.3*(2.0*ASQ - 1.0)*ESQ)/3.0	XYF0091
92	BSQ = Y**2/R0SQ	XYF0092
93	VYC = ESQ*(2.0*BSQ - 1.0 - 0.5*A4)/6.0	XYF0093
94	VYP = (1.0 + (ASQ - 0.25)*ESQ*0.6)/6.0	XYF0094
95	VXP = AE*VYP	XYF0095
96	VYP = BE*VYP	XYF0096
97	C	XYF0097
98	C	XYF0098
99	30 CONTINUE	XYF0099
100	XYFORM = -S*VX + C*VY	XYF0100
101	BO = C*VX + S*VY	XYF0101
102	IF (IIFST.EQ. 0) .AND. (ISEC.EQ. 0) RETURN	XYF0102
103	A1 = -S*VX1 + C*VY1	XYF0103
104	AC = -S*VXC + C*VYC	XYF0104
105	AP = -S*VXP + C*VYP	XYF0105
106	C	XYF0106
107	B1 = C*VX1 + S*VY1	XYF0107
108	BC = C*VXC + S*VYC	XYF0108
109	BP = C*VXP + S*VYP	XYF0109
110	C	XYF0110
111	C	XYF0111
112	RETURN	XYF0112
113	END	XYF0113

Program NOZZLEC

M NOZZLEC

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1      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
2      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
3      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
4      3      ELND,ANG(700),AR(700),AROFF(200)
5      COMMON /VELOC/ V1(700),V2(700),V3(700),V4(700),V5(700),V1X(200),
6      1      V2X(200),V3X(200),V4X(200),V5X(200),V1Y(200),
7      2      V2Y(200),V3Y(200),V4Y(200),V5Y(200)
8      COMMON /CONT/ VC,V51,V52,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
9      1      TITLE(3),VINP,ALF,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
10     2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
11     COMMON /COUT/ NT,NS1,NH,NP,IV,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
12     1      NST2,NST3,NST7,NPPR(30),IRAK(30),M1,M2,ICOMP1,IPL,
13     2      IHUB
14     COMMON /CONDIT/ TTCTAL,PT,PSTAT,1STAT,PSTATC,ATOTAL,PTC,RHOST,
15     1      RHO101,ASTAT,QCINF,RSORTC
16     COMMON /WRIT/ AA1C,AA2C,AA3C,AA4C,AA5C,AA11,AA21,AA31,AA41,AA51,
17     1      AA12,AA22,AA32,AA42,AA52
18     COMMON /SOLUT/ VBAR(700),VBARO(200),VINCI(700),VXINC(200),
19     1      VYINC(200),RHOBI(700),RBOIT(700),RHOEC(200),
20     2      VCOM(700),RBOOT(200),VRE(200),VRECO(200),
21     3      VXCCH(200),VYCOM(200),THETA(200),PSCP(700),
22     4      PSOPT(700),CHACH(700),XMACH(700),CPI(700),CPC(700),
23     5      RHOI(700)
24     COMMON /SOLUTO/ PSOFPC(200),PSOFF(200),CHACO(200),XPACO(200),
25     1      RMC01(200)
26     COMMON /PICT/ VPERIN,XX,XMIN,EXEF,VV,VMIN,ORD,EMSTCR,AL,AAAA
27     COMMON /CLPLOT/ XPEN,YPEN,NX6,NY,IPEN,XLABEL(10),YLABEL(10)
28     COMMON/FINDER/ JCHK1,JCHK2,JCHK3
29     C -----
30     C
31     C
32     C THIS IS THE MAIN PROGRAM WHICH CALLS THE SUBROUTINES TO
33     C COMPUTE THE 2-D COMBINATION SOLUTIONS FOR BOTH COMPRESSIBLE
34     C AND INCOMPRESSIBLE VERSIONS.
35     C
36     C
37     DIMENSION XPL0T(700),YPL0T(700),NKK(8),P(14)
38     CALL INPTR
39     AAAA=VINP
40     VSAVE = VINP
41     IF(ICOIMP1.EQ.1) ICOIMP=0
42     IF(ICOIMP1.EQ.0) ICOIMP=1
43     IF(ICOIMP.EQ.1) VINP=VINP*(1.0-0.2*(VINP/ATOTAL)**2)**2.5
44     CALL SEARCH
45     CALL ANGLEF
46     VCSAVE=VC
47     VC1S=V51
48     VC2S=V52
49     VC=VIC
50     V51=VIC1
51     V52=VIC2
52     10 CALL SOLVE
53     VC = VCSAVE
54     V51=VC1S
55     V52=VC2S
56     VINFP=VINP

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57      VINP=VSAVE
58      IF (INUB.EQ.0) GO TO 11
59      WRITE(6,100)
60      IF (M1.EQ.1) WRITE(6,105) A11,A21,A31,A41,A51,A12,A22,A32,
61      1      A42,A52
62      IF (M2.EQ.1) WRITE(6,110) A1C,A2C,A3C,A4C,A5C,A12,A22,A32,
63      1      A42,A52
64      IF ((M1.EQ.0).AND.(M2.EQ.0)) WRITE(6,115) A1C,A2C,A3C,A4C,A5C,
65      1      A11,A21,A31,A41,A51
66      WRITE(6,120) A,B,C,D,VINP
67      GO TO 12
68      11 WRITE(6,101)
69      WRITE(6,102) A1C,A2C,A3C,A4C
70      WRITE(6,103) A,B,C,VINP
71      12 IF (ICOMP.EQ.1) CALL COMCOR
72      CALL ONBODY
73      CALL OFRDY
74      C
75      C THE FOLLOWING CODE PLOTS PS/PT VS S AND MACH NUMBER VS S.
76      C
77      IF (CUTOF1.LE.0.0) GO TO 20
78      DO 15 I=1,NS1
79      IF (S(I).LT.0.0) S1(I)=-S(I)/CUTOF1
80      IF (S(I).GE.0.0) S1(I)= S(I)/CUTOF1
81      15 CONTINUE
82      J=0
83      J1=0
84      DO 16 I=1,NS1
85      IF (S1(I).GT.1.0) GO TO 16
86      IF (I.LT.JCHK1) J=J+1
87      IF (J.EQ.1) IS=1
88      IF (I.GE.JCHK1) J1=J+1
89      16 CONTINUE
90      IPL=1
91      CALL PLYR(J1,J,JCHK1,IS,1)
92      20 IF (CUTOFH.LE.0.0) GO TO 30
93      IN=NS1+1
94      IT=NH
95      DO 25 I=IN,IT
96      IF (S(I).LT.0.0) S1(I)=-S(I)/CUTOFH
97      IF (S(I).GE.0.0) S1(I)= S(I)/CUTOFH
98      25 CONTINUE
99      J=0
100     J1=0
101     DO 26 I=IN,IT
102     IF (S1(I).GT.1.0) GO TO 26
103     IF (I.LE.JCHK2) J=J+1
104     IF (J.EQ.1) IS=1
105     IF (I.GT.JCHK2) J1=J+1
106     26 CONTINUE
107     IPL=2
108     CALL PLYR(J1,J,JCHK2,IS,1)
109     30 IF (CUTOF2.LE.0.0) GO TO 37
110     IN=NH+1
111     DO 35 I=IN,NT
112     IF (S(I).LT.0.0) S1(I)=-S(I)/CUTOF2
113     IF (S(I).GE.0.0) S1(I)= S(I)/CUTOF2

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114 35 CONTINUE
115 J=0
116 J1=0
117 DO 36 I=IN,NI
118 IF(I(1).GT.1.0) GO TO 36
119 IF(I(1).EQ.0) J=J+1
120 IF(I(1).EQ.1) I5=1
121 IF(I(1).GT.1) JCHK3=J1=J+1
122 36 CONTINUE
123 IPL=3
124 CALL PLTER(J,J1,JCHK3,I5,I2)
125 IPL=IPL-1
126 IF(IPL.EQ.0) GO TO 40
127 MKK(1) = 4
128 MKK(2) = 0
129 MKK(3) = 3
130 IF(I(1).EQ.0) MKK(3) = 2
131 MKK(4) = 1
132 MKK(5) = 1
133 P(1) = 3.0
134 P(2) = XX
135 P(3) = XMIN
136 P(4) = XMIN*XX*EXEF
137 P(5) = YY
138 P(6) = YMIN
139 P(7) = YMIN*YY*ORD
140 P(8) = 10.0
141 P(9) = 0.0
142 P(10) = 0.0
143 P(11) = 0.0
144 P(12) = 0.0
145 P(13) = 0.0
146 P(14) = 90.0
147 I1=0
148 I11=0
149 DO 200 I=1,MS1
150 IF(XON(I).LT.XMIN) GO TO 200
151 IF(YON(I).LT.YMIN) GO TO 200
152 IF(XON(I).XX*EXEF-YMIN) 130,130,200
153 130 IF(YON(I)-YY*ORD-YMIN) 135,135,200
154 135 I1=I1+1
155 I11=I11+1
156 XPLOT(I11)=XON(I)
157 YPLOT(I11)=YON(I)
158 200 CONTINUE
159 MKK(6)=I1
160 I1=0
161 IF(I(1).EQ.0) GO TO 250
162 NN=NS1+1
163 DO 240 I=NN,NH
164 IF(XON(I).LT.XMIN) GO TO 240
165 IF(YON(I).LT.YMIN) GO TO 240
166 IF(XON(I).XX*EXEF-YMIN) 230,230,240
167 230 IF(YON(I)-YY*ORD-YMIN) 235,235,240
168 235 I1=I1+1
169 I11=I11+1
170 XPLOT(I11)=XON(I)

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171      YPLOT(III)=YON(I)
172      240 CONTINUE
173      KKK(7) = II
174      250 II=0
175      NN=NN+1
176      DO 270 I=NN,NT
177          IF(XON(I).LT.XMIN) 60 TO 270
178          IF(YON(I).LT.YMIN) 60 TO 270
179          IF(XON(I)-XX*EXEP-XMIN) 260,260,270
180      260 IF(YON(I)-YY*ORD-YMIN) 265,265,270
181      265 II=II+1
182      III=III+1
183      XPLOT(III)=XON(I)
184      YPLOT(III)=YON(I)
185      270 CONTINUE
186      IF(IMUB.EQ.0) KKK(7)=II
187      IF(IMUB.NE.0) KKK(8)=II
188      DO 280 I=1,3
189      XLABEL(I)=TITLE(I)
190      280 CONTINUE
191      XPEN=0.0
192      YPEN=0.0
193      IPEN=-3
194      MX6=-18
195      MY=0
196      CALL CALPLT(XPLOT,YPLOT,KKK,P)
197      40 STOP
198      100 FORMAT(/,16X,'V1',12X,'V2',12X,'V3',12X,'V4',12X,'V5',/)
199      101 FORMAT(/,16X,'V1',12X,'V2',12X,'V3',12X,'V4',/)
200      102 FORMAT(2X,'CONTROL',3X,4(1PE10.3,4X))
201      103 FORMAT(/,10X,'A',13X,'B',13X,'C',11X,'VINFP',/,6X,4(1PE10.3,4X))
202      105 FORMAT(2X,'LOWER',/,2X,'PASSAGE',3X,5(1PE10.3,4X),/,2X,'UPPER',/,
203      1 2X,'PASSAGE',3X,5(1PE10.3,4X))
204      110 FORMAT(2X,'CONTROL',3X,5(1PE10.3,4X),/,2X,'UPPER',/,2X,'PASSAGE',
205      1 3X,5(1PE10.3,4X))
206      115 FORMAT(2X,'CONTROL',3X,5(1PE10.3,4X),/,2X,'LOWER',/,2X,'PASSAGE',
207      1 3X,5(1PE10.3,4X))
208      120 FORMAT(/,10X,'A',13X,'B',13X,'C',13X,'D',11X,
209      1 'VINFP',/,6X,5(1PE10.3,4X))
210      END

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1      SUBROUTINE INPTR
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCM,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XPH,YP1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELND,ANG(700),AR(700),AROFF(200)
6      COMMON /VELOC/ V1(700),V2(700),V3(700),V4(700),V5(700),V1X(200),
7      1      V2X(200),V3X(200),V4X(200),V5X(200),V1Y(200),
8      2      V2Y(200),V3Y(200),V4Y(200),V5Y(200)
9      COMMON /CONT/ VC,V51,V52,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
10     1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
11     2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
12     COMMON /COUT/ NT,NS1,NH,NP,IW,NX,KND,ICOMP,M,NXH11,NXH12,NXH13,
13     1      NST2,NST3,NST7,NPFR(30),TRAK(30),M1,P2,ICOMP1,IPL,
14     2      IHUB
15     COMMON /CONDIT/ TTCTAL,PT,PSTAT,ISTAT,PSTATC,ATOTAL,PTC,RH0ST,
16     1      RH0T01,ASTAT,QCINF,RSORTC
17     COMMON/PICT/ VPERIN,XX,XMIN,EXEP,YY,YMIN,ORD,EMSTOR,AL,AAAA
18     C -----
19     C
20     C THIS SUBROUTINE READS DATA FROM BOTH CARDS AND DISK FILES.
21     C
22     READ(5,100) TITLE
23     WRITE(6,101) TITLE
24     READ(5,110) NT,NS1,NH,NP,IW,NX,KND,ICOMP1,IHUB
25     WRITE(6,111) NT,NS1,NH,NP,IW,NX,KND,ICOMP1,IHUB
26     READ(5,120) VC,V51,V52,VINP,ALFA,XMC,XMC1,XMC2,TTCTAL,PT
27     WRITE(6,121) VC,V51,V52,VINP,ALFA,XMC,XMC1,XMC2,TTCTAL,PT
28     READ(5,120) ELND,WDOTC,WDOTC1,WDOTC2,PSTAT,YSTAT,CUTOF1,CUTOF2,
29     1      CUTOFH,VPERIN
30     WRITE(6,121) ELND,WDOTC,WDOTC1,WDOTC2,PSTAT,YSTAT,CUTOF1,CUTOF2,
31     1      CUTOFH,VPERIN
32     IF (VPERIN.NE.0) READ(5,130) XX,XPIN,EXEP,YY,YMIN,ORD
33     IF (VPERIN.NE.0) WRITE(6,131) XX,XMIN,EXEP,YY,YMIN,ORD
34     C
35     C NT = TOTAL NUMBER OF CN-BODY POINTS.
36     C NS1 = TOTAL NUMBER OF ON-BODY POINTS ON BODY 1
37     C NH = TOTAL NUMBER OF CN-BODY POINTS ON BODIES 1 AND 2
38     C IW = 0 - WEIGHT FLOW DATA AT CONTROL STATIONS INPUT
39     C      = 1 - MACH NUMBERS AT CONTROL STATIONS INPUT
40     C      = 2 - VELOCITIES AT CONTROL STATIONS INPUT
41     C NX = 1 SUPERSONIC VELOCITY CORRECTION APPLIES.
42     C KND = 0 DATA NOT SCALED.
43     C      = 6 DATA SCALED BY CONTROL STATION PASSAGE HEIGHT.
44     C ICOMP1 = 0 COMPRESSIBLE VERSION
45     C          = 1 INCOMPRESSIBLE VERSION
46     C NOTE -- TWO OF THE THREE CONTROL STATIONS' DATA MUST BE INPUT, IF IHUB NE 0
47     C IHUB = 0 NO BODY 2
48     C
49     C
50     READ(5,140) XTEST,YCL,YCU
51     WRITE(6,141) XTEST,YCL,YCU
52     IF (IHUB.EQ.0) GO TO 3
53     READ(5,140) XTEST1,YCL1,YCU1
54     WRITE(6,141) XTEST1,YCL1,YCU1
55     READ(5,140) XTEST2,YCL2,YCU2
56     WRITE(6,141) XTEST2,YCL2,YCU2

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57      3 READ(5,140)   XR1,XR2,XRH
58      WRITE(6,141)   XR1,XR2,XRH
59      READ(5,140)    YR1,YR2,YRH
60      WRITE(6,141)   YR1,YR2,YRH
61      C XTEST,XTEST1,XTEST2 ARE THE LOCATIONS OF THE CONTROL STATIONS
62      C DOWNSTREAM OF BODY 2, BETWEEN BODIES 1 AND 2.
63      C AND BETWEEN BODIES 2 AND 3, RESPECTIVELY.
64      C XR1,XRH,XR2 ARE THE POINTS ON EACH OF THE BODIES WHERE
65      C SURFACE DISTANCE EQUALS ZERO.
66      READ(7,150) (XON(J),J=1,NT)
67      READ(7,150) (YON(J),J=1,NT)
68      READ(7,150) (V1(J),J=1,NT)
69      READ(7,150) (V2(J),J=1,NT)
70      READ(7,150) (V3(J),J=1,NT)
71      READ(7,150) (V4(J),J=1,NT)
72      READ(7,150) (V5(J),J=1,NT)
73      IF(NP.GT.100) GO TO 4
74      READ(7,150) (XOFF(J),J=1,NP)
75      READ(7,150) (YOFF(J),J=1,NP)
76      READ(7,150) (V1X(J),J=1,NP)
77      READ(7,150) (V2X(J),J=1,NP)
78      READ(7,150) (V3X(J),J=1,NP)
79      READ(7,150) (V4X(J),J=1,NP)
80      READ(7,150) (V5X(J),J=1,NP)
81      READ(7,150) (V1Y(J),J=1,NP)
82      READ(7,150) (V2Y(J),J=1,NP)
83      READ(7,150) (V3Y(J),J=1,NP)
84      READ(7,150) (V4Y(J),J=1,NP)
85      READ(7,150) (V5Y(J),J=1,NP)
86      GO TO 5
87      4 READ(7,150) (XOFF(J),J=1,100)
88      READ(7,150) (YOFF(J),J=1,100)
89      READ(7,150) (V1X(J),J=1,100)
90      READ(7,150) (V2X(J),J=1,100)
91      READ(7,150) (V3X(J),J=1,100)
92      READ(7,150) (V4X(J),J=1,100)
93      READ(7,150) (V5X(J),J=1,100)
94      READ(7,150) (V1Y(J),J=1,100)
95      READ(7,150) (V2Y(J),J=1,100)
96      READ(7,150) (V3Y(J),J=1,100)
97      READ(7,150) (V4Y(J),J=1,100)
98      READ(7,150) (V5Y(J),J=1,100)
99      READ(7,150) (XOFF(J),J=101,NP)
100     READ(7,150) (YOFF(J),J=101,NP)
101     READ(7,150) (V1X(J),J=101,NP)
102     READ(7,150) (V2X(J),J=101,NP)
103     READ(7,150) (V3X(J),J=101,NP)
104     READ(7,150) (V4X(J),J=101,NP)
105     READ(7,150) (V5X(J),J=101,NP)
106     READ(7,150) (V1Y(J),J=101,NP)
107     READ(7,150) (V2Y(J),J=101,NP)
108     READ(7,150) (V3Y(J),J=101,NP)
109     READ(7,150) (V4Y(J),J=101,NP)
110     READ(7,150) (V5Y(J),J=101,NP)
111     5 WRITE(6,160) TITLE
112     WRITE(6,170)
113     IF(ICOMP1.EQ.0) WRITE(6,180)

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114      IF(I COMP1.EQ.1) WRITE(6,175)
115      IF(IHUB.EQ.0) WRITE(6,189)
116      IF(IMUB.NE.0) WRITE(6,190)
117      IF(ELND.EQ.0.0) ELND=1.0
118      AL = ALFA
119      CALL CONST
120      C THE FOLLOWING SCALES THE DATA BY THE VALUE OF ELND.
121      DO 10 I=1,NT
122      XOM(I)=XOM(I)/ELND
123      YOM(I)=YOM(I)/ELND
124      10 CONTINUE
125      DO 20 I=1,NP
126      XOFF(I)=XOFF(I)/ELND
127      YOFF(I)=YOFF(I)/ELND
128      20 CONTINUE
129      RETURN
130      C *****FORMATS*****
131      100 FORMAT(3A6)
132      101 FORMAT(1H ,3A6)
133      110 FORMAT(9I4)
134      111 FORMAT(1H ,9I4)
135      120 FORMAT(10P10F8.0)
136      121 FORMAT(1H ,10P10F8.3)
137      130 FORMAT(10P6F10.0)
138      131 FORMAT(1H ,10P6F10.3)
139      140 FORMAT(10P3F10.0)
140      141 FORMAT(1H ,10P3F10.3)
141      150 FORMAT(10P6E13.8)
142      160 FORMAT(1H1,40X,3A6)
143      170 FORMAT(///,2X,'2-D COMBINATION SOLUTION')
144      175 FORMAT(//,6X,'INCOMPRESSIBLE VERSION')
145      180 FORMAT(//,6X,'COMPRESSIBLE VERSION')
146      189 FORMAT(//,6X,'COMBINATION OF THE FOLLOWING BASIC SOLUTIONS',/,9X,
147      1      '1. UNIFORM AXIAL',/,9X,'2. UNIFORM CROSSFLOW',/,9X,
148      2      '3. VORTICITY ABOUT BODY 1',/,9X,'4. VORTICITY ABOUT '
149      3      'BODY 3',/)
150      190 FORMAT(//,6X,'COMBINATION OF THE FOLLOWING BASIC SOLUTIONS',/,9X,
151      1      '1. UNIFORM AXIAL',/,9X,'2. UNIFORM CROSSFLOW',/,9X,
152      2      '3. VORTICITY ABOUT BODY 1',/,9X,'4. VORTICITY ABOUT '
153      3      'BODY 2',/,9X,'5. VORTICITY ABOUT BODY 3',/)
154      END

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57      VSONCC = ATOTAL/SQRT(1.2)
58      IF(IHUB.EQ.0) GO TO 16
59      AS1 = YCU1-YCL1
60      AS2 = YCU2-YCL2
61      IF(IIW.EQ.0).AND.(WDOTC.EQ.0.0)) WDOTC = WDOTC1+WDOTC2
62      IF(IIW.EQ.0).AND.(WDOTC1.EQ.0.0)) WDOTC1 = WDOTC-WDOTC2
63      IF(IIW.EQ.0).AND.(WDOTC2.EQ.0.0)) WDOTC2 = WDOTC-WDOTC1
64      16 IF(IIW.EQ.1) GO TO 40
65      GO TO 50
66      40 VC = ATOTAL*XMC/SQRT(1.0*XMC**2/5.0)
67      IF(IHUB.EQ.0) GO TO 25
68      VS1 = ATOTAL*XMC1/SQRT(1.0*XMC1**2/5.0)
69      VS2 = ATOTAL*XMC2/SQRT(1.0*XMC2**2/5.0)
70      GO TO 25
71      50 IF(IIW.EQ.2) GO TO 25
72      VIC = WDOTC/(16*RHOTOT*AC)*12.0
73      CALL VBARIT(VIC,ATOTAL,RHOTOT,RHCC)
74      VC = WDOTC/(16*RHOC*AC)*12.0
75      IF(IHUB.EQ.0) GO TO 25
76      VIC1 = WDOTC1/(16*RHOTOT*AS1)*12.0
77      VIC2 = WDOTC2/(16*RHOTOT*AS2)*12.0
78      CALL VBARIT(VIC1,ATOTAL,RHOTOT,RHOC1)
79      CALL VBARIT(VIC2,ATOTAL,RHOTOT,RHOC2)
80      VS1 = WDOTC1/(16*RHCC1*AS1)*12.0
81      VS2 = WDOTC2/(16*RHCC2*AS2)*12.0
82      25 IF(VC.GT.VSONCC) WRITE(6,116) VC
83      IF(VC.GT.VSONCC) VC = VSONCC
84      IF(IHUB.EQ.0) GO TO 26
85      IF(VS1.GT.VSONCC) WRITE(6,117) VS1
86      IF(VS1.GT.VSONCC) VS1 = VSONCC
87      IF(VS2.GT.VSONCC) WRITE(6,118) VS2
88      IF(VS2.GT.VSONCC) VS2 = VSONCC
89      26 IF(KND.EQ.8) GO TO 35
90      30 IF(KND.EQ.-1).OR.(KND.EQ.4)) ELND = YCU
91      IF(KND.EQ. 1).OR.(KND.EQ.6)) ELND = YCU-YCL
92      IF(KND.EQ. 0).OR.(KND.EQ.5)) ELND = 1.0
93      AC = AC/ELND
94      YCU = YCU/ELND
95      YCL = YCL/ELND
96      XR1 = XR1/ELND
97      XR2 = XR2/ELND
98      YR1 = YR1/ELND
99      YR2 = YR2/ELND
100     XTEST = XTEST/ELND
101     IF(IHUB.EQ.0) GO TO 35
102     AS1 = AS1/ELND
103     AS2 = AS2/ELND
104     YCU1 = YCU1/ELND
105     YCU2 = YCU2/ELND
106     YCL1 = YCL1/ELND
107     YCL2 = YCL2/ELND
108     XRH = XRH/ELND
109     YRH = YRH/ELND
110     XTEST1 = XTEST1/ELND
111     XTEST2 = XTEST2/ELND
112     35 RTTOT = R21568*TTOTAL
113     GRHO = G*PT/TTOTAL

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1      SUBROUTINE CONST
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTFST,XTFST1,XTFST2,YCL,YCM,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELND,ANG(700),AR(700),AROFF(200)
6      COMMON /CONT/ VC,V51,V52,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
7      1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
8      2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
9      COMMON /COUT/ NT,N51,NH,NP,IM,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
10     1      NST2,NST3,NST7,NPFR(30),IRAK(30),M1,P2,ICOMP1,IPL,
11     2      IMCE
12     COMMON /CONDIT/ TTOTAL,PT,PSTAT,TSTAT,PSTATC,ATOTAL,PTC,RHOST,
13     1      RHOTOT,ASTAT,QCINF,RSQRTC
14     C -----
15     C
16     C THIS SUBROUTINE CALCULATES THE CONSTANTS USED IN THE PROGRAM
17     C
18     PI=3.141592654
19     M1=0
20     M2=0
21     IF(IINUB.EQ.0) GO TO 5
22     IF((IM.EQ.0).AND.(WDOTC.EQ.0.0)) M1=1
23     IF((IM.EQ.0).AND.(WDOTC1.EQ.0.0)) M2=1
24     IF((IM.EQ.1).AND.(XMC.EQ.0.0)) M1=1
25     IF((IM.EQ.1).AND.(XMC1.EQ.0.0)) M2=1
26     IF((IM.EQ.2).AND.(VC.EQ.0.0)) M1=1
27     IF((IM.EQ.2).AND.(V51.EQ.0.0)) M2=1
28     5 P10180=PI/180.0
29     R21568= 1716.06
30     B = 32.174
31     PSTATC = PSTAT
32     IF((PSTAT.NE.0.0).AND.(TSTAT.NE.C.0)) GO TO 10
33     IF(PT.EQ.0.0) PT = 2116.23
34     IF(TTOTAL.EQ.0.0) TTOTAL = 518.69
35     ATOTAL = 49.009*SQRT(TTOTAL)
36     CATOT = 1.0-0.2*(VINP/ATOTAL)**2
37     PSTATC = PT*CATOT**3.5
38     PTC = PT
39     RHOTOT = PT/(R21568*TTOTAL)
40     TSTAT = TTOTAL*CATOT
41     RHOST = PSTATC/(R21568*TSTAT)
42     PSTAT = PT-0.5*RHOTOT*VINP*VINP
43     ASTAT = 49.009*SQRT(TSTAT)
44     GO TO 15
45     10 ASTAT = 49.009*SQRT(TSTAT)
46     RHOST = PSTAT/(R21568*TSTAT)
47     AMINF = VINP/ASTAT
48     CAMINF = 1.0-0.2*AMINF**2
49     PTC = PSTAT*CAMINF**3.5
50     PSTAT = PTC-0.5*RHOTOT*VINP*VINP
51     PT=PTC
52     TTOTAL = TSTAT*CAMINF
53     RHOTOT = PT/(R21568*TTOTAL)
54     ATOTAL = 49.009*SQRT(TTOTAL)
55     15 AC = YCU-YCL
56     VSONIC = ATOTAL/1.728

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114 C102RT = 0.5*RHOT1/P1
115 VFOAT = VINFL/ATOTAL
116 VCOAT = VC/ATOTAL
117 VS10AT = VS1/ATOTAL
118 VS20AT = VS2/ATOTAL
119 CON1 = 1.0-0.2*VCOAT**2
120 CON2 = 1.0-0.2*VFOAT**2
121 CON11 = 1.0-0.2*VS10AT**2
122 CON21 = 1.0-0.2*VS20AT**2
123 RSORTF = CON2**2.5
124 RSORTC = CON1**2.5
125 RSORT1 = CON11**2.5
126 RSORT2 = CON21**2.5
127 IF(IIM.EQ.1).OR.(Ib.EQ.2)) VIC = VC*RSORTC
128 IF(IIM.EQ.1).OR.(Ib.EQ.2)) VIC1 = VS1*RSORT1
129 IF(IIM.EQ.1).OR.(Ib.EQ.2)) VIC2 = VS2*RSORT2
130 IF(IIM.EQ.1).OR.(Ib.EQ.2)) WDOTC = VIC*6*RHOTOT*AC/12.0
131 IF(IIM.EQ.1).OR.(Ib.EQ.2)) WDOTC1 = VIC1*6*RHOTOT*AS1/12.0
132 IF(IIM.EQ.1).OR.(Ib.EQ.2)) WDOTC2 = VIC2*6*RHOTOT*AS2/12.0
133 IF(IHUB.EQ.0) GO TO 36
134 F = WDOTC1*WDOTC2
135 IF(WDOTC.EQ.0.0) VIC = F/(6*RHOTOT*AC/12.0)
136 IF(WDOTC.EQ.0.0) CALL VBARIT(VIC,ATOTAL,RHOTOT,RHOC)
137 IF(WDOTC.EQ.0.0) VC = F/(6*RHOC*AC)*12.0
138 IF(WDOTC.EQ.0.0) VCOAT = VC/ATOTAL
139 IF(WDOTC.EQ.0.0) CON1 = 1.0-0.2*VCOAT**2
140 IF(WDOTC.EQ.0.0) RSORTC = CON1**2.5
141 IF(WDOTC.EQ.0.0) WDOTC = F
142 F = WDOTC-WDOTC2
143 IF(WDOTC1.EQ.0.0) VIC1 = F/(6*RHOTOT*AS1/12.0)
144 IF(WDOTC1.EQ.0.0) CALL VBARIT(VIC1,ATOTAL,RHOTOT,RHOC)
145 IF(WDOTC1.EQ.0.0) VS1 = F/(6*RHOC*AS1)*12.0
146 IF(WDOTC1.EQ.0.0) VS10AT = VS1/ATOTAL
147 IF(WDOTC1.EQ.0.0) CON11 = 1.0-0.2*VS10AT**2
148 IF(WDOTC1.EQ.0.0) RSORT1 = CON11**2.5
149 IF(WDOTC1.EQ.0.0) WDOTC1 = F
150 F = WDOTC-WDOTC1
151 IF(WDOTC2.EQ.0.0) VIC2 = F/(6*RHOTOT*AS2/12.0)
152 IF(WDOTC2.EQ.0.0) CALL VBARIT(VIC2,ATOTAL,RHOTOT,RHOC)
153 IF(WDOTC2.EQ.0.0) VS2 = F/(6*RHOC*AS2)*12.0
154 IF(WDOTC2.EQ.0.0) VS20AT = VS2/ATOTAL
155 IF(WDOTC2.EQ.0.0) CON21 = 1.0-0.2*VS20AT**2
156 IF(WDOTC2.EQ.0.0) RSORT2 = CON21**2.5
157 IF(WDOTC2.EQ.0.0) WDOTC2 = F
158 IF(ICOMP1.EQ.1) VS1 = VIC1
159 IF(ICOMP1.EQ.1) VS2 = VIC2
160 36 IF(ICOMP1.EQ.1) VC = VIC
161 VNFOVC = VINFL/VC
162 IF(KND.LT.8) WDOTC = WDOTC/ELND
163 IF(KND.LT.8) WDOTC1 = WDOTC1/ELND
164 IF(KND.LT.8) WDOTC2 = WDOTC2/ELND
165 PSPTC = 1.0-C102RT*VC**2
166 PSPTC1 = 1.0-C102RT*VS1**2
167 PSPTC2 = 1.0-C102RT*VS2**2
168 PSPTIF = 1.0-C102RT*VINFL**2
169 PSPTC1 = RSORTF*CON2
170 PSPT1 = RSORT1*CON11

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171 PSPT2 = RSORT2*CON21
172 PSPTCC = RSORTC*CON1
173 XMINF = VFOAT/(CON2**0.5)
174 IF(XMC.EQ.0.0) XMC = VCOAT/(CON1**0.5)
175 IF(XMC1.EQ.0.0) XMC1 = VS1OAT/(CCN11**0.5)
176 IF(XMC2.EQ.0.0) XMC2 = VS2OAT/(CCN21**0.5)
177 QINF = PT*(1.0-PSPTIF)
178 QCINF = PTC*(10.7*VFOAT**2*RSORTF)
179 QC = PT*(1.0-PSPTC)
180 QCC = PTC*(10.7*VCOAT**2*RSORTC)
181 QC1 = PT*(1.0-PSPTC1)
182 QC2 = PT*(1.0-PSPTC2)
183 QC1C = PTC*(10.7*VS1OAT**2*RSORT1)
184 QC2C = PTC*(10.7*VS2OAT**2*RSORT2)
185 THETC = TTOTAL/518.69
186 DEL = PTC/2116.23
187 IF(IHUB.EQ.0) GO TO 37
188 WDOT1R = WDOTC1*SQR(THETC)/DEL
189 WDOT2R = WDOTC2*SQR(THETC)/DEL
190 37 WDOTCR = WDOTC*SQR(THETC)/DEL
191 IF(ICOMP1.EQ.0) GO TO 41
192 WRITE(6,101)
193 WRITE(6,111) VC,XMC,QC,PSPTC
194 IF(IHUB.EQ.0) GO TO 38
195 WRITE(6,121) VS1,XMC1,QC1,PSPTC1
196 WRITE(6,131) VS2,XMC2,QC2,PSPTC2
197 38 WRITE(6,141) VINI,XMINF,QINF,PSPTIF
198 GO TO 51
199 41 WRITE(6,100)
200 WRITE(6,110) VC,XMC,QC,QCC,PSPTC,PSPTCC,RSORTC
201 IF(IHUB.EQ.0) GO TO 39
202 WRITE(6,120) VS1,XMC1,QC1,QC1C,PSPTC1,PSPT1,RSORT1
203 WRITE(6,130) VS2,XMC2,QC2,QC2C,PSPTC2,PSPT2,RSORT2
204 39 WRITE(6,140) VINI,XMINF,QINF,QCINF,PSPTIF,PSPTC1,RSORTF
205 51 WRITE(6,150)
206 WRITE(6,155) ALFA,VNFOVC,VSONIC,VSONCC,WDOTCR,WDOT1R,WDOT2R
207 WRITE(6,160)
208 WRITE(6,165)
209 WRITE(6,170) YSTAT,PSYAT,PSYATC,ASTAT,RHOSY,WDOTC,WDOTC1,WDOTC2
210 WRITE(6,175) VIC,VIC1,VIC2
211 WRITE(6,180)
212 WRITE(6,155) TTOTAL,PT,PTC,ATOTAL,RHOTOT,THETC,DEL
213 WRITE(6,160)
214 WRITE(6,185)
215 WRITE(6,170) XR1,YR1,XR2,YR2,XTEST,YCL,YCU,ELND
216 WRITE(6,190)
217 WRITE(6,170) XTEST1,YCL1,YCU1,XTEST2,YCL2,YCU2
218 WRITE(6,210) CUTOF1,CUTOFH,CUTOF2
219 WRITE(6,160)
220 WRITE(6,200) NT,NP,NS1,NH,KND,IW,NX,ICOMP1,IHUB
221 WRITE(6,160)
222 ALFA = ALFA*PI0180
223 C *****FORMATS*****
224 100 FORMAT(1H0,27X,'MACH',11X,'DYNAMIC PRESSURE',13X,'PRESSURE RATIO',
225 1 RX,'DENSITY RATIO',/,14X,'VELOCITY',7X,'NO.',9X,'INC',10X,
226 2 'COMP',11X,'INC',8X,'COMP',/)
227 101 FORMAT(1H0,27X,'MACH',11X,'DYNAMIC PRESSURE',13X,'PRESSURE RATIO',

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228      1 /,14X,'VELOCITY',7X,'NO.',17X,'INC',25X,'INC',/)
229      110 FORMAT(3X,'CONTROL ',1PE10.3,2X,2(1PE10.3,5X,1PE10.3,3X),1PE10.3,
230      1      6X,1PE10.3,/)
231      111 FORMAT(3X,'CONTROL ',1PE10.3,2X,1PE10.3,9X,1PE10.3,20X,1PE10.3,/)
232      116 FORMAT(2X,'CONTROL STATION VELOCITY = ',1PE13.4,' EXCEEDS VSONIC.'
233      1      ,/, 'PROCEEDING WITH VC = VSONIC. IF THE CONTROL STATION '
234      2      'IS NOT AT THE THROAT, RESUBMIT WITH A LOWER VC.')
235      117 FORMAT(2X,'CONTROL STATION VELOCITY = ',1PE13.4,' EXCEEDS VSONIC.'
236      1      ,/, 'PROCEEDING WITH VS1 = VSONIC. RESUBMIT WITH A LOWER '
237      2      'VS1,')
238      118 FORMAT(2X,'CONTROL STATION VELOCITY = ',1PE13.4,' EXCEEDS VSONIC.'
239      1      ,/, 'PROCEEDING WITH VS2 = VSONIC. RESUBMIT WITH A LOWER '
240      2      'VS2,')
241      120 FORMAT(3X,'LOWER',/,3X,'PASSAGE ',1PE10.3,2X,2(1PE10.3,5X,1PE10.3
242      1      ,3X),1PE10.3,6X,1PE10.3,/)
243      121 FORMAT(3X,'LOWER',/,3X,'PASSAGE ',1PE10.3,2X,1PE10.3,9X,1PE10.3,
244      1      20X,1PE10.3,/)
245      130 FORMAT(3X,'UPPER',/,3X,'PASSAGE ',1PE10.3,2X,2(1PE10.3,5X,1PE10.3
246      1      ,3X),1PE10.3,6X,1PE10.3,/)
247      131 FORMAT(3X,'UPPER',/,3X,'PASSAGE ',1PE10.3,2X,1PE10.3,9X,1PE10.3,
248      1      20X,1PE10.3,/)
249      140 FORMAT(3X,'FREE ',/,3X,'STREAM ',1PE10.3,2X,2(1PE10.3,5X,1PE10.3
250      1      ,3X),1PE10.3,6X,1PE10.3,/)
251      141 FORMAT(3X,'FREE ',/,3X,'STREAM ',1PE10.3,2X,1PE10.3,9X,1PE10.3,
252      1      20X,1PE10.3,/)
253      150 FORMAT(/,9X,'ALPHA',9X,'VINF/VC',7X,'VSONIC',8X,'VSONICC',7X,
254      1      'WDOTCR',8X,'WDOTLCR',8X,'WDOTUCR')
255      155 FORMAT(7X,7(1PE10.3,4X))
256      160 FORMAT(/,1X,'-----',/)
257      1
258      165 FORMAT(/,9X,'TSTAT',9X,'PSTAT',9X,'PSTATC',8X,'ASTAT',9X,'RHSTAT'
259      1      ,7X,'WDOTC',9X,'WDOTL',7X,'WDOTU')
260      170 FORMAT(7X,8(1PE10.3,4X))
261      175 FORMAT(/,9X,'VIC',11X,'VICL',10X,'VICU',/,7X,3(1PE10.3,4X))
262      180 FORMAT(/,9X,'TTOT',10X,'PTOT',10X,'PTOTC',9X,'ATOT',10X,'RHOTOT',
263      1      8X,'THET',1CX,'DEL')
264      185 FORMAT(/,1CX,'XRI1',11X,'YRI1',1CX,'XRI2',9X,'YRI2',9X,'XTEST',10X
265      1      ,9X,'YCL',11X,'YCU',10X,'LND')
266      190 FORMAT(/,9X,'XTEST1',10X,'YCL1',10X,'YCU1',8X,'XTEST2',10X,'YCL2',
267      1      10X,'YCU2')
268      200 FORMAT(/,10X,'NT',5X,'NP',5X,'NS1',5X,'NH',5X,'MND',5X,'IW',5X,
269      1      'NX',5X,'ICOMP1',5X,'IHUB',/,5X,I3,4X,I3,5X,I3,4X,I3,5X,
270      2      I3,4X,I3,5X,I2,7X,I1,9X,I1)
271      210 FORMAT(/,7X,'P-S CUTOFF 1',3X,'P-S CUTOFF 2 P-S CUTOFF 3',/,
272      1      8X,1PE10.3,5X,1PE10.3,5X,1PE10.3)
273      END

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1      SUBROUTINE SEARCH
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELND,ANG(700),AR(700),AROFF(200)
6      COMMON /COUT/ NT,NS1,NH,NP,IW,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
7      1      NS12,NST3,NST7,NPFR(30),IRAK(30),M1,M2,ICOMP1,IPL,
8      2      IHUB
9      COMMON /CONT/ VC,V51,V52,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
10     1      YITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
11     2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
12     COMMON /CONDIT/ YTCAL,PT,PSTAT,ISTAT,PSTATC,ATOTAL,PTC,RHGST,
13     1      RHO10T,ASTAT,QCINF
14     C -----
15     C
16     C THIS SUBROUTINE FINDS THE HIGHLIGHT ON EACH BODY AND
17     C CALCULATES AREAS FOR ALL THE ON- AND OFF-BODY POINTS.
18     C
19     C PI = 3.141592654
20     C
21     C ON-BODY POINT CALCULATIONS
22     C
23     JJ = NS1+1
24     JJJ = NH + 1
25     NXH11=1
26     DO 10 J = 2,NS1
27     IF(XON(J).GE.XON(J-1)) GO TO 10
28     NXH11 = J
29     10 CONTINUE
30     IF(IHUB.EQ.0) GO TO 21
31     J1 = JJ + 1
32     J2 = JJJ-1
33     DO 20 J = J1,J2
34     IF(XON(J).GE.XON(J-1)) GO TO 20
35     NXH12 = J
36     XH12=XON(NXH12)
37     20 CONTINUE
38     21 J2 = JJJ+1
39     IF(IHUB.EQ.0) XH12=99999.0
40     DO 30 J = J2,NT
41     IF(XON(J).GE.XON(J-1)) GO TO 30
42     NXH13 = J
43     30 CONTINUE
44     NST3=NXH11
45     CALL SURF
46     C
47     C OFF-BODY POINT CALCULATIONS
48     C
49     K = 0
50     NP1=NP-1
51     DO 105 I = 1,NP1
52     IF(XOFF(I).LT.XH12) GO TO 101
53     IF(I.EQ.1) GO TO 101
54     L=L+1
55     IF(L.EQ.L1) GO TO 101
56     AA=YOFF(I+1)-YOFF(I)

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57      AB=YOFF(I)-YOFF(I-1)
58      IF(ABS(AA-AB).GT.0.001) GO TO 102
59      101 IF(XOFF(I).EC.XOFF(I+1)) GO TO 101
60      102 K=K+1
61      NPPR(K) = I
62      L=NPPR(K)
63      L=L+1
64      105 CONTINUE
65      K=K+1
66      NPPR(K)=NP
67      DO 110 I=1,NP
68      IC=0
69      IF(I.LE.NPPR(1)) K1=1
70      IF(I.LE.NPPR(1)) GO TO 112
71      DO 111 J=2,K
72      IF(I.LE.NPPR(J)).AND.(I.GT.NPPR(J-1)) K1=J
73      111 CONTINUE
74      112 IF(XOFF(I).LT.XON(NXHI1)) GO TO 200
75      NN=NS1+1
76      IF(INUB.NE.0) CALL INTERIXON,YON,NN,NXHI2,XOFF(I),Y5)
77      IF(XOFF(I).GT.XHI2) GO TO 180
78      IRAK(K1)=3
79      GO TO 110
80      180 IF(YOFF(I).LT.Y5) GO TO 190
81      IRAK(K1)=4
82      GO TO 110
83      190 IRAK(K1)=5
84      GO TO 110
85      200 IRAK(K1)=1
86      110 CONTINUE
87      RETURN
88      END

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1      SUBROUTINE ANGLEF
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCN,YCL1,YCL2,YCU1,
4      2      YCU2,XP1,XP2,XPH,YF1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELNG,ANG(700),AR(700),AROFF(200)
6      COMMON /COUT/ NT,NS1,NH,MP,IM,NX,KND,ICOMP,N,NXM11,NXM12,NXM13,
7      1      NS12,NS13,NS17,NPPF(30),IRAK(30),M1,M2,ICOMP1,IPL,
8      2      IHUE
9      C -----
10     C
11     C THIS SUBROUTINE CALCULATES THE ANGLE OF THE BODY SURFACE
12     C
13     NS = NS1-1
14     DO 10 I = 1,NS
15     IF(XON(I+1).EQ.XON(I)) ANG(I)=3.14159/2.0
16     IF(XON(I+1).EQ.XON(I)) GO TO 10
17     ANG(I) = ATAN((YON(I+1)-YON(I))/(XON(I+1)-XON(I)))
18     10 CONTINUE
19     NT2 = NH-1
20     IF(IHUE.EQ.0) GO TO 21
21     N = NS1+1
22     DO 20 I = N,NT2
23     IF(XON(I+1).EQ.XON(I)) ANG(I)=3.14159/2.0
24     IF(XON(I+1).EQ.XON(I)) GO TO 20
25     ANG(I) = ATAN((YON(I+1)-YON(I))/(XON(I+1)-XON(I)))
26     20 CONTINUE
27     21 N = NT2+2
28     NT3 = NT-1
29     DO 30 I = N,NT3
30     IF(XON(I+1).EQ.XON(I)) ANG(I)=3.14159/2.0
31     IF(XON(I+1).EQ.XON(I)) GO TO 30
32     ANG(I) = ATAN((YON(I+1)-YON(I))/(XON(I+1)-XON(I)))
33     30 CONTINUE
34     RETURN
35     END

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1      SUBROUTINE SURF
2      COMMON /CGORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(200),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YF1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELAC,ANG(700),AR(700),AROFF(200)
6      COMMON /COUT/ NT,NS1,NH,NP,IW,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
7      1      NST1,NST3,NST7,NPFF(30),IRAK(30),M1,M2,ICOMP1,IPL,
8      2      INLE
9      COMMON/FINDER/ JCHK1,JCHK2,JCHK3
10     C -----
11     C
12     C THIS SUBROUTINE CALCULATES THE SURFACE DISTANCE ON EACH BODY
13     C FROM ITS HIGHLIGHT
14     C
15     CALL CHECK(1,NS1,IR1,XON)
16     XRS=XR1
17     IF(XON(IR1).GT.XR1) XR1=XON(IR1)
18     DO 2 I=2,NS1
19     IF((((YON(I).GE.YR1).AND.(YON(I-1)).LE.YR1)).OR.((YCH(I)).LE.YR1),
20     1 AND.(YON(I-1).GE.YR1)).AND.(((XON(I)).LE.XR1).AND.(XON(I-1).GE.
21     2 XR1)).OR.((XON(I).GE.XR1).AND.(XON(I-1).LE.XR1)))) JCHK1 = I
22     2 CONTINUE
23     XR1=XRS
24     IF(INHUB.EQ.0) GO TO 5
25     NK=NS1+2
26     N2=NK-1
27     CALL CHECK(N2,NH,IR2,XON)
28     XRS=XRH
29     IF(XON(IR2).GT.XRH) XRH=XON(IR2)
30     DO 4 I=NK,NH
31     IF((((YON(I).GE.YRH).AND.(YON(I-1)).LE.YRH)).OR.((YCH(I)).LE.YRH),
32     1 AND.(YON(I-1).GE.YRH)).AND.(((XON(I)).LE.XRH).AND.(XON(I-1).GE.
33     2 XRH)).OR.((XON(I).GE.XRH).AND.(XON(I-1).LE.XRH)))) JCHK2 = I
34     4 CONTINUE
35     XRH=XRS
36     5 NK=NH+1
37     NN2=NNK+1
38     CALL CHECK(NK,NT,IR2,XON)
39     XRS=XR2
40     IF(IR2.EQ.0) IR2=NT
41     IF(XON(IR2).GT.XR2) XR2=XON(IR2)
42     DO 6 I=NN2,NT
43     IF((((YON(I).GE.YR2).AND.(YON(I-1)).LE.YR2)).OR.((YCH(I)).LE.YR2),
44     1 AND.(YON(I-1).GE.YR2)).AND.(((XON(I)).LE.XR2).AND.(XON(I-1).GE.
45     2 XR2)).OR.((XON(I).GE.XR2).AND.(XON(I-1).LE.XR2)))) JCHK3 = I
46     6 CONTINUE
47     XR2=XRS
48     IF(YON(JCHK1).GE.YF1) GO TO 10
49     S(JCHK1) = -SQRT((XCN(JCHK1)-XR1)**2+(YON(JCHK1)-YR1)**2)
50     NN1 = JCHK1+2
51     NN2 = JCHK1-1
52     S(NN1-1) = SQRT((XCN(NN1-1)-XR1)**2+(YON(NN1-1)-YR1)**2)
53     GO TO 20
54     10 S(JCHK1) = SQRT((XCN(JCHK1)-XR1)**2+(YON(JCHK1)-YR1)**2)
55     NN1 = JCHK1+1
56     NN2 = JCHK1-2

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57      S(NN2+1) = -SQRT((XCN(NN2+1)-XR1)**2+(YON(NN2+1)-YR1)**2)
58      DO 30 I = NN1,NS1
59          S(I) = S(I-1)+SQRT((XON(I)-XON(I-1))**2+(YON(I)-YON(I-1))**2)
60      30 CONTINUE
61      DO 40 II = 1,NN2
62          I = NN2+1-II
63          S(I) = S(I+1)-SQRT((XON(I)-XON(I+1))**2+(YON(I)-YON(I+1))**2)
64      40 CONTINUE
65      IF (IHUP.EQ.0) GO TO 81
66      IF (YON(JCHK2).GT.YRM) GO TO 50
67      S(JCHK2) = -SQRT((XCN(JCHK2)-XRH)**2+(YON(JCHK2)-YR1)**2)
68      NN1 = JCHK2+2
69      NN2 = JCHK2-1
70      S(NN1-1) = SQRT((XCN(NN1-1)-XRH)**2+(YON(NN1-1)-YR1)**2)
71      GO TO 60
72      50 S(JCHK2) = SQRT((XCN(JCHK2)-XRH)**2+(YON(JCHK2)-YR1)**2)
73      NN1 = JCHK2+1
74      NN2 = JCHK2-2
75      S(NN2+1) = -SQRT((XCN(NN2+1)-XRH)**2+(YON(NN2+1)-YR1)**2)
76      60 IS = NS1+1
77      IS2 = NH
78      DO 70 I = NN1,IS2
79          S(I) = S(I-1)+SQRT((XON(I)-XON(I-1))**2+(YON(I)-YON(I-1))**2)
80      70 CONTINUE
81      DO 80 II = IS,NN2
82          I = NN2+IS-II
83          S(I) = S(I+1)-SQRT((XON(I)-XON(I+1))**2+(YON(I)-YON(I+1))**2)
84      80 CONTINUE
85      81 IF (YON(JCHK3).LT.YR2) GO TO 90
86      S(JCHK3) = -SQRT((XCN(JCHK3)-XR2)**2+(YON(JCHK3)-YR2)**2)
87      NN1 = JCHK3+1
88      NN2 = JCHK3-2
89      S(NN2+1) = SQRT((XCN(NN2+1)-XR2)**2+(YON(NN2+1)-YR2)**2)
90      GO TO 100
91      90 S(JCHK3) = SQRT((XCN(JCHK3)-XR2)**2+(YON(JCHK3)-YR2)**2)
92      NN1 = JCHK3+2
93      NN2 = JCHK3-1
94      S(NN1-1) = -SQRT((XCN(NN1-1)-XR2)**2+(YON(NN1-1)-YR2)**2)
95      100 IS = NH+1
96      DO 110 I=NN1,N1
97          S(I) = S(I-1)+SQRT((XON(I)-XON(I-1))**2+(YON(I)-YON(I-1))**2)
98      110 CONTINUE
99      DO 120 II = IS,NN2
100          I = NN2+IS-II
101          S(I) = S(I+1)+SQRT((XON(I)-XON(I+1))**2+(YON(I)-YON(I+1))**2)
102      120 CONTINUE
103      RETURN
104      END

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210

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1      SUBROUTINE CHECK(IS,IF,IRT,X0)
2      DIMENSION X0(500)
3      IT=0
4      IST=IS+1
5      DO 5 I=IST,IF
6          IF (IT.EQ.1) GO TO 5
7          IF (X0(I).GT.X0(I-1)) IRT=I-1
8          IF (X0(I).GT.X0(I-1)) IT=1
9      5 CONTINUE
10     RETURN
11     END

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1      SUBROUTINE SOLVE
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELNE,ANG(700),ARI(700),AROFF(200)
6      COMMON /VELOC/ V1(700),V2(700),V3(700),V4(700),V5(700),V1X(200),
7      1      V2X(200),V3X(200),V4X(200),V5X(200),V1Y(200),
8      2      V2Y(200),V3Y(200),V4Y(200),V5Y(200)
9      COMMON /CONT/ VC,VS1,VS2,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
10     1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
11     2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
12     COMMON /COUT/ NT,NS1,NH,NP,IN,NX,MND,ICOMP,K,NXH11,NXH12,NXH13,
13     1      NST2,NST3,NST7,NPPR(30),IRAK(30),M1,M2,ICOMP1,IPL,
14     2      THLB
15     COMMON /CONDIT/ ITOTAL,PT,PSTAT,ISTAT,PSTATC,ATOTAL,PTC,RHOST,
16     1      RHOST1,ASTAT,QCINF,RSORTC
17     COMMON /WRIT/ AA1C,AA2C,AA3C,AA4C,AA5C,AA11,AA21,AA31,AA41,AA51,
18     1      AA12,AA22,AA32,AA42,AA52
19     COMMON /SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
20     1      VYINC(200),RHOB(700),RBOPT(700),RHOEO(200),
21     2      VCOM(700),RBOOT(200),VRE(200),VRECOM(200),
22     3      VXCOM(200),VYCOM(200),THETA(200),PSCPTC(700),
23     4      PSOPT(700),CMACH(700),XMACH(700),CPI(700),CPC(700),
24     5      RHOI(700)
25     C -----
26     C
27     C THIS SUBROUTINE SOLVES FOR THE COEFFICIENTS A,B,C, AND D AND
28     C SOLVES FOR VBAR AND V INCOMPRESSIBLE FOR BOTH ON- AND OFF-BODY
29     C POINTS
30     C
31     DIMENSION YNEW(50),VNEW1(50),VNEW2(50),VNEW3(50),VAEW4(50),
32     1      VNEW5(50)
33     6 = 32.174
34     IF(I1.EQ.1) GO TO 30
35     C THE FOLLOWING CALCULATIONS ARE FOR THE CONTROL STATION
36     C UPSTREAM OF THE HUB
37     A = XTEST
38     NPL = 0
39     NH1 = NH+1
40     DO 5 I10=NXH11,NS1
41     IF((A-GE.XON(I10)).AND.(A-LT.XON(I10+1))) THEAL = ANG(I10)
42     5 CONTINUE
43     DO 6 I10=NH1,NXH1
44     IF((A-LE.XON(I10)).AND.(A-GT.XON(I10+1))) THEAH = ANG(I10)
45     6 CONTINUE
46     CALL INTER2(NXH11,NS1,A,YL,V1L,V2L,V3L,V4L,V5L)
47     CALL INTER2(NH1,NXH13,A,YH,V1H,V2H,V3H,V4H,V5H)
48     DO 10 I1 = 1,K
49     IF(I1.EQ.1) I=1
50     IF(I1.EQ.1) GO TO 12
51     I=NPPR(I1-1)+1
52     12 IF(A-NE.XOFF(I)) GO TO 10
53     IF(I1.EQ.1) NPL = 1
54     IF(I1.EQ.1) NPH = APPR(I1)
55     IF(I1.EQ.1) J3 = NPPR(I1)
56     DO 15 J=2,K

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57      IF(II.EQ.J) NPL = 1
58      IF(II.EQ.J) NPH = NPPR(J)-NPPR(J-1)
59      IF(II.EQ.J) J3 = NPPR(J)
60      15 CONTINUE
61      10 CONTINUE
62      IF(NPL.EQ.0) WRITE(6,1000)
63      IF(NPL.FQ.0) STOP
64      DO 20 JJ = NPL,J3
65          VNEW(JJ+2-NPL) = YOFF(JJ)
66          VNEW1(JJ+2-NPL) = V1X(JJ)
67          VNEW2(JJ+2-NPL) = V2X(JJ)
68          VNEW3(JJ+2-NPL) = V3X(JJ)
69          VNEW4(JJ+2-NPL) = V4X(JJ)
70          IF(IHUB.NE.0) VNEW5(JJ+2-NPL) = V5X(JJ)
71      20 CONTINUE
72      VNEW(1) = YL
73      VNEW1(1) = V1L*COS(THCAL)
74      VNEW2(1) = V2L*COS(THCAL)
75      VNEW3(1) = V3L*COS(THCAL)
76      VNEW4(1) = V4L*COS(THCAL)
77      IF(IHUB.NE.0) VNEW5(1) = V5L*COS(THCAL)
78      VNEW(NPH+2) = YH
79      VNEW1(NPH+2) = -V1H*COS(THCAH)
80      VNEW2(NPH+2) = -V2H*COS(THCAH)
81      VNEW3(NPH+2) = -V3H*COS(THCAH)
82      VNEW4(NPH+2) = -V4H*COS(THCAH)
83      IF(IHUB.NE.0) VNEW5(NPH+2) = -V5H*COS(THCAH)
84      NPT = NPH+2
85      CALL INTEG(VNEW1,YNEW,AA1C,NPT)
86      CALL INTEG(VNEW2,YNEW,AA2C,NPT)
87      CALL INTEG(VNEW3,YNEW,AA3C,NPT)
88      CALL INTEG(VNEW4,YNEW,AA4C,NPT)
89      IF(IHUB.NE.0) CALL INTEG(VNEW5,YNEW,AA5C,NPT)
90      AROFF(NPL)=YCU-YCL
91      A1C = AA1C/AROFF(NPL)
92      A2C = AA2C/AROFF(NPL)
93      A3C = AA3C/AROFF(NPL)
94      A4C = AA4C/AROFF(NPL)
95      IF(IHUB.NE.0) A5C = AA5C/AROFF(NPL)
96      IF(IHUB.EQ.0) GO TO 215
97      C THE FOLLOWING CALCULATIONS ARE FOR THE LOWER CONTROL STATION
98      30 IF(I1.EQ.1) GO TO 40
99      IF(I2.EQ.1) GO TO 70
100     40 A = XTEST1
101     NPL = 0
102     NH1 = NS1+1
103     CALL INTER2(NS13,NS1,A,VNEW(1),VNEW1(1),VNEW2(1),VNEW3(1),VNEW4(1),
104     1,VNEW5(1))
105     DO 41 I10=NS13,NS1
106     IF((A.GE.XON(I10)).AND.(A.LT.XON(I10+1))) THEAL = AAC(I10)
107     41 CONTINUE
108     DO 42 I10=NH1,NXH12
109     IF((A.LE.XON(I10)).AND.(A.GT.XON(I10+1))) THEAH = AAG(I10)
110     42 CONTINUE
111     VNEW1(1) = VNEW1(1)*COS(THCAL)
112     VNEW2(1) = VNEW2(1)*COS(THCAL)
113     VNEW3(1) = VNEW3(1)*COS(THCAL)

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114 VNEW4(I) = VNEW4(I)*COS(THEAL)
115 VNEW5(I) = VNEW5(I)*COS(THEAL)
116 CALL INTER2(NH1,NXH12,A,YH,V1H,V2H,V3H,V4H,V5H)
117 DO 50 I1 = 1,M
118 IF (I1.EQ.1) I=1
119 IF (I1.EQ.1) GO TO 45
120 I = NPPR(I1-1)+1
121 45 IF (IA.NE.XOFF(I)).CR.(YOFF(I),GT.YCU1) GO TO 50
122 IF (I1.EQ.1) NPL = 1
123 IF (I1.EQ.1) NPH = APPR(1)
124 IF (I1.EQ.1) J3 = APPR(1)
125 DO 46 J=2,M
126 IF (I1.EQ.J) NPL = APPR(J-1)+1
127 IF (I1.EQ.J) NPH = APPR(J)-NPPR(J-1)
128 IF (I1.EQ.J) J3 = APPR(J)
129 46 CONTINUE
130 50 CONTINUE
131 IF (NPL.EQ.0) WRITE (6,1001)
132 IF (NPL.EQ.0) STOP
133 DO 60 JJ = NPL,J3
134 VNEW1(JJ+2-NPL) = YCFF(JJ)
135 VNEW1(JJ+2-NPL) = V1X(JJ)
136 VNEW2(JJ+2-NPL) = V2X(JJ)
137 VNEW3(JJ+2-NPL) = V3X(JJ)
138 VNEW4(JJ+2-NPL) = V4X(JJ)
139 VNEW5(JJ+2-NPL) = V5X(JJ)
140 60 CONTINUE
141 VNEW(NPH+2) = YH
142 VNEW1(NPH+2) = -V1H*COS(THEAH)
143 VNEW2(NPH+2) = -V2H*COS(THEAH)
144 VNEW3(NPH+2) = -V3H*COS(THEAH)
145 VNEW4(NPH+2) = -V4H*COS(THEAH)
146 VNEW5(NPH+2) = -V5H*COS(THEAH)
147 NPT=NPH+2
148 CALL INTEG(VNEW1,VNEW,AA11,NPT)
149 CALL INTEG(VNEW2,VNEW,AA21,NPT)
150 CALL INTEG(VNEW3,VNEW,AA31,NPT)
151 CALL INTEG(VNEW4,VNEW,AA41,NPT)
152 CALL INTEG(VNEW5,VNEW,AA51,NPT)
153 AROFF(NPL)=YCU1-YCL1
154 A11 = AA11/AROFF(NPL)
155 A21 = AA21/AROFF(NPL)
156 A31 = AA31/AROFF(NPL)
157 A41 = AA41/AROFF(NPL)
158 A51 = AA51/AROFF(NPL)
159 C THE FOLLOWING CALCULATIONS ARE FOR THE UPPER CONTROL STATION
160 70 IF (I1.EQ.0).AND.(I2.EQ.0) GO TO 80
161 A = XTEST2
162 NPL = 0
163 NH1 = NH+1
164 NSH= NH
165 CALL INTER2(NXH12,NSH,A,YNEW1),VNEW1(1),VNEW2(1),VNEW3(1),
166 VNEW4(1),VNEW5(1))
167 DO 71 I10 = NXH12,NSH
168 IF (IA.GE.XON(I10)).AND.(IA.LT.XON(I10+1))) THEAL = AN6(I10)
169 71 CONTINUE
170 DO 72 I10=NH1,NXH12

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171 IF (IA.LE.XON(I10)).AND.(IA.GT.XON(I10+1))) THEAH = ANG(I10)
172 72 CONTINUE
173 VNEW1(I) = VNEW1(I)*COS(THAL)
174 VNEW2(I) = VNEW2(I)*COS(THAL)
175 VNEW3(I) = VNEW3(I)*COS(THAL)
176 VNEW4(I) = VNEW4(I)*COS(THAL)
177 VNEW5(I) = VNEW5(I)*COS(THAL)
178 DO 90 JI = 1,K
179 IF (II.EQ.1) I=1
180 IF (II.EQ.1) 60 TO 55
181 I = NPPR(II-1)+1
182 95 IF (IA.NE.XOFF(I)).OR.(IRAK(II).NE.4)) 60 TO 90
183 IF (II.EQ.1) NPL=1
184 IF (II.EQ.1) NPH = NPPR(1)
185 IF (II.EQ.1) J3 = NPPR(1)
186 DO 96 J=2,K
187 IF (II.EQ.J) NPL=I
188 IF (II.EQ.J) NPH = NPPR(J)-NPPR(J-1)
189 IF (II.EQ.J) J3 = NPPR(J)
190 96 CONTINUE
191 90 CONTINUE
192 IF (NPL.EQ.0) WRITE (6,1002)
193 IF (NPL.EQ.0) STOP
194 NPT = NPH*2
195 CALL INTER2(NH1,NH+1,3,A,VNEW(NPT),VNEW1(NPT),VNEW2(NPT),VNEW3(NPT)
196 1 ,VNEW4(NPT),VNEW5(NPT))
197 VNEW1(NPT) = -VNEW1(NPT)*COS(THAH)
198 VNEW2(NPT) = -VNEW2(NPT)*COS(THAH)
199 VNEW3(NPT) = -VNEW3(NPT)*COS(THAH)
200 VNEW4(NPT) = -VNEW4(NPT)*COS(THAH)
201 VNEW5(NPT) = -VNEW5(NPT)*COS(THAH)
202 DO 100 JJ = NPL,J3
203 VNEW1(JJ+2-NPL) = VCF1(JJ)
204 VNEW1(JJ+2-NPL) = V1X(JJ)
205 VNEW2(JJ+2-NPL) = V2X(JJ)
206 VNEW3(JJ+2-NPL) = V3X(JJ)
207 VNEW4(JJ+2-NPL) = V4X(JJ)
208 VNEW5(JJ+2-NPL) = V5X(JJ)
209 100 CONTINUE
210 CALL INTEG(VNEW1,VNEW,AA12,NPT)
211 CALL INTEG(VNEW2,VNEW,AA22,NPT)
212 CALL INTEG(VNEW3,VNEW,AA32,NPT)
213 CALL INTEG(VNEW4,VNEW,AA42,NPT)
214 CALL INTEG(VNEW5,VNEW,AA52,NPT)
215 AROFF(NPL)=VCU2-VCL2
216 A12 = AA12/AROFF(NPL)
217 A22 = AA22/AROFF(NPL)
218 A32 = AA32/AROFF(NPL)
219 A42 = AA42/AROFF(NPL)
220 A52 = AA52/AROFF(NPL)
221 80 IF (M1.EQ.1) 60 TO 200
222 IF (M2.EQ.1) 60 TO 210
223 C THE FOLLOWING IS FOR THE CASE WHERE UPSTREAM AND LOWER
224 C CONTROL STATIONS ARE INPUT
225 VCL = ATC-A4C
226 VL1= A31-A41
227 VU1 = A51-A41

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228      VCU = A5C-A4C
229      A = VINFCOS(ALFA)
230      B = VINFSIN(ALFA)
231      D = (VS1*VCL-VC*VL1+(A*A1C+B*A2C)*VL1-(A*A11+B*A21)*VCL)/(VU1*VCL-
232      1   VCU*VL1)
233      C = (VS1-(A*A11+B*A21)-D*VU1)/VL1
234      GO TO 220
235      C THE FOLLOWING IS FOR THE CASE WHERE UPPER AND LOWER
236      C CONTROL STATIONS ARE INPUT
237      200 A = VINFCOS(ALFA)
238      B = VINFSIN(ALFA)
239      VL2 = A32-A42
240      VL1 = A31-A41
241      VU1 = A51-A41
242      VU2 = A52-A42
243      D = (VS1*VL2-VS2*VL1+(A*A12+B*A22)*VL1-(A*A11+B*A21)*VL2)/(VU1*VL2-
244      1   -VU2*VL1)
245      C = (VS1-(A*A11+B*A21)-D*VU1)/VL1
246      GO TO 220
247      C THE FOLLOWING IS FOR THE CASE WHERE UPSTREAM AND UPPER
248      C CONTROL STATIONS ARE INPUT
249      210 A = VINFCOS(ALFA)
250      B = VINFSIN(ALFA)
251      VCL = A3C-A4C
252      VL2 = A32-A42
253      VU2 = A52-A42
254      VCU = A5C-A4C
255      D = (VS2*VCL-VC*VL2+(A*A1C+B*A2C)*VL2-(A*A12+B*A22)*VCL)/(VU2*VCL-
256      1   VCU*VL2)
257      C = (VS2-(A*A12+B*A22)-D*VU2)/VL2
258      GO TO 220
259      215 A = VINFCOS(ALFA)
260      B = VINFSIN(ALFA)
261      C = (VC-A*A1C-B*A2C)/(A3C-A4C)
262      C THE FOLLOWING ARE THE CALCULATIONS FOR V AND VBAR
263      220 DO 230 I=1,N1
264          IF (IHUB.EQ.0) VINC(I)=A*V1(I)+B*V2(I)+C*(V3(I)-V4(I))
265          IF (IHUB.EQ.0) GO TO 230
266          VINC(I) = A*V1(I)+B*V2(I)+C*(V3(I)-V4(I))+D*(V5(I)-V4(I))
267      230 CONTINUE
268      DO 240 I=1,NP
269          IF (IHUB.EQ.0) VXINC(I)=A*V1X(I)+B*V2X(I)+C*(V3X(I)-V4X(I))
270          IF (IHUB.EQ.0) VYINC(I)=A*V1Y(I)+B*V2Y(I)+C*(V3Y(I)-V4Y(I))
271          IF (IHUB.EQ.0) GO TO 235
272          VXINC(I)=A*V1X(I)+B*V2X(I)+C*(V3X(I)-V4X(I))+D*(V5X(I)-V4X(I))
273          VYINC(I)=A*V1Y(I)+B*V2Y(I)+C*(V3Y(I)-V4Y(I))+D*(V5Y(I)-V4Y(I))
274      235 VRE(I) = SORT(VXINC(I)**2+VYINC(I)**2)
275          THETA(I) = ATAN(VYINC(I)/VXINC(I))/3.141592654*180.0
276      240 CONTINUE
277      1000 FORMAT(///,1CX,'THERE IS NO RAKE LOCATED AT THE DOWNSTREAM '
278      1          'CONTROL STATION. ',/,10X,'CHECK INPUT LOCATION OF '
279      2          'CONTROL STATION. RUN TERMINATED.')
280      1001 FORMAT(///,1CX,'THERE IS NO RAKE LOCATED AT THE LOWER '
281      1          'CONTROL STATION. ',/,10X,'CHECK INPUT LOCATION OF '
282      2          'CONTROL STATION. RUN TERMINATED.')
283      1002 FORMAT(///,1CX,'THERE IS NO RAKE LOCATED AT THE UPPER '
284      1          'CONTROL STATION. ',/,10X,'CHECK INPUT LOCATION OF '
285      2          'CONTROL STATION. RUN TERMINATED.')
286      RETURN
287      END

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1      SUBROUTINE COMCOR
2      COMMON /COUT/ NT,NS1,NH,NP,IN,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
3      1      NST2,NST3,NST7,NPPR(30),IRAK(30),M1,P2,ICOMP1,IPL,
4      2      INUB
5      COMMON /CONDIT/ TICTAL,PT,PSTAT,TSTAT,PSTATC,ATOTAL,PTC,RHOST,
6      1      RHO TOT,ASTAT,OCINF,RSORTC
7      COMMON /SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
8      1      VYINC(200),RHOB(700),RBORT(700),RHOEC(200),
9      2      VCOF(700),RBOOT(200),VRE(200),VRECOP(200),
10     3      VXCCH(200),VYCOM(200),THETA(200),PSCPIC(700),
11     4      PSOFT(700),CMACH(700),XMACH(700),CPI(700),CPC(700),
12     5      RHC(700)
13     COMMON /CONT/ VC,VCI,VS2,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
14     1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,ANC,ASC,A11,
15     2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIE,VIC1,VIC2
16     C -----
17     C
18     C THIS SUBROUTINE APPLIES THE COMPRESSIBILITY CORRECTION TO THE SOLUTION
19     C
20     EXCON = 1.095*ATOTAL*125.0/216.0
21     PI = 3.141592654
22     P1180 = 180.0/PI
23     VCHK = ATOTAL/RSORT(1,2)
24     DO 10 I=1,NT
25     A9 = 1.0
26     VCOM(I) = VINC(I)/RSORTC**((ABS(VINC(I))/VC))
27     VA = 0.2*(VCOM(I)/ATOTAL)**2
28     IF(VA.GT.1.0) GO TO 25
29     PSO = (1.0-0.2*(VCCM(I)/ATOTAL)**2)**3.5
30     GO TO 30
31     25 PSO = 0.0
32     30 IF((ABS(VCOM(I)).LT.VCHK).OR.(ABS(INX).NE.1)) GO TO 16
33     VSAVE = ABS(VCOM(I))
34     IF((VSAVE/VCHK).GT.2.0) GO TO 10
35     RHORTC = VSAVE*PSO**0.715/EXCON
36     IF(RHORTC.EQ.0.0) RHORTC = 1.0
37     IF(VCOM(I).LT.0.0) A9=-1.0
38     VCOM(I) = VCHK*(1.0*(VSAVE/VCHK-1.0)**(1.0/RHORTC))+A9
39     10 CONTINUE
40     DO 20 I=1,NP
41     VRECOM(I) = VRE(I)/RSORTC**((VRE(I)/VC))
42     VA = 0.2*(VRECOM(I)/ATOTAL)**2
43     IF(VA.GT.1.0) GO TO 35
44     PSI = (1.0-0.2*(VRECOM(I)/ATOTAL)**2)**3.5
45     GO TO 40
46     35 PSI = 0.0
47     40 IF((VRECOM(I).LT.VCHK).OR.(ABS(INX).NE.1)) GO TO 15
48     VSAVE = VRECOM(I)
49     IF((VSAVE/VCHK).GT.2.0) GO TO 15
50     RHORTC = VSAVE*PSI**0.715/EXCON
51     IF(RHORTC.EQ.0.0) RHORTC = 1.0
52     VRECOM(I) = VCHK*(1.0*(VSAVE/VCHK-1.0)**(1.0/RHORTC))
53     15 VXCOM(I) = VXINC(I)*VRECOM(I)/VRE(I)
54     VYCOM(I) = VYINC(I)*VRECOM(I)/VRE(I)
55     THETA(I) = ATAN(VYCOM(I)/VXCOM(I))*P1180
56     20 CONTINUE
57     RETURN
58     END

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1      SUBROUTINE ONBODY
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YP1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      FLND,ANG(700),ARI(700),AROFF(200)
6      COMMON /CONT/ VC,VS1,VS2,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
7      1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,ASC,A11,
8      2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
9      COMMON /COUT/ NT,NS1,MH,NP,IN,NX,MND,ICOMP,M,NXH11,NXH12,NXH13,
10     1      NST2,NST3,NST7,NPFR(30),IRAK(30),M1,P2,ICOMP1,IPL,
11     2      INUB
12     COMMON /CONDIT/ TTCTAL,PT,PSTAT,ISTAT,PSTATC,ATOTAL,PTC,RHOST,
13     1      RHOTOT,ASTAT,QCINF,RSQRTC
14     COMMON /SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
15     1      VYINC(200),RHOBI(700),RBOBI(700),RHOBO(200),
16     2      VCOPI(700),RBOOT(200),VRE(200),VRECOPI(200),
17     3      VXCOM(200),VYCOM(200),THETA(200),PSOPTIC(700),
18     4      PSOPT(700),CMACH(700),XMACH(700),CPI(700),CPC(700),
19     5      RHOI(700)
20     C -----
21     C -----
22     C THIS SUBROUTINE CALCULATES THE ON-BODY PROPERTIES
23     C -----
24     DIMENSION DIMDUM(5)
25     DO 10 I = 1,NT
26     VCONC = 0.2*(ABS(VCOM(I))/ATOTAL)**2
27     VCON = 0.2*(ABS(VINC(I))/ATOTAL)**2
28     IF(VINF.EQ.0.0) CPI(I) = 9999.0
29     IF(VINF.EQ.0.0) CPC(I) = 9999.0
30     IF(VINF.EQ.0.0) GO TO 4
31     CPI(I) = 1.0-(ABS(VINC(I)/VINP))**2
32     CPC(I) = (PY-PSTAT-10.5*RHOBI(I)*VCOM(I)**2)/QCINF
33     4 IF(VCONC.GT.1.0) PSOPTC(I) = 0.0
34     IF(VCONC.GT.1.0) CMACH(I) = 999.0
35     IF(VCON.GT.1.0) XMACH(I) = 999.0
36     IF(VCON.GT.1.0) PSOPT(I) = 0.0
37     IF(VCON.GT.1.0) ROCI(I) = 2000.0
38     IF(VCONC.GT.1.0).AND.(VCON.GT.1.0) GO TO 10
39     IF(VCONC.GT.1.0) GO TO 5
40     PSOPTC(I) = (1.0-VCONC)**3.5
41     CMACH(I) = ABS(VCCPI(I))/ATOTAL/SCRT(1.0-VCONC)
42     IF(VCON.GT.1.0) GO TO 10
43     5 PSOPT(I) = (1.0-0.5*RHOTOT*VINC(I)**2/PT)
44     XMACH(I) = ABS(VINC(I))/ATOTAL/SCRT(1.0-VCON)
45     10 CONTINUE
46     WRITE(6,100)
47     J=0
48     DO 15 I=1,NS1
49     IF(ICOMP.EQ.0) GO TO 11
50     IF(ICMACH(I).LT.1.0).OR.(ABS(INX).NE.1) GO TO 12
51     IF(ICMACH(I-1).LT.1.0) WRITE(6,160) I
52     IF(ICMACH(I-1).LT.1.0) J = 3
53     GO TO 12
54     11 IF(IMACH(I).LT.1.0).OR.(ABS(INX).NE.1) GO TO 12
55     IF(IMACH(I-1).LT.1.0) WRITE(6,160) I
56     IF(IMACH(I-1).LT.1.0) J = 3

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57      12 IF(IICOMP.EQ.0) GO TO 13
58      IF((CMACH(I).LT.1.C).AND.(CMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
59      1 WRITE(6,170) I
60      IF((CMACH(I).LT.1.C).AND.(CMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
61      1 J = J+2
62      GO TO 15
63      13 IF((XMACH(I).LT.1.C).AND.(XMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
64      1 WRITE(6,170) I
65      IF((XMACH(I).LT.1.C).AND.(XMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
66      1 J = J+2
67      15 CONTINUE
68      WRITE(6,105)
69      DO 20 I = 1,NS1
70      J=J+1
71      IF(J.EQ.51) WRITE(6,95)
72      IF(J.EQ.51) WRITE(6,105)
73      IF(J.EQ.51) J=1
74      IF(IICOMP.EQ.0) WRITE(6,110) I,XON(I),YON(I),S(I),V1AC(I),XMACH(I),
75      1 CPI(I),PSOPT(I)
76      IF(IICOMP.EQ.1) WRITE(6,110) I,XON(I),YON(I),S(I),VCCM(I),
77      1 CMACH(I),CPC(I),PSOPTC(I)
78      20 CONTINUE
79      IF(IHUB.EQ.0) GO TO 26
80      WRITE(6,120)
81      IS = NS1+1
82      IST = NH
83      J=0
84      DO 25 I=IS,IST
85      IF(IICOMP.EQ.0) GO TO 21
86      IF((CMACH(I).LT.1.C).OR.(ABS(NX).NE.1)) GO TO 22
87      IF(CMACH(I-1).LT.1.0) WRITE(6,16C) I
88      IF(CMACH(I-1).LT.1.0) J = 3
89      GO TO 22
90      21 IF((XMACH(I).LT.1.C).OR.(ABS(NX).NE.1)) GO TO 22
91      IF(XMACH(I-1).LT.1.0) WRITE(6,16C) I
92      IF(XMACH(I-1).LT.1.0) J = 3
93      22 IF(IICOMP.EQ.0) GO TO 23
94      IF((CMACH(I).LT.1.C).AND.(CMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
95      1 WRITE(6,170) I
96      IF((CMACH(I).LT.1.C).AND.(CMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
97      1 J = J+2
98      GO TO 25
99      23 IF((XMACH(I).LT.1.C).AND.(XMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
100      1 WRITE(6,170) I
101      IF((XMACH(I).LT.1.C).AND.(XMACH(I-1).GE.1.0).AND.(ABS(NX).EQ.1))
102      1 J = J+2
103      25 CONTINUE
104      WRITE(6,105)
105      DO 30 I = IS,IST
106      J=J+1
107      IF(J.EQ.51) WRITE(6,95)
108      IF(J.EQ.51) WRITE(6,105)
109      IF(J.EQ.51) J=1
110      IF(IICOMP.EQ.0) WRITE(6,110) I,XON(I),YON(I),S(I),V1AC(I),XMACH(I),
111      1 CPI(I),PSOPT(I)
112      IF(IICOMP.EQ.1) WRITE(6,110) I,XON(I),YON(I),S(I),VCCM(I),
113      1 CMACH(I),CPC(I),PSOPTC(I)

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114      30 CONTINUE
115      26 IS = NM+1
116      IF(IHUB.EQ.0) WRITE(6,131)
117      IF(IHUB.NE.0) WRITE(6,130)
118      J=0
119      DO 35 I=IS,N1
120      IF(ICOMP.EQ.0) GO TO 31
121      IF((CMACH(I).LT.1.0).OR.(ABS(INX).NE.1)) GO TO 32
122      IF(CMACH(I-1).LT.1.0) WRITE(6,16C) I
123      IF(CMACH(I-1).LT.1.0) J = 3
124      GO TO 32
125      31 IF((XMACH(I).LT.1.0).OR.(ABS(INX).NE.1)) GO TO 32
126      IF(XMACH(I-1).LT.1.0) WRITE(6,16C) I
127      IF(XMACH(I-1).LT.1.0) J = 3
128      32 IF(ICOMP.EQ.0) GO TO 33
129      IF((CMACH(I).LT.1.0).AND.(CMACH(I-1).GE.1.0).AND.(ABS(INX).EQ.1))
130      1 WRITE(6,170) I
131      IF((CMACH(I).LT.1.0).AND.(CMACH(I-1).GE.1.0).AND.(ABS(INX).EQ.1))
132      1 J = J+2
133      GO TO 35
134      33 IF((XMACH(I).LT.1.0).AND.(XMACH(I-1).GE.1.0).AND.(ABS(INX).EQ.1))
135      1 WRITE(6,170) I
136      IF((XMACH(I).LT.1.0).AND.(XMACH(I-1).GE.1.0).AND.(ABS(INX).EQ.1))
137      1 J = J+2
138      35 CONTINUE
139      WRITE(6,105)
140      DO 40 I=IS,N1
141      J=J+1
142      IF(J.EQ.51) WRITE(6,95)
143      IF(J.EQ.51) WRITE(6,105)
144      IF(J.EQ.51) J=1
145      IF(ICOMP.EQ.0) WRITE(6,110) I,XON(I),YON(I),S(I),VJAC(I),XMACH(I),
146      1 CPI(I),PSOPT(I)
147      IF(ICOMP.EQ.1) WRITE(6,110) I,XON(I),YON(I),S(I),VCOM(I),
148      1 CMACH(I),CPC(I),PSOPTC(I)
149      40 CONTINUE
150      C THE FOLLOWING WRITES DATA TAPES FOR THE BOUNDARY LAYER PROGRAM
151      ISTAG = 0
152      ISTAF = 0
153      REWIND 12
154      REWIND 14
155      DO 50 II = 1,NS1
156      I = NS1+1-II
157      IF(VINC(I).LT.0.0) GO TO 45
158      IF((ISTAF.EQ.0).AND.(ICOMP.EQ.0)) WRITE(14,140) XON(I),YON(I),
159      1 VINC(I),XMACH(I),PSOPT(I)
160      IF((ISTAF.EQ.0).AND.(ICOMP.EQ.1)) WRITE(14,140) XON(I),YON(I),
161      1 VCOM(I),CMACH(I),PSOPTC(I)
162      IF(ISTAF.EQ.0) ISTAG=ISTAG+1
163      GO TO 50
164      45 IF((VINC(I)+VINC(I+1)).LT.0.0).AND.(ICOMP.EQ.0).AND.(ISTAF.EQ.0))
165      1 WRITE(14,140) XON(I),YON(I),VINC(I),XMACH(I),PSOPT(I)
166      IF((VINC(I)+VINC(I+1)).LT.0.0).AND.(ICOMP.EQ.1).AND.(ISTAF.EQ.0))
167      1 WRITE(14,140) XON(I),YON(I),VCOM(I),CMACH(I),PSOPTC(I)
168      ISTAF=1
169      50 CONTINUE
170      REWIND 14

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171      IF(ISTAF.EQ.1) ISTAG = ISTAG+1
172      WRITE(12,150) ISTAG
173      DO 55 I=1,ISTAG
174      READ(14,140) DIMDLP
175      WRITE(12,140) DIMDLM
176      55 CONTINUE
177      ISTAG = 0
178      ISTAF = 0
179      NSTA = NH+1
180      REWIND 12
181      REWIND 13
182      REWIND 14
183      DO 60 I=NSTA,NT
184      IF(VINC(I),6T,0.0) GO TO 56
185      IF((ISTAF.EQ.0).AND.(ICOMP.EQ.0)) WRITE(14,140) XON(I),YON(I),
186      1 VINC(I),XMACH(I),PSOPT(I)
187      IF((ISTAF.EQ.0).AND.(ICOMP.EQ.1)) WRITE(14,140) XON(I),YON(I),
188      1 VCOM(I),CMACH(I),PSOPT(I)
189      IF(ISTAF.EQ.0) ISTAG=ISTAG+1
190      GO TO 60
191      56 IF((VINC(I)*VINC(I-1),LT,0.0).AND.(ISTAF.EQ.0).AND.(ICOMP.EQ.0))
192      1 WRITE(14,140) XON(I),YCN(I),VINC(I),XMACH(I),PSOPT(I)
193      IF((VINC(I)*VINC(I-1),LT,0.0).AND.(ISTAF.EQ.0).AND.(ICOMP.EQ.1))
194      1 WRITE(14,140) XON(I),YCN(I),VCOM(I),CMACH(I),PSOPT(I)
195      ISTAF=1
196      60 CONTINUE
197      REWIND 14
198      IF(ISTAF.EQ.1) ISTAG=ISTAG+1
199      WRITE(13,150) ISTAG
200      DO 65 I=1,ISTAG
201      READ(14,140) DIMDLP
202      WRITE(13,140) DIMDLM
203      65 CONTINUE
204      REWIND 13
205      RETURN
206      C *****FORMATS*****
207      95 FORMAT(1H1)
208      100 FORMAT(1H1,10X,'BODY 1',/,10X,'ON-BODY POINTS')
209      105 FORMAT(7,2X,'I',7X,'X',11X,'Y',11X,'S',11X,' V ',8X,'MACH',8X,
210      1 'CP',6X,'PS/PT',/)
211      110 FORMAT(14,1P6E12.3,OPF8.4)
212      120 FORMAT(1H1,10X,'BODY 2',/,10X,'ON-BODY POINTS')
213      130 FORMAT(1H1,10X,'BODY 3',/,10X,'ON-BODY POINTS')
214      131 FORMAT(1H1,10X,'BODY 2',/,10X,'ON-BODY POINTS')
215      140 FORMAT(OPSF10.4)
216      150 FORMAT(13)
217      160 FORMAT(7,2X,'ON-BODY SUPERSONIC VELOCITY CORRECTION START. '
218      1 'I = ',14,/)
219      170 FORMAT(2X,'SUPERSONIC VELOCITY CORRECTION STOP, I = ',14,/)
220      END

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1      SUBROUTINE OFBODY
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCL2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CLTOF2,CUTOFH,
5      3      ELAE,ANG(700),AR(700),AROFF(200)
6      COMMON /COUT/ NT,NS1,NH,NP,IW,NX,FND,ICOMP,K,NXH11,NXH12,NXH13,
7      1      NST2,NST3,NST7,NPPR(30),IRAK(30),M1,P2,ICOMP1,IPL,
8      2      IHUP
9      COMMON /CONDIT/ TTOTAL,PT,PSTAT,TSTAT,PSTATC,ATOTAL,PTC,RHOST,
10     1      RHO10T,ASTAT,OCINF,RSORTC
11     COMMON /SOLUT/ VBAR(700),VBARC(200),VINC(700),VXINC(200),
12     1      VYINC(200),RHOB(700),REORT(700),RHOB(200),
13     2      VCOM(700),PBOT(200),VRE(200),VRECOM(200),
14     3      VXCM(200),VYCM(200),THETA(200),PSCPTC(700),
15     4      PSCFT(700),CMACH(700),XMACH(700),CP1(700),CPC(700),
16     5      RHO1(700)
17     COMMON /SOLUTO/ PSOFPC(200),PSOFF(200),CHACO(200),XPACO(200),
18     1      RHO1(200)
19     COMMON/PICT/VPERIN,XX,XMIN,EXEP,YY,YMIN,ORD,ENSTOR,AL,AAAA
20
21     C
22     C THIS SUBROUTINE CALCULATES OFF-BODY PROPERTIES
23     C
24     DIMENSION WYOT(30),YINT(200),PVI(200),WFRAC(200)
25     WRITE(6,1)
26     DO 10 I=1,NP
27     VCONC = 0.2*(VRECOM(I)/ATOTAL)**2
28     VCON = 0.2*(VRE(I)/ATOTAL)**2
29     IF(VCONC.GT.1.0) PSOFPC(I) = 0.0
30     IF(VCONC.GT.1.0) CHACO(I) = 999.0
31     IF(VCON.GT.1.0) XMACH(I) = 999.0
32     IF(VCON.GT.1.0) PSCFT(I) = 0.0
33     IF(VCON.GT.1.0) RHOC(I) = 2000.0
34     IF((VCONC.GT.1.0).AND.(VCON.GT.1.0)) GO TO 10
35     IF(VCONC.GT.1.0) GO TO 5
36     PSOFPC(I) = (1.0-VCONC)**3.5
37     CHACO(I) = VRECOM(I)/ATOTAL/SQRT(1.0-VCONC)
38     IF(VCON.GT.1.0) GO TO 10
39     5 PSOFPI = (1.0-0.5*RHO10T*VRE(I)**2/PT)
40     XMACH(I) = VRE(I)/ATOTAL/SQRT(1.0-VCON)
41     10 CONTINUE
42     DO 50 I=1,K
43     IF(I.EQ.1) NPH = NPPR(1)
44     IF(I.EQ.1) NPL = 1
45     IF(I.EQ.1) J3 = NPH
46     II = NPPR(I-1)+1
47     20 DO 30 J=2,K
48     IF(I.NE.J) GO TO 30
49     NPL = II
50     NPH = NPPR(J)-NPPR(J-1)
51     J3 = NPPR(J)
52     30 CONTINUE
53     C
54     C IRAK = 1 RAKE IS UPSTREAM OF BODY
55     C = 2 RAKE IS ON LOWER EXTENDED LIP
56     C = 3 RAKE IS COMPLETELY IN LEFT UPSTREAM OF HUE

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57 C = 4 RAKE IS DOWNSTREAM OF AND ABOVE HUB
58 C = 5 RAKE IS DOWNSTREAM OF AND BELOW HUB
59 C = 6 RAKE IS OUTSIDE AND BELOW INLET
60 C = 7 RAKE IS OUTSIDE AND ABOVE INLET
61 THEAL = 0.0
62 THEAH = 0.0
63 IF (IRAK(1).EQ.1) GC TO 40
64 IF (IRAK(1).EQ.2) GC TO 60
65 IF (IRAK(1).EQ.3) GC TO 70
66 IF (IRAK(1).EQ.4) GC TO 80
67 IF (IRAK(1).EQ.5) GC TO 90
68 IF (IRAK(1).EQ.6) GC TO 220
69 IF (IRAK(1).EQ.7) CC TO 230
70 40 WRITE(6,500) I
71 WRITE(6,521)
72 DO 45 L = NPL,J3
73 LL = L-NPL+1
74 YINT(LL) = YOFF(L)
75 IF (ICOMP.EQ.0) RV(LL) = VXINC(L)*RHOTOT
76 RHO = (PSOFFC(L)**(1./1.4))
77 IF (ICOMP.EQ.1) RV(LL) = VXCOM(L)*RHO*RHOTOT
78 45 CONTINUE
79 CALL INTEG(RV,YINT,WTOT(I),LL)
80 LI = NPL+1
81 DO 46 L = LI,J3
82 LL = L-NPL+1
83 CALL INTEG(RV,YINT,W,LL)
84 WPRAC(LL) = W/WTOT(I)
85 46 CONTINUE
86 WFRAC(1) = 0.0
87 DO 100 M = NPL,J3
88 NM = M-NPL+1
89 IF (ICOMP.EQ.0) WRITE(6,550) NM,XCFF(N),YOFF(N),VXIAC(N),VYINC(N),
90 1 VRE(N),THETA(N),XMACO(N),PSOFF(N),WFRAC(N)
91 IF (ICOMP.EQ.1) WRITE(6,550) NM,XCFF(N),YOFF(N),VXCCP(N),VYCOM(N),
92 1 VRECOM(N),THETA(N),CMACO(N),PSOFFC(N),WFRAC(N)
93 100 CONTINUE
94 GO TO 50
95 60 WRITE(6,500) I
96 WRITE(6,521)
97 DO 110 I10 = NXMI,NSI
98 IF ((XOFF(NPL).GE.XON(I10)).AND.(XOFF(NPL).LT.XON(I10+1))) THEAL =
99 1 ANG(I10)
100 110 CONTINUE
101 CALL INTER3(XON,NX+1,NSI,XOFF(NPL),YINT(1),VLC,VL),ROC,VPL,
102 1 CML,XML,PL,PIL)
103 VLIx = VLI*Cos(THAL)
104 VLIy = VLI*Sin(THAL)
105 VLCx = VLC*Cos(THAL)
106 VLCy = VLC*Sin(THAL)
107 THEAL = THEAL+180./3.141592654
108 IF (ICOMP.EQ.0) RV(I1) = VLIx*RHOTCT
109 IF (ICOMP.EQ.1) RV(I1) = VLCx*PL**(.1./1.4)*RHOTCT
110 DO 120 L = NPL,J3
111 NM = L-NPL+2
112 YINT(NM) = YOFF(L)
113 IF (ICOMP.EQ.0) RV(NM) = VXINC(L)*RHOTCT

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114      RHO=(PSOFFP(L)**(1./1.4))
115      IF(ICOMP.EQ.1) RV(NW) = RHO*VXCOP(L)*RHOTOT
116      120 CONTINUE
117      121 X3 = 0.0
118      CALL INTEG(RV,YINT,WTOT(I),NW)
119      IF(ICOMP.EQ.0) WRITE(6,530) XOFF(NPL),YINT(I),VLIX,VLIV,VLIJ,THEAL,
120      1      XML,PIL,X3
121      IF(ICOMP.EQ.1) WRITE(6,530) XOFF(NPL),YINT(I),VLCX,VLCY,VLC,THEAL,
122      1      CML,PL,X3
123      DO 130 L=NPL,J3
124      LL = L-NPL+2
125      CALL INTEG(RV,YINT,W,LL)
126      LI = L-NPL+1
127      WFRAC(L) = W/WTOT(I)
128      IF(ICOMP.EQ.0) WRITE(6,550) LI,XCFF(L),YOFF(L),VXINC(L),VYINC(L),
129      1      VRE(L),THEAL(L),XMACO(L),PSOFFP(L),WFRAC(L)
130      IF(ICOMP.EQ.1) WRITE(6,550) LI,XCFF(L),YOFF(L),VXCOP(L),VYCOM(L),
131      1      VRECOM(L),THEAL(L),CMACO(L),PSOFFP(L),
132      2      WFRAC(L)
133      130 CONTINUE
134      GO TO 50
135      70 DO 140 I10 = NXH1,NS1
136      IF((XOFF(NPL).GE.XCN(I10)).AND.(XOFF(NPL).LT.XON(I10+1))) THEAL =
137      1      ANG(I10)
138      140 CONTINUE
139      NNS = NH+1
140      DO 150 I10 = NNS,NXH13
141      IF((XOFF(NPL).LE.XCN(I10)).AND.(XOFF(NPL).GT.XON(I10+1))) THEAH =
142      1      ANG(I10)
143      150 CONTINUE
144      NH1 = NPH+2
145      CALL INTER3(XON,NXH1,NS1,XOFF(NPL),YINT(I),VLC,VLI,ROC,VBL,
146      1      CML,XPL,PL,PIL)
147      CALL INTER3(XON,NNS,NXH13,XOFF(NPL),YINT(NH1),VUC,VUI,ROCU,
148      1      VRU,CML,XMUT,PU,PIU)
149      160 VUC = -VUC
150      VUI = -VUI
151      VLIX = VLI*COS(THEAL)
152      VLIV = VLI*SIN(THEAL)
153      VLCX = VLC*COS(THEAL)
154      VLCY = VLC*SIN(THEAL)
155      VUIX = VUI*COS(THEAH)
156      VUIY = VUI*SIN(THEAH)
157      VUCX = VUC*COS(THEAH)
158      VUCY = VUC*SIN(THEAH)
159      THEAL = THEAL*180./3.141592654
160      THEAH = THEAH*180./3.141592654
161      WRITE(6,500) I
162      WRITE(6,521)
163      IF(ICOMP.EQ.1)RV(I) = VLCX*PL**(.1./1.4)*RHOTOT
164      IF(ICOMP.EQ.0)RV(I) = VLIX*RHOTOT
165      IF(ICOMP.EQ.0)RV(NH1) = VUIX*RHOTOT
166      IF(ICOMP.EQ.1)RV(NH1) = VUCX*PU**(.1./1.4)*RHOTOT
167      DO 165 L = NPL,J3
168      LL = L-NPL+2
169      IF(ICOMP.EQ.0) RV(LL) = VXINC(L)*RHOTOT
170      VINT(LL) = YOFF(L)

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171      RHO=(PSOFPC(I)*.1/.4)
172      IF(IComp.EQ.1) RVILL = VXCOM(I)*RHO*RHOTOT
173      165 CONTINUE
174      NW=LL
175      IRT=NPPR(I)
176      IF((IRAK(I).EQ.4).AND.(XOFF(IRT).LT.XON(NXH3))) GO TO 121
177      X3=0.0
178      CALL INTEG(RV,YINT,WTOT(I),NH1)
179      IF(IComp.EQ.0) WRITE(6,530) XOFF(NPL),YINT(I),VLIY,VLI,THEAL.
180      1      XML,PIL,X3
181      IF(IComp.EQ.1) WRITE(6,530) XOFF(NPL),YINT(I),VLCX,VLCY,VLC,THEAL.
182      1      CML,PL,X3
183      DO 170 L = NPL,J3
184      LL = L-NPL+2
185      CALL INTEG(RV,YINT,W,LL)
186      LI = L-NPL+1
187      WFRAC(L) = W/WTOT(I)
188      IF(IComp.EQ.0) WRITE(6,550) LI,XCFF(L),YOFF(L),VXIAC(L),VYINC(L),
189      1      VRE(L),THETA(L),XMACO(L),PSGFC(L),WFRAC(L)
190      IF(IComp.EQ.1) WRITE(6,550) LI,XCFF(L),YOFF(L),VXC(L),VYCOM(L),
191      1      VRECOM(L),THETA(L),CMACO(L),PSOFPC(L),
192      2      WFRAC(L)
193      170 CONTINUE
194      LI=LI+1
195      WFRAC(L) = 1.0
196      IF(IComp.EQ.0) WRITE(6,530) XOFF(NPL),YINT(NH1),VUIX,VUIY,VUI.
197      1      TPEAH,XMUI,PIU,WFRAC(L)
198      IF(IComp.EQ.1) WRITE(6,530) XOFF(NPL),YINT(NH1),VUCX,VUCY,VUC.
199      1      TPEAH,CMU,PU,WFRAC(L)
200      GO TO 50
201      80 NT1 = NH
202      DO 180 I10 = NXH12,NT1
203      IF((XOFF(NPL).GE.XCN(I10)).AND.(XOFF(NPL).LT.XON(I1C+1))) THEAL =
204      1      ANG(I10)
205      180 CONTINUE
206      MNN = NH+1
207      DO 190 I10 = MNN,NXH13
208      IF((XOFF(NPL).LE.XCN(I10)).AND.(XOFF(NPL).GT.XON(I1C+1))) THEAH =
209      1      ANG(I10)
210      190 CONTINUE
211      NH1 = NPH+2
212      CALL INTER3(XON,NXH12,NT1,XOFF(NPL),YINT(1),VLC,VLI,ROC,VPL,
213      1      CML,XML,PL,PIL)
214      CALL INTER3(XON,MNN,NXH13,XOFF(NPL),YINT(NH1),VUC,VUI,ROCU,
215      1      VBU,CML,XMUI,PU,PIU)
216      GO TO 160
217      90 DO 200 I10 = NXH11,NS1
218      IF((XOFF(NPL).GE.XCN(I10)).AND.(XOFF(NPL).LT.XON(I1C+1))) THEAL =
219      1      ANG(I10)
220      200 CONTINUE
221      MNN = NS1+1
222      DO 210 I10 = MNN,NXH12
223      IF((XOFF(NPL).LE.XCN(I10)).AND.(XOFF(NPL).GT.XON(I1C+1))) THEAH =
224      1      ANG(I10)
225      210 CONTINUE
226      NH1 = NPH+2
227      CALL INTER3(XON,NXH11,NS1,XOFF(NPL),YINT(1),VLC,VLI,ROC,VPL,

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228      1      CML,XML,PL,PIL)
229      CALL INTER3(XON,NNN,NXH12,XOFF(NPL),YINT(NH1),VUC,VUI,ROCU,
230      1      VBU,CPL,XMUI,PU,PIU)
231      60 TO 160
232      220 DO 221 I10 = 1,NXP11
233      IF((XOFF(NPL).LT.XCN(I10)).AND.(XOFF(NPL).GT.XON(I1C+1))) THEAH =
234      1      ANG(I10)
235      221 CONTINUE
236      WRITE(6,500) I
237      WRITE(6,521)
238      NH1 = NPH+1
239      CALL INTER3(XON,1, NXH11,XOFF(NPL),YINT(NH1),VUC,VUI,ROCU,
240      1      VBU,CPL,XMUI,PU,PIU)
241      IF(ICOMP.EQ.C) RV(NH1) = -VUI*COS(THEAH)*RHOTOT
242      IF(ICOMP.EQ.1) RV(NH1) = -VUC*COS(THEAH)*PU*(1./1.4)*RHOTOT
243      VUIX = -VUI*COS(THEAH)
244      VUIY = -VUI*SIN(THEAH)
245      VUCX = -VUC*COS(THEAH)
246      VUCY = -VUC*SIN(THEAH)
247      VUI = -VUI
248      VUC = -VUC
249      THEAH = THEAH*180./3.141592654
250      DO 222 JJ = NPL,J3
251      J1 = JJ-NPL +1
252      IF(ICOMP.EQ.0) RV(J1) = VXINC(JJ)*RHOTOT
253      RHO=(PSOFPC(J1)**(1./1.4))
254      IF(ICOMP.EQ.1) RV(J1) = RHO*VXCOP(JJ)*RHOTOT
255      YINT(J1) = YOFF(JJ)
256      222 CONTINUE
257      CALL INTEG(RV,YINT,WTOT(I),NH1)
258      DO 223 J = NPL,J3
259      J1 = J-NPL+2
260      CALL INTEG(RV,YINT,W,J1)
261      WFRAC(J1) = W/WTOT(I)
262      WFRAC(I) = 0.0
263      J2 = J1-1
264      IF(ICOMP.EQ.0) WRITE(6,550) J2,XCFF(J),YOFF(J),VXINC(J),VYINC(J),
265      1      VRE(J),THETA(J),XPACO(J),PSOFP(J),WFRAC(J2)
266      IF(ICOMP.EQ.1) WRITE(6,550) J2,XCFF(J),YOFF(J),VXCOP(J),VYCOM(J),
267      1      VRECOM(J),THETA(J),CHACO(J),PSOFP(J),WFRAC(J2)
268      223 CONTINUE
269      J2 = J2+1
270      IF(ICOMP.EQ.C) WRITE(6,550) J2,XCFF(NPL),YINT(NH1),VUIX,VUIY,VUI,
271      1      THEAH,XMUI,PIU,WFRAC(J2)
272      IF(ICOMP.EQ.1) WRITE(6,550) J2,XCFF(NPL),YINT(NH1),VUCX,VUCY,VUC,
273      1      THEAH,CMU,PU,WFRAC(J2)
274      60 TO 50
275      230 WRITE(6,500) I
276      WRITE(6,521)
277      DO 231 I10 = NXH13,NT
278      IF((XOFF(NPL).GE.XCN(I10)).AND.(XOFF(NPL).LT.XON(I1C+1))) THEAL =
279      1      ANG(I10)
280      231 CONTINUE
281      NH1 = NPH+1
282      CALL INTER3(XON,NXH13,NT ,XOFF(NPL),YINT(I),VLC,VLI,ROC,VBL,
283      1      CML,XML,PL,PIL)
284      VLIX = VLI*COS(THEAL)

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285      VLIY = VLI*SIN(THAL)
286      VLCX = VLC*COS(THAL)
287      VLCY = VLC*SIN(THAL)
288      THEAL = THEAL*180./3.141592654
289      IF(IComp.EQ.0) RV(1) = VLI*RHOTCT
290      IF(IComp.EQ.1) RV(1) = VLC*PL**1./1.4)*RHOTCT
291      DO 232 J = NPL,J3
292      J1 = J-NPL+2
293      IF(IComp.EQ.0) RV(1) = VXINC(J)*PHOTOT
294      RHO=(PSOFPC(J)**1./1.4)
295      IF(IComp.EQ.1) RV(1) = RHO*VXCOM(J)*RHOTOT
296      YINT(J) = YOFF(J)
297      232 CONTINUE
298      CALL INTEG(RV,YINT,WTOT(I),NH1)
299      WFRAC(1) = 0.0
300      IF(IComp.EQ.0) WRITE(6,530) XOFF(NPL),YINT(1),VLI*VLIY,VLI,THEAL,
301      1      XML,PIL,WFRAC(1)
302      IF(IComp.EQ.1) WRITE(6,530) XOFF(NPL),YINT(1),VLCX,VLCY,VLC,THEAL,
303      1      CML,PL,WFRAC(1)
304      DO 233 J = NPL,J3
305      J1 = J-NPL+2
306      CALL INTEG(RV,YINT,W,J1)
307      WFRAC(J1) = W/WTOT(I)
308      J2 = J1-1
309      IF(IComp.EQ.0) WRITE(6,550) J2,XOFF(J1),YOFF(J1),VXINC(J1),VYINC(J1),
310      1      VRE(J1),THETA(J1),XPAO(J1),PSOFPC(J1),WFRAC(J1)
311      IF(IComp.EQ.1) WRITE(6,550) J2,XOFF(J1),YOFF(J1),VXCOM(J1),VYCOM(J1),
312      1      VRECOM(J1),THETA(J1),CNACO(J1),PSOFPC(J1),WFRAC(J1)
313      233 CONTINUE
314      50 CONTINUE
315      C THE FOLLOWING CALCULATES INTEGRATED PAKE WEIGHT FLOW DATA
316      WRITE(6,570)
317      DO 240 I=1,K
318      WTOT(I) = WTOT(I)*32.174/12.0
319      IF(I.EQ.1) NPL = 1
320      IF(I.NE.1) NPL = NPPR(I-1) + 1
321      I9 = NPPR(I)
322      IF(IRAK(I).EQ.1) AREA = YOFF(I9)-YOFF(NPL)
323      IF(IRAK(I).EQ.2) CALL INTER(XON,YON,NXH11,NS13,XOFF(NPL),YL)
324      IF(IRAK(I).EQ.2) AREA = YOFF(I9)-YL
325      NN=NH+1
326      NR=NS1+1
327      IF(IRAK(I).EQ.3).OR.(IRAK(I).EQ.4)
328      1CALL INTER(XON,YON,NN,NXH13,XOFF(NPL),YH)
329      IF(IRAK(I).EQ.5) CALL INTER(XON,YON,NR,NXH12,XOFF(NPL),YH)
330      IF(IRAK(I).EQ.3).OR.(IRAK(I).EQ.5)
331      1CALL INTER(XON,YON,NXH11,NS1,XOFF(NPL),YL)
332      IF(IRAK(I).EQ.4) CALL INTER(XON,YON,NXH12,NN,XOFF(NPL),YL)
333      IF(IRAK(I).EQ.3).OR.(IRAK(I).EQ.4).OR.(IRAK(I).EQ.5) AREA =
334      1      YH-YL
335      IF(IRAK(I).EQ.6) CALL INTER(XON,YON,1, NXH11,XOFF(NPL),YH)
336      IF(IRAK(I).EQ.6) AREA = YH-YOFF(NPL)
337      IF(IRAK(I).EQ.7) CALL INTER(XON,YON,NXH13,NY, XOFF(NPL),YL)
338      IF(IRAK(I).EQ.7) AREA = YOFF(I9)-YL
339      WDOTCA = WTOT(I)/AREA*12.0*SQRT(11TOTAL/519.67)/PTC*2116.23
340      EM=0.1
341      235 EF = 85.3848*EM/(1.0+0.2*EM*EM)**1.0-WDOTCA

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342      DFDN = RS.3848/(1.(+0.2*EM*EM)**2.C*(1.0-1.2*EF)/1.0+0.2*EM*EM)
343      1
344      EM = EM-EF/DFDN
345      IF(ABS(EF/DFDN/(EM+EF/DFDN)).GT.C.01) GO TO 235
346      WRITE(6,580) 1,XOFF(NPL),IRAK(I),WTOT(I),WDOTCA,EM
347      IF(XOFF(NPL).EQ.XTEST) EMSTOR=EM
348      240 CONTINUE
349      RETURN
350      C *****FORMATS*****
351      1 FORMAT(1H1)
352      500 FORMAT(////,10X,'RAKE NUMBER ',I2)
353      521 FORMAT(/,2X,'I',6X,'X',10X,'Y',9X,'VX',9X,'VY',9X,'VRE',7X,'THETA'
354      1      ,6X,'MACH',6X,'PS/PT',4X,'WFRAC',/)
355      530 FORMAT(4X,1P7E11.3,OPF8.4,1PE11.3)
356      550 FORMAT(14,1P7E11.3,OPF8.4,1PE11.3)
357      570 FORMAT(////,10X,'RAME WEIGHT FLOW DATA',/,3X,'I',8X,'X',4X,'IRAK',
358      1      ,4X,'WDOT',10X,'WDOTCA',8X,'MACH',/)
359      580 FORMAT(14,5X,OPF8.4,14,1P3E14.5)
360      END

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1      SUBROUTINE VPARIT(VBAR,ATOTAL,RHOTOT,RHOBAR)
2      C
3      C THIS SUBROUTINE ITERATIVELY CALCULATES RHOBAR
4      C
5      VCRIT = ATOTAL/SQRT(1.2)
6      I=0
7      VGUES = VBAR
8      10 VGUESA = (VGUES/ATOTAL)**2
9      A = 1.0-0.2*VGUESA
10     B = A -VGUESA
11     VCOMP = (VBAR-A**2.5*VGUES)/(A**1.5*B)*VGUES
12     IF(ABS(VCOMP-VGUES)/VCOMP.LT.0.0001) GO TO 15
13     I =I+1
14     IF(VCOMP.GE.VCRIT) VCOMP = 0.5*(VGUES + VCRIT)
15     VGUES = VCOMP
16     IF(I.GT.20) GO TO 15
17     GO TO 10
18     15 RHOBAR = 11.0-0.2*(VCOMP/ATOTAL)**2)**2.5*RHOTOT
19     IF(I.GT.20) WRITE(6,20) VBAR,VCOMP,RHOBAR
20     IF(I.GT.20) VBAR = VCOMP*RHOBAR/RHOTOT
21     RETURN
22     C *****FORMATS*****
23     20 FORMAT(1H0,'I EXCEEDS 20 ITERATIONS FOR RHOBAR',5X,'VBAR = ',
24     1      1PE10.3,2X,'VCOMP = ',1PE10.3,2X,'RHOBAR = ',1PE10.3,/,
25     2      'VBAR HAS BEEN REDUCED TO VCOMP*RHOBAR/RHOTOT, WHERE ',
26     3      'VCOMP = VCRITICAL')
27     END

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1      SUBROUTINE INTER(A,B,N1,N2,C,D)
2      C
3      C THIS SUBROUTINE INTERPOLATES ARRAY A FOR THE VALUE C
4      C AND RETURNS IN D THE CORRESPONDING VALUE FROM B .
5      C
6      DIMENSION X(700),Y(700),A(1),P(1)
7      N = N2-N1+1
8      DO 10 I=1,N
9      X(I)=A(N1+I-1)
10     Y(I)=B(N1+I-1)
11     10 CONTINUE
12     CALL SORTXY(X,Y,N)
13     DO 15 I=1,N
14     K=I
15     IF(C-X(I)) 25,20,15
16     15 CONTINUE
17     20 D = Y(K)
18     GO TO 30
19     25 IF(K.EQ.1) GO TO 35
20     IF(K.EQ.N) K=N-1
21     IF(X(K).EQ.X(K+1)) K=K-1
22     W1 = (C-X(K))*((C-X(K+1))/(X(K)-X(K+1)))/(X(K)-X(K+1))
23     W2 = ((C-X(K-1))*((C-X(K+1))/(X(K)-X(K-1)))/(X(K)-X(K+1))
24     W3 = ((C-X(K-1))*((C-X(K))/(X(K+1)-X(K-1)))/(X(K+1)-X(K))
25     D = Y(K-1)*W1+Y(K)*W2+Y(K+1)*W3
26     30 RETURN
27     35 D = Y(1)
28     RETURN
29     END

```

228

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1      SUBROUTINE SORTXY(X,Y,NPTS)
2      C
3      C THIS SUBROUTINE SORTS X INTO ASCENDING ORDER
4      C
5      DIMENSION X(300),Y(300)
6      N = NPTS
7      NN = N-1
8      DO 10 KT = 1,NN
9      XMIN = X(KT)
10     JAD = KT
11     JKL = KT+1
12     DO 20 JK = JKL,N
13     IF(XMIN-X(JK)) 20,20,25
14     25 XMIN = X(JK)
15     JAD = JK
16     20 CONTINUE
17     YMIN = Y(JAD)
18     X(JAD) = X(KT)
19     Y(JAD) = Y(KT)
20     X(KT) = XMIN
21     Y(KT) = YMIN
22     10 CONTINUE
23     RETURN
24     END

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4      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELAP,ANG(700),AR(700),AROFF(200)
6      COMMON /VELOC/ V1(700),V2(700),V3(700),V4(700),V5(700),V1X(200),
7      1      V2X(200),V3X(200),V4X(200),V5X(200),V1Y(200),
8      2      V2Y(200),V3Y(200),V4Y(200),V5Y(200)
9      COMMON /COUT/ NT,NS1,NH,NP,IW,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
10     1      NST1,NST3,NST7,NPFF(30),IRAK(30),M1,P2,ICOMP1,IPL,
11     2      IMLE
12     C -----
13     CALL INTER(XON,YON,I1,I2,A,B)
14     CALL INTER(XON,V1,I1,I2,A,C)
15     CALL INTER(XON,V2,I1,I2,A,D)
16     CALL INTER(XON,V3,I1,I2,A,E)
17     CALL INTER(XON,V4,I1,I2,A,F)
18     IF(IHUB.NE.0) CALL INTER(XON,V5,I1,I2,A,G)
19     IF(IHUB.EQ.0) G=0.C
20     RETURN
21     END

```

```

1      SUBROUTINE INTER3(A,I1,I2,C,D,E,F,H,P,Q,R,S2,T)
2      COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S1(700),
3      1      S1(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
4      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CUTOF2,CUTOFH,
5      3      ELAP,ANG(700),AR(700),AROFF(200)
6      COMMON /SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
7      1      VYINC(200),RHOB(700),RBOBT(700),RHOB(200),
8      2      VCOM(700),RBOOT(200),VRE(200),VRECOM(200),
9      3      VXCM(200),VYCOM(200),THETA(200),PSOPTC(700),
10     4      PSOPT(700),CMACH(700),XMACH(700),CPI(700),CPC(700),
11     5      RMO(700)
12     C -----
13     DIMENSION A(700)
14     CALL INTER(A,YON, I1,I2,C,D)
15     CALL INTER(A,VCOM, I1,I2,C,E)
16     CALL INTER(A,VINC, I1,I2,C,F)
17     CALL INTER(A,RBOBT,I1,I2,C,H)
18     CALL INTER(A,VBAR, I1,I2,C,P)
19     CALL INTER(A,CMACH,I1,I2,C,Q)
20     CALL INTER(A,XMACH,I1,I2,C,R)
21     CALL INTER(A,PSOPTC,I1,I2,C,S2)
22     CALL INTER(A,PSOPT, I1,I2,C,T)
23     RETURN
24     END

```

```

1      SUBROUTINE INTEG(P,A,C,I1)
2      C
3      C THIS SUBROUTINE IS A TRAPEZOIDAL INTEGRATION ROUTINE
4      C
5      DIMENSION A(700),B(700)
6      SUM = 0.0
7      DO 10 I=2,I1
8      SUM = SUM + 0.5*(B(I)+B(I-1))*(A(I)-A(I-1))
9      10 CONTINUE
10     C = SUM
11     RETURN
12     END

```

```

1  SUBROUTINE CALIT
2  COMMON/VELOC/ V1(700),V2(700),V3(700),V4(700),V5(700),V1X(200),
3  1 V2X(200),V3X(200),V4X(200),V5X(200),V1Y(200),
4  2 V2Y(200),V3Y(200),V4Y(200),V5Y(200)
5  COMMON/PICT/PPERIN,XX,XMIN,XXEP,YY,YMIN,ORD,EMSTOR,PI,AAAA
6  COMMON /COORD/ XON(700),YON(700),XOFF(200),YOFF(200),S1(700),
7  1 S11(700),XTEST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
8  2 YCU2,XR1,XR2,XPH,YR1,YR2,YRH,CUTOF1,CLTOF2,CUTOFH,
9  3 ELNC,ANG(700),ARI(700),AROFF(200)
10 COMMON /COUT/ NT,NS1,NH,NP,IV,NX,NND,ICOMP,K,NXH11,NXH12,NXH13,
11 1 NST2,NST3,NST7,NPPP(30),IRAN(30),M1,P2,ICOMP1,IPL,
12 2 IHLB
13 COMMON/SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
14 1 VYINC(200),RHOB(700),RBORT(700),RHOB(200),
15 2 VCOM(700),RROOT(200),VRE(200),VRECOM(200),
16 3 VXCOP(200),VYCOM(200),THETA(200),PSOFTC(700),
17 4 PSOFT(700),CMACH(700),XPACH(700),CPI(700),CPC(700),
18 5 RHOI(700)

```

```

19 C
20 C -----
21 C THIS SUBROUTINE TITLES THE PLOTS OF PS/PT AND MACH NUMBER VS S

```

```

22 C
23 DIMENSION BD(2),BC(1),BD2(2),BT(1),BTIT(1),X(30),Y(30)
24 DATA AA,AB,AC,AD,AE/'ANGLE=','VINP =','MACH =','AT X =',
25 1 'SCALE= '/
26 DATA BD/'OUTER ','WALL '/
27 DATA BD1/' HUB '/
28 DATA BD2/'INNER ','WALL '/
29 DATA BT2/'SREF = '/
30 DATA BTIT/'S/SREF'/
31 IF(IPL.EQ.10) GO TO 60
32 VM = 0.0
33 NS = 12
34 HE = 0.25
35 DO 10 I = 1,19
36 X(I) = 0.0
37 10 CONTINUE
38 Y(1) = 1.0
39 DO 20 I = 1,19,4
40 X(I) = 10.0
41 X(I+1) = 10.0
42 20 CONTINUE
43 K = 1
44 DO 30 I = 2,19,2
45 K = K+1
46 Y(I) = 1.0*K
47 Y(I+1) = 1.0*K
48 30 CONTINUE
49 CALL PLOT(0.C,1.0,3)
50 DO 40 I = 1,19,2
51 CALL PLOT(X(I),Y(I),2)
52 CALL PLOT(X(I+1),Y(I+1),3)
53 40 CONTINUE
54 CALL PLOT(0.C,10.C,2)
55 CALL PLOT(1.C,0.0,2)
56 DO 50 I = 1,19,2

```

```

57      CALL PLOT(Y(I),X(I),2)
58      CALL PLOT(Y(I+1),X(I+1),3)
59      50 CONTINUE
60      CALL PLOT(10.0,0.0,2)
61      CALL PLOT(10.0,0.0,3)
62      A = 7.1
63      B = 1.3
64      IF(IPL.EQ.1) CALL SYMBOL(A,B,HE,BD2,TH,12)
65      IF(IPL.EQ.2) CALL SYMBOL(A,B,HE,BD1,TH,6)
66      IF(IPL.EQ.3) CALL SYMBOL(A,B,HE,BD,TH,12)
67      55 A=7.1
68      B=0.6
69      HE=0.25
70      CALL SYMBOL(A,B,HE,BT2,TH,6)
71      A=8.4
72      IF(IPL.EQ.1) CALL NUMBER(A,B,HE,CUTOF1,TH,3)
73      IF(IPL.EQ.2) CALL NUMBER(A,B,HE,CUTOFH,TH,3)
74      IF(IPL.EQ.3) CALL NUMBER(A,B,HE,CUTOF2,TH,3)
75      A=4.65
76      B=0.5
77      HE=0.15
78      CALL SYMBOL(A,B,HE,BT1,TH,6)
79      60 TO 100
80      60 XST=0.0
81      YST=0.0
82      YST1=YST
83      CALL PLOT(XST,YST,1)
84      IX=INT(XX)
85      IY=INT(YV)
86      CALL PLOT(IX,YST,2)
87      DO 70 I=1,IY,2
88      YST=YST+1.0
89      CALL PLOT(IX,YST,1)
90      CALL PLOT(10.0,YST,2)
91      YST=YST+1.0
92      IF(YST.GT.YY) GO TO 70
93      CALL PLOT(10.0,YST,1)
94      CALL PLOT(IX,YST,2)
95      70 CONTINUE
96      CALL PLOT(10.0,YST1,3)
97      CALL PLOT(10.0,YY,2)
98      DO 75 I=1,IX,2
99      XST=XST+1.0
100     CALL PLOT(XST,YY,3)
101     CALL PLOT(XST,0.0,2)
102     XST=XST+1.0
103     IF(XST.GT.XX) GO TO 75
104     CALL PLOT(XST,0.0,3)
105     CALL PLOT(XST,YY,2)
106     75 CONTINUE
107     CALL PLOT(10.0,YST1,3)
108     PIO=3.141592654/180.0
109     DO 80 I=1,NP
110     IF(1COMP.EQ.0) GO TO 76
111     VX=VXCOM(I)
112     VY=VYCOM(I)
113     VRES=VRECOM(I)

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```

114      GO TO 77
115      76 VX=VXINC(I)
116      VY=VYINC(I)
117      VRES=VRE(I)
118      77 SIZE=VRES/VPERIN
119      IF(VX.EQ.0) VX=.00000000
120      ANGLE=-SIGN(90.,VX)*ATAN(VY/VX)/F10
121      XP=(XOFF(I)-XMIN)/EXEP
122      YP=(YOFF(I)-YMIN)/CPD
123      IF(XP.GT.XX.OR.XP.LT.0.D.OR.YP.GT.YY.OR.YP.LT.0.D) GO TO 80
124      SI2=AMIN1(4./21.*SIZE,.15)
125      SI22=SIZE-D.5*SI2
126      XHEAD=XP+VX/VPERIN/SIZE*SI22
127      YHEAD=YP+VY/VPERIN/SIZE*SI22
128      CALL SYMBOL(XP,YP,SI22,16,ANGLE,-1)
129      CALL SYMBOL(XHEAD,YHEAD,SI2,2,ANGLE,-1)
130      80 CONTINUE
131      CALL SYMBOL(XX+.15,4.,0.2,AA,0.,6)
132      CALL NUMBER(XX+1.25,4.0,0.2,AL,0.,3)
133      CALL SYMBOL(XX+.15,3.5,0.2,AB,0.,6)
134      CALL NUMBER(XX+1.25,3.5,0.2,AAA,C.,3)
135      CALL SYMBOL(XX+.15,3.,0.2,AC,0.,6)
136      CALL NUMBER(XX+1.25,3.0,0.2,EMSTCP,0.,3)
137      CALL SYMBOL(XX+.15,2.7,0.2,AD,0.,6)
138      CALL NUMBER(XX+1.25,2.7,0.2,XTEST,0.,3)
139      CALL SYMBOL(XX+.15,2.2,0.2,AE,0.,6)
140      CALL NUMBER(XX+1.25,2.2,0.2,VPERIN,0.,3)
141      100 RETURN
142      END

```



```

1      SUBROUTINE PLTER(N1,N11,NH1,IS,K1)
2      COMMON /CLPLOT/ XPEN,YPEN,NX6,NY,IPEN,XLABEL(10),YLABEL(10)
3      COMMON /COORD/ XON(700),YCN(700),XOFF(200),YOFF(200),SI(700),
4      1      S1(700),XTFST,XTEST1,XTEST2,YCL,YCU,YCL1,YCL2,YCU1,
5      2      YCU2,XR1,XR2,XRH,YR1,YR2,YRH,CUTOF1,CLTOF2,CUTOFH,
6      3      ELNC,ANG(700),AR(700),AROFF(200)
7      COMMON /CONT/ VC,VS1,VS2,XMC,XMC1,XMC2,WDOTC,WDOTC1,WDOTC2,
8      1      TITLE(3),VINP,ALFA,A,B,C,D,A1C,A2C,A3C,A4C,A5C,A11,
9      2      A21,A31,A41,A51,A12,A22,A32,A42,A52,VIC,VIC1,VIC2
10     COMMON /COUT/ NT,NS1,NH,NP,IW,NX,KND,ICOMP,K,NXH11,NXH12,NXH13,
11     1      NST2,NST3,NST7,NPFR(30),IRAK(30),M1,P2,ICOMP1,IPL,
12     2      INUB
13     COMMON /SOLUT/ VBAR(700),VBARO(200),VINC(700),VXINC(200),
14     1      VYINC(200),RHOB(700),RBOT(700),RHORE(200),
15     2      VCOM(700),RBOOT(200),VRE(200),VRECOM(200),
16     3      VXCOM(200),VYCOM(200),THETA(200),PSCPTC(700),
17     4      PSOPT(700),CMACH(700),XMACH(700),CP(700),CPC(700),
18     5      RHC(700)
19     C -----
20     C
21     C THIS SUBROUTINE PLOTS PS/PT AND MACH NUMBER VS S
22     C
23     DIMENSION YD(4),YDC(3),XPLOT(500),YPLLOT(500),KKK(7),P(14)
24     KKK(1) = 4
25     KKK(2) = 1
26     KKK(3) = 2
27     KKK(4) = 1
28     KKK(5) = 1
29     KKK(6) = N1
30     KKK(7) = N11
31     P(1) = 3.0
32     P(2) = 10.0
33     P(3) = 0.0
34     P(4) = 1.0
35     P(5) = 10.0
36     P(6) = 0.0
37     P(7) = 1.0
38     P(8) = 10.0
39     P(9) = 0.0
40     P(10) = 0.0
41     P(11) = 0.0
42     P(12) = 0.0
43     P(13) = 0.0
44     P(14) = 90.0
45     DATA YD(1),YD(2),YD(3),YD(4) / 'PRESSU','RE RAT','JO, PS','/PTC '
46     1 /
47     DATA YDD(1),YDD(2),YDD(3) / 'LOCAL ','MACH N','0. ' /
48     DO 10 I=1,4
49     YLABEL(I) = YD(I)
50     10 CONTINUE
51     XLABEL(1) = TITLE(1)
52     XLABEL(2) = TITLE(2)
53     XLABEL(3) = TITLE(3)
54     XPEN = 0.0
55     YPEN = 0.0
56     IPEN = -3

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57      NX6=-18
58      NY=24
59      IF(NI.EQ.2) GO TO 31
60      DO 20 I=1,N1
61      XPLCT(I) = S1(NHI+I-1)
62      IF(ICOMP.EQ.0) YPLCT(I) = PSOPT(NHI+I-1)
63      IF(ICOMP.EQ.1) YPLCT(I) = PSOPTC(NHI+I-1)
64      20 CONTINUE
65      IF(IPL.EQ.1) KKK(3)=1
66      IF(IPL.EQ.1) GO TO 35
67      DO 30 I=1,N11
68      XPLCT(N1+I) = S1(IS+I-1)
69      IF(ICOMP.EQ.0) YPLCT(N1+I) = PSOPT(IS+I-1)
70      IF(ICOMP.EQ.1) YPLCT(N1+I) = PSOPTC(IS+I-1)
71      30 CONTINUE
72      GO TO 35
73      31 DO 32 I=1,N1
74      XPLCT(I) = S1(NHI-I+1)
75      IF(ICOMP.EQ.0) YPLCT(I) = PSOPT(NHI-I+1)
76      IF(ICOMP.EQ.1) YPLCT(I) = PSOPTC(NHI-I+1)
77      32 CONTINUE
78      35 CALL CALPLT(XPLCT,YPLCT,KKK,P)
79      IF(NI.EQ.2) GO TO 61
80      DO 40 I=1,N1
81      IF(ICOMP.EQ.0) YPLCT(I) = XMACH(NHI+I-1)
82      IF(ICOMP.EQ.1) YPLCT(I) = CMACH(NHI+I-1)
83      40 CONTINUE
84      IF(IPL.EQ.1) KKK(3)=1
85      IF(IPL.EQ.1) GO TO 65
86      DO 50 I=1,N11
87      IF(ICOMP.EQ.0) YPLCT(N1+I) = XMACH(IS+I-1)
88      IF(ICOMP.EQ.1) YPLCT(N1+I) = CMACH(IS+I-1)
89      50 CONTINUE
90      GO TO 65
91      61 DO 62 I=1,N1
92      IF(ICOMP.EQ.0) YPLCT(I) = XMACH(NHI-I+1)
93      IF(ICOMP.EQ.1) YPLCT(I) = CMACH(NHI-I+1)
94      62 CONTINUE
95      DO 60 I=1,3
96      YLABEL(I)=YDD(I)
97      60 CONTINUE
98      NY = 18
99      P(7) = 2.0
100     CALL CALPLT(XPLCT,YPLCT,KKK,P)
101     RETURN
102     END

```

TEST CASE INPUT AND OUTPUT

Program SCIRCL

INPUT FILE DUMP

NOZZLE WITH CENTERBODY

6. -2. 4. 6. -10. 4.

1331

284

3. .4 4. 0.

8. .4 11.6 20

6. 2 4.8 10

6. 7.2 11.0 10

12. -5. -3. 2

14. -5. -3. 2

16. -5. -3. 2

18. -5. -3. 2

20. -5. -3. 2

1. 4.

10. 0.

-45. 0.

0. 0.

1. 0.

0. 0.

0. 2.

0. 6.

0. 0.

9. 9.

-3. -40.

2. 3.

1. 6.

6. 5.

0. 2.

10. 6.

5. 5.

-4. 3.

6. 6.

3. 4.

1. 6.

21. 21.

-40. -3.

0. 2.

21. 21.

-4. -3.

1. 0.

12. 12.

1. 0.

0. -45.

12. 12.

17. 17.

CASE

NOZZLE WITH CENTERBODY

GEOMETRY OUTPUT IS SAVED ON UNIT 17 FOR
PROGRAM 20Y WHICH WILL GENERATE
HIGHER ORDER SOLUTION
WITH STRIP VORTEX FLOW.

SCIRCLE WILL PRODUCE

NO CROSSSECTIONAL AREA PLOTS.

NO CURVATURE PLOTS.

NO REMARKING OF GEOMETRY.

NO. OF BODIES = 3. DELS = .400 DELSMX = 5.000 XRI = .000000

BODY 1 *****

----- STRAIGHT LINE

EMREED

10.000 X -4.5000+01 0.0000
Y 0.0000 0.0000

LAST POINT N= 20, X= .00000 Y= .00000 KAPPA= .00000 DY/DX= .00000 ALPHA= .00000

----- STRAIGHT LINE

EMREED

1.000 X 0.0000 6.0000+00
Y 0.0000 0.0000

LAST POINT N= 35, X= .6000+01, Y= .00000 KAPPA= .00000 DY/DX= .00000 ALPHA= .00000

----- SUPERELLIPSE, EMREED= .00

EXPONENTS

P = 2.000 X 0.0000 6.0000+00 0.0000 0.0000 9.0000+00 9.0000+00
Q = 2.000 Y 0.0000 0.0000 0.0000 0.0000 -3.0000+00 -5.0000+00

P = .2000000+01 A = .30000000+01 X0 = -.60000000+01
Q = .2000000+01 B = .30000000+01 Y0 = .30000000+01 QMEGA = .00000000

3 ITERATIONS----

DELS IM = .40000 DELS = .30940 DELS OUT = .30940 DSEST = .00079

LAST POINT N= 48, X= .90000+01, Y= -.30000+01, KAPPA= -.33333+00, DY/DX= .99999+05, ALPHA= .89999+02

----- STRAIGHT LINE

EMREED

1.000 X 9.0000+00 9.0000+00
Y -3.0000+00 -4.0000+01

LAST POINT N= 66, X= .90000+01, Y= -.50000+02, KAPPA= .00000 DY/DX= -.99999+05, ALPHA= -.90000+02

BODY 2 *****

----- STRAIGHT LINE

EMREED

1.000 X 3.1000+01 6.0000+00
Y 6.0000+00 5.0000+00

LAST POINT K= 74, X= .6000+01, Y= .5000+01, KAPPA= .00000 ,DY/DX= .00000 ,ALPHA= .00000

----- SUPERELLIPSE, EMREED= .00

EXPONENTS

P = 2.000 X 1.000+01 6.000+00 0.0000 0.0000 3.0000+00 3.0000+00
Q = 2.000 Y 5.000+00 5.000+00 0.0000 0.0000 6.0000+00 8.0000+00

P = .2000000+01 A = .3000000+01 X0 = .6000000+01 OMEGA = .00000000
Q = .2000000+01 B = .1000000+01 Y0 = .6000000+01

6 ITERATIONS----

DELS IN = .40000 DELS OUT = .32757 DISTEST = .00276

LAST POINT K= 84, X= .3000+01, Y= .6000+01, KAPPA= -.29997+01,DY/DX= -.99999+05,ALPHA= -.89999+02

----- MIRROR IMAGE

EMREED THIS SEGMENT IS A MIRROR IMAGE ABOUT THE Y= 6.0000+00 LINE
THE ORIGINAL SEGMENT LIES BETWEEN THE FOLLOWING POINTS

***** X 1.000+01 3.000+00
Y 6.000+00 6.000+00

LAST POINT K= 101, X= .1100+02, Y= .6000+01, KAPPA= .00000 ,DY/DX= -.2000+00,ALPHA= -.11110+02

BODY 3 *****

----- STRAIGHT LINE

EMREED

1.000 X 2.100+01 2.100+01
Y -9.000+01 -3.000+00

237

LAST POINT K= 120, X= .2100+02, Y= -.3000+01, KAPPA= .00000 ,DY/DX= .00000 ,ALPHA= .00000

----- SUPERELLIPSE, EMREED= .00

EXPONENTS

P = 2.000 X 2.100+01 2.100+01 0.0000 0.0000 6.0000+00 0.0000
Q = 2.000 Y -9.000+00 -3.000+00 0.0000 0.0000 1.2000+01 1.2000+01

P = .2000000+01 A = .1500000+02 X0 = .2000000+02 OMEGA = .00000000
Q = .2000000+01 B = .1500000+02 Y0 = .1100000+02

3 ITERATIONS----

DELS IN = .40000 DELS OUT = .39792 DISTEST = .39259

LAST POINT K= 180, X= .6000+01, Y= .1200+02, KAPPA= -.66667-01,DY/DX= .00000 ,ALPHA= .00000

----- STRAIGHT LINE

EMREED

1.000 X 6.000+00 0.0000
Y 1.200+01 1.200+01

LAST POINT K= 195, X= .28312+06, Y= .1200+02, KAPPA= .00000 ,DY/DX= .00000 ,ALPHA= .00000

----- STRAIGHT LINE

EMREED

1.000 X 0.0000 -9.5000+01
Y 1.200+01 1.200+01

LAST POINT K= 214, X= -.45000+02, Y= .12000+02, KAPPA= .00000, 0Y/0X= .00000, ALPHA= .00000

INPUT FOR THE COMBINE PROGRAM NT(1)= 65 NT(2)= 99 NT(3)= 211 NP= 50

BODY 1 CO-ORDINATES- X Y KAPPA DY/DX ALPHA S S-S(2) DELTAS

1	-.45000+02	.00000	.00000	.00000	.00000	.00000	.45000+02	.00000
2	-.40733+02	.00000	.00000	.00000	.00000	.42669+01	.42669+01	.00000
3	-.36466+02	.00000	.00000	.00000	.00000	.36466+02	.42669+01	.00000
4	-.32199+02	.00000	.00000	.00000	.00000	.32199+02	.42669+01	.00000
5	-.27932+02	.00000	.00000	.00000	.00000	.27932+02	.42669+01	.00000
6	-.23666+02	.00000	.00000	.00000	.00000	.23666+02	.42669+01	.00000
7	-.19399+02	.00000	.00000	.00000	.00000	.19399+02	.42669+01	.00000
8	-.15832+02	.00000	.00000	.00000	.00000	.15832+02	.42669+01	.00000
9	-.12860+02	.00000	.00000	.00000	.00000	.12860+02	.42669+01	.00000
10	-.10363+02	.00000	.00000	.00000	.00000	.10363+02	.42669+01	.00000
11	-.83198+01	.00000	.00000	.00000	.00000	.83198+01	.42669+01	.00000
12	-.65998+01	.00000	.00000	.00000	.00000	.65998+01	.42669+01	.00000
13	-.51664+01	.00000	.00000	.00000	.00000	.51664+01	.42669+01	.00000
14	-.39720+01	.00000	.00000	.00000	.00000	.39720+01	.42669+01	.00000
15	-.29768+01	.00000	.00000	.00000	.00000	.29768+01	.42669+01	.00000
16	-.21472+01	.00000	.00000	.00000	.00000	.21472+01	.42669+01	.00000
17	-.14560+01	.00000	.00000	.00000	.00000	.14560+01	.42669+01	.00000
18	-.88000+00	.00000	.00000	.00000	.00000	.88000+00	.42669+01	.00000
19	-.40000+00	.00000	.00000	.00000	.00000	.40000+00	.42669+01	.00000
20	.00000	.00000	.00000	.00000	.00000	.00000	.42669+01	.00000
21	.40000+00	.00000	.00000	.00000	.00000	.40000+00	.42669+01	.00000
22	.80000+00	.00000	.00000	.00000	.00000	.80000+00	.42669+01	.00000
23	.12000+01	.00000	.00000	.00000	.00000	.12000+01	.42669+01	.00000
24	.16000+01	.00000	.00000	.00000	.00000	.16000+01	.42669+01	.00000
25	.20000+01	.00000	.00000	.00000	.00000	.20000+01	.42669+01	.00000
26	.24000+01	.00000	.00000	.00000	.00000	.24000+01	.42669+01	.00000
27	.28000+01	.00000	.00000	.00000	.00000	.28000+01	.42669+01	.00000
28	.32000+01	.00000	.00000	.00000	.00000	.32000+01	.42669+01	.00000
29	.36000+01	.00000	.00000	.00000	.00000	.36000+01	.42669+01	.00000
30	.40000+01	.00000	.00000	.00000	.00000	.40000+01	.42669+01	.00000
31	.44000+01	.00000	.00000	.00000	.00000	.44000+01	.42669+01	.00000
32	.48000+01	.00000	.00000	.00000	.00000	.48000+01	.42669+01	.00000
33	.52000+01	.00000	.00000	.00000	.00000	.52000+01	.42669+01	.00000
34	.56000+01	.00000	.00000	.00000	.00000	.56000+01	.42669+01	.00000
35	.60000+01	.00000	.00000	.00000	.00000	.60000+01	.42669+01	.00000
36	.63659+01	.00000	.00000	.00000	.00000	.63659+01	.42669+01	.00000
37	.67290+01	.00000	.00000	.00000	.00000	.67290+01	.42669+01	.00000
38	.70839+01	.00000	.00000	.00000	.00000	.70839+01	.42669+01	.00000
39	.74161+01	.00000	.00000	.00000	.00000	.74161+01	.42669+01	.00000
40	.77271+01	.00000	.00000	.00000	.00000	.77271+01	.42669+01	.00000
41	.80122+01	.00000	.00000	.00000	.00000	.80122+01	.42669+01	.00000
42	.82671+01	.00000	.00000	.00000	.00000	.82671+01	.42669+01	.00000
43	.84798+01	.00000	.00000	.00000	.00000	.84798+01	.42669+01	.00000
44	.86607+01	.00000	.00000	.00000	.00000	.86607+01	.42669+01	.00000
45	.88160+01	.00000	.00000	.00000	.00000	.88160+01	.42669+01	.00000
46	.89125+01	.00000	.00000	.00000	.00000	.89125+01	.42669+01	.00000
47	.89778+01	.00000	.00000	.00000	.00000	.89778+01	.42669+01	.00000
48	.90000+01	.00000	.00000	.00000	.00000	.90000+01	.42669+01	.00000
49	.90000+01	.00000	.00000	.00000	.00000	.90000+01	.42669+01	.00000
50	.90000+01	.00000	.00000	.00000	.00000	.90000+01	.42669+01	.00000

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BODY 2 CO-ORDINATES- X

BODY 3 CO-ORDINATES- X

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102	.21000+02	-.40000+02	.00000	-.99999+05	-.90000+02	.00000	.66561+02	.00000
103	.21000+02	-.36980+02	.00000	-.99999+05	-.90000+02	.35203+01	.63001+02	.35203+01
104	.21000+02	-.32959+02	.00000	-.99999+05	-.90000+02	.70705+01	.59521+02	.35203+01
105	.21000+02	-.29439+02	.00000	-.99999+05	-.90000+02	.10361+02	.56000+02	.35203+01
106	.21000+02	-.25919+02	.00000	-.99999+05	-.90000+02	.14061+02	.52480+02	.35203+01
107	.21000+02	-.22399+02	.00000	-.99999+05	-.90000+02	.17601+02	.48960+02	.35203+01
108	.21000+02	-.18832+02	.00000	-.99999+05	-.90000+02	.21168+02	.45393+02	.35664+01
109	.21000+02	-.15388+02	.00000	-.99999+05	-.90000+02	.24120+02	.42921+02	.29720+01
110	.21000+02	-.13383+02	.00000	-.99999+05	-.90000+02	.26617+02	.39945+02	.24767+01
111	.21000+02	-.11320+02	.00000	-.99999+05	-.90000+02	.28880+02	.37881+02	.20639+01
112	.21000+02	-.95994+01	.00000	-.99999+05	-.90000+02	.30800+02	.36161+02	.17199+01
113	.21000+02	-.81665+01	.00000	-.99999+05	-.90000+02	.32824+02	.34728+02	.14133+01
114	.21000+02	-.69720+01	.00000	-.99999+05	-.90000+02	.33020+02	.33533+02	.11944+01
115	.21000+02	-.59766+01	.00000	-.99999+05	-.90000+02	.34023+02	.32538+02	.99533+00
116	.21000+02	-.51472+01	.00000	-.99999+05	-.90000+02	.34853+02	.31708+02	.82944+00
117	.21000+02	-.44560+01	.00000	-.99999+05	-.90000+02	.35544+02	.31017+02	.82120+00
118	.21000+02	-.38800+01	.00000	-.99999+05	-.90000+02	.36120+02	.30441+02	.57600+00
119	.21000+02	-.34000+01	.00000	-.99999+05	-.90000+02	.36600+02	.29861+02	.48000+00
120	.21000+02	-.30000+01	-.66667-01	-.99999+03	-.90000+02	.37193+02	.29561+02	.40000+00
121	.20995+02	-.26073+01	-.66667-01	-.38185+02	.68500+02	.37786+02	.29169+02	.39273+00
122	.20979+02	-.22148+01	-.66667-01	-.19076+02	.86999+02	.38179+02	.28776+02	.39286+00
123	.20954+02	-.18228+01	-.66667-01	-.12701+02	.85998+02	.38572+02	.28330+02	.39299+00
124	.20918+02	-.14311+01	-.66666-01	-.95085+01	.83996+02	.38965+02	.27990+02	.39313+00
125	.20872+02	-.10408+01	-.66666-01	-.75498+01	.82494+02	.39358+02	.27594+02	.39326+00
126	.20815+02	-.65125+00	-.66667-01	-.63076+01	.80991+02	.39758+02	.27203+02	.39742+00
127	.20748+02	-.28340+00	-.66667-01	-.53993+01	.79988+02	.39752+02	.26809+02	.39354+00
128	.20711+02	-.12270+00	-.66667-01	-.46983+01	.78984+02	.40146+02	.26414+02	.39368+00
129	.20584+02	-.50678+00	-.66667-01	-.41589+01	.76880+02	.40539+02	.26022+02	.39383+00
130	.20487+02	-.88859+00	-.66667-01	-.37956+01	.74975+02	.40933+02	.25628+02	.39397+00
131	.20380+02	-.12679+01	-.66667-01	-.33894+01	.73470+02	.41328+02	.25234+02	.39411+00
132	.20263+02	-.16443+01	-.66667-01	-.30710+01	.71964+02	.41728+02	.24839+02	.39424+00
133	.20136+02	-.20177+01	-.66667-01	-.28172+01	.70570+02	.42116+02	.24445+02	.39441+00
134	.19999+02	-.23878+01	-.66667-01	-.25983+01	.69500+02	.42511+02	.24050+02	.39457+00
135	.19852+02	-.27543+01	-.66667-01	-.24071+01	.68442+02	.42905+02	.23656+02	.39472+00
136	.19696+02	-.31169+01	-.66667-01	-.22391+01	.67394+02	.43300+02	.23261+02	.39489+00
137	.19530+02	-.34755+01	-.66667-01	-.20825+01	.66355+02	.43695+02	.22866+02	.39505+00
138	.19355+02	-.38297+01	-.66667-01	-.19354+01	.65315+02	.44091+02	.22471+02	.39522+00
139	.19170+02	-.41793+01	-.66667-01	-.18365+01	.64280+02	.44486+02	.22076+02	.39539+00
140	.18976+02	-.45241+01	-.66667-01	-.17246+01	.63242+02	.44882+02	.21680+02	.39557+00
141	.18773+02	-.48638+01	-.66667-01	-.16243+01	.62192+02	.45277+02	.21284+02	.39575+00
142	.18561+02	-.51982+01	-.66667-01	-.15322+01	.61156+02	.45673+02	.20888+02	.39594+00
143	.18344+02	-.55271+01	-.66667-01	-.14472+01	.60120+02	.46069+02	.20492+02	.39618+00
144	.18111+02	-.58501+01	-.66667-01	-.13684+01	.59082+02	.46466+02	.20095+02	.39634+00
145	.17873+02	-.61672+01	-.66667-01	-.12951+01	.58042+02	.46862+02	.19699+02	.39656+00
146	.17626+02	-.64780+01	-.66667-01	-.12266+01	.56999+02	.47259+02	.19302+02	.39678+00
147	.17371+02	-.67824+01	-.66667-01	-.11624+01	.55956+02	.47656+02	.18905+02	.39701+00
148	.17108+02	-.70801+01	-.66667-01	-.11020+01	.54913+02	.48053+02	.18508+02	.39725+00
149	.16837+02	-.73708+01	-.66667-01	-.10450+01	.53869+02	.48451+02	.18110+02	.39751+00
150	.16558+02	-.76546+01	-.66667-01	-.99097+00	.52826+02	.48849+02	.17713+02	.39777+00
151	.16279+02	-.79240+01	-.66667-01	-.94100+00	.51780+02	.49236+02	.17325+02	.39776+00
152	.15993+02	-.81862+01	-.66667-01	-.89338+00	.50737+02	.49624+02	.16937+02	.39800+00
153	.15701+02	-.84411+01	-.66667-01	-.84788+00	.49794+02	.50013+02	.16549+02	.39824+00
154	.15401+02	-.86884+01	-.66667-01	-.80431+00	.48950+02	.50401+02	.16160+02	.39846+00
155	.15095+02	-.89281+01	-.66667-01	-.76249+00	.48105+02	.50790+02	.15771+02	.39867+00
156	.14783+02	-.91598+01	-.66667-01	-.72228+00	.47260+02	.51179+02	.15383+02	.39888+00
157	.14464+02	-.93836+01	-.66667-01	-.68353+00	.46413+02	.51568+02	.14994+02	.39908+00
158	.14140+02	-.95990+01	-.66667-01	-.64611+00	.45567+02	.51957+02	.14604+02	.39927+00

159	-13810+02	-98061+01	-66667-01	-60990+00	-31379+02	-52366+02	-14215+02	-38946+00
160	-13875+02	-10005+02	-66667-01	-57481+00	-89891+02	-52366+02	-13825+02	-38964+00
161	-13135+02	-10195+02	-66667-01	-54074+00	-28022+02	-53126+02	-13435+02	-38981+00
162	-12789+02	-10378+02	-66667-01	-50739+00	-28912+02	-53136+02	-12655+02	-38998+00
163	-12039+02	-10548+02	-66667-01	-47530+00	-25422+02	-53906+02	-12655+02	-39115+00
164	-12084+02	-10711+02	-66667-01	-44378+00	-83931+02	-54236+02	-12285+02	-39031+00
165	-11726+02	-10868+02	-66667-01	-41297+00	-22439+02	-54677+02	-11874+02	-39074+00
166	-11363+02	-11009+02	-66667-01	-38221+00	-80947+02	-55017+02	-11684+02	-39062+00
167	-10996+02	-11146+02	-66667-01	-35322+00	-39454+02	-55968+02	-11093+02	-39078+00
168	-10626+02	-11249+02	-66667-01	-32817+00	-17961+02	-55959+02	-10702+02	-39093+00
169	-10252+02	-11385+02	-66667-01	-29559+00	-16467+02	-56280+02	-10311+02	-39107+00
170	-98754+01	-11491+02	-66667-01	-26749+00	-34973+02	-56441+02	-99198+01	-39122+00
171	-94961+01	-11587+02	-66667-01	-23967+00	-13478+02	-57033+02	-95284+01	-39136+00
172	-91182+01	-11673+02	-66667-01	-21228+00	-11983+02	-57228+02	-91369+01	-39150+00
173	-87301+01	-11749+02	-66667-01	-18610+00	-10487+02	-57816+02	-87853+01	-39164+00
174	-83539+01	-11816+02	-66667-01	-15821+00	-89801+01	-58208+02	-83535+01	-39178+00
175	-79561+01	-11872+02	-66667-01	-13153+00	-74930+01	-58600+02	-79616+01	-39192+00
176	-75447+01	-11918+02	-66667-01	-10582+00	-59254+01	-58922+02	-75485+01	-39205+00
177	-71762+01	-11954+02	-66667-01	-78655+01	-44973+01	-59384+02	-71774+01	-39219+00
178	-67847+01	-11979+02	-66667-01	-52346+01	-29887+01	-59776+02	-67850+01	-39232+00
179	-63924+01	-11985+02	-66667-01	-26179+01	-14994+01	-60169+02	-63926+01	-39246+00
180	-60000+01	-12000+02	-00000	-00000	-00000	-60364+02	-60000+01	-39259+00
181	-56000+01	-12000+02	-00000	-00000	-00000	-60961+02	-56000+01	-40000+00
182	-52000+01	-12000+02	-00000	-00000	-00000	-61361+02	-52000+01	-40000+00
183	-48000+01	-12000+02	-00000	-00000	-00000	-61764+02	-48000+01	-40000+00
184	-44000+01	-12000+02	-00000	-00000	-00000	-62161+02	-44000+01	-40000+00
185	-40000+01	-12000+02	-00000	-00000	-00000	-62561+02	-40000+01	-40000+00
186	-36000+01	-12000+02	-00000	-00000	-00000	-62961+02	-36000+01	-40000+00
187	-32000+01	-12000+02	-00000	-00000	-00000	-63361+02	-32000+01	-40000+00
188	-28000+01	-12000+02	-00000	-00000	-00000	-63761+02	-28000+01	-40000+00
189	-24000+01	-12000+02	-00000	-00000	-00000	-64161+02	-24000+01	-40000+00
190	-20000+01	-12000+02	-00000	-00000	-00000	-64561+02	-20000+01	-40000+00
191	-16000+01	-12000+02	-00000	-00000	-00000	-64961+02	-16000+01	-40000+00
192	-12000+01	-12000+02	-00000	-00000	-00000	-65361+02	-12000+01	-40000+00
193	-80000+00	-12000+02	-00000	-00000	-00000	-65761+02	-80000+01	-40000+00
194	-40000+00	-12000+02	-00000	-00000	-00000	-66161+02	-40000+00	-40000+00
195	-00000	-12000+02	-00000	-00000	-00000	-66561+02	-00000	-40000+00
196	-40000+00	-12000+02	-00000	-00000	-00000	-66961+02	-40000+00	-40000+00
197	-80000+00	-12000+02	-00000	-00000	-00000	-67441+02	-80000+00	-40000+00
198	-12000+01	-12000+02	-00000	-00000	-00000	-68017+02	-12000+01	-40000+00
199	-16000+01	-12000+02	-00000	-00000	-00000	-68708+02	-16000+01	-40000+00
200	-20000+01	-12000+02	-00000	-00000	-00000	-69538+02	-20000+01	-40000+00
201	-24000+01	-12000+02	-00000	-00000	-00000	-70333+02	-24000+01	-40000+00
202	-28000+01	-12000+02	-00000	-00000	-00000	-71128+02	-28000+01	-40000+00
203	-32000+01	-12000+02	-00000	-00000	-00000	-71961+02	-32000+01	-40000+00
204	-36000+01	-12000+02	-00000	-00000	-00000	-72801+02	-36000+01	-40000+00
205	-40000+01	-12000+02	-00000	-00000	-00000	-73645+02	-40000+01	-40000+00
206	-44000+01	-12000+02	-00000	-00000	-00000	-74481+02	-44000+01	-40000+00
207	-48000+01	-12000+02	-00000	-00000	-00000	-75321+02	-48000+01	-40000+00
208	-52000+01	-12000+02	-00000	-00000	-00000	-76161+02	-52000+01	-40000+00
209	-56000+01	-12000+02	-00000	-00000	-00000	-77001+02	-56000+01	-40000+00
210	-60000+01	-12000+02	-00000	-00000	-00000	-77841+02	-60000+01	-40000+00
211	-64000+01	-12000+02	-00000	-00000	-00000	-78681+02	-64000+01	-40000+00
212	-68000+01	-12000+02	-00000	-00000	-00000	-79521+02	-68000+01	-40000+00
213	-72000+01	-12000+02	-00000	-00000	-00000	-80361+02	-72000+01	-40000+00
214	-76000+01	-12000+02	-00000	-00000	-00000	-81201+02	-76000+01	-40000+00
215	-80000+01	-12000+02	-00000	-00000	-00000	-82041+02	-80000+01	-40000+00
216	-84000+01	-12000+02	-00000	-00000	-00000	-82881+02	-84000+01	-40000+00
217	-88000+01	-12000+02	-00000	-00000	-00000	-83721+02	-88000+01	-40000+00
218	-92000+01	-12000+02	-00000	-00000	-00000	-84561+02	-92000+01	-40000+00
219	-96000+01	-12000+02	-00000	-00000	-00000	-85401+02	-96000+01	-40000+00
220	-100000+01	-12000+02	-00000	-00000	-00000	-86241+02	-100000+01	-40000+00
221	-104000+01	-12000+02	-00000	-00000	-00000	-87081+02	-104000+01	-40000+00
222	-108000+01	-12000+02	-00000	-00000	-00000	-87921+02	-108000+01	-40000+00
223	-112000+01	-12000+02	-00000	-00000	-00000	-88761+02	-112000+01	-40000+00
224	-116000+01	-12000+02	-00000	-00000	-00000	-89601+02	-116000+01	-40000+00
225	-120000+01	-12000+02	-00000	-00000	-00000	-90441+02	-120000+01	-40000+00
226	-124000+01	-12000+02	-00000	-00000	-00000	-91281+02	-124000+01	-40000+00
227	-128000+01	-12000+02	-00000	-00000	-00000	-92121+02	-128000+01	-40000+00
228	-132000+01	-12000+02	-00000	-00000	-00000	-92961+02	-132000+01	-40000+00
229	-136000+01	-12000+02	-00000	-00000	-00000	-93801+02	-136000+01	-40000+00
230	-140000+01	-12000+02	-00000	-00000	-00000	-94641+02	-140000+01	-40000+00
231	-144000+01	-12000+02	-00000	-00000	-00000	-95481+02	-144000+01	-40000+00
232	-148000+01	-12000+02	-00000	-00000	-00000	-96321+02	-148000+01	-40000+00
233	-152000+01	-12000+02	-00000	-00000	-00000	-97161+02	-152000+01	-40000+00
234	-156000+01	-12000+02	-00000	-00000	-00000	-98001+02	-156000+01	-40000+00
235	-160000+01	-12000+02	-00000	-00000	-00000	-98841+02	-160000+01	-40000+00
236	-164000+01	-12000+02	-00000	-00000	-00000	-99681+02	-164000+01	-40000+00
237	-168000+01	-12000+02	-00000	-00000	-00000	-100521+02	-168000+01	-40000+00
238	-172000+01	-12000+02	-00000	-00000	-00000	-101361+02	-172000+01	-40000+00
239	-176000+01	-12000+02	-00000	-00000	-00000	-102201+02	-176000+01	-40000+00
240	-180000+01	-12000+02	-00000	-00000	-00000	-103041+02	-180000+01	-40000+00
241	-184000+01	-12000+02	-00000	-00000	-00000	-103881+02	-184000+01	-40000+00
242	-188000+01	-12000+02	-00000	-00000	-00000	-104721+02	-188000+01	-40000+00
243	-192000+01	-12000+02	-00000	-00000	-00000	-105561+02	-192000+01	-40000+00
244	-196000+01	-12000+02	-00000	-00000	-00000	-106401+02	-196000+01	-40000+00
245	-200000+01	-12000+02	-00000	-00000	-00000	-107241+02	-200000+01	-40000+00
246	-204000+01	-12000+02	-00000	-00000	-00000	-108081+02	-204000+01	-40000+00
247	-208000+01	-12000+02	-00000	-00000	-00000	-108921+02	-208000+01	-40000+00
248	-212000+01	-12000+02	-00000	-00000	-00000	-109761+02	-212000+01	-40000+00
249	-216000+01	-12000+02	-00000	-00000	-00000	-110601+02	-216000+01	-40000+00
250	-220000+01	-12000+02	-00000	-00000	-00000	-111441+02	-220000+01	-40000+00
251	-224000+01	-12000+02	-00000	-00000	-00000	-112281+02	-224000+01	-40000+00
252	-228000+01	-12000+02	-00000	-00000	-00000	-113121+02	-228000+01	-40000+00
253	-232000+01	-12000+02	-00000	-00000	-00000	-113961+02	-232000+01	-40000+00
254	-236000+01	-12000+02	-00000	-00000	-00000	-114801+02	-236000+01	-40000+00
255	-240000+01	-12000+02	-00000	-00000	-00000	-115641+02	-240000+01	-40000+00
256	-244000+01	-12000+02	-00000	-00000	-00000	-116481+02	-244000+01	-40000+00
257	-248000+01	-12000+02	-00000	-00000	-00000	-117321+02	-248000+01	-40000+00
258	-252000+01	-12000+02	-00000	-00000	-00000	-118161+02	-252000+01	-40000+00
259	-256000+01	-12000+02	-00000	-00000	-00000	-119001+02	-256000+01	-40000+00
260	-260000+01	-12000+02	-00000	-00000	-00000	-119841+02	-260000+01	-40000+00
261	-264000+01	-12000+02	-00000	-00000	-00000	-120681+02	-264000+01	-40000+00
262	-268000+01	-12000+02	-00000	-00000	-00000	-121521+02	-268000+01	-40000+00
263	-272000+01	-12000+02	-00000	-00000	-00000	-122361+02	-272000+01	-40000+00
264	-276000+01	-12000+02	-00000	-00000	-00000	-123201+02	-276000+01	-40000+00
265	-280000+01	-12000+02	-00000	-00000	-00000	-124041+02	-280000+01	-40000+00
266	-284000+01	-12000+02	-00000	-00000	-00000	-124881+02	-284000+01	-40000+00
267	-288000+01	-12000+02	-00000	-00000	-00000	-125721+02	-288000+01	-40000+00
268	-292000+01	-12000+02	-00000	-00000	-00000	-126561+02	-292000+01	-40000+00
269	-296000+01	-12000+02	-00000	-00000	-00000	-127401+02	-296000+01	-40000+00
270	-3000							

-1.10000+01	5.00000+00	1.14000+02	20
6.00000+01	2.00000+00	9.00000+01	10
6.00000+01	1.20000+01	1.18000+02	10
1.20000+02	-5.00000+01	-3.00000+01	2
1.18000+02	-5.00000+01	-3.00000+01	2
1.60000+02	-5.00000+01	-3.00000+01	2
1.60000+02	-5.00000+01	-3.00000+01	2
2.00000+02	-5.00000+01	-3.00000+01	2

I	XOM	YOM	YOMH	AREA	DISC AREA	ENSUBH
XMIN = -9.50000+01 XMAX = 9.00000+00 JMIN = 1 JMAX = 98						
XMIN = 3.00000+00 XMAX = 1.10000+01 JMIN = 84 JMAX = 67						
102	2.10000+01	-9.00000+01	-9.00000+01	0.0000	-9.0000+01	0.0000

8PM0.1

SCRIPT PRINTS

Program 24Y

UNTRANSFORMED COORDINATE DATA FOR BODY ID = 1, NOZZLE WITH CENTERBO

I	X(I)	Y(I)	I	X(I)	Y(I)
1	-85.000000	.000000	34	5.600000	.000000
2	-40.733110	.000000	35	6.000000	.000000
3	-36.866210	.000000	36	6.365880	-.022190
4	-32.199320	.000000	37	6.720020	-.089930
5	-27.932830	.000000	38	7.083830	-.202660
6	-23.665530	.000000	39	7.416150	-.355200
7	-19.398660	.000000	40	7.721130	-.587030
8	-15.032200	.000000	41	8.012190	-.774890
9	-12.860170	.000000	42	8.267050	-1.035190
10	-10.303470	.000000	43	8.470830	-1.311680
11	-8.319560	.000000	44	8.665700	-1.618120
12	-6.559830	.000000	45	8.808970	-1.938620
13	-5.166160	.000000	46	8.912520	-2.280830
14	-3.971970	.000000	47	8.977840	-2.636040
15	-2.976650	.000000	48	9.000000	-3.000000
16	-2.147200	.000000	49	9.000000	-3.389400
17	-1.455600	.000000	50	9.000000	-3.856690
18	-.880000	.000000	51	9.000000	-4.417430
19	-.400000	.000000	52	9.000000	-5.090120
20	.000000	.000000	53	9.000000	-5.897790
21	.400000	.000000	54	9.000000	-6.866750
22	.800000	.000000	55	9.000000	-8.029500
23	1.200000	.000000	56	9.000000	-9.428800
24	1.600000	.000000	57	9.000000	-11.099170
25	2.000000	.000000	58	9.000000	-13.108800
26	2.400000	.000000	59	9.000000	-15.519490
27	2.800000	.000000	60	9.000000	-18.412790
28	3.200000	.000000	61	9.000000	-21.884750
29	3.600000	.000000	62	9.000000	-25.950700
30	4.000000	.000000	63	9.000000	-29.130850
31	4.400000	.000000	64	9.000000	-32.753900
32	4.800000	.000000	65	9.000000	-36.376950
33	5.200000	.000000	66	9.000000	-40.000000

ELEMENT COORDINATE DATA FOR BODY ID = 1. NOZZLE WITH CENTERB

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
1	-85.000000	.000000					
2	-42.866555	.000000	9.266890	9.266890	.000000	1.000000	.000000
3	-30.733110	.000000					
4	-30.599660	.000000	9.266900	9.266900	.000000	1.000000	.000000
5	-36.866210	.000000					
6	-39.332765	.000000	9.266890	9.266890	.000000	1.000000	.000000
7	-32.199320	.000000					
8	-30.066875	.000000	9.266890	9.266890	.000000	1.000000	.000000
9	-27.933210	.000000					
10	-25.799980	.000000	9.266900	9.266900	.000000	1.000000	.000000
11	-23.666530	.000000					
12	-21.532085	.000000	9.266890	9.266890	.000000	1.000000	.000000
13	-19.398640	.000000					
14	-17.615420	.000000	3.566440	3.566440	.000000	1.000000	.000000
15	-15.832200	.000000					
16	-19.398185	.000000	2.972030	2.972030	.000000	1.000000	.000000
17	-12.840170	.000000					
18	-11.621820	.000000	2.976700	2.976700	.000000	1.000000	.000000
19	-10.318170	.000000					
20	-9.351515	.000000	2.063910	2.063910	.000000	1.000000	.000000
21	-8.319540	.000000					
22	-7.459595	.000000	1.719930	1.719930	.000000	1.000000	.000000
23	-6.599630	.000000					
24	-5.802995	.000000	1.933270	1.933270	.000000	1.000000	.000000
25	-5.164360	.000000					
26	-4.589165	.000000	1.194390	1.194390	.000000	1.000000	.000000
27	-3.971970	.000000					
28	-3.474305	.000000	.995330	.995330	.000000	1.000000	.000000
29	-2.976440	.000000					
30	-2.561920	.000000	.829440	.829440	.000000	1.000000	.000000
31	-2.197200	.000000					
32	-1.801600	.000000	.691200	.691200	.000000	1.000000	.000000
33	-1.456000	.000000					
34	-1.168000	.000000	.576000	.576000	.000000	1.000000	.000000
35	-.880000	.000000					
36	-.690000	.000000	.480000	.480000	.000000	1.000000	.000000
37	-.480000	.000000					
38	-.200000	.000000	.400000	.400000	.000000	1.000000	.000000
39	.000000	.000000					
40	.200000	.000000	.400000	.400000	.000000	1.000000	.000000
41	.400000	.000000					
42	.600000	.000000	.400000	.400000	.000000	1.000000	.000000
43	.800000	.000000					
44	1.000000	.000000	.400000	.400000	.000000	1.000000	.000000
45	1.200000	.000000					
46	1.400000	.000000	.400000	.400000	.000000	1.000000	.000000
47	1.600000	.000000					
48	1.800000	.000000	.400000	.400000	.000000	1.000000	.000000
49	2.000000	.000000					

ELEMENT COORDINATE DATA FOR BODY ID = 1.				NOZZLE WITH CENTERBO			
I	X(I)	Y(I)	DL	DS	SIN(IALF)	COS(IALF)	CURVATURE
	2.200000	.000000	.000000	.000000	.000000	1.000000	.000000
26	2.400000	.000000	.000000	.000000	.000000	1.000000	.000000
	2.600000	.000000	.000000	.000000	.000000	1.000000	.000000
27	2.800000	.000000	.000000	.000000	.000000	1.000000	.000000
	3.000000	.000000	.000000	.000000	.000000	1.000000	.000000
28	3.200000	.000000	.000000	.000000	.000000	1.000000	.000000
	3.400000	.000000	.000000	.000000	.000000	1.000000	.000000
29	3.600000	.000000	.000000	.000000	.000000	1.000000	.000000
	3.800000	.000000	.000000	.000000	.000000	1.000000	.000000
30	4.000000	.000000	.000000	.000000	.000000	1.000000	.000000
	4.200000	.000000	.000000	.000000	.000000	1.000000	.000000
31	4.400000	.000000	.000000	.000000	.000000	1.000000	.000000
	4.600000	.000000	.000000	.000000	.000000	1.000000	.000000
32	4.800000	.000000	.000000	.000000	.000000	1.000000	.000000
	5.000000	.000000	.000000	.000000	.000000	1.000000	.000000
33	5.200000	.000000	.000000	.000000	.000000	1.000000	.000000
	5.400000	.000000	.000000	.000000	.000000	1.000000	.000000
34	5.600000	.000000	.000000	.000000	.000000	1.000000	.000000
	5.800000	.000000	.000000	.000000	.000000	1.000000	.000000
35	6.000000	.000000	.000000	.000000	.000000	1.000000	.000000
	6.183177	-.007327	.366569	.866679	-.061081	.998133	-.230700
36	6.365880	-.022390	.369367	.869601	-.182853	.983140	-.333695
	6.548490	-.050566	.372383	.872623	-.302726	.953078	-.333618
37	6.729020	-.089930	.375599	.875826	-.417451	.908699	-.333656
	6.908226	-.140789	.378887	.879089	-.529897	.851197	-.333658
38	7.083930	-.202660	.382397	.882600	-.629381	.781120	-.333669
	7.252367	-.273904	.386166	.886389	-.719539	.699801	-.333639
39	7.416150	-.355280	.390287	.890409	-.792492	.609883	-.333593
	7.574562	-.446416	.394709	.894799	-.858236	.513256	-.333608
40	7.727130	-.547030	.399414	.899628	-.912714	.408598	-.333691
	7.873128	-.656621	.404533	.904792	-.959790	.297282	-.333624
41	8.012190	-.774890	.410166	.910389	-.998152	.060773	-.231994
	8.143575	-.901168	.416339	.916409	-1.000000	.000000	.000000
42	8.267050	-1.035190	.423087	.922908			
	8.377462	-1.170339	.430414	.930829			
43	8.479830	-1.311680	.438339	.939209			
	8.574709	-1.460292	.446869	.948089			
44	8.660700	-1.614120	.456014	.957542			
	8.738146	-1.774216	.465769	.967599			
45	8.805970	-1.938620	.476139	.978309			
	8.869360	-2.108132	.487124	.989628			
46	8.912520	-2.280830	.498769	.999628			
	8.950530	-2.457851	.511166				
47	8.977840	-2.636040	.524479				
	8.992769	-2.817786	.538639				
48	9.000000	-3.000000	.553600				
	9.000000	-3.194700	.569400				
49	9.000000	-3.389400	.586000				
	9.000000	-3.623045	.603400				
50	9.000000	-3.856690	.621600				

ELEMENT COORDINATE DATA FOR BODY ID = 1,

NOZZLE WITH CENTERBO

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
	9.000000	-8.137060	.560740	.560740	-1.000000	.000000	.000000
51	9.000000	-4.417430					
	9.000000	-4.753875	.672890	.672890	-1.000000	.000000	.000000
52	9.000000	-5.090320					
	9.000000	-5.424055	.807470	.807470	-1.000000	.000000	.000000
53	9.000000	-5.897790					
	9.000000	-6.382270	.968960	.968960	-1.000000	.000000	.000000
54	9.000000	-6.866750					
	9.000000	-7.448125	1.162750	1.162750	-1.000000	.000000	.000000
55	9.000000	-8.029500					
	9.000000	-8.727150	1.395300	1.395300	-1.000000	.000000	.000000
56	9.000000	-9.424800					
	9.000000	-10.241985	1.678370	1.678370	-1.000000	.000000	.000000
57	9.000000	-11.099170					
	9.000000	-12.103785	2.009230	2.009230	-1.000000	.000000	.000000
58	9.000000	-13.108400					
	9.000000	-14.311985	2.411090	2.411090	-1.000000	.000000	.000000
59	9.000000	-15.519490					
	9.000000	-16.966180	2.893300	2.893300	-1.000000	.000000	.000000
60	9.000000	-18.412790					
	9.000000	-20.148770	3.471960	3.471960	-1.000000	.000000	.000000
61	9.000000	-21.884750					
	9.000000	-23.696275	3.623050	3.623050	-1.000000	.000000	.000000
62	9.000000	-25.507800					
	9.000000	-27.319325	3.623050	3.623050	-1.000000	.000000	.000000
63	9.000000	-29.130850					
	9.000000	-30.942375	3.623050	3.623050	-1.000000	.000000	.000000
64	9.000000	-32.753900					
	9.000000	-34.565425	3.623050	3.623050	-1.000000	.000000	.000000
65	9.000000	-36.376950					
	9.000000	-38.188475	3.623050	3.623050	-1.000000	.000000	.000000
66	9.000000	-40.000000					

SUMDS = 92.712139

UNTRANSFORMED COORDINATE DATA FOR BODY ID = 2, NOZZLE WITH CENTERBO

I	X(I)	Y(I)	I	X(I)	Y(I)
1	11.00000	6.000000	19	3.000100	6.283280
2	9.894830	5.778970	21	3.315830	6.446620
3	8.918850	5.683770	21	3.608100	6.603580
4	8.105500	5.621100	22	3.932980	6.724570
5	7.422730	5.285580	23	4.267840	6.816380
6	6.862910	5.172580	24	4.608830	6.885980
7	6.392230	5.078850	25	4.958110	6.933720
8	6.000000	5.000000	26	5.301760	6.972540
9	5.650680	5.006800	27	5.650680	6.993200
10	5.301760	5.027460	28	6.000000	7.000000
11	4.958110	5.062780	29	6.392230	6.921550
12	4.608830	5.110200	30	6.862910	6.827820
13	4.267840	5.183620	31	7.422730	6.718460
14	3.932980	5.275430	32	8.105500	6.578900
15	3.608100	5.396820	33	8.918850	6.416230
16	3.315830	5.553380	34	9.894830	6.221030
17	3.000100	5.756760	35	11.000000	6.000000
18	3.000000	6.000000			

ELEMENT COORDINATE DATA FOR BODY ID = 2,				NOZZLE WITH CENTERBO			
I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
1	11.000000	6.000000					
2	10.447415	5.889486	1.127056	1.127056	-.196113	-.980581	.000005
3	9.898830	5.778970					
4	9.406835	5.681370	.995319	.995319	-.196118	-.980580	.000001
5	8.918840	5.583770					
6	8.512170	5.502435	.829448	.829449	-.196118	-.980580	.000002
7	8.105500	5.421100					
8	7.766615	5.353320	.691194	.691194	-.196125	-.980579	.000000
9	7.427730	5.285580					
10	7.145320	5.229059	.576005	.576005	-.196109	-.980582	-.000016
11	6.862910	5.172580					
12	6.627570	5.125515	.480000	.480000	-.196184	-.980583	.000000
13	6.392230	5.078450					
14	6.196115	5.039225	.399998	.399998	-.196126	-.980579	.000000
15	6.000000	5.000000					
16	5.825264	4.999488	.349386	.349503	.019443	-.999811	-.256431
17	5.658680	5.000800					
18	5.476115	5.015349	.349531	.349555	.059108	-.998252	-.116807
19	5.301760	5.027460					
20	5.127741	5.043183	.349436	.349464	.100963	-.994890	-.126222
21	4.959110	5.062740					
22	4.781150	5.086228	.349067	.349103	.146986	-.989151	-.142858
23	4.608830	5.114020					
24	4.437616	5.146273	.348412	.348464	.199763	-.979844	-.171316
25	4.267440	5.183620					
26	4.099025	5.226294	.347411	.347497	.264269	-.964449	-.222055
27	3.932380	5.275430					
28	3.768555	5.331409	.346116	.346295	.349565	-.936912	-.321881
29	3.610100	5.396420					
30	3.458378	5.468221	.331750	.332212	.473127	-.880994	-.551048
31	3.315830	5.553380					
32	3.194132	5.645266	.303838	.305366	.669370	-.742929	-1.143543
33	3.090100	5.756760					
34	3.027862	5.872013	.259391	.262845	.937735	-.347352	-2.179330
35	3.000000	6.000000					
36	3.027862	6.127987	.259391	.262845	.937735	.347352	-2.179330
37	3.090100	6.243290					
38	3.194132	6.354734	.303838	.305366	.669370	.742929	-1.143543
39	3.315830	6.466620					
40	3.458378	6.531779	.331750	.332212	.473127	.880994	-.551048
41	3.608100	6.603580					
42	3.768555	6.668591	.346116	.346295	.549565	.936912	-.321881
43	3.932380	6.724570					
44	4.099025	6.773706	.347411	.347497	.264269	.964449	-.222055
45	4.267440	6.816380					
46	4.437616	6.853727	.348412	.348464	.199763	.979844	-.171316
47	4.608830	6.885980					
48	4.781150	6.913772	.349067	.349103	.146986	.989151	-.142858
49	4.959110	6.937260					

ELEMENT COORDINATE DATA FOR BODY ID = 20

NOZZLE WITH CENTERBO

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
	5.127281	6.956817	.389836	.889868	.100983	.998890	-.126222
26	5.301760	6.972540					
	5.878115	6.988651	.389531	.889558	.059108	.998252	-.116807
27	5.650600	6.993200					
	5.825264	7.000512	.389386	.889508	.019463	.999811	-.256331
28	6.000000	7.000000					
	6.196115	6.960775	.389998	.889998	-.196126	.980579	.000000
29	6.392230	6.921550					
	6.627570	6.878885	.880000	.880008	-.196108	.980583	.000000
30	6.862910	6.827420					
	7.145320	6.770941	.576005	.876005	-.196109	.980582	-.000016
31	7.427730	6.714460					
	7.766615	6.656680	.681198	.881198	-.196125	.980579	.000000
32	8.105500	6.578900					
	8.512170	6.497565	.829448	.829448	-.196118	.980580	.000002
33	8.918840	6.416230					
	9.306835	6.338632	.995319	.995319	-.196118	.980580	.000001
34	9.694830	6.221030					
	10.447415	6.110518	1.127056	1.127056	-.196113	.980581	.000005
35	11.000000	6.000000					

SUMDS = 16.878647

UNTRANSFORMED COORDINATE DATA FOR BODY ID = 3.

NOZZLE WITH CENTERBO

I	K(I)	V(I)	I	K(I)	V(I)
1	21.000000	-80.000000	51	15.283820	8.106180
2	21.000000	-36.979730	52	15.700620	8.481070
3	21.000000	-32.958660	53	15.901100	8.688430
4	21.000000	-29.939180	54	15.095100	8.928080
5	21.000000	-25.918910	55	19.782830	9.159850
6	21.000000	-22.398690	56	19.464490	9.383550
7	21.000000	-20.832200	57	18.110300	9.599030
8	21.000000	-15.860170	58	13.810460	9.806120
9	21.000000	-13.383370	59	13.828210	10.008660
10	21.000000	-11.319560	60	13.139750	10.194520
11	21.000000	-9.888630	61	12.789820	10.375580
12	21.000000	-8.166360	62	12.439260	10.547590
13	21.000000	-6.971870	63	12.088800	10.710580
14	21.000000	-5.976490	64	11.725560	10.864270
15	21.000000	-5.147200	65	11.362400	11.008660
16	21.000000	-4.456000	66	10.995860	11.143600
17	21.000000	-3.888800	67	10.625890	11.268990
18	21.000000	-3.400000	68	10.252030	11.384720
19	21.000000	-3.000000	69	9.875850	11.490720
20	20.994860	-2.607310	70	9.496090	11.586890
21	20.979390	-2.213750	71	9.112210	11.673160
22	20.953790	-1.822600	72	8.730080	11.749460
23	20.917790	-1.433120	73	8.343950	11.815730
24	20.871590	-1.040580	74	7.956080	11.871910
25	20.819970	-0.651250	75	7.566780	11.917950
26	20.768260	-0.263800	76	7.176190	11.953810
27	20.711360	-0.127100	77	6.784710	11.979460
28	20.654320	-0.06780	78	6.392560	11.994860
29	20.597200	-0.088390	79	6.000700	12.000000
30	20.540030	1.267860	80	5.600000	12.000000
31	20.482890	1.688320	81	5.200000	12.000000
32	20.425860	2.017720	82	4.800000	12.000000
33	19.368290	2.387190	83	4.400000	12.000000
34	19.852360	2.750270	84	4.000000	12.000000
35	19.696110	3.116720	85	3.600000	12.000000
36	19.530270	3.475980	86	3.200000	12.000000
37	19.358270	3.826900	87	2.800000	12.000000
38	19.170320	4.179320	88	2.400000	12.000000
39	18.976330	4.529110	89	2.000000	12.000000
40	18.773020	4.863030	90	1.600000	12.000000
41	18.561910	5.198230	91	1.200000	12.000000
42	18.340590	5.527080	92	0.800000	12.000000
43	18.110990	5.850150	93	0.400000	12.000000
44	17.872760	6.167200	94	0.000000	12.000000
45	17.626130	6.478030	95	-0.400000	12.000000
46	17.371230	6.782390	96	-0.800000	12.000000
47	17.108190	7.080090	97	-1.200000	12.000000
48	16.837190	7.370890	98	-1.600000	12.000000
49	16.558380	7.655600	99	-2.000000	12.000000
50	16.279450	7.923960	100	-2.400000	12.000000

UNTRANSFORMED COORDINATE DATA FOR BODY ID = 3, NOZZLE WITH CENTER80

I	X(I)	Y(I)	I	X(I)	Y(I)
101	-5.166360	12.000000	108	-23.665830	12.000000
102	-6.599630	12.000000	109	-27.932830	12.000000
103	-8.319560	12.000000	110	-32.199320	12.000000
104	-10.383870	12.000000	111	-36.466210	12.000000
105	-12.860170	12.000000	112	-40.733110	12.000000
106	-15.832200	12.000000	113	-45.000300	12.000000
107	-19.398680	12.000000			

ELEMENT COORDINATE DATA FOR BODY ID = 3.

NOZZLE WITH CENTERO

I	X(I)	Y(I)	DL	DS	SIGALF3	COSIALF1	CURVATURE
1	21.000000	-20.000000	3.520270	9.820270	1.000000	.000000	.000000
2	21.000000	-38.239885	3.520270	3.520270	1.000000	.000000	.000000
3	21.000000	-36.879733	3.520270	3.520270	1.000000	.000000	.000000
4	21.000000	-34.719595	3.520270	3.520270	1.000000	.000000	.000000
5	21.000000	-32.959460	3.520270	3.520270	1.000000	.000000	.000000
6	21.000000	-31.199320	3.520270	3.520270	1.000000	.000000	.000000
7	21.000000	-29.439180	3.520270	3.520270	1.000000	.000000	.000000
8	21.000000	-27.679045	3.520270	3.520270	1.000000	.000000	.000000
9	21.000000	-25.918910	3.520270	3.520270	1.000000	.000000	.000000
10	21.000000	-24.158775	3.520270	3.520270	1.000000	.000000	.000000
11	21.000000	-22.398640	3.520270	3.520270	1.000000	.000000	.000000
12	21.000000	-20.638505	3.520270	3.520270	1.000000	.000000	.000000
13	21.000000	-18.878370	3.520270	3.520270	1.000000	.000000	.000000
14	21.000000	-17.118235	3.520270	3.520270	1.000000	.000000	.000000
15	21.000000	-15.358100	3.520270	3.520270	1.000000	.000000	.000000
16	21.000000	-13.597965	3.520270	3.520270	1.000000	.000000	.000000
17	21.000000	-11.837830	3.520270	3.520270	1.000000	.000000	.000000
18	21.000000	-10.077695	3.520270	3.520270	1.000000	.000000	.000000
19	21.000000	-8.317560	3.520270	3.520270	1.000000	.000000	.000000
20	21.000000	-6.557425	3.520270	3.520270	1.000000	.000000	.000000
21	21.000000	-4.797290	3.520270	3.520270	1.000000	.000000	.000000
22	21.000000	-3.037155	3.520270	3.520270	1.000000	.000000	.000000
23	21.000000	-1.277020	3.520270	3.520270	1.000000	.000000	.000000
24	21.000000	0.483115	3.520270	3.520270	1.000000	.000000	.000000
25	21.000000	2.243230	3.520270	3.520270	1.000000	.000000	.000000

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ELEMENT COORDINATE DATA FOR BODY ID = 3, NOZZLE WITH CENTERBO

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
26	20.782883	-4.57107	.393545	.893554	.885528	-.169510	.066656
26	20.780260	-.263000	.393684	.393695	.980717	-.195335	.066707
27	20.671360	.122700	.393819	.393830	.975270	-.221015	.066638
28	20.584320	.506780	.393969	.393980	.869138	-.286517	.066674
29	20.487200	.888590	.394121	.394132	.662319	-.271922	.066676
30	20.380030	1.267860	.394261	.894272	.958880	-.297087	.066645
31	20.262900	1.644320	.394420	.594431	.986704	-.122094	.066689
32	20.135660	2.017720	.394570	.594581	.937908	-.395734	.066666
33	19.998990	2.387790	.394718	.394729	.928461	-.371930	.066656
34	19.852380	2.754270	.394866	.894898	.918365	-.395734	.066685
35	19.696110	3.116920	.395055	.395064	.907621	-.419790	.066696
36	19.530270	3.475480	.395215	.595226	.896286	-.443356	.066652
37	19.354970	3.829690	.395394	.895406	.884284	-.467002	.066652
38	19.170320	4.179320	.395567	.395579	.871684	-.490157	.066665
39	18.976430	4.524110	.395756	.895787	.858408	-.512867	.066679
40	18.773420	4.863830	.395944	.895954	.844564	-.535855	.066680
41	18.561410	5.198230	.396139	.396150	.830138	-.557557	.066660
42	18.340540	5.527080	.396346	.396358	.815121	-.579291	.066676
43	18.110940	5.850150	.396548	.396560	.799525	-.600633	.066673
44	17.872760	6.167200	.396789	.396801	.783363	-.621564	.066659
45	17.626130	6.478030	.397000	.597012	.766650	-.642066	.066658
46	17.371230	6.782390	.397260	.597271	.749384	-.662136	.066683
47	17.108190	7.080090	.397599	.897511	.731574	-.681762	.066694
48	16.837190	7.370890	.397777	.397788	.713239	-.700921	.066652
49	16.558380	7.654600	.397759	.397769	.694659	-.719339	.066645
50	16.279450	7.923960					

ELEMENT COORDINATE DATA FOR BODY ID = 3,

NOZZLE WITH CENTERBO

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
51	16.137308	8.055895	.388000	.388013	.675825	-.737042	.066676
	15.993470	8.186180					
	15.887870	8.316573	.388240	.388250	.656528	-.759302	.066682
52	15.700620	8.441070					
	15.551661	8.565720	.388457	.388488	.636725	-.771050	.066690
53	15.401100	8.688430					
	15.288876	8.809244	.388675	.388686	.616582	-.787290	.066653
54	15.095100	8.928080					
	14.938716	9.044977	.388883	.388894	.595989	-.802992	.066671
55	14.782830	9.159850					
	14.629385	9.272732	.389078	.389089	.574998	-.818190	.066667
56	14.464490	9.383550					
	14.303098	9.492381	.389270	.389280	.553580	-.832816	.066658
57	14.140300	9.599830					
	13.976052	9.703688	.389462	.389473	.531733	-.846812	.066690
58	13.810460	9.806120					
	13.643479	9.906878	.389629	.389640	.509562	-.860939	.066657
59	13.475210	10.004660					
	13.305597	10.100696	.389820	.389831	.487085	-.873377	.066673
60	13.134750	10.194520					
	12.962623	10.286153	.389987	.389998	.464916	-.885797	.066673
61	12.789320	10.375540					
	12.615799	10.462703	.390195	.390156	.440990	-.897512	.066666
62	12.439160	10.547590					
	12.262355	10.630219	.390312	.390323	.417986	-.908683	.066669
63	12.084490	10.710540					
	11.905525	10.788578	.390466	.390477	.393709	-.919235	.066663
64	11.725560	10.864270					
	11.544550	10.937697	.390626	.390637	.369638	-.929176	.066679
65	11.362600	11.008660					
	11.179669	11.077324	.390777	.390789	.345312	-.938988	.066658
66	10.995860	11.143600					
	10.811139	11.207502	.390925	.390936	.320752	-.947163	.066695
67	10.625590	11.268990					
	10.439187	11.328072	.391076	.391087	.295927	-.955211	.066666
68	10.252030	11.384720					
	10.064085	11.438997	.391219	.391225	.270952	-.962593	.066652
69	9.875450	11.490720					
	9.686084	11.540093	.391360	.391371	.245733	-.969338	.066695
70	9.496090	11.586890					
	9.305431	11.631271	.391503	.391514	.220396	-.975920	.066664
71	9.114210	11.673160					
	8.922399	11.712563	.391639	.391646	.194825	-.980838	.066656
72	8.730080	11.749460					
	8.537231	11.783856	.391776	.391787	.169153	-.985590	.066667
73	8.343950	11.815730					
	8.150198	11.845087	.391918	.391929	.143346	-.989673	.066680
74	7.956080	11.871910					
	7.761560	11.896202	.392053	.392064	.117433	-.993081	.066675
75	7.566740	11.917950					

ELEMENT COORDINATE DATA FOR BODY ID = 3,

NOZZLE WITH CENTERBO

I	X(I)	Y(I)	DL	DS	SIN(ALF)	COS(ALF)	CURVATURE
	-8.569165	12.000000	1.188380	1.188390	.000000	-1.000000	.000000
101	-5.166360	12.000000					
	-5.882995	12.000000	1.933270	1.933270	.000000	-1.000000	.000000
102	-6.599630	12.000000					
	-7.859595	12.000000	1.719930	1.719930	.000000	-1.000000	.000000
103	-8.319560	12.000000					
	-9.351515	12.000000	2.063910	2.063910	.000000	-1.000000	.000000
104	-10.383470	12.000000					
	-11.621820	12.000000	2.976700	2.976700	.000000	-1.000000	.000000
105	-12.860170	12.000000					
	-14.346185	12.000000	2.972030	2.972030	.000000	-1.000000	.000000
106	-15.832200	12.000000					
	-17.615820	12.000000	3.566980	3.566980	.000000	-1.000000	.000000
107	-19.398600	12.000000					
	-21.532085	12.000000	4.266890	4.266890	.000000	-1.000000	.000000
108	-23.665530	12.000000					
	-25.798980	12.000000	4.266900	4.266900	.000000	-1.000000	.000000
109	-27.932430	12.000000					
	-30.065875	12.000000	4.266890	4.266890	.000000	-1.000000	.000000
110	-32.199320	12.000000					
	-34.332765	12.000000	4.266890	4.266890	.000000	-1.000000	.000000
111	-36.466210	12.000000					
	-38.599660	12.000000	4.266900	4.266900	.000000	-1.000000	.000000
112	-40.733110	12.000000					
	-42.866555	12.000000	4.266890	4.266890	.000000	-1.000000	.000000
113	-45.000000	12.000000					

SUMDS = 111.561906

BODY GEOMETRY SUMMARY

BODY DESCRIPTION	BODY ID	LIFT TYPE	N/O	SID	TFORM	NORM	CMORD	TYPE	SIGMA				ELEMENT STORAGE	
									F	C	S	C	N/O	FIRST NO.
NOZZLE WITH CENTERBO	1	YES	NEW		NO	NO	.300000	P	1	1	1	1	NEW	1 65
NOZZLE WITH CENTERBO	2	YES	NEW		NO	NO	.090000	P	1	1	1	1	NEW	66 34
NOZZLE WITH CENTERBO	3	YES	NEW		NO	NO	.000000	P	1	1	1	1	NEW	100 112

TOTAL NUMBER OF BODIES = 3

TOTAL NUMBER OF ELEMENTS = 211

ELFORM COMPLETE, CALL MAFORM, T = .0005SECONDS.

VORTICITY WEIGHTING FUNCTION = S/L*(1 - S/L)

.022882	.068269	.101819	.135134	.168212	.189054
.208127	.221315	.230405	.236663	.240969	.243903
.245916	.247284	.248214	.248883	.249259	.249536
.249718	.249845	.249933	.249985	.250000	.249977
.249917	.249820	.249686	.249514	.249305	.249059
.248776	.248455	.248097	.247702	.247289	.246860
.246395	.245901	.245340	.244829	.244288	.243689
.243032	.242379	.241692	.240969	.240209	.239388
.238415	.237190	.235640	.233665	.231302	.227831
.223532	.217872	.210356	.202294	.186780	.168271
.144929	.118068	.088162	.055182	.019168	

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VORTICITY WEIGHTING FUNCTION = S/L*(1 - S/L)

.032272	.086993	.127720	.157195	.178657	.194389
.206005	.214826	.222165	.228645	.234265	.239022
.242918	.245954	.248107	.249394	.249989	.249939
.249394	.248107	.245954	.242918	.239022	.234265
.228645	.222165	.214826	.206005	.194390	.178657
.157195	.127720	.086993	.032272		

VERTICITY WEIGHTING FUNCTION = $S/L \cdot (1 - S/L)$

.015528	.085091	.032663	.098248	.821832	.193565
.161827	.175733	.166411	.194676	.201124	.206193
.210205	.213604	.215962	.218028	.219685	.221033
.222249	.223410	.224546	.225658	.226745	.227808
.228847	.229861	.230850	.231816	.232755	.233670
.234561	.235428	.236269	.237086	.237878	.238646
.239388	.240106	.240799	.241463	.242110	.242729
.243322	.243891	.244434	.244958	.245466	.245914
.246352	.246760	.247184	.247608	.248020	.248452
.248839	.249203	.249542	.249857	.249548	.249514
.249656	.249773	.249867	.249935	.249979	.249999
.249994	.249964	.249910	.249831	.249728	.249600
.249887	.249829	.249767	.249680	.249588	.249431
.2498007	.249704	.249515	.249300	.2490520	.2486084
.2495623	.249316	.249023	.248688	.248319	.247828
.2492313	.248722	.2481005	.247312	.246520	.245528
.249280	.248496	.247673	.246808	.245889	.244887
.247434	.246675	.245871	.245012	.244108	.243149
.245985	.245075	.244079	.243058	.242018	.240949

UNIFORM COMPLETE, CALL SOLVE, Y = .000SECONDS.

SOLVE TIME = .000 SECONDS.

THE 211 X 211 MATRIX WITH 5 RIGHT SIDES WAS SOLVED DIRECTLY IN .000 MINUTES.

SOLVE TIME = .000 SECONDS.

SOLVE COMPLETE, READ FLOW TITLE & CONTROL CARD, CALL COMBO, Y = .000SECONDS.

ON RETURN FROM MSL, MERR = 0

COMBINATION CONSTANTS

ALPHA =	0	90	.000000
1	-.151631	.134504	-.151631
2	-.007417	.008360	-.007417
3	.026002	-.780141	.926002

LIFT CURVE CONSTANTS

PK1 = 14.050192 PK2 = -12.556113 PK3 = .000000

ALPHA0 = 130.200014

ALPHA = .000000

CLT = 373.425842

COMBO COMPLETE. CALL FLOWS.Y = .0005ECONDS.

INDIVIDUAL FLOW NUMBER 1

PT.NO.	VN	VT	SIGMA
1	.000000	.538021	.012913
2	.000000	.540255	.016034
3	.000000	.540333	.019160
4	.000000	.540226	.022292
5	.000000	.540219	.025521
6	.000000	.540345	.028969
7	.000000	.540278	.032068
8	.000000	.541715	.035762
9	.000000	.543885	.038886
10	.000000	.546240	.041853
11	.000000	.550332	.044665
12	.000000	.555838	.047302
13	.000000	.562712	.049787
14	.000000	.570770	.051977
15	.000000	.579785	.053978
16	.000000	.589258	.055739
17	.000000	.598858	.057265
18	.000000	.608207	.058569
19	.000000	.617890	.059673
20	.000000	.625960	.060684
21	.000000	.635761	.061696
22	.000000	.646456	.062705
23	.000000	.658100	.063708
24	.000000	.670656	.064695
25	-.000000	.684123	.065669
26	.000000	.698524	.066632
27	.000000	.713907	.067587
28	-.000000	.730400	.068546
29	-.000000	.748326	.069527
30	-.000000	.768270	.070545
31	-.000000	.791309	.071620
32	-.000000	.819573	.072757
33	-.000000	.857507	.073897
34	-.000000	.927918	.074656
35	-.000000	1.033690	.069889
36	-.000000	1.111885	.049184
37	.000000	1.147150	.036263
38	.000000	1.167324	.024533
39	.000000	1.176368	.013698
40	.000000	1.177476	.003347
41	.000000	1.171941	-.006742
42	.000000	1.160240	-.016577
43	.000000	1.142017	-.026391
44	.000000	1.115539	-.036611
45	.000000	1.078498	-.047432
46	.000000	1.029065	-.059144
47	.000000	.952387	-.073725
48	.000000	.835184	-.083479
49	.000000	.759448	-.083074
50	.000000	.711291	-.081730
51	.000000	.673120	-.080001
52	-.000000	.641805	-.078079
53	.000000	.615959	-.076093
54	.000000	.594805	-.074155

55	-.000000	.57896	-.072380
56	-.000000	.564728	-.070884
57	.000000	.555823	-.069803
58	-.000000	.548215	-.069295
59	.000000	.543512	-.069547
60	-.000000	.538635	-.070841
61	.000000	.534203	-.073433
62	-.000000	.522242	-.077458
63	.000000	.492128	-.083271
64	-.000000	.413823	-.091368
65	.000000	.175095	-.101797
66	-.000000	.128287	-.101799
67	-.000000	-.284804	-.181558
68	-.000000	-.807845	-.138063
69	.000000	-.512437	-.115352
70	-.000000	-.601277	-.101374
71	.000000	-.687209	-.091875
72	-.000000	-.815228	-.083518
73	-.000000	-.921697	-.050925
74	-.000000	-.875905	-.089581
75	-.000000	-.890119	-.044736
76	-.000000	-.902535	-.039302
77	-.000000	-.914280	-.033330
78	-.000000	-.924847	-.026589
79	-.000000	-.934267	-.018236
80	-.000000	-.939054	-.006800
81	-.000000	-.914670	.011705
82	-.000000	-.704562	.081834
83	-.000000	-.198528	.069112
84	-.000000	.205329	.067164
85	-.000000	.380406	.062368
86	-.000000	.462808	.059035
87	.000000	.508965	.057078
88	-.000000	.538365	.056064
89	-.000000	.559628	.055680
90	-.000000	.572258	.055702
91	-.000000	.597152	.055863
92	-.000000	.627039	.053962
93	-.000000	.644518	.042098
94	-.000000	.563233	.055935
95	-.000000	.546409	.070896
96	-.000000	.540562	.089295
97	-.000000	.546928	.117221
98	-.000000	.561381	.184766
99	.000000	.844420	.209031
100	.000001	-.241931	.135812
101	.000001	-.677051	.131870
102	.000001	-.593330	.130208
103	.000001	-.561299	.130746
104	.000001	-.548412	.133109
105	.000001	-.542555	.137073
106	-.000001	-.538662	.142153
107	.000001	-.534322	.147617
108	-.000001	-.528342	.153266
109	.000001	-.520296	.158971
110	.000001	-.510249	.164640
111	.000000	-.498564	.170209

112	.000001	-.485777	.175648
113	.000001	-.472385	.180960
114	.000000	-.458751	.186180
115	.000000	-.445113	.191389
116	.000000	-.431615	.196740
117	.000000	-.415808	.202689
118	.000000	-.400723	.209635
119	.000000	-.389419	.218196
120	.000000	-.379369	.217003
121	.000000	-.371508	.219190
122	.000000	-.364935	.220927
123	.000000	-.359592	.222231
124	.000000	-.354545	.223238
125	.000000	-.350313	.223992
126	-.000000	-.346519	.224480
127	-.000000	-.343116	.224788
128	-.000000	-.340178	.224798
129	-.000000	-.337529	.224652
130	-.000000	-.335255	.224322
131	-.000000	-.333295	.223811
132	-.000000	-.331634	.223132
133	-.000000	-.330254	.222293
134	-.000000	-.329131	.221295
135	-.000000	-.328279	.220184
136	-.000001	-.327674	.218851
137	-.000000	-.327371	.217417
138	-.000000	-.327294	.215848
139	-.000000	-.327538	.214145
140	-.000001	-.328022	.212314
141	-.000001	-.328805	.210362
142	-.000001	-.329884	.208289
143	-.000001	-.331282	.206103
144	-.000001	-.332990	.203807
145	-.000001	-.335009	.201410
146	-.000001	-.337365	.198911
147	-.000001	-.340082	.196316
148	-.000001	-.343041	.193672
149	-.000001	-.346280	.190987
150	-.000001	-.349796	.188230
151	-.000001	-.353560	.185410
152	-.000001	-.357548	.182526
153	-.000000	-.361667	.179595
154	-.000000	-.365906	.176612
155	-.000001	-.370100	.173587
156	-.000001	-.374238	.170519
157	-.000001	-.378199	.167405
158	-.000000	-.381903	.164249
159	-.000000	-.385312	.161082
160	-.000001	-.388366	.157781
161	-.000000	-.391051	.154459
162	-.000000	-.393334	.151073
163	-.000000	-.395284	.147615
164	-.000000	-.396938	.144079
165	-.000000	-.398153	.140463
166	-.000000	-.399655	.136758
167	-.000000	-.400888	.132969
168	-.000000	-.402080	.129092

169	-000000	-403401	.125121
170	-000000	-404915	.121059
171	-000000	-406674	.116900
172	-000000	-408198	.112436
173	-000000	-411398	.108256
174	-000000	-414682	.103742
175	-000000	-418760	.099063
176	-000000	-423311	.094159
177	-000000	-433470	.088863
178	-000000	-446155	.083853
179	-000000	-455308	.080201
180	-000000	-460967	.077478
181	-000000	-464982	.075486
182	-000000	-467960	.073508
183	-000000	-470288	.071679
184	-000000	-472233	.069963
185	-000000	-474014	.068336
186	-000000	-475815	.066788
187	-000000	-477725	.065310
188	-000000	-479816	.063899
189	-000000	-482129	.062555
190	-000000	-484866	.061274
191	-000000	-487354	.060064
192	-000000	-490205	.058915
193	-000000	-493162	.057832
194	-000000	-496337	.056708
195	-000000	-500350	.055450
196	-000000	-504921	.054058
197	-000000	-510013	.052537
198	-000000	-515474	.050890
199	-000000	-520966	.049117
200	-000000	-526191	.047205
201	-000000	-530659	.045138
202	-000000	-534245	.042881
203	-000000	-536830	.040400
204	-000000	-538490	.037658
205	-000000	-539198	.034612
206	-000000	-539832	.031237
207	-000000	-539979	.027822
208	-000000	-539974	.024569
209	-000000	-539730	.021809
210	-000000	-538493	.018246
211	-000000	-527832	.015214

INDIVIDUAL FLOW NUMBER 2

PT.NO.	WM	VT	SIGNA
1	.000000	-.151918	-.099427
2	.000000	-.375604	-.087920
3	.000000	-.AA0228	-.020860
4	.000000	-.461093	-.073536
5	.000000	-.367888	-.069271
6	.000000	-.970207	-.066059
7	.000000	-.871248	-.063784
8	.000000	-.472329	-.062896
9	.000000	-.873938	-.061511
10	.000000	-.476422	-.061138
11	.000000	-.880088	-.061113
12	.000000	-.484822	-.061341
13	.000000	-.490825	-.061739
14	.000000	-.997873	-.062232
15	.000000	-.505206	-.062761
16	.000000	-.513988	-.063281
17	.000000	-.522366	-.063768
18	.000000	-.530512	-.064192
19	.000000	-.538218	-.064563
20	.000000	-.545996	-.064906
21	.000000	-.554587	-.065289
22	.000000	-.563884	-.065588
23	.000000	-.573025	-.065917
24	.000000	-.582966	-.066235
25	.000000	-.592710	-.066540
26	.000000	-.602270	-.066833
27	.000000	-.612687	-.067118
28	.000000	-.623073	-.067405
29	.000000	-.632710	-.067707
30	.000000	-.640107	-.068037
31	.000000	-.650208	-.068310
32	.000000	-.714853	-.068825
33	.000000	-.782924	-.069224
34	.000000	-.809232	-.069237
35	.000000	-.901684	-.069016
36	.000000	-.969975	-.065902
37	.000000	-1.000616	-.038404
38	.000000	-1.018204	-.024023
39	.000000	-1.026066	-.018872
40	.000000	-1.027015	-.005897
41	.000000	-1.022189	-.003417
42	.000000	-1.011967	-.011975
43	.000000	-.996060	-.020480
44	.000000	-.972950	-.029298
45	.000000	-.950605	-.038586
46	.000000	-.897366	-.048576
47	.000000	-.821837	-.060879
48	.000000	-.728541	-.088690
49	.000000	-.662367	-.067639
50	.000000	-.620371	-.065702
51	.000000	-.587077	-.063308
52	.000000	-.559766	-.060587
53	.000000	-.537232	-.057609
54	.000000	-.518830	-.054428

55	-.000000	-.504061	.051085
56	-.000000	-.492678	.087614
57	-.000000	-.484340	.044043
58	-.000000	-.478663	.080381
59	-.000000	-.475139	.036622
60	-.000000	-.473198	.032710
61	-.000000	-.472401	.028835
62	-.000000	-.472261	.025176
63	-.000000	-.472519	.021657
64	-.000000	-.472663	.018180
65	-.000000	-.465196	.014764
66	-.000000	-.471296	.024310
67	-.000000	.298480	.158381
68	-.000000	.355761	.120480
69	-.000000	.446998	.100631
70	-.000000	.524931	.088438
71	-.000000	.599446	.080152
72	-.000000	.711113	.072859
73	-.000000	.804160	.044431
74	-.000000	.764081	.081223
75	-.000000	.776427	.039032
76	-.000000	.787289	.038294
77	-.000000	.797507	.029385
78	-.000000	.806828	.023167
79	-.000000	.814952	.015916
80	-.000000	.819135	.005939
81	-.000000	.797824	-.010202
82	-.000000	.614642	-.038232
83	-.000000	.173251	-.060285
84	-.000000	-.119231	-.058588
85	-.000000	-.331736	-.054406
86	-.000000	-.403690	-.051500
87	-.000000	-.443829	-.049793
88	-.000000	-.469444	-.048923
89	-.000000	-.488046	-.048594
90	-.000000	-.503378	-.048612
91	-.000000	-.520871	-.048755
92	-.000000	-.536986	-.047101
93	-.000000	-.562363	-.036748
94	-.000000	-.491496	-.048811
95	-.000000	-.476729	-.041849
96	-.000000	-.471633	-.048508
97	-.000000	-.477130	-.102257
98	-.000000	-.489648	-.143734
99	-.000000	-.736559	-.252188
100	-.000000	.419277	.019259
101	-.000000	.467179	.023335
102	-.000000	.470295	.026875
103	-.000000	.471039	.030559
104	-.000000	.471012	.034440
105	-.000000	.470339	.038641
106	-.000000	.468566	.042902
107	-.000000	.465427	.046870
108	-.000000	.460891	.050611
109	-.000000	.453642	.054192
110	-.000000	.444990	.057471
111	-.000000	.434892	.061095

112	-.000000	.423794	.064492
113	-.000000	.412150	.062878
114	-.000000	.400335	.071266
115	-.000000	.388493	.074670
116	-.000000	.376606	.078151
117	-.000000	.363709	.081886
118	-.000000	.349296	.087252
119	-.000000	.337635	.092876
120	-.000000	.330334	.097848
121	-.000000	.323862	.102628
122	-.000000	.318355	.107278
123	-.000000	.313078	.111812
124	-.000000	.309377	.116230
125	-.000000	.305581	.120400
126	-.000000	.302302	.124887
127	-.000000	.299391	.129108
128	-.000000	.296780	.133268
129	-.000000	.294526	.137360
130	-.000000	.292521	.141395
131	-.000000	.290780	.145366
132	-.000000	.289323	.149272
133	-.000000	.288102	.153116
134	-.000001	.287111	.156894
135	-.000000	.286375	.160602
136	-.000000	.285877	.164241
137	-.000000	.285576	.167810
138	-.000000	.285500	.171305
139	-.000000	.285719	.174724
140	-.000001	.286162	.178061
141	-.000000	.286895	.181317
142	-.000001	.287791	.184485
143	-.000000	.288939	.187562
144	-.000000	.290471	.190545
145	-.000000	.292236	.193431
146	-.000000	.294283	.196213
147	-.000000	.296620	.198879
148	-.000000	.299221	.201400
149	-.000000	.302036	.203773
150	-.000000	.305107	.206024
151	-.000000	.308398	.208146
152	-.000000	.311861	.210129
153	-.000000	.315483	.211974
154	-.000000	.319122	.213673
155	-.000000	.322832	.215219
156	-.000001	.326417	.216614
157	-.000001	.329880	.217849
158	-.000000	.333131	.218930
159	-.000000	.336185	.219855
160	-.000000	.339039	.220625
161	-.000000	.341653	.221245
162	-.000000	.344072	.221711
163	-.000000	.346369	.222029
164	-.000001	.348417	.222193
165	-.000000	.349369	.222208
166	-.000000	.349560	.222061
167	-.000001	.349661	.221745
168	-.000001	.350709	.221259

169	-.000001	.351871	.220577
170	-.000000	.353192	.219689
171	-.000000	.354749	.218572
172	-.000001	.356616	.217199
173	-.000001	.358934	.215527
174	-.000001	.361838	.213852
175	-.000001	.365566	.211024
176	-.000001	.371682	.207957
177	-.000000	.378320	.203194
178	-.000001	.387666	.196169
179	-.000001	.396761	.190647
180	-.000001	.401570	.186800
181	-.000001	.405105	.182844
182	-.000001	.407748	.178767
183	-.000001	.409806	.174555
184	-.000001	.411528	.170334
185	-.000001	.413192	.172451
186	-.000001	.414679	.170366
187	-.000001	.416363	.168645
188	-.000001	.418207	.168260
189	-.000001	.420247	.165388
190	-.000001	.422467	.163908
191	-.000001	.424854	.162506
192	-.000001	.427380	.161168
193	-.000001	.429907	.159882
194	-.000001	.432774	.158523
195	-.000001	.436194	.156956
196	-.000001	.440172	.155155
197	-.000001	.444651	.153097
198	-.000001	.449813	.150761
199	-.000001	.454221	.148140
200	-.000001	.458760	.145237
201	-.000001	.462740	.142074
202	-.000001	.465925	.138689
203	-.000001	.468243	.135133
204	-.000001	.469787	.131673
205	-.000001	.470841	.127791
206	-.000001	.471806	.124237
207	-.000001	.474285	.121366
208	-.000001	.481035	.119695
209	-.000001	.502040	.119699
210	-.000001	.568037	.122161
211	-.000001	.808716	.127634

INDIVIDUAL FLOW NUMBER 3

PT.NO.	VM	VT	SIGMA
1	-.000001	1.181783	-.187672
2	-.000001	.979528	-.201942
3	-.000001	.918760	-.203768
4	-.000000	.901838	-.199696
5	-.000001	.896538	-.191191
6	-.000000	.899965	-.179444
7	-.000000	.895086	-.166586
8	-.000000	.896582	-.159769
9	-.000000	.899128	-.182282
10	-.000080	.903702	-.135147
11	-.000000	.910843	-.122187
12	-.000000	.919530	-.120245
13	-.000000	.930823	-.113189
14	-.000000	.940285	-.108901
15	-.000000	.959170	-.108308
16	-.000000	.979871	-.100320
17	-.000000	.990778	-.096897
18	-.000000	1.004276	-.093973
19	-.000000	1.020802	-.091883
20	-.000000	1.035664	-.089178
21	-.000000	1.051889	-.086834
22	-.000000	1.069599	-.084451
23	-.000000	1.088893	-.082311
24	-.000000	1.109610	-.079517
25	-.000000	1.131889	-.076938
26	-.000000	1.155710	-.074242
27	-.000000	1.181158	-.071379
28	-.000000	1.208423	-.068278
29	-.000000	1.238071	-.064834
30	-.000000	1.271043	-.060908
31	-.000000	1.309157	-.056302
32	-.000000	1.352936	-.050727
33	-.000000	1.401858	-.044313
34	-.000000	1.522506	-.038289
35	-.000000	1.707970	-.022810
36	-.000000	1.836622	-.015827
37	-.000000	1.897893	-.011064
38	-.000000	1.931045	-.007013
39	-.000000	1.945970	-.003203
40	-.000000	1.947758	.000591
41	-.000000	1.938600	.002856
42	-.000000	1.919236	.008390
43	-.000000	1.889065	.012539
44	-.000000	1.845190	.017239
45	-.000000	1.783809	.022877
46	-.000000	1.698285	.030128
47	-.000000	1.556255	.041173
48	-.000000	1.370364	.057560
49	-.000000	1.255161	.071917
50	-.000000	1.176700	.085296
51	-.000000	1.113612	.098279
52	-.000000	1.061852	.111046
53	-.000000	1.019144	.123659
54	-.000000	.984177	.136137

55	-.000000	.956183	.198478
56	-.000000	.314630	.160676
57	-.000000	.918951	.172677
58	-.000000	.908812	.188813
59	-.000000	.902257	.195702
60	-.000000	.899980	.206153
61	-.000000	.902068	.214382
62	-.000000	.911281	.219114
63	-.000000	.937200	.219660
64	-.000000	1.010611	.218663
65	-.000000	1.258577	.199943
66	-.000000	.213285	.899299
67	-.000000	.870938	-.300368
68	-.000000	.867419	-.228420
69	-.000000	-.847442	-.190859
70	-.000000	-.995262	-.167738
71	-.000000	-1.136642	-.152027
72	-.000000	-1.138825	-.138189
73	-.000000	-1.524895	-.084279
74	-.000000	-1.848882	-.081932
75	-.000000	-1.472382	-.074097
76	-.000000	-1.492950	-.065061
77	-.000000	-1.512476	-.055183
78	-.000000	-1.530336	-.043968
79	-.000000	-1.545854	-.030211
80	-.000000	-1.553854	-.011283
81	-.000000	-1.513480	-.019340
82	-.000000	-1.168047	.072513
83	-.000000	-.328796	.118355
84	-.000000	.339427	.111180
85	-.000000	.629128	.103209
86	-.000000	.765635	.097896
87	-.000000	.841819	.094459
88	-.000000	.890421	.092790
89	-.000000	.925637	.092159
90	-.000000	.958835	.092184
91	-.000000	.987779	.092461
92	-.000000	1.037211	.089316
93	-.000000	1.066236	.086691
94	-.000000	.931860	.092589
95	-.000000	.903910	.117319
96	-.000000	.894191	.148911
97	-.000000	.904692	.193948
98	-.000001	.928529	.272598
99	-.000001	1.396762	.478195
100	-.000000	-.622165	-.049791
101	-.000000	-.789117	-.038423
102	-.000000	-.852048	-.028415
103	-.000000	-.877386	-.020000
104	-.000000	-.887057	-.012749
105	-.000000	-.889664	-.008066
106	-.000000	-.887853	-.000005
107	-.000000	-.882381	.005250
108	-.000000	-.873249	.009334
109	-.000000	-.860336	.014131
110	-.000000	-.843915	.017857
111	-.000000	-.824723	.021113

112	.000000	-.803690	.023915
113	.000000	-.781521	.026297
114	.000000	-.759041	.028311
115	.000000	-.736521	.030028
116	.000000	-.714057	.031494
117	.000000	-.688728	.032934
118	.000000	-.662518	.032934
119	.000000	-.641702	.031388
120	.000000	-.626592	.029620
121	.000000	-.619217	.027806
122	.000000	-.603590	.025967
123	.000000	-.598898	.028088
124	.000000	-.586476	.022236
125	.000000	-.579887	.020366
126	.000000	-.573886	.018496
127	.000000	-.567526	.016627
128	.000000	-.562616	.014757
129	.000000	-.558291	.012895
130	.000000	-.554503	.011034
131	.000000	-.551280	.009176
132	.000000	-.548460	.007323
133	.000000	-.546138	.005471
134	.000000	-.544279	.003626
135	.000000	-.542850	.001785
136	.000000	-.541867	-.000050
137	.000000	-.541317	-.001882
138	.000000	-.541214	-.003708
139	.000000	-.541570	-.005530
140	.000000	-.542391	-.007343
141	.000000	-.543682	-.009188
142	.000000	-.545465	-.010943
143	.000000	-.547759	-.012728
144	.000000	-.550565	-.014491
145	.000000	-.553905	-.016240
146	.000000	-.557795	-.017966
147	.000000	-.562236	-.019662
148	.000000	-.567129	-.021303
149	.000000	-.572463	-.022894
150	.000000	-.578294	-.024435
151	.000000	-.584590	-.025929
152	.000000	-.591140	-.027364
153	.000000	-.597983	-.028736
154	.000000	-.604954	-.030046
155	.000000	-.611942	-.031284
156	.000000	-.618771	-.032459
157	.000000	-.625325	-.033570
158	.000000	-.631482	-.034623
159	.000000	-.637110	-.035631
160	.000000	-.642145	-.036599
161	.000000	-.646586	-.037584
162	.000000	-.650367	-.038478
163	.000000	-.653608	-.039410
164	.000000	-.656341	-.040352
165	.000000	-.658710	-.041308
166	.000000	-.660833	-.042288
167	.000000	-.662858	-.043244
168	.000000	-.664858	-.044311

169	•000000	-.667070	-.045354
170	•000000	-.669574	-.046412
171	•000000	-.672534	-.047480
172	•000000	-.676070	-.048553
173	•000000	-.680415	-.049619
174	•000000	-.685842	-.050645
175	•000000	-.692760	-.051667
176	•000000	-.702235	-.052598
177	•000000	-.717097	-.053170
178	•000000	-.736855	-.051909
179	•000000	-.752776	-.049732
180	•000000	-.762077	-.047811
181	•000000	-.768762	-.046322
182	•000000	-.773719	-.044917
183	•000000	-.777605	-.043673
184	•000000	-.780853	-.042567
185	•000000	-.783828	-.041581
186	•000000	-.786819	-.040698
187	•000000	-.789999	-.039884
188	•000000	-.793501	-.039135
189	•000000	-.797395	-.038422
190	•000000	-.801591	-.037729
191	•000000	-.806081	-.037099
192	•000000	-.810823	-.036371
193	•000000	-.815718	-.035682
194	•000000	-.821137	-.034915
195	•000000	-.827656	-.033978
196	•000000	-.835220	-.032819
197	•000000	-.843698	-.031392
198	•000000	-.852744	-.029659
199	•000000	-.861858	-.027552
200	•000000	-.870912	-.025071
201	•000000	-.879886	-.022192
202	•000000	-.888386	-.018909
203	•000000	-.8968136	-.015213
204	•000000	-.890803	-.011055
205	•000000	-.892196	-.006326
206	•000000	-.892316	-.000792
207	•000000	-.891031	•005419
208	•000000	-.885864	•012272
209	•000000	-.869864	•020405
210	•000000	-.821758	•030555
211	•000000	-.674881	•042807

INDIVIDUAL FLOW NUMBER 4

PT.NO.	VN	VT	SIGMA
1	-.000000	.061620	-.013095
2	-.000000	.039010	-.013617
3	-.000000	.032766	-.018661
4	-.000000	.030848	-.016221
5	-.000000	.029360	-.018150
6	-.000000	.027924	-.021551
7	-.000000	.023908	-.025847
8	-.000000	.015510	-.030180
9	-.000000	.000354	-.035820
10	-.000000	-.023722	-.040952
11	-.000000	-.058053	-.046678
12	-.000000	-.102705	-.052256
13	-.000000	-.156382	-.057808
14	-.000000	-.216523	-.061900
15	-.000000	-.280033	-.065613
16	-.000000	-.343874	-.068517
17	-.000000	-.405367	-.070663
18	-.000000	-.462592	-.072163
19	-.000000	-.518259	-.073132
20	-.000000	-.569177	-.073737
21	-.000000	-.616682	-.074058
22	-.000000	-.671207	-.074025
23	-.000000	-.727278	-.073620
24	-.000000	-.784144	-.072814
25	-.000000	-.841032	-.071596
26	-.000000	-.897100	-.069971
27	-.000000	-.951532	-.067862
28	-.000000	-1.003650	-.065619
29	-.000000	-1.052941	-.063010
30	-.000000	-1.099325	-.060227
31	-.000000	-1.143422	-.057383
32	-.000000	-1.186959	-.054617
33	-.000000	-1.238281	-.052111
34	-.000000	-1.315299	-.050030
35	-.000000	-1.422903	-.035757
36	-.000000	-1.465075	-.012003
37	-.000000	-1.483365	.008671
38	-.000000	-1.336928	.028407
39	-.000000	-1.228306	.040945
40	-.000000	-1.085419	.052424
41	-.000000	-.943196	.061013
42	-.000000	-.805416	.068973
43	-.000000	-.677185	.079926
44	-.000000	-.558916	.073524
45	-.000000	-.453013	.075309
46	-.000000	-.362277	.076918
47	-.000000	-.276053	.080784
48	-.000000	-.202970	.085222
49	-.000000	-.153689	.088360
50	-.000000	-.114984	.081643
51	-.000000	-.082772	.077829
52	-.000000	-.055797	.073214
53	-.000000	-.033300	.067992
54	-.000000	-.014968	.062330

55	-0.00000	-0.00095	0.05600
56	-0.00000	0.01047	0.05039
57	-0.00000	0.01837	0.04463
58	-0.00000	0.23729	0.01876
59	-0.00000	0.27088	0.03415
60	-0.00000	0.29072	0.28630
61	-0.00000	0.03021	0.24452
62	-0.00000	0.01816	0.21544
63	-0.00000	0.03506	0.01906
64	-0.00000	0.04357	0.01211
65	-0.00000	0.07139	0.01633
66	-0.00000	2.73110	-1.59233
67	-0.00000	1.36985	-0.85141
68	-0.00000	1.38339	-0.58730
69	-0.00000	1.34877	-0.42636
70	-0.00000	1.39535	-0.31806
71	-0.00000	1.48090	-0.23906
72	-0.00000	1.70817	-0.17381
73	-0.00000	1.75061	-0.14401
74	-0.00000	1.78552	-0.12555
75	-0.00000	1.77190	-0.09889
76	-0.00000	1.82958	-0.06908
77	-0.00000	1.91750	-0.03756
78	-0.00000	2.05173	-0.00382
79	-0.00000	2.26750	0.03218
80	-0.00000	2.68206	0.08820
81	-0.00000	3.33918	0.09224
82	-0.00000	4.24621	0.09102
83	-0.00000	4.28081	-0.05104
84	-0.00000	3.41582	-0.08628
85	-0.00000	2.73601	-0.06333
86	-0.00000	2.37152	-0.02784
87	-0.00000	2.16526	0.00790
88	-0.00000	2.04182	0.04170
89	-0.00000	1.96594	0.07330
90	-0.00000	1.92225	0.10282
91	-0.00000	1.91383	0.13008
92	-0.00000	1.93951	0.14865
93	-0.00000	1.92822	0.17205
94	-0.00000	1.69615	0.23772
95	-0.00000	1.62807	0.31663
96	-0.00000	1.60685	0.42360
97	-0.00000	1.63858	0.58430
98	-0.00000	1.68398	0.86067
99	-0.00000	3.05881	1.53706
100	-0.00000	0.07923	-0.01023
101	-0.00000	-0.01647	-0.01074
102	-0.00000	-0.02748	-0.01029
103	-0.00000	-0.02094	-0.01042
104	-0.00000	-0.02658	-0.01086
105	-0.00000	-0.03088	-0.01169
106	-0.00000	-0.03297	-0.01275
107	-0.00000	-0.03526	-0.01395
108	-0.00000	-0.04060	-0.01522
109	-0.00000	-0.04606	-0.01607
110	-0.00000	-0.05180	-0.01789
111	-0.00000	-0.06160	-0.01910

112	.000000	-.070068	-.020190
113	.000000	-.078288	-.021147
114	.000000	-.085701	-.022096
115	.000000	-.082078	-.022839
116	.000000	-.097321	-.023624
117	.000000	-.100927	-.024887
118	.000000	-.103220	-.025713
119	.000000	-.106103	-.027045
120	.000000	-.110308	-.028200
121	.000000	-.114892	-.029315
122	.000000	-.119957	-.030412
123	.000000	-.125821	-.031496
124	.000000	-.131436	-.032564
125	.000000	-.137878	-.033636
126	.000000	-.144924	-.034726
127	.000000	-.152631	-.035815
128	.000000	-.160506	-.036913
129	.000000	-.169202	-.038018
130	.000000	-.178591	-.039134
131	.000000	-.188586	-.040261
132	.000000	-.199306	-.041397
133	.000000	-.211001	-.042543
134	.000000	-.223495	-.043698
135	.000000	-.236837	-.044858
136	.000000	-.251927	-.046023
137	.000000	-.267800	-.047188
138	.000000	-.283776	-.048351
139	.000000	-.301838	-.049506
140	.000000	-.321291	-.050643
141	.000000	-.342219	-.051759
142	.000000	-.364729	-.052840
143	.000000	-.388915	-.053876
144	.000000	-.414862	-.054851
145	.000000	-.442648	-.055751
146	.000000	-.472339	-.056554
147	.000000	-.503955	-.057232
148	.000000	-.536955	-.057763
149	.000000	-.571282	-.058132
150	.000000	-.607227	-.058305
151	.000000	-.644535	-.058261
152	.000000	-.682928	-.057974
153	.000000	-.722018	-.057419
154	.000000	-.761325	-.056586
155	.000000	-.800353	-.055454
156	.000000	-.838964	-.054030
157	.000000	-.875111	-.052324
158	.000000	-.909735	-.050356
159	.000000	-.943861	-.048163
160	.000000	-.977134	-.045776
161	.000000	-.997345	-.043284
162	.000000	-1.020376	-.040617
163	.000000	-1.046029	-.037934
164	.000000	-1.057283	-.035230
165	.000000	-1.071603	-.032582
166	.000000	-1.083601	-.029899
167	.000000	-1.093606	-.027300
168	.000000	-1.101929	-.024774

169	0.000000	-1.108937	-0.023110
170	0.000000	-1.114867	-0.019910
171	0.000000	-1.119982	-0.017571
172	0.000000	-1.124512	-0.015292
173	0.000000	-1.128676	-0.013068
174	0.000000	-1.132799	-0.010899
175	0.000000	-1.137289	-0.008788
176	0.000000	-1.143048	-0.006763
177	0.000000	-1.155495	-0.004840
178	0.000000	-1.172057	-0.002885
179	0.000000	-1.176839	-0.005667
180	0.000000	-1.185659	-0.011774
181	0.000000	-1.195097	-0.017606
182	0.000000	-1.117286	-0.021103
183	0.000000	-1.082096	-0.028208
184	0.000000	-1.080087	-0.032864
185	0.000000	-0.994573	-0.037033
186	0.000000	-0.884423	-0.040670
187	0.000000	-0.891506	-0.043760
188	0.000000	-0.816943	-0.046299
189	0.000000	-0.781797	-0.048303
190	0.000000	-0.727008	-0.049799
191	0.000000	-0.673391	-0.050625
192	0.000000	-0.621638	-0.051423
193	0.000000	-0.572251	-0.051661
194	0.000000	-0.520977	-0.051538
195	0.000000	-0.464188	-0.050986
196	0.000000	-0.403209	-0.049859
197	0.000000	-0.339415	-0.048046
198	0.000000	-0.276083	-0.045891
199	0.000000	-0.216023	-0.042203
200	0.000000	-0.162464	-0.038303
201	0.000000	-0.117658	-0.033996
202	0.000000	-0.083554	-0.029549
203	0.000000	-0.059888	-0.025237
204	0.000000	-0.044338	-0.021245
205	0.000000	-0.035945	-0.017877
206	0.000000	-0.031925	-0.015054
207	0.000000	-0.030279	-0.012994
208	0.000000	-0.029184	-0.011652
209	0.000000	-0.027075	-0.010829
210	0.000000	-0.020947	-0.010503
211	0.000000	-0.000238	-0.010624

INDIVIDUAL FLOW NUMBER 5

PT.NO.	VN	VT	SIGMA
1	-.000000	-.195852	-.057685
2	-.000000	-.343500	-.050286
3	-.000000	-.388880	-.044972
4	-.000000	-.403790	-.041649
5	-.000000	-.408676	-.039752
6	-.000000	-.410334	-.038866
7	-.000000	-.411163	-.038780
8	-.000000	-.412060	-.039228
9	-.000000	-.413489	-.040009
10	-.000000	-.415681	-.041001
11	-.000000	-.418839	-.042121
12	-.000000	-.423064	-.043501
13	-.000000	-.428326	-.045184
14	-.000000	-.434482	-.046624
15	-.000000	-.441358	-.048688
16	-.000000	-.448600	-.047648
17	-.000000	-.455915	-.048895
18	-.000000	-.463043	-.049226
19	-.000000	-.469773	-.049889
20	-.000000	-.476569	-.050423
21	-.000000	-.483437	-.050997
22	-.000000	-.492191	-.051569
23	-.000000	-.501074	-.052134
24	-.000000	-.510623	-.052690
25	-.000000	-.520872	-.053237
26	-.000000	-.531834	-.053775
27	-.000000	-.543588	-.054307
28	-.000000	-.556099	-.054843
29	-.000000	-.569746	-.055394
30	-.000000	-.584930	-.055973
31	-.000000	-.602472	-.056593
32	-.000000	-.624009	-.057257
33	-.000000	-.652873	-.057815
34	-.000000	-.706434	-.058267
35	-.000000	-.787027	-.059587
36	-.000000	-.846589	-.038486
37	-.000000	-.873396	-.028570
38	-.000000	-.888756	-.019587
39	-.000000	-.895647	-.011300
40	-.000000	-.896464	-.003405
41	-.000000	-.892249	.004882
42	-.000000	-.883337	.011763
43	-.000000	-.869481	.019216
44	-.000000	-.849303	.026964
45	-.000000	-.821084	.035153
46	-.000000	-.783390	.043993
47	-.000000	-.717809	.054951
48	-.000000	-.635848	.062135
49	-.000000	-.578198	.061588
50	-.000000	-.541481	.060503
51	-.000000	-.512409	.058684
52	-.000000	-.488555	.056866
53	-.000000	-.468860	.054933
54	-.000000	-.452731	.052958

55	-.000000	-.439801	.051011
56	-.000000	-.429789	.049165
57	-.000000	-.422395	.047509
58	-.000000	-.417184	.046133
59	-.000000	-.413637	.045170
60	-.000000	-.411079	.044409
61	-.000000	-.408854	.043735
62	-.000000	-.406947	.043242
63	-.000000	-.405342	.042953
64	-.000000	-.403982	.042841
65	-.000000	-.402826	.042841
66	-.000000	-.401821	.042841
67	-.000000	-.400912	.042841
68	-.000000	-.400000	.042841
69	-.000000	-.399088	.042841
70	-.000000	-.398176	.042841
71	-.000000	-.397264	.042841
72	-.000000	-.396352	.042841
73	-.000000	-.395440	.042841
74	-.000000	-.394528	.042841
75	-.000000	-.393616	.042841
76	-.000000	-.392704	.042841
77	-.000000	-.391792	.042841
78	-.000000	-.390880	.042841
79	-.000000	-.389968	.042841
80	-.000000	-.389056	.042841
81	-.000000	-.388144	.042841
82	-.000000	-.387232	.042841
83	-.000000	-.386320	.042841
84	-.000000	-.385408	.042841
85	-.000000	-.384496	.042841
86	-.000000	-.383584	.042841
87	-.000000	-.382672	.042841
88	-.000000	-.381760	.042841
89	-.000000	-.380848	.042841
90	-.000000	-.379936	.042841
91	-.000000	-.379024	.042841
92	-.000000	-.378112	.042841
93	-.000000	-.377200	.042841
94	-.000000	-.376288	.042841
95	-.000000	-.375376	.042841
96	-.000000	-.374464	.042841
97	-.000000	-.373552	.042841
98	-.000000	-.372640	.042841
99	-.000000	-.371728	.042841
100	-.000000	-.370816	.042841
101	-.000000	-.369904	.042841
102	-.000000	-.368992	.042841
103	-.000000	-.368080	.042841
104	-.000000	-.367168	.042841
105	-.000000	-.366256	.042841
106	-.000000	-.365344	.042841
107	-.000000	-.364432	.042841
108	-.000000	-.363520	.042841
109	-.000000	-.362608	.042841
110	-.000000	-.361696	.042841
111	-.000000	-.360784	.042841

112	-.000000	.369835	-.069535
113	-.000000	.359675	-.063877
114	-.000000	.349354	-.058973
115	-.000000	.339049	-.054768
116	.000000	.328730	-.051212
117	.000000	.320103	-.048333
118	.000000	.304927	-.048254
119	.000000	.295079	-.049938
120	.000000	.288184	-.051130
121	.000000	.282557	-.052106
122	.000000	.277676	-.052925
123	.000000	.273566	-.053648
124	.000000	.269869	-.054179
125	.000000	.266619	-.054660
126	.000000	.263763	-.055048
127	.000000	.261207	-.055353
128	.000000	.258968	-.055600
129	.000000	.256988	-.055772
130	.000000	.255271	-.055894
131	.000000	.253782	-.055965
132	.000000	.252513	-.055987
133	.000000	.251448	-.055969
134	.000000	.250593	-.055906
135	.000000	.249928	-.055806
136	.000000	.249484	-.055667
137	.000000	.249240	-.055506
138	.000000	.249190	-.055311
139	.000000	.249363	-.055099
140	.000000	.249732	-.054857
141	.000000	.250350	-.054598
142	.000000	.251164	-.054324
143	.000000	.252233	-.054039
144	.000000	.253580	-.053743
145	.000000	.255066	-.053440
146	.000000	.256683	-.053140
147	.000000	.258410	-.052840
148	.000000	.261151	-.052542
149	.000000	.263614	-.052265
150	.000000	.266292	-.052001
151	.000000	.269161	-.051758
152	.000000	.272204	-.051537
153	.000000	.275358	-.051341
154	.000000	.278585	-.051184
155	.000000	.281753	-.051047
156	.000000	.284896	-.050953
157	.000000	.287917	-.050889
158	.000000	.290745	-.050854
159	.000000	.293351	-.050853
160	.000000	.295676	-.050876
161	.000000	.297692	-.050917
162	.000000	.299435	-.050972
163	.000000	.300922	-.051040
164	.000000	.302195	-.051115
165	.000000	.303271	-.051195
166	.000000	.304252	-.051282
167	.000000	.305170	-.051358
168	.000000	.306111	-.051446

169	.000000	.307117	-.051532
170	.000000	.308256	-.051616
171	.000000	.309584	-.051699
172	.000000	.311177	-.051782
173	.000000	.313125	-.051854
174	.000000	.315562	-.051908
175	.000000	.318607	-.051931
176	.000000	.322146	-.051893
177	.000000	.330007	-.051525
178	.000000	.333187	-.051361
179	.000000	.346802	-.043436
180	.000000	.351112	-.038875
181	.000000	.359112	-.034775
182	.000000	.356136	-.030755
183	.000000	.358079	-.026869
184	.000000	.359539	-.023092
185	.000000	.360884	-.019404
186	.000000	.362237	-.015799
187	.000000	.363673	-.012266
188	.000000	.365254	-.008805
189	.000000	.367007	-.005413
190	.000000	.368951	-.002088
191	.000000	.370997	.001163
192	.000000	.373173	.004351
193	.000000	.375425	.007466
194	.000000	.377909	.010825
195	.000000	.380802	.015759
196	.000000	.383343	.021352
197	.000000	.386218	.028699
198	.000000	.392367	.030898
199	.000000	.396519	.038072
200	.000000	.400431	.044371
201	.000000	.403847	.055928
202	.000000	.406578	.066894
203	.000000	.408565	.079379
204	.000000	.409887	.093816
205	.000000	.410755	.109861
206	.000000	.411635	.125200
207	.000000	.413396	.139995
208	.000001	.418414	.151180
209	.000000	.433943	.158074
210	.000000	.448363	.159377
211	.000000	.665693	.151122

DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW

NOZZLE WITH CENTERBODY

ALPHA = .000000 ALPHA 0 = 130.200014 NO. OF BODIES 3

CL = 373.425842 CHORD = 1.000000 TOTAL ELEMENTS 211

BODY ID = 1 NOZZLE WITH CENTERBO NO. OF ELEMENTS 65

I	X	Y	S	VT	CP	J	SIGMA	VM
1	-42.866555	.000000	.023011	.173010	.970068	1	-.011999	.000000
2	-38.599660	.000000	.069035	.074115	.894507	2	.000114	.000000
3	-34.332765	.000000	.115058	.040674	.998346	3	.008522	-.000000
4	-30.065875	.000000	.161081	.029341	.999139	4	.014125	.000000
5	-25.798980	.000000	.207104	.025627	.999843	5	.017838	.000000
6	-21.532085	.000000	.253127	.024463	.999402	6	.020349	.000000
7	-17.615420	.000000	.295172	.024146	.999817	7	.022008	.000000
8	-14.346185	.000000	.330634	.024112	.999439	8	.023128	.000000
9	-11.621820	.000000	.360820	.024215	.999844	9	.023979	.000000
10	-9.351515	.000000	.388507	.024466	.999401	10	.024682	-.000000
11	-7.859595	.000000	.409914	.024865	.999382	11	.025292	-.000000
12	-5.882995	.000000	.421919	.025413	.999354	12	.025826	-.000000
13	-4.569165	.000000	.436090	.026084	.999320	13	.026295	-.000000
14	-3.474305	.000000	.447899	.026862	.999279	14	.026700	-.000000
15	-2.561920	.000000	.457740	.027698	.999233	15	.027048	-.000000
16	-1.8001600	.000000	.465941	.028583	.999183	16	.027337	.000000
17	-1.168000	.000000	.472775	.029449	.999123	17	.027576	-.000000
18	-.640000	.000000	.478470	.030277	.999083	18	.027770	-.000000
19	-.200000	.000000	.483216	.031043	.999056	19	.027926	-.000000
20	.200000	.000000	.487530	.031802	.998989	20	.028062	-.000000
21	.600000	.000000	.491845	.032617	.998956	21	.028189	-.000000
22	1.000000	.000000	.496159	.033480	.998879	22	.028307	-.000000
23	1.400000	.000000	.500474	.034396	.998817	23	.028414	-.000000
24	1.800000	.000000	.504788	.035363	.998748	24	.028501	-.000000
25	2.200000	.000000	.509103	.036402	.998675	25	.028569	-.000000
26	2.600000	.000000	.513417	.037457	.998597	26	.028612	-.000000
27	3.000000	.000000	.517731	.038544	.998514	27	.028625	-.000000
28	3.400000	.000000	.522046	.039660	.998427	28	.028601	-.000000
29	3.800000	.000000	.526360	.040819	.998334	29	.028530	-.000000
30	4.200000	.000000	.530675	.042048	.998252	30	.028396	-.000000
31	4.600000	.000000	.534989	.043391	.998167	31	.028177	-.000000
32	5.000000	.000000	.539304	.044946	.997980	32	.027833	-.000000
33	5.400000	.000000	.543618	.046987	.997792	33	.027282	-.000000
34	5.800000	.000000	.547932	.052655	.997227	34	.026271	-.000000
35	6.183177	-.007327	.552067	.056473	.996811	35	.021768	.000000
36	6.548490	-.050566	.556038	.060319	.996362	36	.016035	.000000
37	6.908226	-.190784	.560041	.061191	.996256	37	.011421	.000000
38	7.252367	-.273904	.564023	.061444	.996225	38	.007263	.000000
39	7.574562	-.446416	.567968	.060979	.996282	39	.003413	.000000
40	7.873128	-.656621	.571908	.060059	.996393	40	-.000284	.000000
41	8.143575	-.901168	.575844	.058761	.996547	41	-.003905	.000000

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DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW

NOZZLE WITH CENTERBODY

ALPHA = .000000

ALPHA 0 = 130.200014

NO. OF BODIES 3

CL = 373.425842

CHORD = 1.000000

TOTAL ELEMENTS 211

BODY ID = 1

NOZZLE WITH CENTERBO

NO. OF ELEMENTS 65

I	X	Y	S	VT	CB	J	SIGMA	VN
42	8.377462	-1.170339	.579692	.057226	.996725	42	-.007453	.000000
43	8.574709	-1.460242	.583476	.055476	.996922	43	-.011024	.000000
44	8.738146	-1.774216	.587296	.053881	.997144	44	-.014802	.000000
45	8.864360	-2.108132	.591149	.051052	.997394	45	-.018907	.000000
46	8.950530	-2.457451	.595032	.048818	.997617	46	-.023545	.000000
47	8.992769	-2.817786	.598948	.044136	.998052	47	-.029682	.000000
48	9.000000	-3.194700	.603015	.040104	.998392	48	-.035301	.000000
49	9.000000	-3.623045	.607635	.034748	.998793	49	-.037574	.000000
50	9.000000	-4.137060	.613180	.032307	.999056	50	-.039429	.000000
51	9.000000	-4.753875	.619833	.030383	.999077	51	-.041139	.000000
52	9.000000	-5.4494055	.627816	.028806	.999120	52	-.042802	.000000
53	9.000000	-6.382270	.637397	.027507	.999243	53	-.044479	.000000
54	9.000000	-7.448125	.648893	.026455	.999300	54	-.046221	.000000
55	9.000000	-8.727150	.662689	.025606	.999344	55	-.048076	.000000
56	9.000000	-10.261985	.679243	.024926	.999379	56	-.050094	.000000
57	9.000000	-12.103785	.699109	.024407	.999404	57	-.052323	.000000
58	9.000000	-14.313945	.722948	.023982	.999425	58	-.054825	.000000
59	9.000000	-16.966140	.751555	.023472	.999449	59	-.057643	.000000
60	9.000000	-20.148770	.785883	.022609	.999482	60	-.060819	.000000
61	9.000000	-23.696275	.824147	.020449	.999582	61	-.064141	.000000
62	9.000000	-27.319325	.863225	.018865	.999729	62	-.067354	.000000
63	9.000000	-30.942375	.902304	.000207	1.000000	63	-.070463	.000000
64	9.000000	-34.565425	.941382	.039242	.998405	64	-.073505	.000000
65	9.000000	-38.188475	.980461	.173010	.970068	65	-.076042	.000000

INTEGRATED VALUES

CY = -53.80214 CX = -39.86365

CL = -53.80214 CD = -39.86365 CM = -169.67242

PARABOLIC INTEGRATION

INTEGRATED VALUES

CY = -53.79242 CX = -39.85365

CL = -53.79242 CD = -39.85365 CM = -169.65198

DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW

NOZZLE WITH CENTERBODY

ALPHA = .00000 ALPHA 0 = 130.200014 NO. OF BODIES 3

CL = 373.425842 CHORD = 1.000000 TOTAL ELEMENTS 211

BODY ID = 2 NOZZLE WITH CENTERBODY NO. OF ELEMENTS 54

I	X	Y	S	VT	CP	J	SIGMA	VM
1	10.447415	5.443846	.03337	-.01420	.992745	66	-.001891	-.000000
2	9.406835	5.601370	.096259	-.022746	.999483	67	-.001570	-.000000
3	8.512170	5.502435	.150314	-.028029	.999248	68	-.001715	-.000000
4	7.766615	5.353320	.195360	-.032662	.998953	69	-.001907	-.000000
5	7.149320	5.229059	.232899	-.036832	.998636	70	-.002094	-.000000
6	6.627570	5.125515	.264181	-.041329	.998292	71	-.002262	-.000000
7	6.194115	5.039225	.290250	-.046466	.997852	72	-.002381	-.000000
8	5.825264	4.999488	.312452	-.053676	.997119	73	-.001163	-.000000
9	5.477415	5.015189	.333161	-.051591	.997134	74	-.001237	-.000000
10	5.127741	5.043183	.353068	-.052902	.997254	75	-.001225	-.000000
11	4.781150	5.086224	.374562	-.053382	.997150	76	-.001204	-.000000
12	4.437616	5.146273	.395226	-.054538	.997026	77	-.001173	-.000000
13	4.099025	5.226298	.415882	-.056248	.996870	78	-.001121	-.000000
14	3.768555	5.331909	.436395	-.057927	.996644	79	-.001024	-.000000
15	3.448378	5.468221	.456894	-.060083	.996293	80	-.000788	-.000000
16	3.194132	5.645266	.475381	-.064915	.995786	81	-.000153	-.000000
17	3.027862	5.872013	.492214	-.062204	.994138	82	.001501	-.000000
18	3.027862	6.127987	.507786	-.040257	.998399	83	-.003417	-.000000
19	3.128132	6.354234	.526618	-.016114	.999740	84	.003584	-.000000
20	3.458378	6.531779	.543506	-.003419	.999988	85	.003200	-.000000
21	3.768555	6.668591	.563805	.002882	.999742	86	.002790	-.000000
22	4.099025	6.773706	.586158	.006416	.999959	87	.002438	-.000000
23	4.437616	6.853727	.607774	.008205	.999924	88	.002145	-.000000
24	4.781150	6.913772	.625438	.010233	.999895	89	.001493	-.000000
25	5.127741	6.956817	.646132	.011311	.999872	90	.001674	-.000000
26	5.477415	6.984651	.666839	.012243	.999850	91	.001478	-.000000
27	5.825264	7.000512	.687544	.013325	.999822	92	.001259	-.000000
28	6.194115	6.960775	.709750	.014212	.999798	93	.000556	-.000000
29	6.627570	6.874885	.735819	.012221	.999851	94	.000685	-.000000
30	7.149320	6.770941	.767101	.011972	.999887	95	.000759	-.000000
31	7.766615	6.646680	.803480	.011874	.999869	96	.000803	-.000000
32	8.512170	6.497565	.849686	.011965	.999897	97	.000816	-.000000
33	9.406835	6.318630	.903741	.012250	.999850	98	.000861	-.000000
34	10.447415	6.110514	.966613	.014670	.999785	99	.001313	-.000000

282

INTEGRATED VALUES

CY = -.01276 CX = .00617

CL = -.01276 CD = .00617 CM = .10526

PARABOLIC INTEGRATION

INTEGRATED VALUES

CY = -.01276 CX = -.00299

CL = -.01276 CD = -.00299 CM = .05020

DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW

ALPHA = .000000

MOZZLE WITH CENTERBODY

ALPHA 0 = 130.200014

NO. OF BODIES 3

CL = 373.425842

CHORD = 1.300000

TOTAL ELEMENTS 211

BODY 10 = 3

MOZZLE WITH CENTERBO

NO. OF ELEMENTS 112

I	X	Y	S	VT	CP	J	SIGMA	VN
1	21.000000	-38.239845	.015777	-.190811	.063501	100	-.060000	.000000
2	21.000000	-34.719595	.047332	-.083741	.992987	101	-.007361	.000000
3	21.000000	-31.199320	.078886	-.048086	.997832	102	-.010973	.000000
4	21.000000	-27.679045	.110441	-.033454	.998881	103	-.007940	.000000
5	21.000000	-24.158775	.141995	-.027583	.999259	104	-.000298	.000000
6	21.000000	-20.615420	.173756	-.025225	.999364	105	.012857	.000000
7	21.000000	-17.1386185	.205060	-.028280	.999410	106	.028082	.000000
8	21.000000	-14.621820	.227481	-.028753	.999436	107	.043312	.000000
9	21.000000	-12.151515	.247831	-.023314	.999456	108	.058018	.000000
10	21.000000	-10.459595	.268789	-.022833	.999479	109	.071881	.000000
11	21.000000	-8.882995	.278921	-.022276	.999584	110	.084794	.000000
12	21.000000	-7.569165	.290698	-.021620	.999593	111	.096747	.000000
13	21.000000	-6.474305	.300512	-.020932	.999562	112	.107782	.000000
14	21.000000	-5.561920	.308690	-.020241	.999590	113	.117978	.000000
15	21.000000	-4.801600	.315506	-.019518	.999619	114	.127441	.000000
16	21.000000	-4.168000	.321185	-.018791	.999647	115	.136290	.000000
17	21.000000	-3.640000	.325918	-.018215	.999668	116	.144717	.000000
18	21.000000	-3.200000	.329862	-.013807	.999809	117	.153134	.000000
19	20.998134	-2.803443	.333415	-.017136	.999706	118	.160149	.000000
20	20.988434	-2.410980	.336935	-.018087	.999673	119	.163394	.000000
21	20.967873	-2.018591	.340458	-.016381	.999732	120	.165375	.000000
22	20.937036	-1.626742	.343981	-.015873	.999748	121	.166941	.000000
23	20.895940	-1.235698	.347506	-.015393	.999743	122	.168206	.000000
24	20.844532	-.845729	.351031	-.015194	.999749	123	.169135	.000000
25	20.782883	-.452107	.354559	-.014743	.999781	124	.169338	.000000
26	20.711077	-.070098	.358087	-.014555	.999788	125	.170538	.000000
27	20.629100	.315026	.361616	-.014302	.999795	126	.170958	.000000
28	20.537014	.698004	.365147	-.014053	.999803	127	.171235	.000000
29	20.434861	1.078277	.368679	-.013873	.999808	128	.171348	.000000
30	20.322701	1.456475	.372213	-.013648	.999814	129	.171334	.000000
31	20.200608	1.831818	.375748	-.013449	.999819	130	.171181	.000000
32	20.068642	2.203205	.379284	-.013309	.999823	131	.170895	.000000
33	19.926890	2.571512	.382821	-.013169	.999827	132	.170484	.000000
34	19.775439	2.936109	.386360	-.013016	.999831	133	.169955	.000000
35	19.614371	3.296746	.389901	-.012894	.999834	134	.169300	.000000
36	19.443786	3.653162	.393443	-.012775	.999837	135	.168530	.000000
37	19.263727	4.005113	.396986	-.012623	.999841	136	.167652	.000000
38	19.074511	4.352354	.400531	-.012514	.999843	137	.166654	.000000
39	18.876045	4.694640	.404078	-.012373	.999847	138	.165551	.000000
40	18.668518	5.031730	.407626	-.012270	.999849	139	.164329	.000000
41	18.452060	5.363394	.411176	-.012143	.999853	140	.163006	.000000

DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW

NOZZLE WITH CENTERBODY

ALPHA = .000000 ALPHA 0 = 130.200014 NO. OF BODIES 3

CL = 373.425842 CHORD = 1.000000 TOTAL ELEMENTS 211

BODY ID = 3 NOZZLE WITH CENTERBO NO. OF ELEMENTS 112

I	X	Y	S	VT	CP	J	SIGMA	VN
42	18.226807	5.689373	.418728	-.012022	.999855	141	.161575	-.000000
43	17.992898	6.009462	.418282	-.011891	.999859	142	.160036	-.000000
44	17.750473	6.323430	.421838	-.011772	.999861	143	.158392	-.000000
45	17.499686	6.631053	.425395	-.011661	.999864	144	.156645	-.000000
46	17.240695	6.932111	.428955	-.011546	.999867	145	.154800	-.000000
47	16.973655	7.226388	.432517	-.011437	.999869	146	.152847	-.000000
48	16.698725	7.513669	.436082	-.011342	.999871	147	.150792	-.000000
49	16.419785	7.790181	.439602	-.011238	.999874	148	.148677	-.000000
50	16.137308	8.055995	.443079	-.011133	.999876	149	.146493	-.000000
51	15.847870	8.314573	.446558	-.011019	.999879	150	.144215	-.000000
52	15.551661	8.565720	.450039	-.010902	.999881	151	.141845	-.000000
53	15.248876	8.809246	.453522	-.010786	.999884	152	.139382	-.000000
54	14.939215	9.044977	.457007	-.010661	.999886	153	.136837	-.000000
55	14.624385	9.272732	.460494	-.010587	.999888	154	.134192	-.000000
56	14.303094	9.492341	.463983	-.010471	.999890	155	.131473	-.000000
57	13.976052	9.703646	.467473	-.010381	.999892	156	.128659	-.000000
58	13.643479	9.906478	.470965	-.010277	.999894	157	.125760	-.000000
59	13.305597	10.100696	.474458	-.010174	.999897	158	.122781	-.000000
60	12.962623	10.286193	.477953	-.010077	.999899	159	.119712	-.000000
61	12.614799	10.462703	.481450	-.009998	.999900	160	.116558	-.000000
62	12.267355	10.630219	.484948	-.009951	.999901	161	.113323	-.000000
63	11.905525	10.788573	.488447	-.009870	.999903	162	.110009	-.000000
64	11.534550	10.937647	.491948	-.009807	.999904	163	.106609	-.000000
65	11.179669	11.077324	.495450	-.009742	.999905	164	.103126	-.000000
66	10.811134	11.207502	.498954	-.009695	.999906	165	.099561	-.000000
67	10.439187	11.328072	.502459	-.009677	.999906	166	.095904	-.000000
68	10.064085	11.438947	.505965	-.009640	.999907	167	.092178	-.000000
69	9.686084	11.540043	.509472	-.009635	.999907	168	.088356	-.000000
70	9.305431	11.631271	.512981	-.009637	.999907	169	.084445	-.000000
71	8.922394	11.712563	.516491	-.009672	.999906	170	.080447	-.000000
72	8.537231	11.783856	.520002	-.009718	.999906	171	.076357	-.000000
73	8.150198	11.845087	.523515	-.009794	.999904	172	.072162	-.000000
74	7.761560	11.896202	.527028	-.009901	.999902	173	.067860	-.000000
75	7.371582	11.937156	.530543	-.010035	.999899	174	.063438	-.000000
76	6.980534	11.967915	.534059	-.010250	.999895	175	.058875	-.000000
77	6.588685	11.988403	.537577	-.010426	.999891	176	.054131	-.000000
78	6.196292	11.998334	.541195	-.010586	.999888	177	.049295	-.000000
79	5.800000	12.000000	.544847	-.007941	.999837	178	.044447	-.000000
80	5.400000	12.000000	.548433	-.011300	.999872	179	.047478	-.000000
81	5.000000	12.000000	.551818	-.011637	.999865	180	.048765	-.000000
82	4.600000	12.000000	.555404	-.012012	.999856	181	.050177	-.000000

DOUGLAS AIRCRAFT COMPANY TWO-DIMENSIONAL POTENTIAL FLOW PROGRAM

COMBINED FLOW NOZZLE WITH CENTERBODY

ALPHA = .000000 ALPHA 0 = 130.200014 NO. OF BODIES 3

CL = 373.425842 CHORD = 1.000000 TOTAL ELEMENTS 211

BODY ID = 3 NOZZLE WITH CENTERBODY NO. OF ELEMENTS 182

I	X	Y	S	VT	CP	J	SIGMA	VN
81	4.200000	12.000000	.558989	-.012386	.999837	182	-.151668	-.000000
84	3.800000	12.000000	.562575	-.012771	.999837	183	.053211	-.000000
85	3.400000	12.000000	.566160	-.013128	.999826	184	.058790	-.000000
86	3.000000	12.000000	.569766	-.013606	.999815	185	.056398	-.000000
87	2.600000	12.000000	.573331	-.014072	.999802	186	.054027	-.000000
88	2.200000	12.000000	.576916	-.014563	.999788	187	.059674	-.000000
89	1.800000	12.000000	.580502	-.015068	.999773	188	.061317	-.000000
90	1.400000	12.000000	.584087	-.015574	.999757	189	.063010	-.000000
91	1.000000	12.000000	.587673	-.016077	.999742	190	.064624	-.000000
92	.600000	12.000000	.591258	-.016589	.999726	191	.066382	-.000000
93	.200000	12.000000	.594844	-.017089	.999718	192	.068078	-.000000
94	-.200000	12.000000	.598429	-.017585	.999691	193	.069773	-.000000
95	-.600000	12.000000	.602013	-.018117	.999672	194	.071484	-.000000
96	-1.000000	12.000000	.605606	-.018711	.999680	195	.073891	-.000000
97	-1.401600	12.000000	.612785	-.019375	.999625	196	.076584	-.000000
98	-2.561920	12.000000	.619600	-.020074	.999597	197	.079812	-.000000
99	-3.874305	12.000000	.627179	-.020792	.999568	198	.083660	-.000000
100	-4.569165	12.000000	.637593	-.021505	.999588	199	.088241	-.000000
101	-5.482985	12.000000	.648369	-.022155	.999509	200	.093662	-.000000
102	-7.459595	12.000000	.663501	-.022708	.999484	201	.100040	-.000001
103	-9.351515	12.000000	.680460	-.023113	.999466	202	.107473	-.000001
104	-11.621820	12.000000	.700810	-.023388	.999453	203	.116025	-.000000
105	-14.386185	12.000000	.725230	-.023531	.999446	204	.125680	-.000001
106	-17.615420	12.000000	.754534	-.023688	.999440	205	.136250	-.000001
107	-21.532085	12.000000	.788682	-.023803	.999466	206	.147191	-.000001
108	-25.798980	12.000000	.827889	-.023841	.999528	207	.156540	-.000000
109	-30.065875	12.000000	.866138	-.023781	.999627	208	.162615	-.000001
110	-34.332765	12.000000	.904383	-.005798	.999968	209	.164612	-.000000
111	-38.599660	12.000000	.942629	.034838	.998786	210	.161158	-.000000
112	-42.866555	12.000000	.980876	.190811	.963591	211	.148583	-.000000

INTEGRATED VALUES

CY = 65.82139 CX = 51.82014

CL = 65.82139 CD = 51.82014 CM = 63.23408

PARABOLIC INTEGRATION

INTEGRATED VALUES

CY = 65.80866 CX = 51.81179

CL = 65.80866 CD = 51.81179 CM = 62.98027

TOTAL CM = -106.33309

TOTAL CM = -106.62152 (PARABOLIC)

FLOW COMPLETE, T = .000SECONDS.

INDIVIDUAL FLOW NO. 1 OFFBODY POINTS

NOZZLE WITH CENTERBOD

I	X(I)	Y(I)	VX	VY	V7	THETA (DEG)
1	-1.000000	-0.000000	-601310	-0.04806	-601309	-6.7988
2	-1.000000	.989470	.599412	-.316936	.599651	-1.610414
3	-1.000000	1.578950	.585891	-.024451	.594887	-2.560867
4	-1.000000	2.168420	.590728	-.035788	.591811	-3.466922
5	-1.000000	2.757890	.583910	-.040377	.585688	-4.312897
6	-1.000000	3.347370	.575500	-.051032	.577758	-5.067389
7	-1.000000	3.936840	.565491	-.054380	.568890	-5.687488
8	-1.000000	4.526320	.554853	-.059891	.558033	-6.119815
9	-1.000000	5.115790	.543608	-.060166	.548925	-6.313605
10	-1.000000	5.705260	.532791	-.068238	.539664	-6.238114
11	-1.000000	6.294740	.523166	-.058069	.528953	-5.900591
12	-1.000000	6.884210	.515308	-.048296	.517567	-5.354214
13	-1.000000	7.473680	.509828	-.041613	.511130	-4.676583
14	-1.000000	8.063160	.505343	-.034997	.508947	-3.951520
15	-1.000000	8.652630	.502495	-.028850	.508099	-3.239226
16	-1.000000	9.242110	.501075	-.022886	.501579	-2.569507
17	-1.000000	9.831580	.500109	-.017083	.500399	-1.951860
18	-1.000000	10.421050	.499564	-.012063	.499710	-1.383218
19	-1.000000	11.010530	.499272	-.007398	.499127	-.848074
20	-1.000000	11.600000	.499142	-.002960	.499151	-.339725
21	6.000000	.500000	.594952	-.866057	.494184	-1.933076
22	6.000000	.711110	.879860	-.119835	.883029	-7.799611
23	6.000000	1.222220	.821629	-.137951	.833047	-9.497109
24	6.000000	1.733330	.783286	-.138793	.795888	-10.048145
25	6.000000	2.244440	.757098	-.130787	.768305	-9.798062
26	6.000000	2.755560	.741719	-.135819	.750707	-8.875051
27	6.000000	3.266670	.737122	-.094428	.731884	-7.2298514
28	6.000000	3.777780	.744753	-.065180	.749599	-5.001722
29	6.000000	4.288890	.769283	-.029227	.769764	-1.803590
30	6.000000	4.800000	.83865	-.044195	.837028	3.020452
31	6.000000	5.311110	.609818	-.037780	.610805	-1.543588
32	6.000000	5.822220	.558464	-.034815	.558558	-3.599814
33	6.000000	6.333330	.525109	-.034053	.526212	-3.710391
34	6.000000	6.844440	.505513	-.034229	.506671	-3.873681
35	6.000000	7.355560	.490427	-.083889	.491596	-3.952922
36	6.000000	7.866670	.478095	-.032835	.479192	-3.877546
37	6.000000	8.377780	.467605	-.029428	.468530	-3.601031
38	6.000000	8.888890	.458387	-.024734	.459054	-3.088590
39	6.000000	9.400000	.450116	-.017831	.450669	-2.268536
40	6.000000	9.911110	.442567	-.007416	.442829	-.959964
41	12.000000	-5.000000	.086198	-.594523	.602699	-81.777357
42	12.000000	-3.000000	.165414	-.603881	.625895	-74.675541
43	12.000000	-1.000000	.089798	-.554545	.552580	-80.687583
44	14.000000	-3.000000	.141743	-.527937	.546834	-74.971422
45	16.000000	-5.000000	.075043	-.507323	.512423	-81.585509
46	16.000000	-3.000000	.110802	-.480951	.493559	-77.026744
47	18.000000	-5.000000	.050663	-.440561	.448224	-83.981828
48	18.000000	-7.000000	.076025	-.447354	.453778	-80.355344
49	20.000000	-5.000000	.018509	-.448815	.445183	-87.719661
50	20.000000	-3.000000	.033462	-.420734	.422063	-85.452641

INDIVIDUAL FLOW NO. 2 OFFBODY POINTS NOZZLE WITH CENTERBOD

I	X(I)	Y(I)	WX	WY	WT	THETA(DEG)
1	-1.000000	.000000	-.521896	.006028	.521530	179.341982
2	-1.000000	.989870	-.522840	.014773	.525048	178.301575
3	-1.000000	1.578950	-.519777	.021259	.520297	177.437778
4	-1.000000	2.168420	-.515262	.031229	.516208	176.531672
5	-1.000000	2.757890	-.509316	.038421	.510763	175.686031
6	-1.000000	3.347370	-.501982	.045550	.503955	174.928358
7	-1.000000	3.936840	-.493427	.049179	.495872	174.308189
8	-1.000000	4.526320	-.483977	.051081	.486756	173.874363
9	-1.000000	5.115790	-.4738170	.052512	.477069	173.680574
10	-1.000000	5.705260	-.464733	.050446	.467506	173.755941
11	-1.000000	6.294740	-.456136	.047232	.462783	174.020837
12	-1.000000	6.884210	-.449500	.042918	.458178	174.634407
13	-1.000000	7.473680	-.444320	.038009	.454176	175.304680
14	-1.000000	8.063160	-.440714	.032981	.450777	176.026943
15	-1.000000	8.652630	-.438365	.028381	.448077	176.738375
16	-1.000000	9.242110	-.436691	.024157	.446133	177.410727
17	-1.000000	9.831580	-.435627	.020221	.444885	178.026991
18	-1.000000	10.421050	-.435224	.016884	.444354	178.601347
19	-1.000000	11.010530	-.435277	.014332	.444326	179.130301
20	-1.000000	11.600000	-.435147	.012624	.444555	179.654552
21	6.000000	.200000	-.825943	.066669	.827885	178.074995
22	6.000000	.711110	-.763087	.104490	.778208	172.202961
23	6.000000	1.222220	-.716662	.119868	.728617	170.503659
24	6.000000	1.733330	-.688220	.121047	.693860	169.953060
25	6.000000	2.244440	-.660375	.118050	.670151	170.201416
26	6.000000	2.755560	-.644964	.101833	.654805	171.124084
27	6.000000	3.266670	-.632267	.082349	.648219	172.701494
28	6.000000	3.777780	-.624962	.058856	.652111	174.908138
29	6.000000	4.288890	-.621117	.021135	.678450	178.196190
30	6.000000	4.800000	-.729116	-.038470	.730130	-176.979725
31	6.000000	7.200000	-.531613	.833012	.532698	176.844820
32	6.000000	7.711110	-.482786	.030613	.488756	176.371830
33	6.000000	8.222220	-.458012	.030008	.458994	176.251450
34	6.000000	8.733330	-.440875	.030192	.441907	176.082367
35	6.000000	9.244440	-.427684	.029911	.438729	175.999477
36	6.000000	9.755560	-.416898	.028637	.417879	176.074642
37	6.000000	10.266670	-.407697	.026031	.408528	176.386651
38	6.000000	10.777780	-.399611	.021946	.400213	176.856590
39	6.000000	11.288890	-.392329	.018112	.392859	177.689294
40	6.000000	11.800000	-.385984	.007637	.385560	178.065036
41	12.000000	-5.000000	-.075146	.521234	.526633	98.219316
42	12.000000	-3.000000	-.144280	.528443	.548856	105.326459
43	12.000000	-1.000000	-.078218	.475471	.481862	99.381839
44	14.000000	-3.000000	-.123593	.460861	.476663	105.027782
45	16.000000	-5.000000	-.065271	.442350	.447140	98.393661
46	16.000000	-3.000000	-.096498	.419332	.430263	102.960416
47	18.000000	-5.000000	-.049590	.419220	.421493	95.978830
48	18.000000	-3.000000	-.066002	.390237	.395779	99.599730
49	20.000000	-5.000000	-.015949	.405613	.405948	82.257294
50	20.000000	-3.000000	-.028761	.367150	.368275	94.479228

INDIVIDUAL FLOW NO. 3 OFFBODY POINTS NOZZLE WITH CENTER880

I	X(I)	Y(I)	VR	VY	VT	THETA (DEG)
1	-1.000000	.000000	.98821	-.01182	-.98848	-.656057
2	-1.000000	.989470	.991658	-.027962	.992052	-1.615174
3	-1.000000	1.578950	.985814	-.089048	.986798	-2.558395
4	-1.000000	2.168420	.977221	-.059129	.979009	-3.462618
5	-1.000000	2.357890	.965963	-.872195	.968702	-4.309618
6	-1.000000	3.347370	.952037	-.084949	.955766	-5.063083
7	-1.000000	3.836880	.935855	-.081114	.938074	-5.882022
8	-1.000000	4.526320	.917960	-.098811	.923210	-6.112912
9	-1.000000	5.115790	.898842	-.088392	.908317	-6.305815
10	-1.000000	5.705260	.881532	-.096271	.884773	-6.232521
11	-1.000000	6.294780	.865613	-.089318	.870219	-5.897144
12	-1.000000	6.884210	.852615	-.079803	.856351	-5.353881
13	-1.000000	7.473680	.842813	-.068866	.845851	-4.679207
14	-1.000000	8.063160	.836036	-.857818	.880594	-3.957405
15	-1.000000	8.652630	.831632	-.887454	.882968	-3.245196
16	-1.000000	9.242110	.828909	-.037897	.829748	-2.577004
17	-1.000000	9.831580	.827281	-.028296	.827165	-1.959570
18	-1.000000	10.421050	.826351	-.822015	.826593	-1.387518
19	-1.000000	11.010530	.825872	-.812284	.825843	-.852149
20	-1.000000	11.600000	.825688	-.004922	.828663	-.341542
21	6.000000	.200000	1.566519	-.105818	1.570116	-3.870610
22	6.000000	.711110	1.447106	-.197998	1.468589	-7.791037
23	6.000000	1.222220	1.358190	-.227331	1.378070	-9.495114
24	6.000000	1.733330	1.298780	-.229559	1.345937	-10.046361
25	6.000000	2.244440	1.252366	-.216282	1.270931	-9.797132
26	6.000000	2.755560	1.226096	-.381591	1.241765	-8.875556
27	6.000000	3.266670	1.219263	-.356150	1.229222	-7.299084
28	6.000000	3.777780	1.231839	-.107895	1.236347	-5.001517
29	6.000000	4.288890	1.272555	-.840057	1.278185	-1.802949
30	6.000000	4.800000	1.382574	-.072975	1.384498	3.021375
31	6.000000	7.200000	1.008118	-.062581	1.010070	-3.587618
32	6.000000	7.711110	.919494	-.885785	.917316	-3.611641
33	6.000000	8.222220	.868556	-.856562	.878196	-3.725952
34	6.000000	8.733330	.836095	-.056862	.83027	-3.890619
35	6.000000	9.244440	.811089	-.056350	.818844	-3.974203
36	6.000000	9.755560	.790668	-.8053876	.792502	-3.898094
37	6.000000	10.266670	.773272	-.088816	.778820	-3.622583
38	6.000000	10.777780	.757993	-.841185	.759111	-3.110032
39	6.000000	11.288890	.748284	-.829884	.749444	-2.299313
40	6.000000	11.800000	.731566	-.813095	.731683	-1.025461
41	12.000000	-5.000000	.82570	-.986298	.927034	-81.778878
42	12.000000	-3.000000	.27378	-.998889	1.083329	-78.669663
43	14.000000	-5.000000	.188360	-.8901837	.918828	-80.658711
44	14.000000	-3.000000	.234408	-.873216	.908131	-78.973667
45	16.000000	-5.000000	.123858	-.859080	.898153	-81.602928
46	16.000000	-3.000000	.183102	-.795379	.816183	-77.035971
47	18.000000	-5.000000	.083359	-.795800	.799368	-84.007076
48	18.000000	-3.000000	.125428	-.740029	.780583	-80.380304
49	20.000000	-5.000000	.030409	-.769104	.769704	-87.735832
50	20.000000	-3.000000	.054847	-.696110	.698267	-85.494925

INDIVIDUAL FLOW NO. 4 OFFBODY POINTS NOZZLE WITH CENTERBOO

I	K(I)	V(I)	VX	VY	VT	THETA(DEG)
1	-1.000000	.800000	-.821815	.842591	.423562	178.228960
2	-1.000000	.989470	-.418358	.105472	.426601	165.685917
3	-1.000000	1.578950	-.398012	.168818	.432139	157.076693
4	-1.000000	2.169420	-.379530	.210971	.440024	148.338053
5	-1.000000	2.757890	-.381785	.232638	.449938	139.829703
6	-1.000000	3.347370	-.298493	.351903	.461447	130.305466
7	-1.000000	3.936880	-.283552	.406888	.473781	120.938614
8	-1.000000	4.526320	-.176712	.458717	.485983	111.322522
9	-1.000000	5.115790	-.093110	.486871	.498856	101.506285
10	-1.000000	5.705260	-.013803	.505117	.508306	91.565322
11	-1.000000	6.294740	.078324	.505088	.510888	81.628258
12	-1.000000	6.884210	.159639	.486847	.512162	71.839522
13	-1.000000	7.473680	.237209	.452287	.518725	62.325070
14	-1.000000	8.063160	.303948	.405746	.506966	53.162766
15	-1.000000	8.652630	.358689	.351059	.501896	44.388117
16	-1.000000	9.242110	.401692	.281658	.496408	35.982430
17	-1.000000	9.831580	.434091	.229869	.489124	27.913391
18	-1.000000	10.421050	.457223	.167438	.486917	20.113021
19	-1.000000	11.010530	.472278	.108849	.483777	12.517078
20	-1.000000	11.600000	.480159	.042821	.482021	5.037048
21	6.000000	.200000	-.1328713	.072321	1.330733	176.852360
22	6.000000	.711110	-.1257271	.114959	1.262516	174.775648
23	6.000000	1.222220	-.126931	.121355	1.232955	178.333691
24	6.000000	1.733330	-.1225434	.117671	1.231071	174.514776
25	6.000000	2.244440	-.1286323	.118872	1.231061	175.011969
26	6.000000	2.755560	-.1286266	.095720	1.289822	175.784045
27	6.000000	3.266670	-.1388035	.076888	1.362214	176.747105
28	6.000000	3.777780	-.1419854	.048564	1.420617	178.121666
29	6.000000	4.288890	-.1518894	-.000733	1.518894	-179.972390
30	6.000000	4.800000	-.1685270	-.073555	1.688866	-177.507637
31	6.000000	5.311110	-.1881176	-.088883	1.888776	-173.010789
32	6.000000	5.822220	-.211110	-.095364	1.691233	-1.537019
33	6.000000	6.333330	-.222220	-.088888	1.578134	-.614005
34	6.000000	6.844440	-.233330	-.079885	1.479895	-.211231
35	6.000000	7.355560	-.244440	-.068885	1.400901	-.162054
36	6.000000	7.866670	-.255560	-.056666	1.334630	-.329459
37	6.000000	8.377780	-.266670	-.043333	1.279539	-.589745
38	6.000000	8.888890	-.277780	-.029820	1.234662	-.831550
39	6.000000	9.399990	-.288890	-.016666	1.199340	-.921059
40	6.000000	9.911110	-.299990	-.003333	1.172507	-.616500
41	12.000000	-5.000000	-.076351	.014828	.077773	169.023811
42	12.000000	-3.000000	-.158820	.021531	.157296	172.143566
43	14.000000	-5.000000	-.257224	-.033338	.082658	-156.365133
44	14.000000	-3.000000	-.124193	-.059296	.137622	-154.477829
45	16.000000	-5.000000	-.056191	-.063824	.085034	-131.360819
46	16.000000	-3.000000	-.079011	-.066921	.125046	-129.187103
47	18.000000	-5.000000	-.032236	-.087833	.085077	-112.265925
48	18.000000	-3.000000	-.038564	-.108100	.114773	-109.633461
49	20.000000	-5.000000	-.009986	-.083314	.083211	-96.818803
50	20.000000	-3.000000	-.008856	-.105479	.106850	-94.799018

INDIVIDUAL FLOW NO. 5

OFFBODY POINTS

NOZZLE WITH CENTERBOG

I	X(I)	Y(I)	VX	VY	VT	TNETHA(DEG)
1	-1.000000	.400000	-.457787	.005264	.457817	179.381282
2	-1.000000	.989470	-.456343	.012938	.456526	178.379734
3	-1.000000	1.578950	-.453651	.020326	.454104	177.434580
4	-1.000000	2.168420	-.449724	.027291	.450552	176.527300
5	-1.000000	2.757890	-.444536	.033581	.445802	175.680042
6	-1.000000	3.347370	-.438105	.038924	.439831	174.922789
7	-1.000000	3.936840	-.430639	.042978	.432778	174.300201
8	-1.000000	4.526320	-.422373	.045596	.424806	173.865511
9	-1.000000	5.115790	-.413811	.046810	.416350	173.669214
10	-1.000000	5.705260	-.405581	.046474	.408013	173.742231
11	-1.000000	6.294740	-.398232	.044328	.400370	174.075697
12	-1.000000	6.884210	-.392227	.040682	.393964	174.617933
13	-1.000000	7.473680	-.387682	.036176	.388890	175.299618
14	-1.000000	8.063160	-.384550	.031871	.385475	176.029579
15	-1.000000	8.652630	-.382523	.027288	.383181	176.745979
16	-1.000000	9.242110	-.381309	.021758	.381695	177.423586
17	-1.000000	9.831580	-.380555	.015813	.380777	178.081491
18	-1.000000	10.421050	-.380178	.009199	.380289	178.613853
19	-1.000000	11.010530	-.379991	.005665	.380033	179.145821
20	-1.000000	11.600000	-.379928	.002272	.379935	179.657417
21	6.000000	.200000	-.720961	.049528	.722658	176.071714
22	6.000000	.711110	-.666093	.091232	.672311	172.200932
23	6.000000	1.222220	-.625565	.104644	.634258	170.503332
24	6.000000	1.733330	-.596373	.105669	.605662	169.952223
25	6.000000	2.244440	-.576419	.099582	.588950	170.202192
26	6.000000	2.755560	-.564710	.088176	.571553	171.125246
27	6.000000	3.266670	-.561229	.071881	.565813	172.701366
28	6.000000	3.777780	-.567041	.049627	.569808	174.998247
29	6.000000	4.288890	-.585795	.018495	.586086	178.196472
30	6.000000	4.800000	-.636415	-.033583	.637300	-176.979362
31	6.000000	5.200000	-.663865	.028786	.664755	176.453913
32	6.000000	5.711110	-.621190	.026528	.622024	176.396105
33	6.000000	6.222220	-.592577	.025951	.608919	176.204027
34	6.000000	6.733330	-.584656	.026079	.585539	176.121418
35	6.000000	7.244440	-.573191	.025805	.574082	176.044495
36	6.000000	7.755560	-.563845	.024649	.564679	176.124414
37	6.000000	8.266670	-.555891	.022824	.556890	176.410637
38	6.000000	8.777780	-.548959	.018680	.549459	176.935886
39	6.000000	9.288890	-.542705	.012302	.542961	177.777174
40	6.000000	9.800000	-.537080	.004657	.537112	179.208387
41	12.000000	-5.000000	-.065668	.454065	.458789	98.229232
42	12.000000	-3.000000	-.125993	.459538	.476497	105.332175
43	14.000000	-5.000000	-.068432	.449286	.470590	99.363956
44	14.000000	-3.000000	-.108002	.401871	.436130	105.042644
45	16.000000	-5.000000	-.057212	.386081	.390277	98.429567
46	16.000000	-3.000000	-.084483	.366014	.375638	102.997381
47	18.000000	-5.000000	-.038505	.365715	.367737	96.010348
48	18.000000	-3.000000	-.057760	.340442	.345308	99.629173
49	20.000000	-5.000000	-.014050	.353910	.354189	92.273434
50	20.000000	-3.000000	-.025137	.320825	.321310	94.486993

COMBINED FLOW FOR OFFBODY POINTS

NOZZLE WITH CENTERBODY

I	X(I)	Y(I)	VX	VY	VT	TWETA(DEG)
1	-1.000000	.800000	.029678	-.000620	.029684	-1.197022
2	-1.000000	.989870	.029537	-.001525	.029576	-2.955497
3	-1.000000	1.578950	.029282	-.002399	.029380	-8.883685
4	-1.000000	2.168420	.028883	-.003264	.029067	-6.846681
5	-1.000000	2.757890	.028334	-.004073	.028625	-8.181103
6	-1.000000	3.347370	.027670	-.004838	.028084	-9.857625
7	-1.000000	3.936840	.026820	-.005538	.027366	-11.860879
8	-1.000000	4.526320	.025855	-.006205	.026520	-12.865062
9	-1.000000	5.115790	.024720	-.006873	.025527	-13.893022
10	-1.000000	5.705260	.023657	-.007524	.024457	-14.694319
11	-1.000000	6.294740	.022598	-.008159	.023378	-15.887136
12	-1.000000	6.884210	.021639	-.008771	.022345	-18.437365
13	-1.000000	7.473680	.020875	-.009350	.021477	-13.599438
14	-1.000000	8.063160	.020223	-.009931	.020702	-12.358459
15	-1.000000	8.652630	.019717	-.010515	.020073	-10.811596
16	-1.000000	9.242110	.019315	-.011103	.019563	-9.131309
17	-1.000000	9.831580	.018954	-.011698	.019203	-7.203524
18	-1.000000	10.421050	.018628	-.012301	.018909	-5.313341
19	-1.000000	11.010530	.018370	-.012914	.018700	-3.272056
20	-1.000000	11.600000	.018174	-.013539	.018579	-1.306921
21	-1.000000	12.189470	.017956	-.014175	.018519	-4.091191
22	-1.000000	12.778940	.017710	-.014820	.018533	-7.347325
23	-1.000000	13.368410	.017440	-.015475	.018611	-8.749097
24	-1.000000	13.957880	.017150	-.016135	.018759	-9.119477
25	-1.000000	14.547350	.016840	-.016800	.018981	-8.772961
26	-1.000000	15.136820	.016510	-.017465	.019277	-6.355803
27	-1.000000	15.726290	.016170	-.018130	.019637	-4.246782
28	-1.000000	16.315760	.015820	-.018795	.020061	-1.351381
29	-1.000000	16.905230	.015470	-.019460	.020554	2.080611
30	-1.000000	17.494700	.015120	-.020125	.021114	-3.982323
31	-1.000000	18.084170	.014770	-.020790	.021741	-5.445735
32	-1.000000	18.673640	.014420	-.021455	.022430	-6.523805
33	-1.000000	19.263110	.014070	-.022120	.023181	-6.987826
34	-1.000000	19.852580	.013720	-.022785	.023995	-7.054152
35	-1.000000	20.442050	.013370	-.023450	.024872	-6.782645
36	-1.000000	21.031520	.013020	-.024115	.025808	-5.419569
37	-1.000000	21.620990	.012670	-.024780	.026803	-4.351880
38	-1.000000	22.210460	.012320	-.025445	.027857	-5.411529
39	-1.000000	22.800000	.011970	-.026110	.028970	-80.711228
40	-1.000000	23.389470	.011620	-.026775	.029146	-72.645399
41	-1.000000	23.978940	.011270	-.027440	.029372	-78.371725
42	-1.000000	24.568410	.010920	-.028105	.029649	-72.790654
43	-1.000000	25.157880	.010570	-.028770	.029976	-80.497566
44	-1.000000	25.747350	.010220	-.029435	.030353	-75.404162
45	-1.000000	26.336820	.009870	-.030100	.030780	-82.890094
46	-1.000000	26.926290	.009520	-.030765	.031257	-78.730321
47	-1.000000	27.515760	.009170	-.031430	.031784	-87.226678
48	-1.000000	28.105230	.008820	-.032095	.032361	-83.789759
49	-1.000000	28.694700	.008470	-.032760	.032988	
50	-1.000000	29.284170	.008120	-.033425	.033665	

OFFBODY POINTS COMPLETE, I = .000 SECONDS.

NOZZLE WITH HUB

2-D COMBINATION SOLUTION

COMPRESSIBLE VERSION

COMBINATION OF THE FOLLOWING BASIC SOLUTIONS

1. UNIFORM AXIAL
2. UNIFORM CROSSFLOW
3. VORTICITY ABOUT BODY 1
4. VORTICITY ABOUT BODY 2
5. VORTICITY ABOUT BODY 3

	VELOCITY	MACH NO.	DYNAMIC PRESSURE INC	COMP	PRESSURE RATIO INC	COMP	DENSITY RATIO
CONTROL	3.564+02	3.226-01	1.510+02	1.434+02	9.287-01	9.304-01	9.498-01
LOWER PASSAGE	4.395+02	4.000-01	2.296+02	2.123+02	8.915-01	8.956-01	9.243-01
UPPER PASSAGE	4.395+02	4.000-01	2.296+02	2.123+02	8.915-01	8.956-01	9.243-01
FREE STREAM	0.000	0.000	0.000	0.000	1.000+00	1.000+00	1.000+00

293

ALPHA	VINF/VC	VSONIC	VSONICC	WDOTCR	WDOTLCR	WDOTUCR
0.000	0.000	6.459+02	1.019+03	2.589+01	3.294+01	1.295+01

TSTAT	PSTAT	PSTATC	ASTAT	RHOSTAT	WDOTC	WDOTL	WDOTU
5.187+02	2.116+03	2.116+03	1.116+03	2.378-03	2.589+01	3.294+01	1.295+01

VIC	VICL	VICU
3.385+02	4.062+02	4.062+02

TTOT	PTOT	PTOTC	ATOT	RHOTO1	THE1	DEL
5.187+02	2.116+03	2.116+03	1.116+03	2.378-03	1.000+00	1.000+00

YR11	YR12	YR17	YTEST	YCL	YCU	LND
0.000	0.000	0.000	1.200+01	-1.000+00	0.000	4.200+01

YCL1	YCU1	XTEST2	YCL2	YCU2
1.000-03	5.000+00	6.000+00	7.000+00	1.200+01

P CUTOFF 1	P-S CUTOFF 2	P-S CUTOFF 3
0.000+01	0.000	4.000+01

BODY 1
ON-BODY POINTS

I	X	Y	S	V	MACH	CP	P6/PI
1	-4.287+01	0.000	-4.287+01	3.589+02	3.249-01	9.999+03	.9295
2	-3.860+01	0.000	-3.860+01	3.528+02	3.192-01	9.999+03	.9318
3	-3.433+01	0.000	-3.433+01	3.530+02	3.194-01	9.999+03	.9317
4	-3.007+01	0.000	-3.007+01	3.534+02	3.198-01	9.999+03	.9316
5	-2.580+01	0.000	-2.580+01	3.536+02	3.200-01	9.999+03	.9315
6	-2.153+01	0.000	-2.153+01	3.536+02	3.200-01	9.999+03	.9315
7	-1.762+01	0.000	-1.762+01	3.538+02	3.200-01	9.999+03	.9315
8	-1.435+01	0.000	-1.435+01	3.536+02	3.200-01	9.999+03	.9315
9	-1.162+01	0.000	-1.162+01	3.537+02	3.201-01	9.999+03	.9315
10	-9.352+00	0.000	-9.352+00	3.536+02	3.202-01	9.999+03	.9314
11	-7.460+00	0.000	-7.460+00	3.541+02	3.205-01	9.999+03	.9313
12	-5.883+00	0.000	-5.883+00	3.545+02	3.209-01	9.999+03	.9311
13	-4.569+00	0.000	-4.569+00	3.552+02	3.215-01	9.999+03	.9309
14	-3.474+00	0.000	-3.474+00	3.563+02	3.225-01	9.999+03	.9305
15	-2.562+00	0.000	-2.562+00	3.577+02	3.238-01	9.999+03	.9299
16	-1.802+00	0.000	-1.802+00	3.594+02	3.254-01	9.999+03	.9293
17	-1.168+00	0.000	-1.168+00	3.614+02	3.273-01	9.999+03	.9285
18	-.640-01	0.000	-.640-01	3.636+02	3.293-01	9.999+03	.9277
19	-2.000-01	0.000	-2.000-01	3.658+02	3.313-01	9.999+03	.9268
20	2.000-01	0.000	2.000-01	3.682+02	3.335-01	9.999+03	.9259
21	6.000-01	0.000	6.000-01	3.710+02	3.361-01	9.999+03	.9248
22	1.000+00	0.000	1.000+00	3.743+02	3.391-01	9.999+03	.9235
23	1.400+00	0.000	1.400+00	3.781+02	3.427-01	9.999+03	.9220
24	1.800+00	0.000	1.800+00	3.824+02	3.467-01	9.999+03	.9202
25	2.200+00	0.000	2.200+00	3.874+02	3.514-01	9.999+03	.9182
26	2.600+00	0.000	2.600+00	3.932+02	3.567-01	9.999+03	.9158
27	3.000+00	0.000	3.000+00	3.997+02	3.628-01	9.999+03	.9131
28	3.400+00	0.000	3.400+00	4.072+02	3.697-01	9.999+03	.9099
29	3.800+00	0.000	3.800+00	4.159+02	3.779-01	9.999+03	.9062
30	4.200+00	0.000	4.200+00	4.262+02	3.875-01	9.999+03	.9016
31	4.600+00	0.000	4.600+00	4.389+02	3.994-01	9.999+03	.8959
32	5.000+00	0.000	5.000+00	4.553+02	4.149-01	9.999+03	.8883
33	5.400+00	0.000	5.400+00	4.744+02	4.367-01	9.999+03	.8772
34	5.800+00	0.000	5.800+00	5.185+02	4.749-01	9.999+03	.8569
35	6.143+00	-7.327-03	6.143+00	5.889+02	5.429-01	9.999+03	.8183
36	6.548+00	-5.057-02	6.551+00	6.423+02	5.955-01	9.999+03	.7866
37	6.908+00	-1.408-01	6.922+00	6.723+02	6.255-01	9.999+03	.7682
38	7.252+00	-2.739-01	7.291+00	6.942+02	6.475-01	9.999+03	.7544
39	7.575+00	-4.464-01	7.656+00	7.103+02	6.639-01	9.999+03	.7441
40	7.873+00	-6.566-01	8.022+00	7.221+02	6.758-01	9.999+03	.7364
41	8.144+00	-9.012-01	8.386+00	7.295+02	6.834-01	9.999+03	.7316
42	8.377+00	-1.170+00	8.743+00	7.320+02	6.859-01	9.999+03	.7300
43	8.575+00	-1.460+00	9.093+00	7.288+02	6.827-01	9.999+03	.7321
44	8.738+00	-1.774+00	9.447+00	7.186+02	6.723-01	9.999+03	.7387
45	8.864+00	-2.108+00	9.804+00	6.996+02	6.529-01	9.999+03	.7518
46	8.951+00	-2.457+00	1.016+01	6.692+02	6.224-01	9.999+03	.7702
47	8.993+00	-2.818+00	1.053+01	6.130+02	5.665-01	9.999+03	.8043
48	9.000+00	-3.195+00	1.090+01	5.391+01	4.947-01	9.999+03	.8460
49	9.000+00	-3.623+00	1.113+01	4.921+01	4.497-01	9.999+03	.8704
50	9.000+00	-4.137+00	1.185+01	4.612+02	4.204-01	9.999+03	.8855

I	X	Y	S	V	MACH	CP	PS/PT
51	9.000+00	-4.754+00	1.246+01	4.369+02	3.976-01	9.999+03	.8968
52	9.000+00	-5.494+00	1.320+01	4.172+02	3.791-01	9.999+03	.9056
53	9.000+00	-6.382+00	1.409+01	4.010+02	3.640-01	9.999+03	.9125
54	9.000+00	-7.448+00	1.516+01	3.878+02	3.517-01	9.999+03	.9180
55	9.000+00	-8.727+00	1.644+01	3.772+02	3.419-01	9.999+03	.9223
56	9.000+00	-1.026+01	1.797+01	3.691+02	3.344-01	9.999+03	.9255
57	9.000+00	-1.210+01	1.981+01	3.631+02	3.288-01	9.999+03	.9278
58	9.000+00	-1.431+01	2.202+01	3.591+02	3.251-01	9.999+03	.9294
59	9.000+00	-1.697+01	2.468+01	3.565+02	3.227-01	9.999+03	.9304
60	9.000+00	-2.015+01	2.786+01	3.551+02	3.214-01	9.999+03	.9309
61	9.000+00	-2.370+01	3.141+01	3.545+02	3.208-01	9.999+03	.9312
62	9.000+00	-2.732+01	3.503+01	3.542+02	3.205-01	9.999+03	.9313
63	9.000+00	-3.094+01	3.865+01	3.541+02	3.205-01	9.999+03	.9313
64	9.000+00	-3.457+01	4.227+01	3.559+02	3.221-01	9.999+03	.9306
65	9.000+00	-3.819+01	4.590+01	3.707+02	3.359-01	9.999+03	.9249

BODY 2
ON-BODY POINTS

I	X	Y	S	V	MACH	CP	PS/PT
66	1.045+01	5.889+00	-7.869+00	2.745+02	2.474-01	9.999+03	.9583
67	9.407+00	5.681+00	-6.808+00	-8.511+01	7.630-02	9.999+03	.9959
68	8.512+00	5.502+00	-5.896+00	-1.657+02	1.488-01	9.999+03	.9847
69	7.767+00	5.353+00	-5.136+00	-2.342+02	2.107-01	9.999+03	.9695
70	7.145+00	5.229+00	-4.502+00	-2.903+02	2.619-01	9.999+03	.9534
71	6.628+00	5.126+00	-3.974+00	-3.418+02	3.091-01	9.999+03	.9359
72	6.196+00	5.039+00	-3.534+00	-4.127+02	3.749-01	9.999+03	.9075
73	5.825+00	4.999+00	-3.161+00	-4.841+02	4.421-01	9.999+03	.8744
74	5.476+00	5.015+00	-2.811+00	-4.522+02	4.119-01	9.999+03	.8898
75	5.128+00	5.043+00	-2.462+00	-4.601+02	4.194-01	9.999+03	.8860
76	4.781+00	5.086+00	-2.113+00	-4.644+02	4.235-01	9.999+03	.8840
77	4.438+00	5.146+00	-1.754+00	-4.660+02	4.250-01	9.999+03	.8852
78	4.099+00	5.226+00	-1.416+00	-4.633+02	4.224-01	9.999+03	.8845
79	3.769+00	5.331+00	-1.069+00	-4.534+02	4.131-01	9.999+03	.8892
80	3.458+00	5.468+00	-7.302-01	-4.283+02	3.895-01	9.999+03	.9007
81	3.194+00	5.645+00	-4.122-01	-3.592+02	3.252-01	9.999+03	.9294
82	3.028+00	5.872+00	-1.310-01	-1.537+02	1.380-01	9.999+03	.9868
83	3.028+00	6.128+00	1.310-01	1.696+02	1.523-01	9.999+03	.9839
84	3.194+00	6.355+00	4.122-01	3.766+02	3.413-01	9.999+03	.9225
85	3.458+00	6.532+00	7.302-01	4.467+02	4.068-01	9.999+03	.8923
86	3.769+00	6.669+00	1.069+00	4.768+02	4.352-01	9.999+03	.8780
87	4.099+00	6.774+00	1.416+00	4.934+02	4.510-01	9.999+03	.8697
88	4.438+00	6.854+00	1.764+00	5.047+02	4.617-01	9.999+03	.8641
89	4.781+00	6.914+00	2.113+00	5.139+02	4.705-01	9.999+03	.8593
90	5.128+00	6.957+00	2.462+00	5.231+02	4.793-01	9.999+03	.8545
91	5.476+00	6.985+00	2.811+00	5.367+02	4.923-01	9.999+03	.8473
92	5.825+00	7.001+00	3.161+00	5.608+02	5.157-01	9.999+03	.8341
93	6.196+00	6.961+00	3.534+00	5.715+02	5.260-01	9.999+03	.8282
94	6.628+00	6.874+00	3.974+00	4.958+02	4.532-01	9.999+03	.8686
95	7.145+00	6.771+00	4.502+00	4.787+02	4.370-01	9.999+03	.8770
96	7.767+00	6.647+00	5.136+00	4.729+02	4.315-01	9.999+03	.8799
97	8.512+00	6.498+00	5.896+00	4.795+02	4.377-01	9.999+03	.8767
98	9.407+00	6.319+00	6.808+00	4.935+02	4.511-01	9.999+03	.8697
99	1.045+01	6.111+00	7.869+00	8.110+02	7.683-01	9.999+03	.6767

BODY 3
ON-BODY POINTS

I	X	Y	S	V	MACH	CP	PS/PT
100	2.100+01	-3.824+01	6.480+01	-3.800+02	3.445-01	9.999+03	.9212
101	2.100+01	-3.472+01	6.128+01	-3.500+02	3.250-01	9.999+03	.9294
102	2.100+01	-3.120+01	5.776+01	-3.552+02	3.215-01	9.999+03	.9309
103	2.100+01	-2.768+01	5.424+01	-3.543+02	3.207-01	9.999+03	.9312
104	2.100+01	-2.416+01	5.072+01	-3.539+02	3.203-01	9.999+03	.9314
105	2.100+01	-2.062+01	4.718+01	-3.532+02	3.197-01	9.999+03	.9316
106	2.100+01	-1.735+01	4.391+01	-3.519+02	3.185-01	9.999+03	.9321
107	2.100+01	-1.462+01	4.118+01	-3.497+02	3.164-01	9.999+03	.9330
108	2.100+01	-1.235+01	3.891+01	-3.461+02	3.131-01	9.999+03	.9343
109	2.100+01	-1.046+01	3.702+01	-3.412+02	3.086-01	9.999+03	.9361
110	2.100+01	-.883+00	3.544+01	-3.350+02	3.029-01	9.999+03	.9384
111	2.100+01	-.756+00	3.413+01	-3.277+02	2.962-01	9.999+03	.9409
112	2.100+01	-.647+00	3.304+01	-3.197+02	2.888-01	9.999+03	.9437
113	2.100+01	-.556+00	3.212+01	-3.113+02	2.811-01	9.999+03	.9466
114	2.100+01	-.480+00	3.136+01	-3.027+02	2.732-01	9.999+03	.9495
115	2.100+01	-.416+00	3.073+01	-2.940+02	2.652-01	9.999+03	.9523
116	2.100+01	-.364+00	3.020+01	-2.853+02	2.573-01	9.999+03	.9550
117	2.100+01	-.320+00	2.976+01	-2.763+02	2.490-01	9.999+03	.9578
118	2.100+01	-.280+00	2.936+01	-2.648+02	2.386-01	9.999+03	.9611
119	2.099+01	-.241+00	2.897+01	-2.566+02	2.311-01	9.999+03	.9635
120	2.097+01	-.201+00	2.858+01	-2.508+02	2.259-01	9.999+03	.9651
121	2.094+01	-.167+00	2.819+01	-2.462+02	2.217-01	9.999+03	.9664
122	2.090+01	-.123+00	2.779+01	-2.423+02	2.181-01	9.999+03	.9674
123	2.084+01	-.085+00	2.740+01	-2.391+02	2.152-01	9.999+03	.9682
124	2.078+01	-.057+00	2.701+01	-2.363+02	2.127-01	9.999+03	.9690
125	2.071+01	-.010+00	2.661+01	-2.339+02	2.105-01	9.999+03	.9696
126	2.063+01	3.150-01	2.622+01	-2.320+02	2.087-01	9.999+03	.9701
127	2.054+01	6.980-01	2.582+01	-2.303+02	2.072-01	9.999+03	.9705
128	2.043+01	1.079+00	2.543+01	-2.289+02	2.059-01	9.999+03	.9709
129	2.032+01	1.456+00	2.504+01	-2.278+02	2.049-01	9.999+03	.9711
130	2.020+01	1.831+00	2.464+01	-2.270+02	2.042-01	9.999+03	.9714
131	2.007+01	2.203+00	2.425+01	-2.264+02	2.037-01	9.999+03	.9715
132	1.993+01	2.572+00	2.385+01	-2.261+02	2.034-01	9.999+03	.9716
133	1.978+01	2.936+00	2.346+01	-2.260+02	2.033-01	9.999+03	.9716
134	1.961+01	3.297+00	2.306+01	-2.261+02	2.034-01	9.999+03	.9716
135	1.944+01	3.653+00	2.267+01	-2.265+02	2.038-01	9.999+03	.9715
136	1.926+01	4.005+00	2.227+01	-2.272+02	2.044-01	9.999+03	.9713
137	1.907+01	4.352+00	2.188+01	-2.281+02	2.052-01	9.999+03	.9711
138	1.888+01	4.695+00	2.148+01	-2.292+02	2.063-01	9.999+03	.9708
139	1.867+01	5.032+00	2.109+01	-2.307+02	2.076-01	9.999+03	.9704
140	1.845+01	5.363+00	2.069+01	-2.324+02	2.091-01	9.999+03	.9700
141	1.823+01	5.689+00	2.029+01	-2.344+02	2.110-01	9.999+03	.9695
142	1.799+01	6.009+00	1.990+01	-2.368+02	2.131-01	9.999+03	.9689
143	1.775+01	6.323+00	1.950+01	-2.394+02	2.155-01	9.999+03	.9682
144	1.750+01	6.631+00	1.910+01	-2.424+02	2.182-01	9.999+03	.9674
145	1.724+01	6.932+00	1.871+01	-2.458+02	2.213-01	9.999+03	.9665
146	1.697+01	7.226+00	1.831+01	-2.495+02	2.246-01	9.999+03	.9655
147	1.670+01	7.514+00	1.791+01	-2.536+02	2.284-01	9.999+03	.9643
148	1.642+01	7.790+00	1.752+01	-2.579+02	2.323-01	9.999+03	.9631
149	1.614+01	8.056+00	1.713+01	-2.626+02	2.365-01	9.999+03	.9618

150	1.585+01	8.315+00	1.674+01	-2.675+02	2.411-01	9.999+03	.9604
151	1.555+01	8.566+00	1.635+01	-2.127+02	2.458-01	9.999+03	.958P
152	1.525+01	8.809+00	1.597+01	-2.782+02	2.508-01	9.999+03	.9572
153	1.494+01	9.045+00	1.558+01	-2.838+02	2.559-01	9.999+03	.9555
154	1.462+01	9.273+00	1.519+01	-2.895+02	2.612-01	9.999+03	.9537
155	1.430+01	9.492+00	1.480+01	-2.952+02	2.664-01	9.999+03	.9519
156	1.398+01	9.704+00	1.441+01	-3.008+02	2.715-01	9.999+03	.9501
157	1.364+01	9.906+00	1.402+01	-3.061+02	2.768-01	9.999+03	.9483
158	1.331+01	1.010+01	1.363+01	-3.112+02	2.810-01	9.999+03	.9464
159	1.296+01	1.029+01	1.324+01	-3.159+02	2.853-01	9.999+03	.9451
160	1.261+01	1.046+01	1.285+01	-3.201+02	2.891-01	9.999+03	.9436
161	1.226+01	1.063+01	1.246+01	-3.238+02	2.926-01	9.999+03	.9423
162	1.191+01	1.079+01	1.207+01	-3.271+02	2.956-01	9.999+03	.9412
163	1.154+01	1.094+01	1.168+01	-3.299+02	2.981-01	9.999+03	.9402
164	1.118+01	1.108+01	1.129+01	-3.322+02	3.003-01	9.999+03	.9393
165	1.081+01	1.121+01	1.090+01	-3.343+02	3.022-01	9.999+03	.9386
166	1.044+01	1.133+01	1.051+01	-3.360+02	3.030-01	9.999+03	.9380
167	1.006+01	1.144+01	1.012+01	-3.376+02	3.033-01	9.999+03	.9374
168	9.686+00	1.154+01	9.724+00	-3.390+02	3.046-01	9.999+03	.9369
169	9.305+00	1.163+01	9.333+00	-3.403+02	3.079-01	9.999+03	.9364
170	8.922+00	1.171+01	8.981+00	-3.419+02	3.093-01	9.999+03	.9358
171	8.537+00	1.178+01	8.599+00	-3.435+02	3.107-01	9.999+03	.9353
172	8.150+00	1.185+01	8.158+00	-3.453+02	3.124-01	9.999+03	.9346
173	7.762+00	1.190+01	7.766+00	-3.474+02	3.143-01	9.999+03	.9338
174	7.372+00	1.194+01	7.371+00	-3.499+02	3.168-01	9.999+03	.9329
175	6.981+00	1.197+01	6.981+00	-3.530+02	3.195-01	9.999+03	.9317
176	6.589+00	1.199+01	6.589+00	-3.571+02	3.233-01	9.999+03	.9302
177	6.196+00	1.200+01	6.196+00	-3.645+02	3.301-01	9.999+03	.9273
178	5.800+00	1.200+01	5.800+00	-3.753+02	3.401-01	9.999+03	.9231
179	5.400+00	1.200+01	5.400+00	-3.810+02	3.458-01	9.999+03	.9208
180	5.000+00	1.200+01	5.000+00	-3.841+02	3.482-01	9.999+03	.9195
181	4.600+00	1.200+01	4.600+00	-3.853+02	3.494-01	9.999+03	.9190
182	4.200+00	1.200+01	4.200+00	-3.852+02	3.493-01	9.999+03	.9191
183	3.800+00	1.200+01	3.800+00	-3.822+02	3.488-01	9.999+03	.9195
184	3.400+00	1.200+01	3.400+00	-3.824+02	3.467-01	9.999+03	.9202
185	3.000+00	1.200+01	3.000+00	-3.802+02	3.447-01	9.999+03	.9211
186	2.600+00	1.200+01	2.600+00	-3.777+02	3.423-01	9.999+03	.9221
187	2.200+00	1.200+01	2.200+00	-3.751+02	3.399-01	9.999+03	.9232
188	1.800+00	1.200+01	1.800+00	-3.724+02	3.374-01	9.999+03	.9242
189	1.400+00	1.200+01	1.400+00	-3.699+02	3.351-01	9.999+03	.9252
190	1.000+00	1.200+01	1.000+00	-3.676+02	3.329-01	9.999+03	.9261
191	6.000-01	1.200+01	6.000-01	-3.654+02	3.310-01	9.999+03	.9270
192	2.000-01	1.200+01	2.000-01	-3.635+02	3.292-01	9.999+03	.9277
193	-2.000-01	1.200+01	-2.000-01	-3.619+02	3.277-01	9.999+03	.9283
194	-6.400-01	1.200+01	-6.400-01	-3.603+02	3.262-01	9.999+03	.9289
195	-1.168+00	1.200+01	-1.168+00	-3.587+02	3.248-01	9.999+03	.9295
196	-1.802+00	1.200+01	-1.802+00	-3.573+02	3.235-01	9.999+03	.9301
197	-2.562+00	1.200+01	-2.562+00	-3.561+02	3.223-01	9.999+03	.9306
198	-3.474+00	1.200+01	-3.474+00	-3.551+02	3.214-01	9.999+03	.9309
199	-4.569+00	1.200+01	-4.569+00	-3.544+02	3.207-01	9.999+03	.9312
200	-5.883+00	1.200+01	-5.883+00	-3.539+02	3.203-01	9.999+03	.9314
201	-7.460+00	1.200+01	-7.460+00	-3.537+02	3.201-01	9.999+03	.9315
202	-9.352+00	1.200+01	-9.352+00	-3.536+02	3.200-01	9.999+03	.9315
203	-1.162+01	1.200+01	-1.162+01	-3.536+02	3.200-01	9.999+03	.9315
204	-4.435+01	1.200+01	-4.435+01	-3.536+02	3.200-01	9.999+03	.9315
205	-1.762+01	1.200+01	-1.762+01	-3.536+02	3.200-01	9.999+03	.9315
206	-2.153+01	1.200+01	-2.153+01	-3.537+02	3.201-01	9.999+03	.9315
207	-2.580+01	1.200+01	-2.580+01	-3.538+02	3.202-01	9.999+03	.9314
208	-3.007+01	1.200+01	-3.007+01	-3.540+02	3.204-01	9.999+03	.9313
209	-3.433+01	1.200+01	-3.433+01	-3.549+02	3.212-01	9.999+03	.9310
210	-3.860+01	1.200+01	-3.860+01	-3.588+02	3.248-01	9.999+03	.9295
211	-4.287+01	1.200+01	-4.287+01	-3.795+02	3.440-01	9.999+03	.9214

RAKE NUMBER 1

I	X	Y	VX	VY	VRE	THE TA	MACH	PS/PT	WFRACT
1	-1.000+00	0.000	3.621+02	0.000	3.621+02	0.000	3.279-01	.9283	0.000
2	-1.000+00	4.000-01	3.619+02	-1.473+00	3.619+02	-2.332-01	3.277-01	.9283	3.404-02
3	-1.000+00	9.895-01	3.612+02	-3.586+00	3.612+02	-5.624-01	3.271-01	.9286	8.415-02
4	-1.000+00	1.579+00	3.600+02	-6.428+00	3.600+02	-8.640-01	3.259-01	.9291	1.341-01
5	-1.000+00	2.168+00	3.582+02	-6.931+00	3.582+02	-1.109+00	3.243-01	.9297	1.839-01
6	-1.000+00	2.758+00	3.559+02	-7.937+00	3.560+02	-1.277+00	3.223-01	.9306	2.335-01
7	-1.000+00	3.347+00	3.534+02	-8.291+00	3.535+02	-1.344+00	3.199-01	.9315	2.827-01
8	-1.000+00	3.937+00	3.508+02	-7.883+00	3.509+02	-1.287+00	3.175-01	.9325	3.317-01
9	-1.000+00	4.526+00	3.484+02	-6.653+00	3.484+02	-1.094+00	3.153-01	.9334	3.803-01
10	-1.000+00	5.116+00	3.465+02	-4.664+00	3.465+02	-7.712-01	3.135-01	.9341	4.286-01
11	-1.000+00	5.705+00	3.454+02	-2.146+00	3.454+02	-3.559-01	3.125-01	.9345	4.768-01
12	-1.000+00	6.295+00	3.454+02	5.416-01	3.454+02	8.983-02	3.125-01	.9346	5.249-01
13	-1.000+00	6.884+00	3.464+02	2.977+00	3.464+02	4.924-01	3.134-01	.9342	5.730-01
14	-1.000+00	7.474+00	3.481+02	4.855+00	3.481+02	7.991-01	3.150-01	.9335	6.214-01
15	-1.000+00	8.063+00	3.502+02	5.961+00	3.503+02	9.751-01	3.170-01	.9327	6.699-01
16	-1.000+00	8.653+00	3.525+02	6.301+00	3.525+02	1.074+00	3.191-01	.9319	7.188-01
17	-1.000+00	9.242+00	3.546+02	5.987+00	3.546+02	9.674-01	3.210-01	.9311	7.679-01
18	-1.000+00	9.832+00	3.563+02	5.163+00	3.563+02	8.302-01	3.226-01	.9305	8.172-01
19	-1.000+00	1.042+01	3.577+02	3.992+00	3.577+02	6.395-01	3.238-01	.9299	8.667-01
20	-1.000+00	1.101+01	3.586+02	2.592+00	3.586+02	4.142-01	3.246-01	.9296	9.164-01
	-1.000+00	1.163+01	3.591+02	1.060+00	3.591+02	1.692-01	3.251-01	.9294	9.662-01
		1.200+01	3.592+02	0.000	3.592+02	0.000	3.252-01	.9294	1.000+00

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RAKE NUMBER 2

I	X	Y	VX	VY	VRE	THE TA	MACH	PS/PT	WFRACT
1	6.000+00	1.033-03	5.570+02	-1.065+01	5.571+02	-1.096+00	5.121-01	.8361	0.000
2	6.000+00	2.000-01	5.346+02	-3.761+01	5.359+02	-4.024+00	4.916-01	.8477	4.824-02
3	6.000+00	7.111-01	4.891+02	-7.107+01	4.942+02	-8.267+00	4.517-01	.8693	1.660-01
4	6.000+00	1.222+00	4.542+02	-8.196+01	4.615+02	-1.073+01	4.207-01	.8854	2.763-01
5	6.000+00	1.733+00	4.275+02	-8.286+01	4.355+02	-1.097+01	3.962-01	.8974	3.805-01
6	6.000+00	2.244+00	4.077+02	-7.801+01	4.151+02	-1.083+01	3.772-01	.9065	4.801-01
7	6.000+00	2.756+00	3.940+02	-6.902+01	4.000+02	-9.936+00	3.631-01	.9130	5.763-01
8	6.000+00	3.267+00	3.864+02	-5.630+01	3.905+02	-8.290+00	3.542-01	.9169	6.703-01
9	6.000+00	3.778+00	3.858+02	-3.930+01	3.878+02	-5.815+00	3.517-01	.9180	7.636-01
10	6.000+00	4.289+00	3.953+02	-1.590+01	3.956+02	-2.304+00	3.590-01	.9148	8.578-01
	6.000+00	4.800+00	4.288+02	2.375+01	4.295+02	3.170+00	3.906-01	.9001	9.565-01
	6.000+00	5.014+00	4.467+02	4.786+01	4.493+02	6.116+00	4.092-01	.8910	1.000+00

RAKE NUMBER 3

I	X	Y	VX	VY	VRE	THETA	MACH	PS/PT	WFRACY
1	6.000+00	6.986+00	5.713+02	-6.122+01	5.746+02	-6.116+00	5.289-01	.8267	0.000
2	6.000+00	7.200+00	5.430+02	-3.237+01	5.439+02	-3.412+00	4.994-01	.8834	5.287-02
3	6.000+00	7.711+00	4.884+02	-2.438+01	4.891+02	-3.092+00	4.468-01	.8719	1.712-01
4	6.000+00	8.222+00	4.595+02	-2.325+01	4.601+02	-2.559+00	4.194-01	.8860	2.821-01
5	6.000+00	8.733+00	4.386+02	-2.298+01	4.392+02	-3.000+00	3.997-01	.8958	3.881-01
6	6.000+00	9.244+00	4.221+02	-2.261+01	4.222+02	-3.066+00	3.883-01	.9032	4.905-01
7	6.000+00	9.756+00	4.086+02	-2.185+01	4.091+02	-3.062+00	3.716-01	.9081	5.898-01
8	6.000+00	1.027+01	3.972+02	-2.024+01	3.977+02	-2.918+00	3.809-01	.9139	6.866-01
9	6.000+00	1.078+01	3.875+02	-1.786+01	3.878+02	-2.581+00	3.518-01	.9180	7.811-01
10	6.000+00	1.129+01	3.791+02	-1.302+01	3.793+02	-1.968+00	3.438-01	.9215	8.738-01
	6.000+00	1.180+01	3.719+02	-5.624+00	3.719+02	-8.664-01	3.370-01	.9244	9.648-01
	6.000+00	1.200+01	3.694+02	-1.553+00	3.694+02	-2.409-01	3.347-01	.9254	1.000+00

RAKE NUMBER 4

I	X	Y	VX	VY	VRE	THETA	MACH	PS/PT	WFRACY
1	1.200+01	-3.819+01	3.707+02	0.000	3.707+02	0.000	3.359-01	.9289	0.000
2	1.200+01	-5.000+00	5.096+01	-3.891+02	3.923+02	-8.254+01	3.560-01	.9161	1.107+00
	1.200+01	5.889+00	9.732+01	-3.940+02	4.058+02	-7.613+01	3.685-01	.9105	1.130+00
			-2.745+02	0.000	-2.745+02	0.000	2.474-01	.9583	1.000+00

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RAKE NUMBER 5

I	X	Y	VX	VY	VRE	THETA	MACH	PS/PT	WFRACY
1	1.400+01	-3.819+01	3.707+02	0.000	3.707+02	0.000	3.359-01	.9289	0.000
2	1.400+01	-5.000+00	5.311+01	-3.575+02	3.615+02	-8.155+01	3.273-01	.9285	1.118+00
	1.400+01	5.889+00	8.351+01	-3.480+02	3.578+02	-7.650+01	3.239-01	.9299	1.140+00
			-2.745+02	0.000	-2.745+02	0.000	2.474-01	.9583	1.000+00

RAKE NUMBER 6

I	X	Y	VX	VY	VRE	THETA	MACH	PS/PT	WFRACY
1	1.600+01	-3.819+01	3.707+02	0.000	3.707+02	0.000	3.359-01	.9289	0.000
2	1.600+01	-5.000+00	4.472+01	-3.339+02	3.369+02	-8.237+01	3.046-01	.9377	1.139+00
	1.600+01	5.889+00	6.626+01	-3.186+02	3.254+02	-7.825+01	2.941-01	.9418	1.157+00
			-2.745+02	0.000	-2.745+02	0.000	2.474-01	.9583	1.000+00

RAKE NUMBER 7

I	X	Y	VX	VY	VRE	THETA	MACH	PS/PT	WFRACY
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	1.800+01	-3.819+01	3.707+02	0.000	3.707+02	0.000	3.359-01	.9249	0.000
1	1.800+01	-5.000+00	3.044+01	-3.168+02	3.183+02	-8.451+01	2.875-01	.9442	1.169+00
2	1.800+01	-3.000+00	4.630+01	-2.967+02	3.003+02	-8.113+01	2.710-01	.9503	1.183+00
	1.800+01	5.889+00	-2.745+02	0.000	-2.745+02	0.000	2.478-01	.9583	1.000+00

RAKE NUMBER 8

I	X	Y	VX	VY	VRE	THE TA	MACH	PS/PT	WFRACT
	2.000+01	-3.819+01	3.707+02	0.000	3.707+02	0.000	3.359-01	.9249	0.000
1	2.000+01	-5.000+00	1.120+01	-3.066+02	3.068+02	-8.791+01	2.770-01	.9481	1.216+00
2	2.000+01	-3.000+00	2.071+01	-2.786+02	2.794+02	-8.575+01	2.519-01	.9568	1.222+00
	2.000+01	5.889+00	-2.745+02	0.000	-2.745+02	0.000	2.478-01	.9583	1.000+00

RAKE WEIGHT FLOW DATA

I	X	IRAK	WDOT	WDOTCA	MACH
1	-1.0000	3	2.57138+01	2.57143+01	3.20048-01
2	6.0000	5	1.26886+01	3.03754+01	3.89041-01
3	6.0000	4	1.26970+01	3.03856+01	3.89200-01
4	12.0000	5	3.80703+01	1.03647+01	1.22483-01
5	14.0000	5	3.79438+01	1.03302+01	1.22069-01
6	16.0000	5	3.65363+01	9.94701+00	1.17463-01
7	18.0000	5	3.43602+01	9.35703+00	1.10390-01
8	20.0000	5	3.14492+01	8.56205+00	1.00890-01

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ABRKPT PRINTS

REFERENCES

1. Stockman, Norbert O.; and Farrell, Charles A., Jr.: Improved Computer Programs for Calculating Potential Flow in Propulsion System Inlets. NASA TM-73728, 1977.
2. Hawk, J. Dennis; Stockman, Norbert O.; and Farrell, Charles A., Jr.: Computer Programs for Calculating Two-Dimensional Potential Flow in and About Propulsion System Inlets. NASA TM-78930, 1978.
3. Hess, J. L.; and Smith, Apollo M. O.: Calculations of Potential Flow About Arbitrary Bodies. Progress in Aeronautical Sciences, Vol. 8, D. Kuchemann, ed., Pergamon Press, 1967, pp. 1-138.

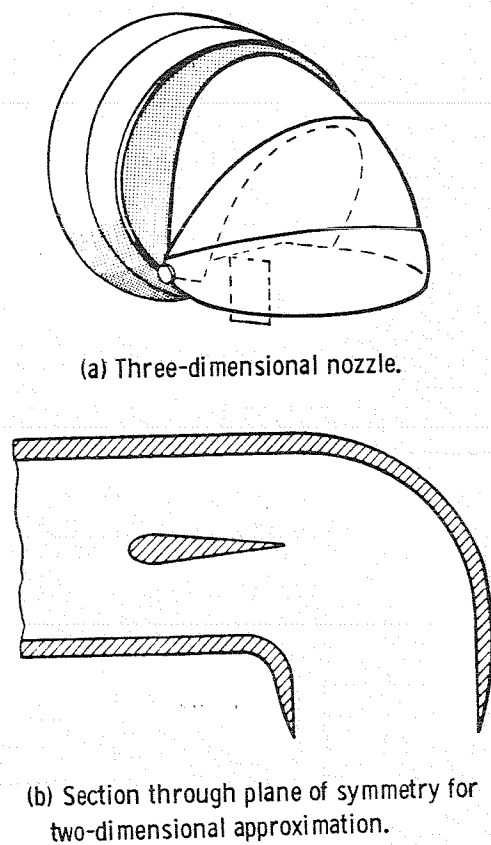


Figure 3. - Two-dimensional approximation of three-dimensional nozzle.

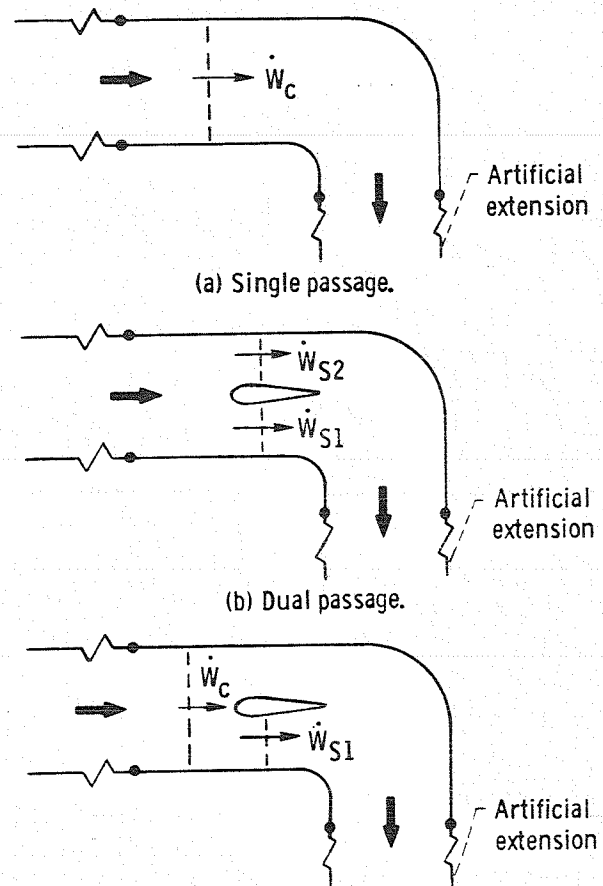


Figure 4. - Nozzle geometries and flow conditions.

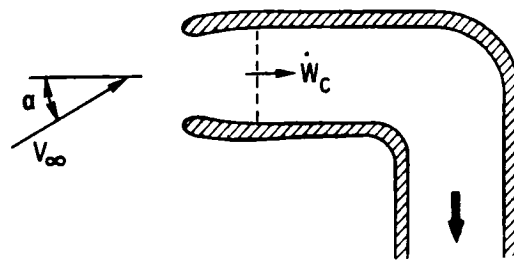


Figure 5. - Inlet and nozzle geometry

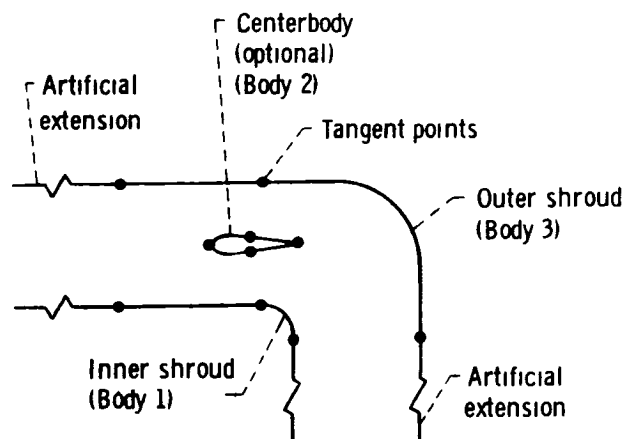


Figure 6. - Typical nozzle segmentation.

At any point

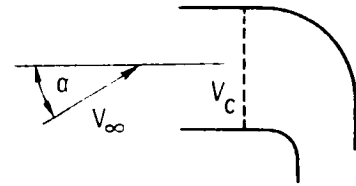
$$\bar{V} = A\bar{V}_1 + B\bar{V}_2 + C(\bar{V}_3 - \bar{V}_4)$$

where A, B, and C are determined by specifying values of

\bar{V}_c average axial velocities at the control station

V_∞ free stream velocity

α direction of free stream velocity relative to nozzle axis



(a) Single passage nozzle

At any point

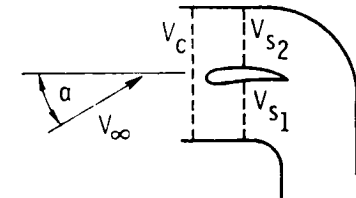
$$\bar{V} = A\bar{V}_1 + B\bar{V}_2 + C(\bar{V}_3 - \bar{V}_4) + D(\bar{V}_5 - \bar{V}_6)$$

where A, B, C, and D are determined by specifying values of

$\bar{V}_c, \bar{V}_{s1}, \bar{V}_{s2}$ average axial velocities at any two of the three control stations

V_∞ free stream velocity

α direction of free stream velocity relative to nozzle axis



(b) Two passage nozzle

Figure 7 - Combined solution, \bar{V}

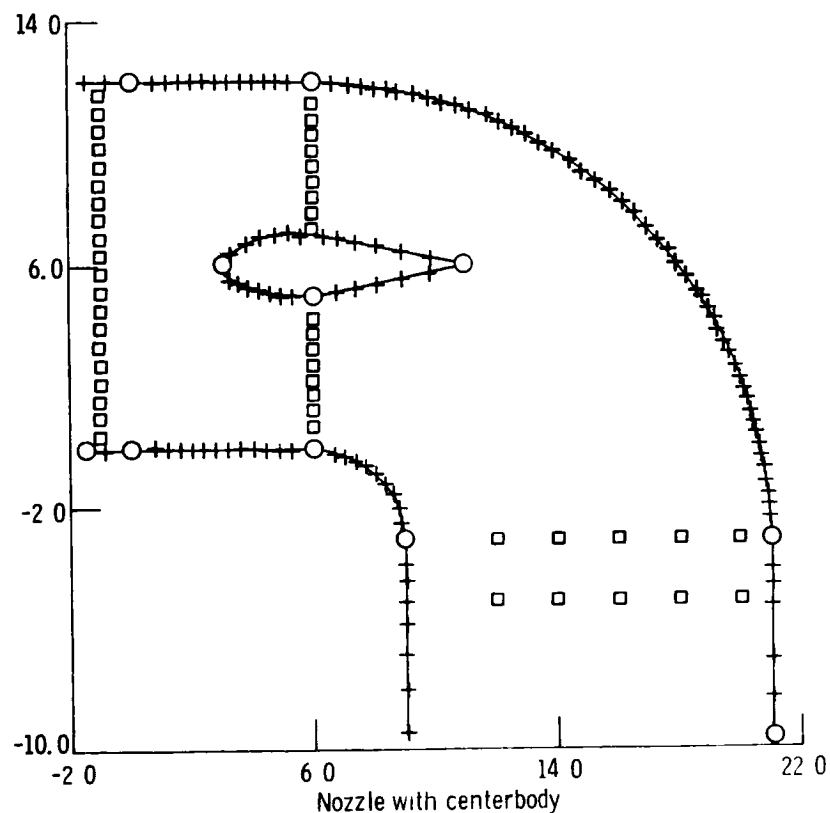


Figure 8 - Graphic output of SCIRCL

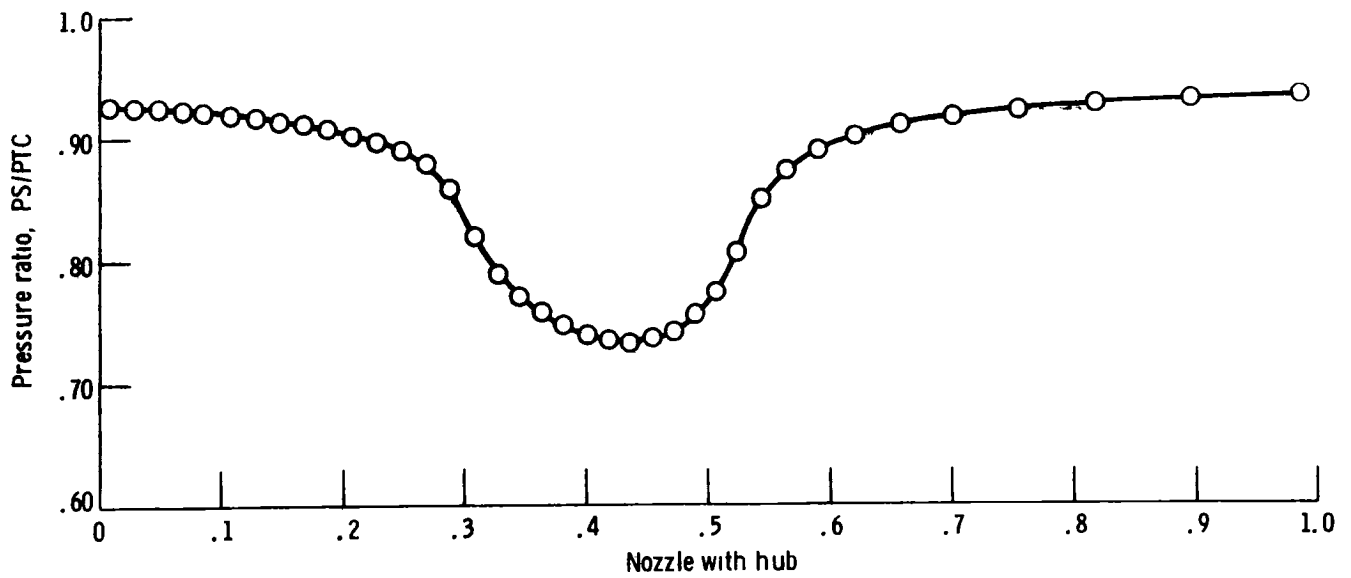


Figure 9. - Surface static pressure ratio from NOZZLE

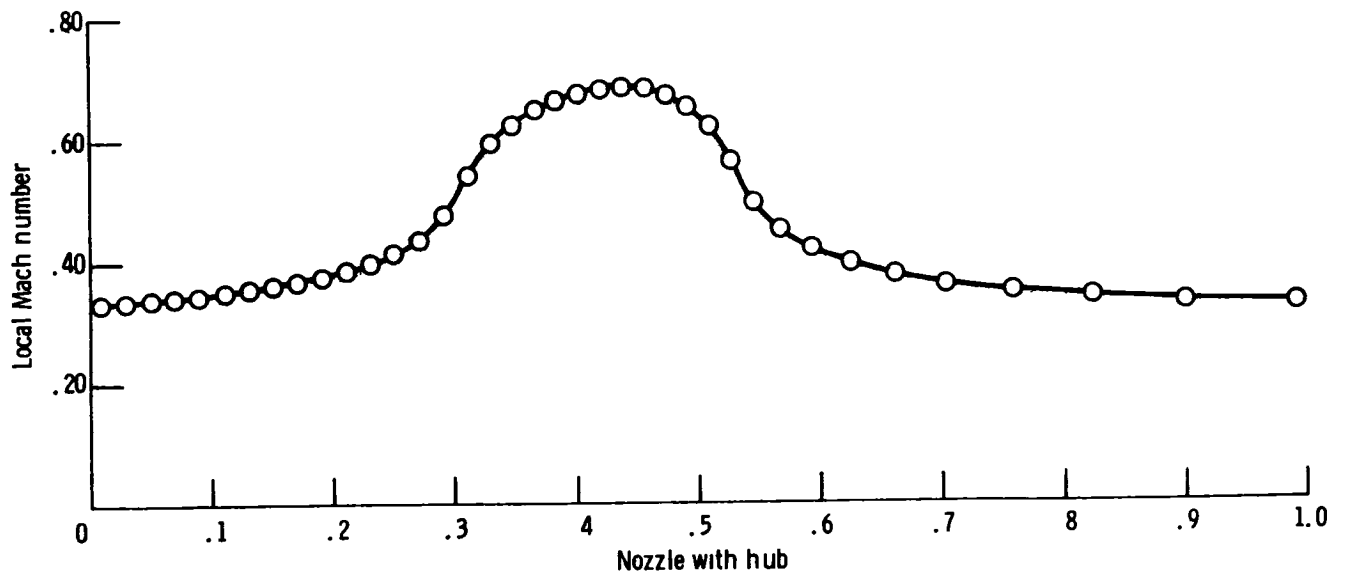


Figure 10. - Surface Mach number distribution from NOZZLE.

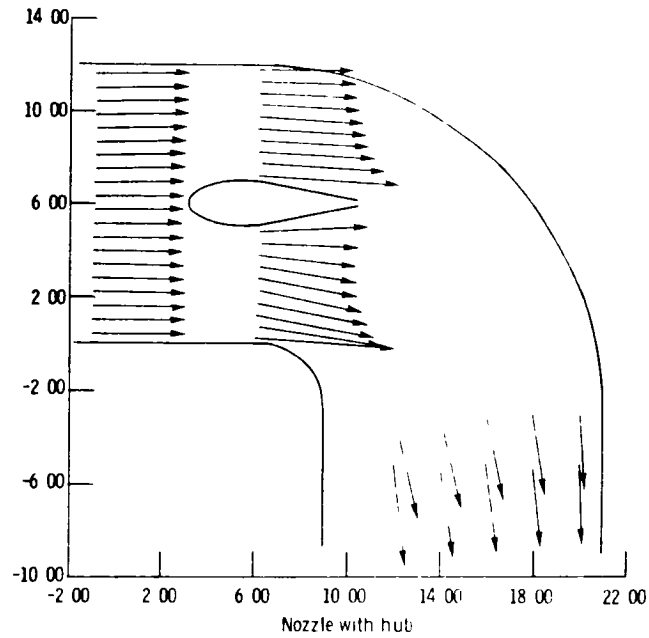


Figure 11 - Vector flow field from nozzle

- Segment endpoints
- ◇ Slope line endpoints (length of slope line is arbitrary)
- △ Optional superellipse point
- ▽ Optional bisuperellipse point
- ▢ Optional bisuperellipse inflection point

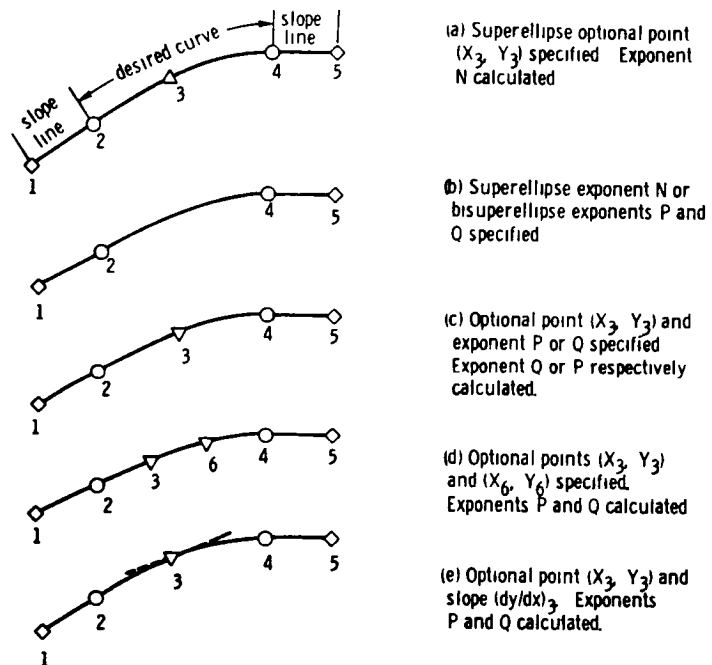
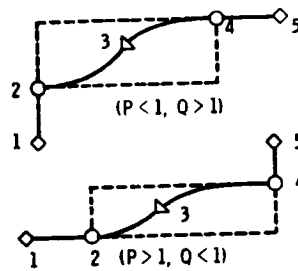
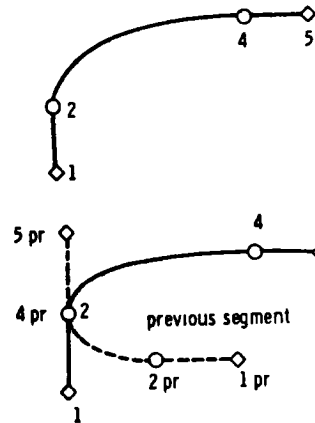


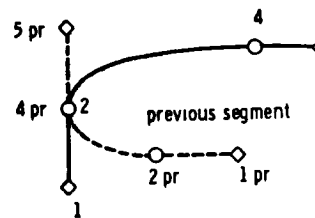
Figure 12 - Sketches for SCIRCL input. Bisuperellipse options



(f) Bisuperellipse with inflection point. Axial location X_3 and slope $(dy/dx)_3$ of inflection point specified. Note that the slope line requirements for this option are different from all other options. One slope line must be perpendicular to curve and one must be tangent, thus there are two possibilities as shown. Both lines must lie away from and outside the 'box' surrounding the desired curve. Also, shown are the exponents that will result in each case



(g) Curvature at either point 2 or point 4 specified

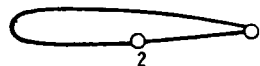


(h) Curvature at endpoint 2 matched to internally calculated curvature at endpoint 4 pr of previous segment

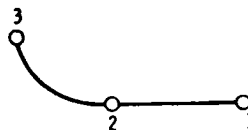
Figure 12 - Concluded



(a) Straight line



(b) Straight line for closed body



(c) Lemniscate



(d) Cubic

Figure 13 - Sketches for SCIRCL input options except bisuperellipse

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(a) Title, control, and rake data cards.

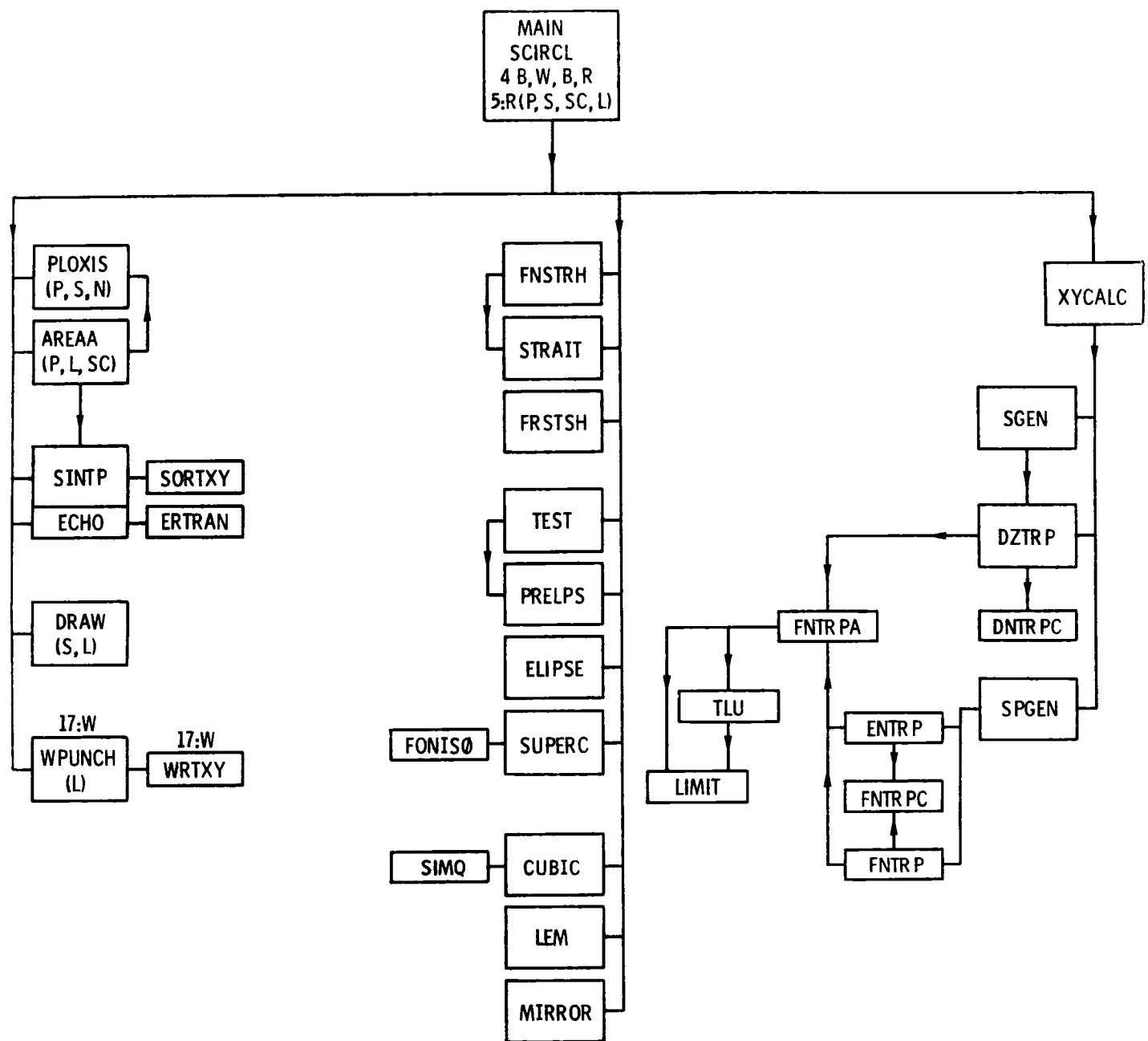
Figure 14. - SCIRCL input layout.

2,

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1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60		
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3	ELND	WDOTC	WDOTC1	WDOTC2	PSTAT	TSTAT	CUTOF1	CUTOF2	CUTOFH	VPERIN																																																			
4																																																													
4A	* INSERT CARD 4A ONLY IF VPERIN > 0 SEE SCIRCL INPUT CARD #2 FOR LAYOUT																																																												
5	XTEST	YCL	YCU																																																										
6	XTEST1	YCL1	YCU1																																																										
7	XTEST2	YCL2	YCU2																																																										
8	XR1	XR2	XRH																																																										
9	YR1	YR2	YRH																																																										

NASA-C-856 (REV 9-14-59)

Figure 15. - NOZZLEC input form



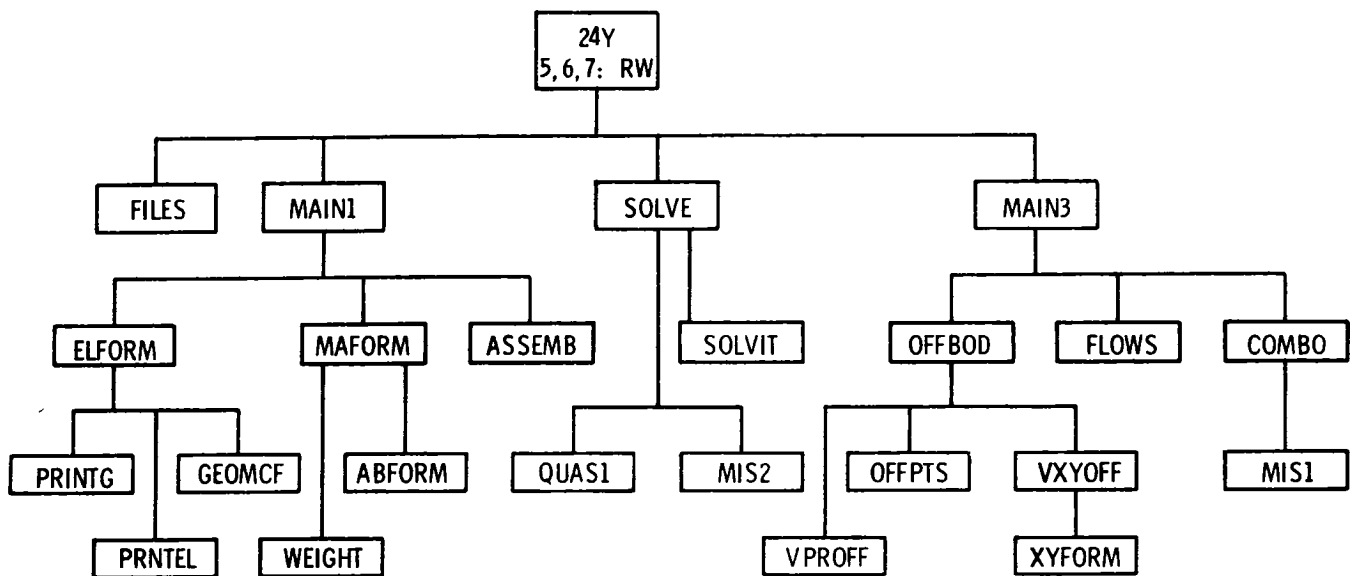
5 - I/O Unit No.
B - Rewind
W - Write
R - Read

Calcomp Routines Referenced

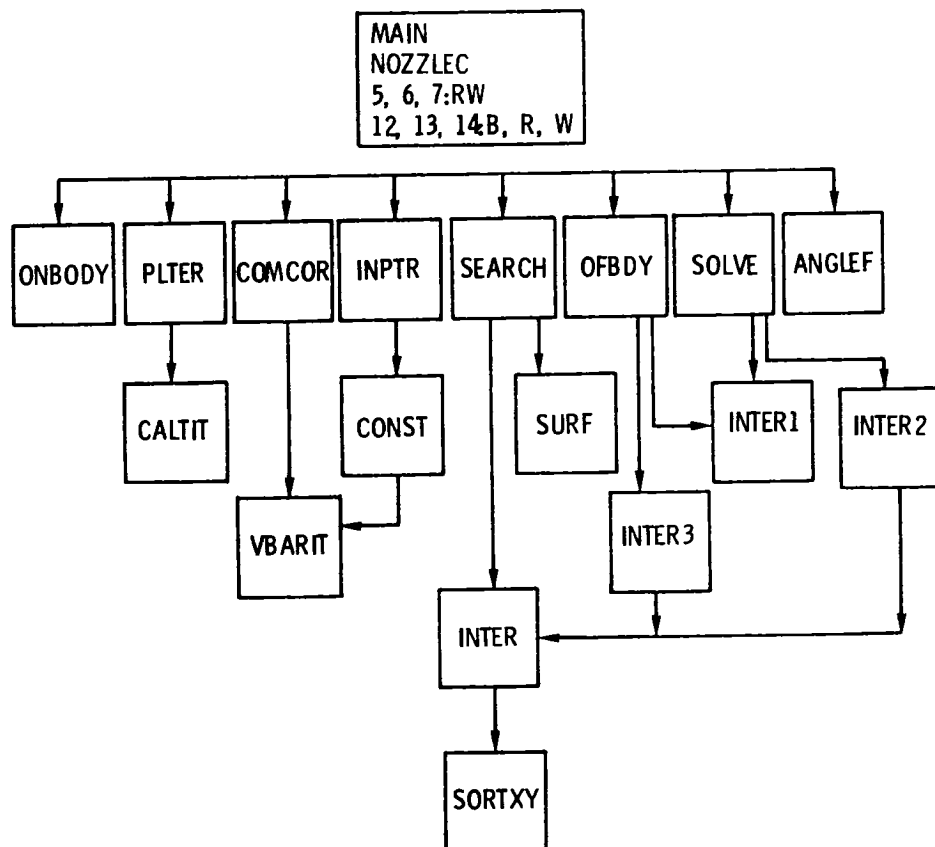
(L) - Line
(P) - Plot
(S) - Symbol
(SC) - Scale
(N) - Number

(a) SCIRCL

Figure 16. - Call Sequences.



(b) 24Y



(c) NOZZLEC.

Figure 16. - Concluded.

End of Document